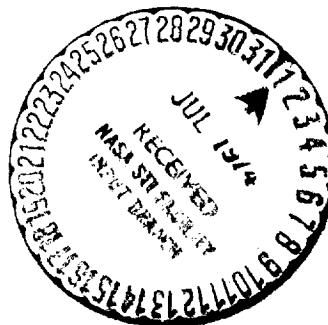


APOLLO 12 MISSION REPORT

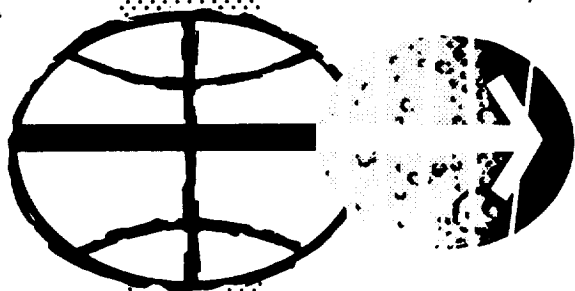
TRAJECTORY RECONSTRUCTION AND POSTFLIGHT ANALYSIS

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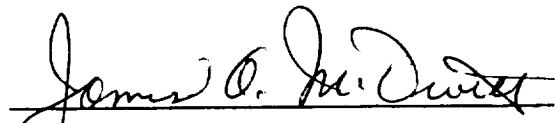
SUPPLEMENT 1

TRAJECTORY RECONSTRUCTION AND POSTFLIGHT ANALYSIS

PREPARED BY

TRW Systems

APPROVED BY


James A. McDivitt
Manager, Apollo Spacecraft Program

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

MANNED SPACECRAFT CENTER

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RECONSTRUCTION AND POSTFLIGHT ANALYSIS
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MANNED SPACECRAFT CENTER
HOUSTON, TEXAS
NAS 9-8166

Approved by

W. P. Girod

W. P. Girod, Manager
MSC/TRW Task A-50

Approved by

J. E. Alexander
for

J. E. Alexander, Manager
Guidance and Control
Systems Department

Approved by

H. L. Moore

H. L. Moore, Manager
MSC/TRW Task A-50

Approved by

D. G. Saile

D. G. Saile, Assistant
Project Manager
Guidance and Performance
Mission Trajectory Control
Program

Approved by

R. K. Petersburg

R. K. Petersburg, Manager
Systems Evaluation
Department

Approved by

R. P. Parten

R. P. Parten, Chief
Mission Planning Support
Office
NASA Manned Spacecraft
Center

TRW
SYSTEMS GROUP

ACKNOWLEDGEMENTS

E. L. Barnett
W. P. Girod
S. C. Jankowski
H. L. Moore
J. B. Moore
J. E. Scheppan
S. L. Smith
R. M. Thomas

FOREWORD

This report is submitted to the NASA Manned Spacecraft Center in accordance with MSC/TRW Task A-50 Contract NAS 9-8166. This report contains the postflight analysis performed in conjunction with the Apollo 12 mission and is issued as Supplement 1 to the Apollo 12 Mission Report.

The report is issued in two volumes. Volume I contains details of the analysis and results obtained, including appendices; Volume II contains a listing of the 45-Day Best Estimated Trajectory (BET) for the Apollo 12 mission in the NASA Apollo Trajectory (NAT) format. The listing is not generally distributed but is available from NASA/MSC upon request. Requests should be made to:

NASA/MSC Computations and Analysis Division
Central Metric Data File
Code ED-5, Building 12, Room 133
Houston, Texas 77058

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SYMBOLS AND NOMENCLATURE

AOS	acquisition of signal
APS	ascent propulsion system
BET	best estimated trajectory
CDH	constant delta height
CM	command module
CSI	coelliptic sequence initiation
CSM	command and service module
DOI	descent orbit insertion
DPS	descent propulsion system
ECI	earth centered inertial
GET	ground elapsed time
GMT	Greenwich mean time
IMU	inertial measurement unit
LGC	LM guidance computer
LLS	lunar landing site
LM	lunar module
LR	landing radar
LOI	lunar orbit insertion
LOPC	lunar orbit plane change
LOS	loss of signal
MCC	midcourse correction
MCI	moon centered inertial
MLR	mean lunar radius
MNBY	mean of nearest Besselian year
MSFN	Manned Space Flight Network
NAT	NASA Apollo Trajectories
PDI	powered descent initiation
PGNCS	primary guidance and navigation control subsystem
RCS	reaction control system
RR	rendezvous radar

