

NOTE: LLET SIDE SUPPORT ARM SHOWN. RIGHT SIDE SUPPORT ARM IS MIRROR IMAGE.

FIGURE 2-1 SUPPORT ARM LATCH MECHANISMS LATCHED CONFIGURATION

Mission	J	Basic Date	12/4/70	Cnange Dat	е	4/19/71	_ouge	2-3

NOTE: DOUBLE BRAKED REEL
DEPLOYMENT TAPE SIMILAR
TO CENTER BRAKED REEL
TAPE BUT STOWED ON
RIGHT SIDE OF VEHICLE

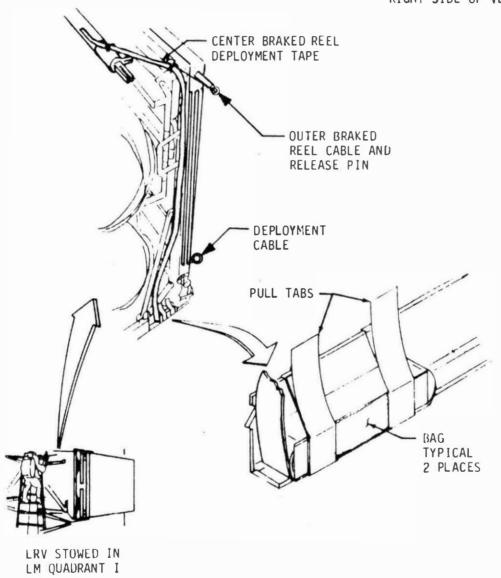


			FIGURE	2-2	LRV	DEPLOYMENT	TAPES	AND C	ABLES	-
Mission	J	IX.	Casi	ic Ju	te	12/4/70	Cnange	Date	4/19/71	Juge 2-4

Lii quadrant, deployment cable may be pulled to initiate

LRV movement.

2-5

REMARKS

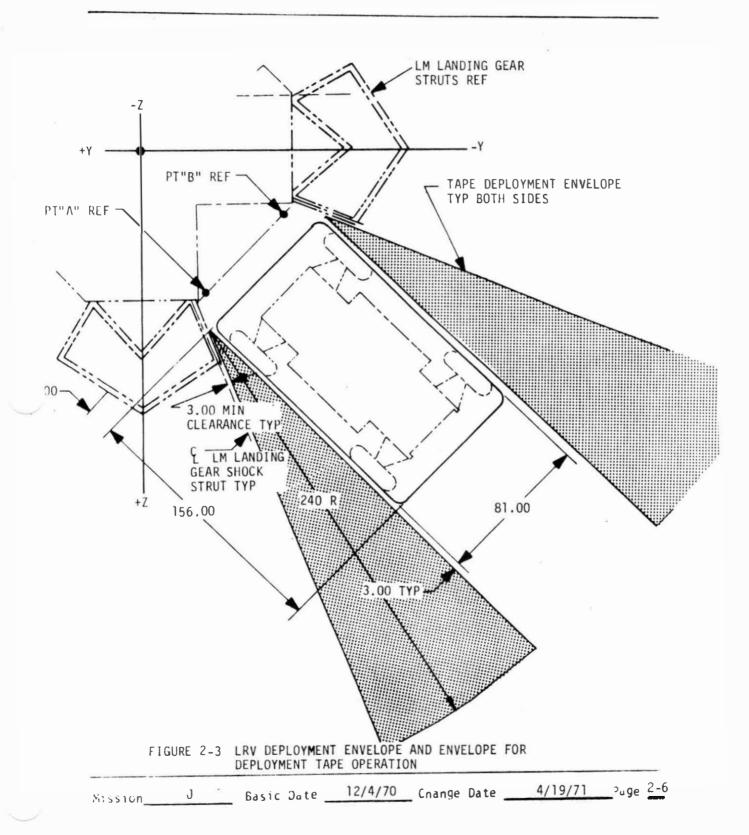
lower half of apex fittings may

fall away immediately or during deployment rotation. The last segment of travel releases the upper pin. As the upper release pin is pulled, LRV rotates out of LM about 4 degrees.