Symbols and Nomenclature (Continued)

RTCC	Real-Time Computer Complex
SOI	sphere of influence
SXT	sextant
T&D	transposition and docking
TEI	transearth injection
TLI	translunar injection
TPI	terminal phase initiation
VHF	very high frequency
wrt	with respect to
ΔV	change in velocity caused by thrusting
ΔT	change in time
$\delta R, \delta V$	square root of the sum of the squares of the differences between position or velocity components

5.0 APOLLO 12 MISSION TRAJECTORY RECONSTRUCTION AND POSTFLIGHT REPORT

5.1 INTRODUCTION AND SUMMARY

Apollo 12 was the sixth manned flight of the Apollo series and the second manned lunar landing. The 10-day Apollo 12 mission has contributed and will continue to contribute to a greater scientific understanding of the lunar environment. The crew were Charles Conrad, Jr., Commander; Richard F. Gordon, Command Module Pilot; and Alan L. Bean, Lunar Module Pilot.

The space vehicle was launched from Kennedy Space Center, Florida, with range zero (the integral second before lift-off) occurring at 16:22:00 Greenwich Mean Time (GMT), November 14, 1969. A sequence of events list for the Apollo 12 mission is presented in Table 5.1.

The descent phase of the Apollo 12 mission was initiated on the thirteenth revolution of the moon at approximately 107 hours 54 minutes Ground Elapsed Time (GET). The Lunar Module (LM) successfully landed on the lunar surface at approximately 110 hours 32 minutes GET.

The rendezvous phase began with ascent ignition during Command and Service Modules (CSM) revolution 30 and ended with docking at 145 hours 36 minutes GET. A summary of the CSM and LM maneuvers performed during descent and rendezvous is presented in Table 5.2. Figure 5-1 shows the CSM and LM ground based and onboard tracking data that were available during the descent and rendezvous phases of the Apollo 12 mission.

The objective of the postflight analysis task is, in general, to generate trajectory parameters and data for the CSM and LM vehicles from S-IVB/CSM separation to the end of the mission. During the early Apollo missions this was accomplished by developing a best estimate of trajectory (BET) from available tracking and telemetry data. Comparisons of the BET and the Real-Time Computer Complex (RTCC) state vectors after the early Apollo missions indicated that the RTCC state vectors were of good quality and that, in general, only small differences existed between the two trajectory sources. Consequently, RTCC state vectors were used to generate the preliminary NAT's for Apollo's 9, 10 and 11. It was decided