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STA/STEP



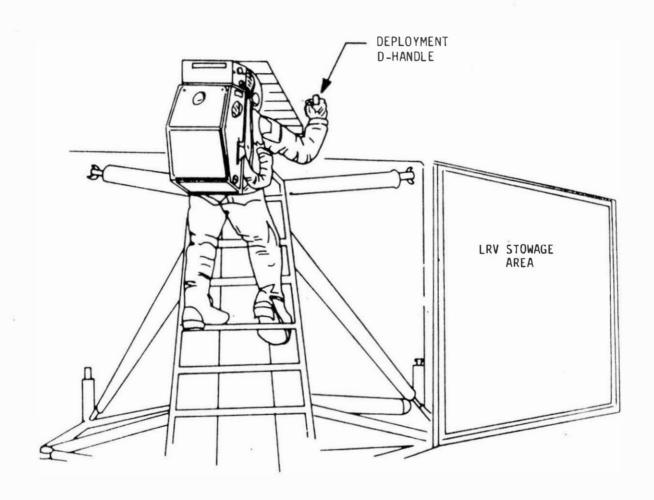
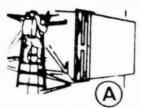


FIGURE 2-4 CREWMAN POSITIONED TO DEPLOY LRV

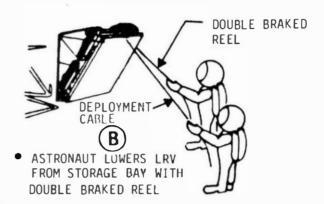
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Miss	STA/STEP PROCEDURE	REMARKS
ion	2.1 (Continued)	
ن	 Descend LM ladder. Grasp deployment cable and monitor deployment activity. 	Crewman operating deployment cable should keep slack out of double braked reel cables.
Gasic Date	Other crewman pulls double braked reel tape at right side of vehicle. Verify LRV rotates outward from LM (Figure 2-5, View B). NOTE: Crewman should remain within defined envelopes for deployment tape operation (Figure 2-3) to ensure that deployment tapes do not contact sensitive LM components.	For first 15 degrees of rotation LRV rotates on apex fittings, thereafter apex fittings lift off spools and rotation point shifts to walking hinge. Lower telescopic tubes ratchet engage at 35 degrees rotation.
12/4/70 Change Date	 n. Continue to pull double braked ree! tape (figure 2-5, View C). When vehicle rotates outboard to about 45 degrees, verify that: (1) Aft chassis unfolds and locks in position. (2) Rear wheels unfold and tethered rear wheel struts fall free. (3) Forward chassis is released from console post and returns to 35 degree position. NOTE: If either aft or forward chassis latch pins fail to pull automatically, deployment cable may be pulled to accomplish pin release. 	At about 45 degrees the 45° cable tightens, pulling the forward and aft chassis latch pins at the console post mount on the center chassis. The aft chassis and wheels fully deploy and the forward chassis returns to the 35° position.
4/19/71 Page 2	 Continue to pull double braked reel tape (Figure 2-5, View D). Verify that: (i) Center/aft chassis rotates until rear wheels contact lunar surface. (2) Rear wheels slide on surface permitting center/aft chassis to move away from LM. 	At about 73 degrees, the cam on forward sides of center chassis strikes latch lock arm, forces arm down out of retaining spring and unlocks latch.
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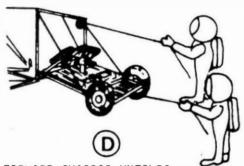
- LRV STOWED IN QUADRANT
- /.STRONAUT REMOVES INSULATION BLANKET, OPERATING TAPES
- ASTRONAUT REMOTELY INITIATES AND EXECUTES DEPLOYMENT



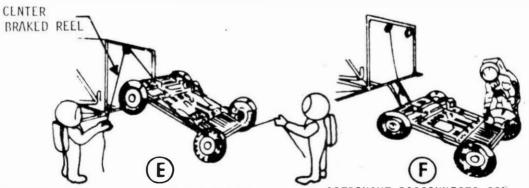


- AFT CHASSIS LOCKS IN
- POSITION





- FORWARD CHASSIS UNFOLDS
- FRONT WHEELS UNFOLD



FORWARD CHASSIS LOCKS IN POSITION. ASTRONAUT LOWERS LRV TO SURFACE WITH CENTER BRAKED REEL

 ASTRONAUT DISCONNECTS SSE ASTRONAUT UNFOLDS SEATS, FOOTRESTS, (FINAL STOP)

FIGURE 2-5 LRV DEPLOYMENT SEQUENCE

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Mission	STA	/STEP PROCEDURE	REMARKS
sion_	2.1	(Continued)	
ر		NOTE: If wheels fail to slide, deployment cable may be pulled to permit center/aft chassis to move away from LM.	
	p.	Continue to pull double braked reel tape (Figure 2-5, View D). Verify that:	Forward wheel lock strut pins release and forward wheels de-
Dasic Date		 (1) Forward chassis continues to unfold and locks in position. (2) Forward wheels unfold. (3) Outer braked reel cables are slack. (4) 45° cable again becomes taut. 	ploy as the angle between the forward and center chassis approaches 170 degrees. (The 45 degree cable again becomes taut).
12 /4/7 0 Change	Ģ.	Release touble braked reel tape and at chassis RR grasp outer braked reel cable in right hand and remove cable pin P8 (Figure 2-6) with left hand.	At this time the forward and aft chassis sections are deployed and locked to the center chassis. All wheels are deployed. The forward chassis is held up by the telescopic tube assembly and the 45 degree cable.
ge Da	r.	Discard cable and pin outside work area.	
Da‡e	S.	At chassis LR grasp outer braked reel cable in left hand and remove cable pin Pl.	Figure 2-6.
4/1	t.	Discard cable and pin outside work area.	
4/19/71	u.	Pull center braked reel tape (Figure 2-5, View E). Verify that forward chassis lowers until all wheels contact linar surface and support vehicle weight and 45° cable is slack.	This take was previously stowed over a LM landing strut for convenient access.
Page 2		NCTE: If wheels fail to slide, deployment cable may be pulled to move LRV away from LM.	Using deployment cable to pull the LRV, with parking brake
2-10	min		

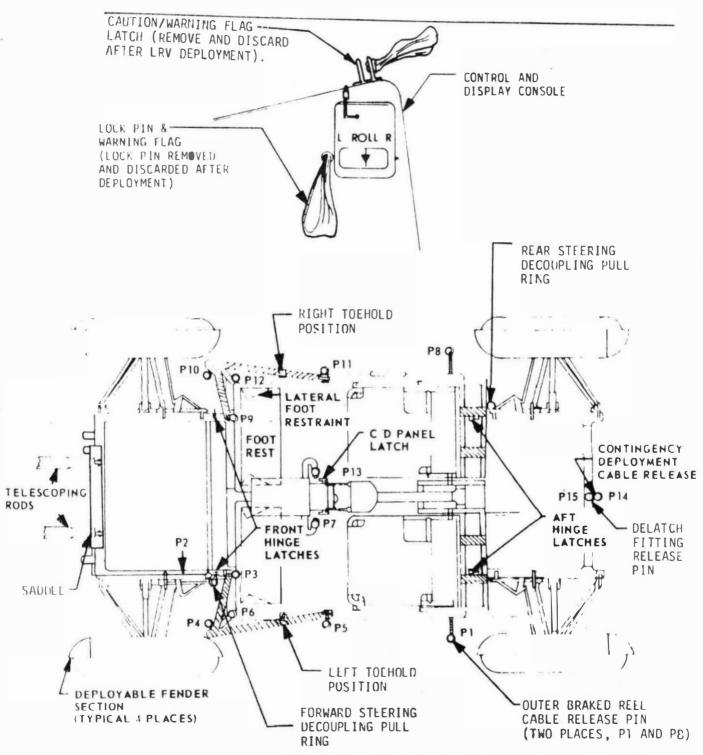
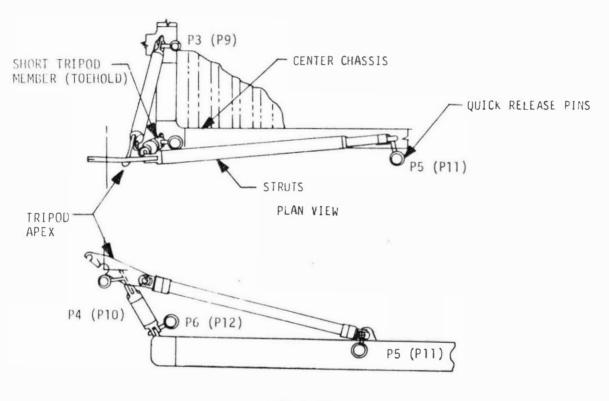


FIGURE 2-6 LRV DEPLOYMENT HARDWARE AND STEERING RING LOCATIONS (SHEET 1 OF 2)

Mission ____ J Basic Date ___12/4/70 Change Date ___4/19/71 Page __2-11

NOTE: LEFT SIDE TRIPOD

ASSEMBLY SHOWN. RIGHT SIDE PULL PIN NUMBERS SHOWN IN PARENTHESIS



SIDE VIEW

FIGURE 2-6 LRV DEPLOYMENT HARDWARE LOCATIONS (SHEET 2 OF 2)

Missian	J	Basic Date	12/4/70	Change Date	4/19/71	Juge 2-12

PROCEDURE

Grasp tripod apex with right hand and remove pin Pll with left

at. Grasp remaining short tripod member in right hand, remove pin P12 with left hand, and discard pin clear of deployment area.

aa. Discard tripod main members and pin clear of deployment area.