TABLE 5.1. APOLLO MISSION 12 SEQUENCE OF EVENTS

	GET(h:m:s)	GMT(d:h:m:s)
Range Zero		14:16:22:00.0
Translunar Injection	2:47:22.7	14:19:09:22.7
$\Delta T = 341.3$ $\Delta V = 10,515$	2:53:14.0	14:19:15:14.0
S-IVB/Command Module Separation	3:18:04.9	14:19:40:04.9
First Docking	3:26:53.3	14:19:48:53.3
Spacecraft Ejection	4:13:00.9	14:20:35:00.9
Evasive Manuever (S-IVB APS)	4:28:01.4	14:20:50:01.4
$\Delta T = 80$	4:28:01.4	14:20:50:01.4
Midcourse Correction #1	30:52:44.4	15:23:14:44.4
$\Delta T = 9.2$ $\Delta V = 61.8$	30:52:53.6	15:23:14:53.6
Enter Moon's Sphere of Influence	68:30:00.	17:12:50:00.
Lunar Orbit Insertion #1	83:25:23.4	18:03:47:23.4
$\Delta T = 352.3$ $\Delta V = 2889.5$	83:31:15.7	18:03:53:15.7
Lunar Orbit Insertion #2	87:48:48.1	18:08:10:48.1
$\Delta T = 16.9$ $\Delta V = 165.2$	87:49:05.0	18:08:11:05.0
Undocking	107:54:02.3	19:04:16:02.3
CSM Active Separation	108:24:36.8	19:04:46:36.8
$\Delta T = 14.4$ $\Delta V = 2.4$	108:24:51.2	19:04:46:51.2
Descent Orbit Insertion	109:23:39.9	19:05:45:39.9
$\Delta T = 29$ $\Delta V = 72.4$	109:24:08.9	19:05:46:08.9
Powered Descent Initiation	110:20:38.1	19:06:42:38.1
$\Delta T = 717$		
Touchdown	110:32:36.2	19:06:54:36.2
CSM Plane Change #1	119:47:13.2	19:16:09:13.2
$\Delta T = 18.2$ $\Delta V = 350$	119:47:31.4	19:16:09:31.4
Ascent	142:03:47.7	20:14:25:47.7
$\Delta T = 423.2$ $\Delta V = 6057$		
Insertion	142:10:50.9	20:14:32:50.9
Coelliptic Sequence Initiation	143:01:51.0	20:15:23:51.0
$\Delta T = 41.1$ $\Delta V = 45$	143:02:32.1	20:15:24:32.1
Constant Differential Height	144:00:02.6	20:16:22:02.6
$\Delta T = 13$ $\Delta V = 13.8$	144:00:15.6	20:16:22:15.6
Terminal Phase Initiation	144:36:26.	20:16:58:26.
$\Delta T = 26$ $\Delta V = 29$	144:36:52.	20:16:58:52.
Lunar Docking	145:36:20.2	20:17:58:20.2
Lunar Module Jettison	147:59:31.6	20:20:21:31.6

TABLE 5.1. APOLLO MISSION 12 SEQUENCE OF EVENTS
(Con't)

	GET (h:m:s)	GMT(d:h:m:s)
CSM Separation	148:04:30.9	20:20:26:30.9
$\Delta T = 5.4$ $\Delta V = 1.0$	148:04:36.3	20:20:26:36.3
Lunar Module Deorbit	149:28:14.8	20:21:50:14.8
$\Delta T = 82.1$ $\Delta V = 196.2$	149:29:36.9	20:21:51:36.9
Lunar Module Impact	149:55:16.4	20:22:17:16.4
CSM Plane Change #2	159:04:45.5	21:07:26:45.5
$\Delta T = 19.3$ $\Delta V = 382$	159:05:04.8	21:07:27:04.8
Transearth Injection	172:27:16.8	21:20:49:15.8
$\Delta T = 130.3$ $\Delta V = 3042$	172:29:27.1	21:20:51:27.1
Midcourse Correction #2	188:27:15.8	22:12:49:15.8
$\Delta T = 4.4$ $\Delta V = 2.0$	188:27:20.2	22:12:49:20.2
Midcourse Correction #3	241:21:59.7	24:17:43:59.7
$\Delta T = 5.7$ $\Delta V = 2.4$	241:22:05.4	24:17:44:05.4
CM/SM Separation	244:07:20.1	24:20:29:20.1
Entry Interface	244:22:19.1	24:20:44:19.1

ΔT burn duration in seconds
 ΔV velocity change in feet per second

Table 5.2 DESCENT AND RENDEZVOUS MANEUVER SUMMARY FOR APOLLO 12

Maneuver	Type of Maneuver	Ignition Time (h:M:S) GET	Cutoff Time (h:M:S) GET	T/M Coverage
Separation	CSM/RCS	108:24:36.8	108:24:51.2	Yes
DOI	LM/DPS	109:23:39.9	109:24:08.9	No
PDI	LM/DPS	110:20:38.1	110:32:35.1	Yes
Ascent	LM/APS	142:03:47.7	142:10:50.9	Yes
CSI	LM/RCS	143:01:51	143:02:32.1	No
CDH	LM/RCS	144:00:02.6	144:00:15.6	Yes
TPI	LM/RCS	144:36:26	144:36:52	No