(Continued)

deployment area.

hand.

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engaged and wheel struts tethered, should be kept to a

minimum.

R	STAZ	PROCEDURE PROCEDURE	PEMARKS
Six Cion	2.1	(Continued)	
ے ا	ac.	Remove short tripod member and insert tripod rember in right twerold josition or stow in underseat stowage bag.	If short tripod member is installed in coehold position, end with hook should be outboard with hook pointing forward. This is also used as wheel decoupling tool.
Basic Date	ad.	Pull right tootrest lift tabs.	Figure 2-7. Tabs pull free of footrests but remain attached to the floor panel.
ס	ae.	Rotate footrest upward and forward and lock into position.	
12/4/70	af.	Release velcro tiedown strap (if necessary), pull out right C/D console "T" handle Pi3 with left hand and turn 90° CW.	Figure 2-8.
	ag.	Release right seat belt from stowage position and stow in temporary location.	Figure 2-9.
Change	ah.	Rotate right seat to stable overcenter position.	
e Date	ai.	Rotate legs to full upright position.	
te	aj.	Attach forward seat legs velcro strap to outboard handhold.	
۵	ak.	Verify underseat stowage bag erects.	,a
4/19/71	al.	Pull seat pan frame forward to engage front legs.	
/71	am.	Verify all seat latches latched.	
Page 2-1	ar.	Verify both hinge pins flush at RR hinge.	If hinge pin is not flush, tap pin with toehold. Verif, pin is latched by pressing down to chassis.

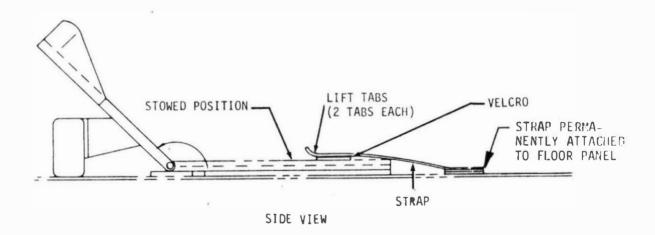
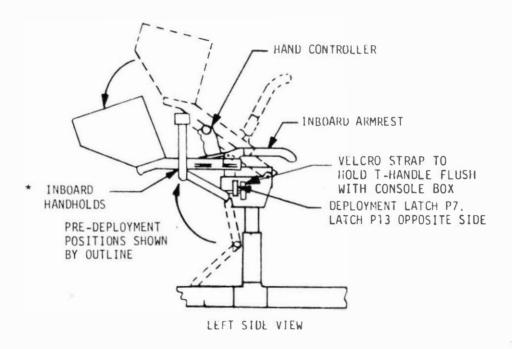


FIGURE 2-7 FOOT REST DEPLOYMENT

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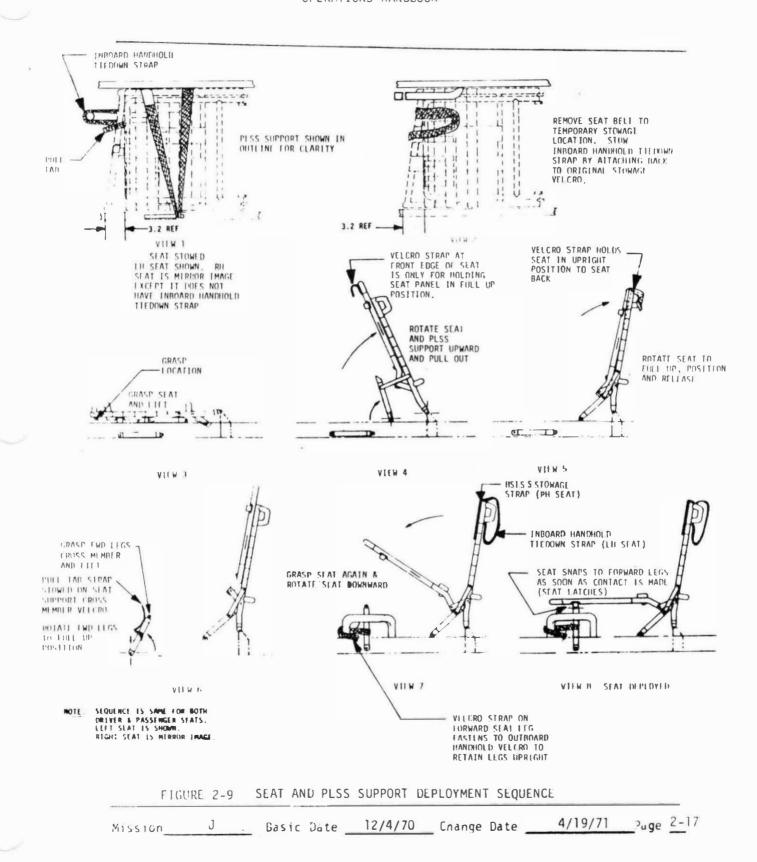


* IG TRAINER INBOARD HANDHOLD CANNOT BE TOEDED DOWN TO SIMBLATE PRE-DEPLOYMENT POSITION DUE TO CABLE RUNS FROM CONSOLE.

FIGURE 2-B CONTROL AND DISPLAY CONSOLE DEPLOYMENT

Sion J Basic Date 12/4/70 Change Date 4/19/71 Page 2-16

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PEMARKS

on chassis.

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2.1

Basic

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PROCEDUPE

(Continued)

au. Rotate seat to stable overcenter position.

aw. Attach forward seat legs velcro strap to outboard handhold.

ay. Pull seat pan frame forward to engage front legs.

av. Rotate legs to full upright position.

ax. Verify underseat stowage big erects.

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Mission	STA/	STEP PRICEDURE	REMARKS
ion i	bl.	Remove short tricco member and use hooked end to pull cable P2.	Figure 2-6. Tool mock inter- faces with cable area color coded gold. Deflection of cable releases telescoping rods saddle and forward wheel strut tethers.
ڦ ا	bm.	Visually verify that telescoping rods saddle falls away from LRY.	
Dasic Date 12	bn.	Either insert short tripod member in left toehold position or stow in underseat stowage bag.	Figure 2-6. If short tripod member is installed in toehold position, end with hook should be outboard with hook pointing forward. This is also used as wheel decoupling tool.
12/4/70	bo.	Pull left footrest lift tabs.	Tabs pull free of footrests, but remain attached to floor panel.
Change	bq.	Rotate footrest upward and forward and lock into position.	
	br.	Verify both hinge sins flush at LF hinge.	
Oa te	bs.	Deploy LF fender extension.	
4/	bt.	Verify battery no. 1 and SPU dust covers closed and secured to velcro patch.	
4/19/71 Page	bu.	Verify the forward steering decoupling pull ring seal has not been broken.	Figure 2-6. If seal is broken and subsequent steering check using hand controller indicates steering is engaged, disregard broken seal. If hand controller check indicates steering is
2-2			

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MISS	STA/	STEP FROCEDUPE	REMARKS
Mission J	2.1	(Centirued)	not engaged, center wheels in neutral steer, verify forward steering lock and continue mission using rear steering only.
l Ga	bv.	Move to right side of vehicle and verify battery no. 2 dust cover closed and secured to velcro patch.	
Basic Date	bw.	At right side of LRV rotate right "T" handle Pl3 90° CW, fold "T" handle flush with console box and secure in position with velcro strap.	"T" handle should snap~in, lock and fold down flush with the console box.
12/4/70			
Change Date			
4/19/71			
Page 2-2		*	

Mission	STA/STEP	PROCECUPE	REMARKS
non_	2.2	LRV POST DEPLOYMENT CHECKOUT AND DRIVE TO MESA	
		fy hand controller in parking brake and neutral throttle tion and reverse inhitit switch is on (pushed down).	Crewman stands along side the vehicle.
Basic Date 12/4/70 Ch	Foll NAV GYRO Syst AUX EUS + 15 F 15 MOTO BATT PIM STEE	fy switches and circuit breakers in pre-launch positions as ows: POWER Circuit Breaker - Open TORQUING Switch - OFF em RESET Switch - OFF Circuit Breaker - Open A, B, C, D, Circuit Breakers - Open DC PRIM and SEC Circuit Breakers - Open DC Switch - OFF R TEMP Switch - FORWARD ERY Switch - ANDS SELECT Switch - BOTH RING FORWARD and REAR Circuit Breakers - Open RING FORWARD and REAR Switches - OFF	Figure 2-11. Crewman stands along side vehicle.
Change Date 4/19/71	DRIV DRIV c. Manu	E POWER LF, RF, LR, RR Circuit Breakers - Open E POWER LF, RF, LR, RR Switches - OFF E ENABLE LF, RF, LR, RR Switches - OFF Mally move the LPV away from the LM. (See remarks for LRV iguration for this operation).	Crew may manually move LRV away from LM prior to powerup; the hand controller should be placed in neutral throttle position and parking brake released. With a crewman standing on either side of vehicle outboard handholds may be used to lift,
Page 2-22	3		move, and tow LRV to any desired location.