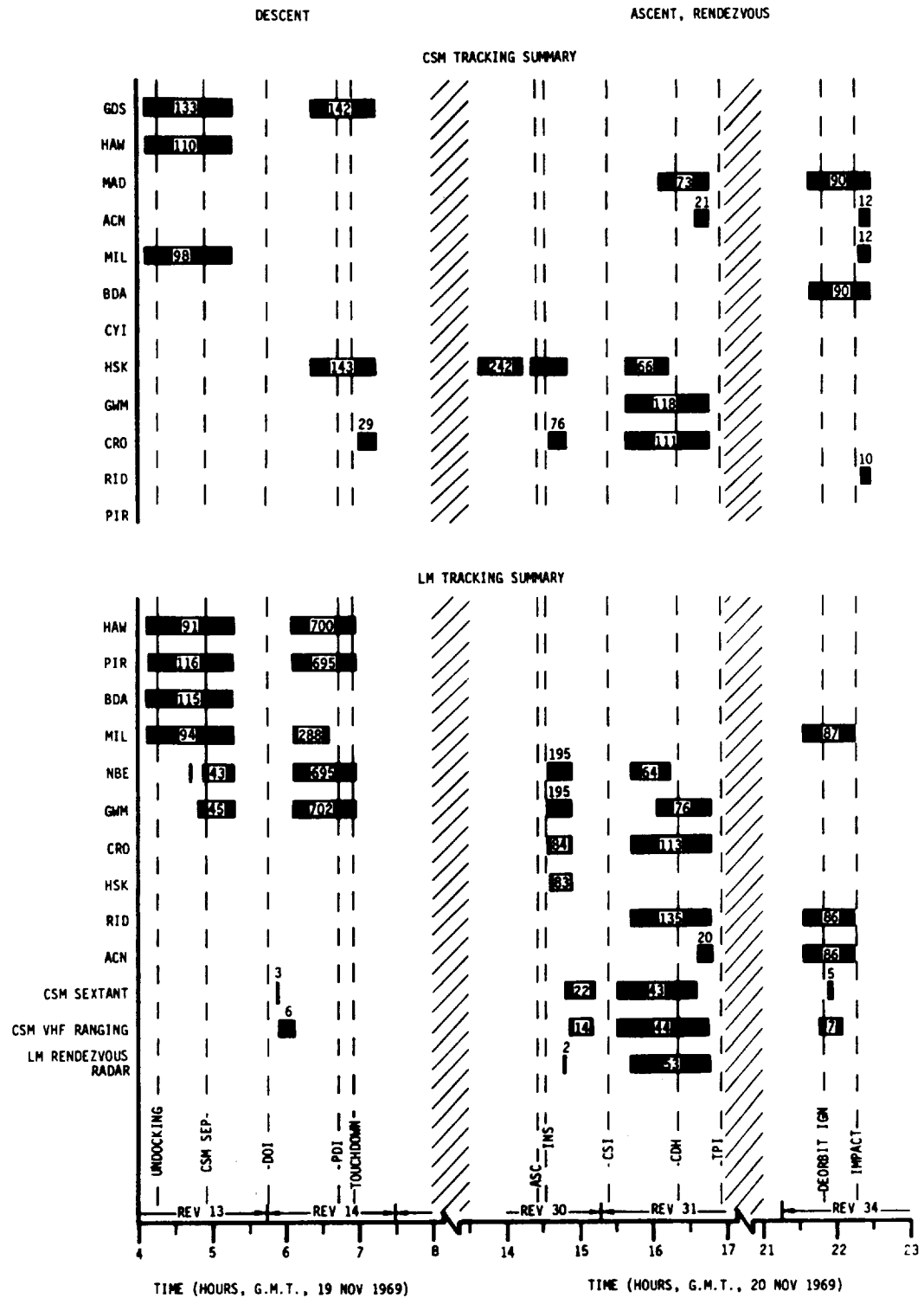


Figure 5-1. Tracking Summary for Descent and Rendezvous

to utilize the RTCC state vectors again for generation of the Apollo 12 preliminary NAT (NASA Apollo Trajectory) and to limit Apollo 12 postflight trajectory reconstruction to the descent and rendezvous phases of the lunar mission. The bulk of the postflight analysis effort was then concentrated on reconstruction of the two periods of flight from LM/CSM undocking to LM touchdown (descent phase) and from LM ascent to LM/CSM docking (rendezvous phase), along with the LM trajectory from deorbit to lunar impact.