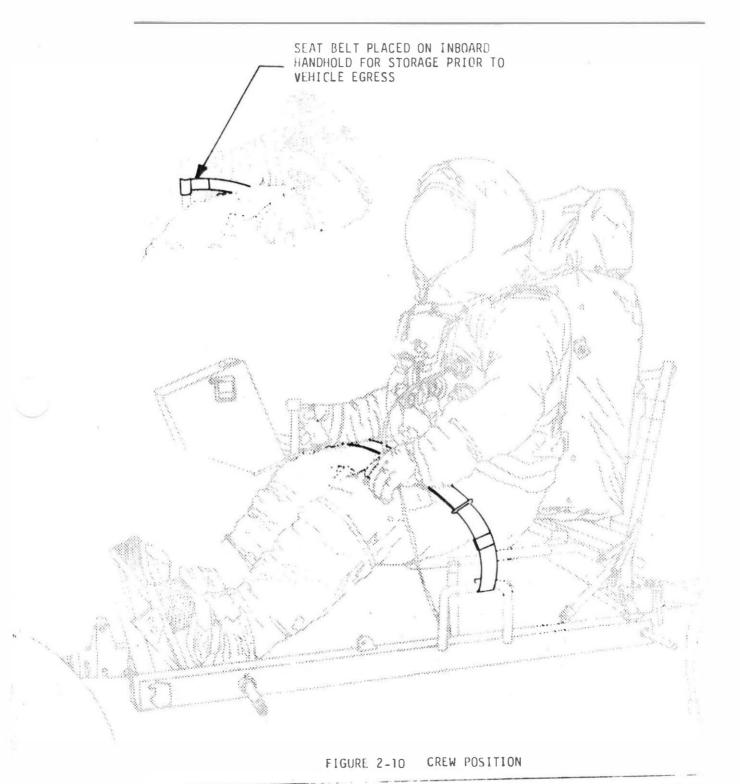
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REMARKS

to permit wheels to roll.

vehicle, and should exercise



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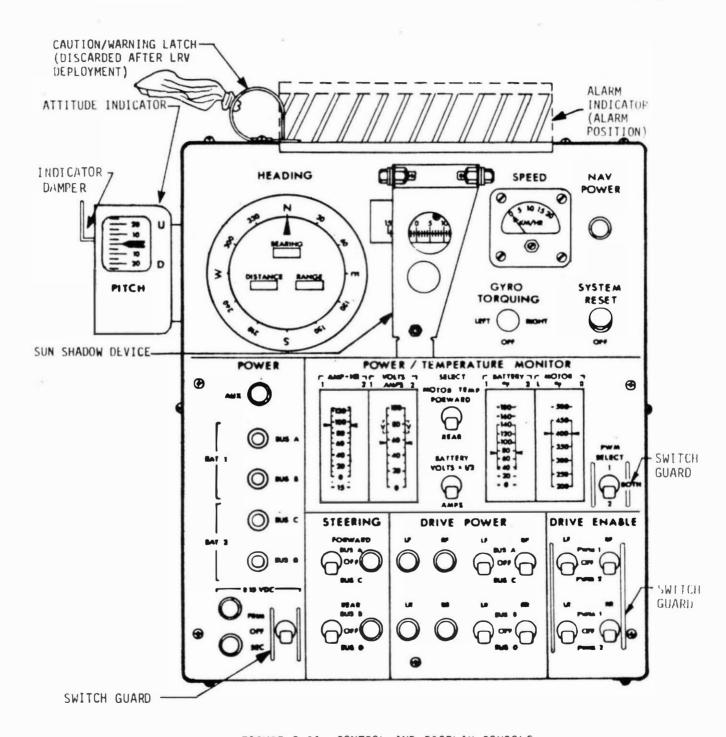


FIGURE 2-11 CONTROL AND DISPLAY CONSOLE

Mission Basic Date 12/4/70 Change Date 4/19/71 Page 2-25

M	STA/S	TEP PROCEDURE	REMARKS
Mission_	2.2	(Continued)	
	р.	DRIVE ENABLE LF and RF Switches - PWM 2.	
ے	٩.	DRIVE ENABLE LR and RR Switches - PWM 1.	
	r.	+ 15 VDC Switch - SEC.	
Sasic	۶.	STEERING FORWARD Switch - BUS C.	Forward steering operates from Battery No. 2.
Date	t.	STEERING REAR Switch - BUS B.	Rear steering operates from Battery No. 1.
12/4/70 Change Date		The hand controller should be in park brake position and the drive enable switches must be set to an active PWM prior to setting any drive power switch to an energized bus. If the drive power switch is turned on and the corresponding drive enable switch is not selected to an active PWM, then full power will be applied to the corresponding drive motor when the hand controller is released from brake position. Should this condition occur the hand controller should be immediately returned to park brake position.	The PWM select switch determines which PWM is active. The hand controller was verified set in park brake position in step 2.2.d. The PWM select switch was verified in "BOTH" position in step 2.2.b. The drive enable switches were set to active PWM positions in steps 2.2.p and 2.2.q.
4/1	u.	DRIVE POWER LF Switch - BUS C.	Front wheels operate from Battery No. 2.
4/19/71	v.	DRIVE POWER RF Switch - BUS C.	
Page	W.	DRIVE POWER LR Switch - BUS B.	Rear wheels operate from Battery No. 1.
2-26			

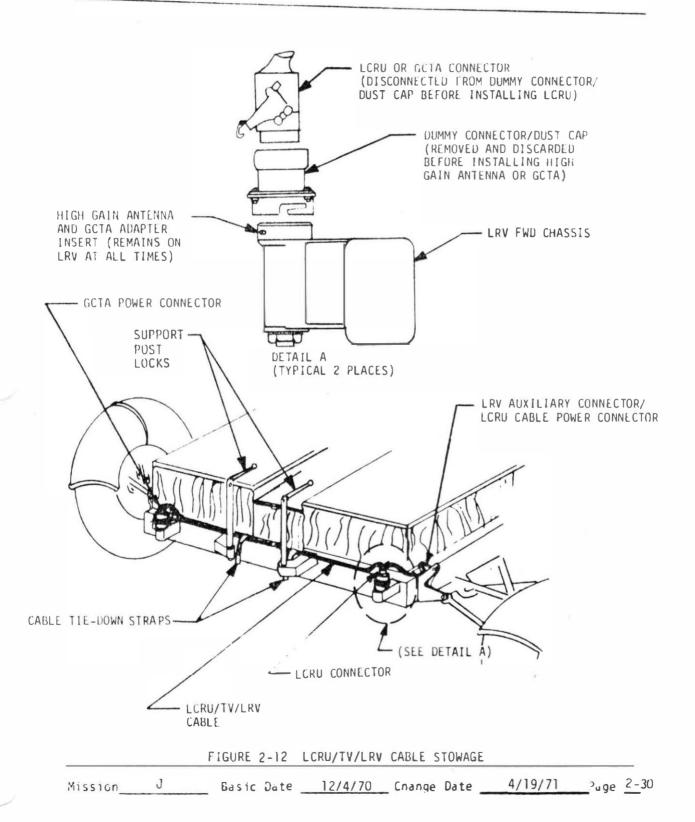
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Mission	STA/STEP PRUCEDURE	REMARKS
non	2.2 (Continued)	
ے	x. DRIVE POWER RR Switch - BUS B.	
	y. Release parking brake.	
1	z. Hand Controller reverse inhibit switch - UP position.	
Basic Da	NOTE: The LRV driver may now back away from LM. LRV driver should request other crewman to direct and monitor any backing operations from an off-vehicle position.	
Date1	aa. Stop LRV and set parking brake. Reset reverse inhibit switch (push switch down).	
12/4/70 Change	ab. Release parking brake and drive to MESA area for equipment load	ing. To the extent possible driver should verify steering, speed control and braking during this brief drive. The off-vehicle crewman should verify all four wheels rotating (not sliding).
Date	ac. Stop LRV and set hand controller in parking brake position; Neutral throttle.	Parking brake should always be set prior to vehicle egress by either crewman.
4/19/71 Page 2-2	ad. Perform LRV partial power down as follows: DRIVE POWER Switches (4) - OFF. STEERING Switches (2) - OFF. + 15 VDC Switch - OFF.	Turning off drive power, steering, and + 15 VDC switches ensures that a failure in the DCE will not apply power to any vehicle motor thereby precluding any unnecessary power drain.