The final NAT was produced by updating the preliminary NAT to include reconstructions of critical maneuvers for which telemetered acceleration data was available and to reflect the results of the trajectory reconstruction efforts performed on the descent and rendezvous periods of the mission. These reconstructions will be discussed in detail in Section 5.3.

The mission was essentially nominal and the analysis was carried out in accordance with the postflight analysis plan. Data quality was satisfactory, and no special difficulties were encountered in the trajectory reconstruction.

Table 5.3 PRELIMINARY NAT CSM SEGMENT 1 - POST TLI TO LUNAR SOI

RTCC Vector ID	Propagation Interval Hr:Min:Sec (GET) Start Stop	δR (ft)	δV (ft/sec)	Comments
HSRC 001	02:53:00 04:03:00	57,275.*	7.3*	Post TLI to Ejection
BDAX 074	04:13:00 11:31:00	16,056.	0.5**	Ejection to Water Dump
MILX 163	11:38:00 30:48:00	19,928.	0.4	Water Dump to MCC2
HSKX 236	30:52:52.5 42:18:00	20,419	0.3	MCC2 to 42:18
BDAX 326	42:20:00 68:28:00	21,766	0.2***	42:20 to Lunar SOI

*The HSRC 001 segment is of low quality because HSRC 001 was a TLI cutoff vector which preceded approximately 10 minutes of unmodeled S-IVB venting. Unmodeled T&D and Ejection maneuvers added to the ΔV also.

** ΔV due to the water dump is not modeled.

***Comparison is made with the first propagation interval of the 'CSM Segment 2' BET.

5.2 PRELIMINARY NAT

The CSM preliminary NAT was generated in four segments; the Command Module (CM) preliminary entry NAT in one segment; and the LM preliminary NAT in three BET segments. Each individual segment will be discussed in later sections.

Best Estimated Trajectory (BET) ephemerides for the CSM and LM vehicles were generated from the best RTCC trajectories determined during the mission (RTCC anchor vectors). A preliminary NAT for each vehicle was then formed by propagating and transforming these BET's into several standard Apollo coordinate systems. The LM preliminary NAT was augmented by the inclusion of the lunar powered descent and ascent trajectories which were reconstructed in near real time, and also by the deorbit to lunar impact trajectory.

The various preliminary NAT BET deliverables were generated in the form of magnetic tapes and listings (hard copy and 16 mm film) according to prescribed delivery schedules.

5.2.1 CSM Segment 1 - TLI Burn Cutoff to Lunar SOI

The "CSM Segment 1" free flight BET for the period from TLI burn cutoff to lunar SOI (sphere of influence) was generated from five selected RTCC state vectors which were propagated at 10-minute intervals. A summary of the five trajectory intervals is given in Table 5.3. As a check on the consistency of the segments, the RSS position and velocity differences (δR , δV) were computed at a common epoch for successive trajectory segments. Unless otherwise noted, δV has been corrected for known thrust velocity increments.

5.2.2 CSM Segment 2 - Lunar SOI to LOPC-1

The "CSM Segment 2" free flight BET for the period from lunar SOI to lunar orbit plane change-1 (LOPC-1) was generated from 19 selected RTCC state vectors. The state vectors were propagated at 1-minute intervals in lunar orbit and at 10-minute intervals during the translunar coast prior to LOI-1. As an indicator of the consistency of the 19 trajectory segments, the RSS position and velocity differences (δR and δV (corrected for thrust velocity across burns)) are computed at a common epoch for successive segments. The 19 free flight intervals are defined in Table 5.4.

NOTE: The selenographic orbit inclination in this segment (data word No. 56) was improperly coded in the NAT program and should be ignored.

Table 5.4 PRELIMINARY NAT CSM SEGMENT 2 - LUNAR SOI TO LOPC-1

RTCC Vector ID	Propagation Interval Hr:Min:Sec (GET)		δR (ft)	δV (ft/sec)	Comments
	Start	Stop			
MILX 473	68:51:00	83:25:22.7*	----	----	Lunar SOI to LOI-1
HSKX 497	83:31:14.7	85:34:00	892	1.6	Rev 1
GMMX 504	85:35:00	87:48:47.4**	3017	1.0	Rev 2
HAWX 519	87:49:04.4	89:39:00	1773	1.8	Rev 3
HSKX 524	89:40:00	91:39:00	2146	1.7	Rev 4
HSKX 530	91:40:00	93:39:00	4906	9.1	Rev 5
NBEX 541	93:40:00	95:34:00	3366	5.0	Rev 6
ACNX 550	95:35:00	97:34:00	3701	2.9	Rev 7
BDAX 556	97:35:00	99:34:00	4087	3.3	Rev 8
CYIX 564	99:35:00	101:29:00	4198	3.8	Rev 9
GDSX 575	101:30:00	103:29:00	2763	6.9	Rev 10
GDSX 586	103:30:00	105:24:00	8494	3.0	Rev 11
HAWX 595	105:25:00	107:24:00	4750	4.4	Rev 12
NBEX 128	107:25:00	108:23:00	2326	2.3	Rev 13 to Sep. Sep. Rev 13 through Rev 14
CROX 609	108:24:00	111:19:00	2886	4.3	Rev 15
CROX 614	111:20:00	113:19:00	3269	4.0	Rev 16
CROX 618	113:20:00	115:19:00	5534	6.0	Rev 17 (short)
CROX 621	115:20:00	116:45:00	7086	3.3	Rev 18 to LOPC-1
HSKX 624	116:46:00	119:47:12.5	6617	1.1	

*LOI-1 ignition (ignore BET time points between ignition and cutoff)

**LOI-2 ignition (ignore BET time points between ignition and cutoff)

5.2.3 CSM Segment 3 - LOPC-1 to Lunar SOI

The "CSM Segment 3" free flight BET for the period from LOPC-1 to lunar SOI was generated from 27 selected RTCC state vectors which were propagated at 1-minute intervals in lunar orbit and at 10-minute intervals during the transearth coast. As a check on the consistency of these vectors, the RSS position and velocity differences (δR , δV) were computed at a common epoch for successive trajectory segments.

The 27 propagation intervals are summarized in Table 5.5.

5.2.4 CSM Segment 4 - Lunar SOI to Entry Interface

The "CSM Segment 4" free flight BET for the period from lunar SOI to entry interface was generated from seven selected RTCC state vectors which were propagated at 10-minute intervals. A summary of the seven trajectory propagation intervals is given in Table 5.6. As a consistency check, the RSS position and velocity differences (δR , δV) are computed at a common epoch for successive segments. Only the ΔV for MCC 2 and MCC 3 are accounted for. The segments do not match as well as in the translunar phase because of unmodeled thrust incurred during numerous attitude maneuvers, water dump, fuel cell purge, and CM/SM separation.

The GWMS 147 vector yields an entry interface (Geodetic Altitude = 400,000 ft) time of 244:22:19.09 GET. Selected trajectory parameters at entry interface are as follows:

Inertial velocity	36116.618	ft/sec
Inertial flight path angle	-6.4834	deg
Inertial heading	98.1699	deg
Geodetic latitude	-13.7947	deg
Geodetic longitude	173.5279	deg

5.2.5 CM Segment 1 - Preliminary Entry

The "CM Segment 1" preliminary entry BET was reconstructed at 2-second intervals from PIPA acceleration data, using the GWMX 164 state vector (determined by RTCC) for initial conditions. The GWMX 164 vector was propagated to 244:22:25.59 GET to initialize the BET. The initial state in ECI (mean of NBY) coordinates at this time is:

$X = 20,431,408.8$ ft	$\dot{X} = 10,403.059$ ft/sec
$Y = -3,193,006.5$	$\dot{Y} = -34,386.006$
$Z = -5,077,174.6$	$\dot{Z} = -3,937.787$