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Mission	STA/STEP	PROCEDURE	REMARKS		
nor	2.2	(Continued)			
J Basic	NOTE:	The above step 2.2.ad assumes that payload loading and first LRV traverse will follow in order. Should crew rest period be scheduled subsequent to step 2.2.ae and prior to first LRV sortie, then Bus A, B, C, and D circuit breakers should be opened.			
c Date	ae. Relea	se and stow seat belt and egress vehicle.			
12/4/70 Change Date					
4/19/71,			7		
Page 2-28		e			

Mi S	STA/	STEP PROCEDURE	REMARKS	
Mission	2.3	PAYLOAD LOADING		
ے	2.3	. CRU Installation		
	а.	Place LCRU support post locks in the up position.	Figure 2-12. LRV arrives on lunar surface with LCRU support	
Basic Date			posts installed in LRV support tubes on forward chassis and with LRV/LCRU power cable connected to LRV auxiliary connector.	
te _	b.	Disconnect GCTA connector from LRV dummy connectors.	Figure 2-12.	
12		NOTES		
12/4/70		 Do not disconnect LCRU power cable from LRV auxiliary connector. Dust contamination could occur if this connector is disconnected. 		
Change		Do not allow GCTA connector of cable to fall to lunar surface.		
Date		3. Do not place payload on battery cover.		
ĺ	c.	Remove dummy connector from LRV GCTA receptacle and discard.		
4/19/71	d.	Remove LCRU from its LM stowage position and place onto LRV forward chassis LCRU support posts.	Figure 2-13.	
	e. When LCRU is bottomed against support posts, position support post locks in horizontal position to secure LCRU.			
Page	f.	Verify LRV AUX power circuit breaker - Open.		
2-29				



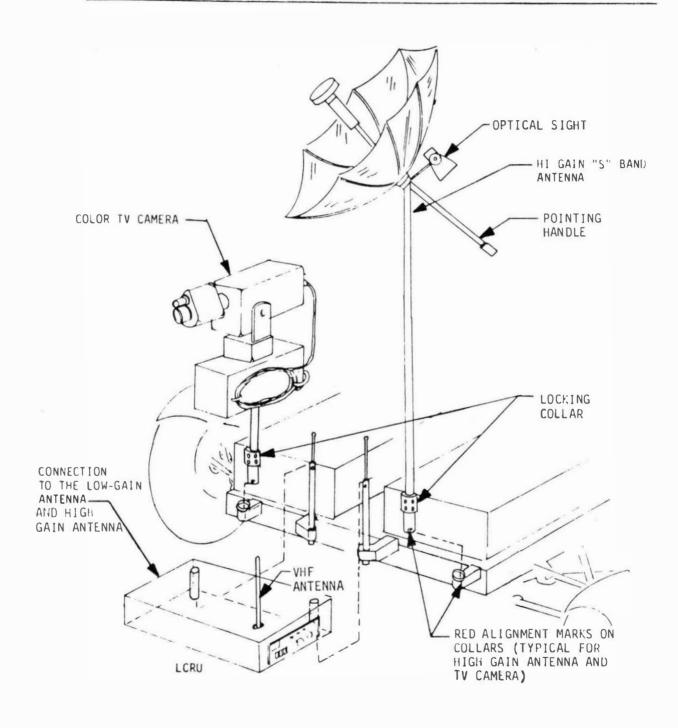


FIGURE 2-13 LCRU, HIGH GAIN ANTENNA, TV CAMERA INSTALLATION

Mission	.1	Basic Date	12/4/70	Change Date	4/19/71	Juge 2-31
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STA/S	TEP PROCEDURE	REMARKS
2.3 g.	Disconnect LCRU power connector from LRV dummy connector and	
h.	Cover connector with thermal boot.	*
i.	Remove dummy connector from LRV HGA receptacle and discard.	
2.3.	2 GCTA Installation	Figure 2-13.
a.	At MESA, pull GCTA control unit pip pin release cable. CAUTION Do not strike GCTA control unit mirror surfaces on MESA.	
b.	Remove GCTA control unit and support staff from MESA.	
с.	Unfold GCTA support staff Verify staff locked. CAUTION If GCTA staff is not properly locked, it could fall on LCRU and cause severe LCRU radiator damage.	
	With connector receptacles inboard, insert GCTA staff into mounting receptacle on right front corner of LRV.	
e.	Rotate staff to assure engagement of staff anti-rotational pins.	
f.	GCTA staff bayonet collar - Lock (CW).	Alignment marks are provided on GCTA staff locking collar.