Geodetic Altitude = 371,820.4 ft

Table 5.5 PRELIMINARY NAT CSM SEGMENT 3 - LOPC-1 TO LUNAR SOI

RTCC Vector ID	Propagation Interval Hr:Min:Sec (GET)		δR (ft)	δV (ft/sec)	Comments
	Start	Stop			
ACNX 634	119:47:30.7	121:14:00	3571	3.4	LOPC-1 through Rev 19
ACNX 637	121:15:00	123:14:00	1949	4.2	Rev 20
MILX 641	123:15:00	125:09:00	2612	1.8	Rev 21
MILX 644	125:10:00	127:09:00	3399	2.5	Rev 22
GDSX 650	127:10:00	129:09:00	3590	5.3	Rev 23
HAWX 654	129:10:00	131:04:00	2816	2.0	Rev 24
HAWX 659	131:05:00	133:04:00	2838	2.4	Rev 25
HSKX 663	133:05:00	134:59:00	2982	2.6	Rev 26
CROX 668	135:00:00	136:59:00	2593	4.2	Rev 27
HAWX 672	137:00:00	138:59:00	3579	3.8	Rev 28
HSKX 678	139:00:00	140:54:00	1731	1.4	Rev 29
CROX 680	140:55:00	142:54:00	1765	2.4	Rev 30
ACNX 687	142:55:00	145:36:00	----	3.8	Rev 31 to Dock
MADX 700	145:37:00	148:49:00	6015	5.7	Dock through Rev 33
MADX 702	148:50:00	150:49:00	2365	6.9	Rev 34
GDSX 712	150:50:00	152:44:00	2967	2.1	Rev 35
HAWX 719	152:45:00	154:44:00	3813	3.4	Rev 36
ACNX 725	154:45:00	156:39:00	3451	3.4	Rev 37
GWMX 731	156:40:00	159:04:44.8	----	4.4	Rev 38 to LOPC-2
GWMX 747	159:05:04.0	160:39:00	5686	6.3	LOPC-2 through Rev 39
GDSX 764	160:40:00	162:34:00	6173	2.4	Rev 40
HAWX 771	162:35:00	164:34:00	1573	2.2	Rev 41
CROX 776	164:35:00	166:34:00	1333	1.2	Rev 42
RIDX 783	166:35:00	168:29:00	1834	1.6	Rev 43
ACNX 794	168:30:00	170:29:00	4538	5.4	Rev 44
BDAX 800	170:30:00	172:27:16.1	----	----	Rev 45 to TEI
HSKX 866	172:29:26.1	186:28:00	----	----	TEI to Lunar SOI

Table 5.6 PRELIMINARY NAT CSM SEGMENT 4 -
LUNAR SOI TO ENTRY INTERFACE

RTCC Vector ID	Propagation Interval Hr:Min:Sec (GET)		δR (ft)	δV (ft/sec)	Comments
	Start	Stop			
HSKX 855	186:38:00	188:18:00	4832	0.4	Lunar SOI to MCC 5
HAWX 881	188:27:18.23	193:48:00	62999	1.2	P23 attitude maneuvers
MADX 909	193:58:00	200:58:00	40980	1.1	Urine dump @ \approx 195:45
HSKX 960	201:08:00	217:18:00	22023	1.6	P23 @ \approx 213:00
MADX 012	217:28:00	223:28:00	39939	1.3	Water dump & fuel cell purge @ \approx 217:30
NBEX 127	223:38:00	241:21:47.5	16838	1.4	Nominal MCC 7 time
GWMS 147	241:30:00	244:22:19.09			Post MCC 7 to Entry Interface

The IMU acceleration data has been corrected for a platform misalignment of 167.7 sec about the Y-axis. This is a "single error fit" which causes the altitudes at drogue deploy, main deploy, and splashdown to be near the nominally expected values and does not necessarily represent the accuracy of the pre-entry P52 alignment.

The coordinates of the spacecraft at drogue deployment are compared with the targeted values and the onboard navigated values, and they are as follows:

Drogue Deploy, 244:30:39.7 GET

	<u>Latitude (deg)</u>	<u>Longitude (deg)</u>
BET	-15.831	-165.168
Target	-15.8125	-165.1740
Onboard Nav.	-15.836	-165.171

Altitudes at drogue deploy, main deploy, and splashdown, and mean descent rate on the main chute are compared with the pre-mission nominal values in Table 5.7.

5.2.6 LM Segment I - Undock to PDI and Insertion to Impact

The "LM Segment I" preliminary free flight BET, covering undock to PDI and insertion to lunar impact, was generated from seven selected RTCC state vectors which were propagated at one-minute intervals. As a check on the consistency of these vectors, the RSS velocity difference, δV , is computed at a common epoch for successive trajectory segments. The seven trajectory segments are summarized in Table 5.8.

The early (real-time) estimate of the Ascent stage impact time was 149:55:15.76 GET. The selenographic coordinates of the MILX 279 vector at this time are:

LAT = -3.9549 deg LONG = -21.1609 deg R = 937.7784 nm

Table 5.7 COMPARISON OF DROGUE DEPLOY, MAIN DEPLOY,
AND SPLASHDOWN ALTITUDES AND THE MAIN CHUTE
MEAN DESCENT RATE WITH PRE-MISSION NOMINAL
VALUES

	BET	Pre-Mission Nominal
Drogue Deploy Altitude 244:30:39.7	23,735 ft	23,300 {+1600 ft -1600 ft
Main Deploy Altitude 244:31:30.2	10,892 ft	10,500 {+1000 ft -1050 ft
Splashdown - 27.4 sec, Altitude 244:35:57.59*	805 ft	770 ft
Mean Descent Rate of Main Chute	31.2 ft/sec	28 ft/sec

*The DSE ran out at this time - nominal altitude is based on
nominal descent rate.

Table 5.8 PRELIMINARY NAT LM SEGMENT 1 -
FREE FLIGHT BET

RTCC Vector ID	Propagation Interval		δV (ft/sec)	Comments
	Hr:Min:Sec Start	Hr:Min:Sec (GET) Stop		
NBEX 128	107:45:00	109:23:39.4	4.6	Undock to DOI
PIRX 159	109:24:08.4	110:20:37		DOI to PDI
GWMX 234	142:11:48	142:01:40.6		Insertion to CSI
RIDX 245	143:02:31.7	144:00:01.5	6.8	CSI to CDH
ACNX 256	144:00:14.0	144:36:29.4	0.4	CDH to TPI
RIDX 273	147:29:00	149:24:00	6.0	Jettison to Deorbit
MTLX 279	149:26:02.4	149:55:15.76		Deorbit to Impact

5.2.7 LM Segment 2 - Real Time Powered Descent

The "LM Segment 2" Real Time LM Powered Descent BET was reconstructed at two-second intervals from LM IMU acceleration measurements transmitted through the MSFN communication network. Initial conditions prior to ullage were obtained from the best rev 14 trajectory determined by the RTCC.

The quality of the LGC downlink data was generally very good, in that only a few isolated dropouts occurred.

Altitudes above the lunar surface are computed with respect to the pre-flight estimate of the radius to Surveyor III. BET altitudes may be adjusted to the current best estimate of the LLS radius by subtracting 924 feet.

Indicated velocities relative to the lunar surface after landing are:

$$V_x \text{ (vertical)} = -0.321 \text{ ft/sec}$$

$$V_y \text{ (north)} = -3.070 \text{ ft/sec}$$

$$V_z \text{ (west)} = 1.500 \text{ ft/sec}$$

The most probable causes of the -3.070 feet per second error in North direction are platform X-axis misalignment and out-of-plane errors in the rev 14 orbit. Correction for these types of errors will move the landing point approximately 2000 - 3000 feet to the North. Correction of the small westward velocity error will move the landing point approximately 1000 feet to the East. The net result of these corrections is to move the BET close to the real-time best estimate of the landing site.

Estimates of LLS coordinates from several sources are shown in Table 5.9.

5.2.8 LM Segment 3 - Real Time Ascent

The "LM Segment 3" Real Time LM Ascent BET, at two-second intervals, was initialized at 142:03:23.78 GET (23.22 sec before ignition) using the current best estimate of LLS coordinates.

Table 5.9 PRELIMINARY ESTIMATES OF LLS COORDINATES

Source of Estimate	Latitude (deg)	Longitude (deg)	Radius (n mi)
BET	-3.0442	-23.4286	937.31188 (-7179.8 ft wrt MLR) *
Best Estimate (real time - Combination of LM P22 tracking and CSM P22 sightings)	-3.036	-23.418	937.3643 (-6861.3 ft wrt MLR)
P57A/T 2 (Least square fit of two alignments)	-3.069	-23.414	

*wrt - with respect to
MLR - mean lunar radius

LAT = -3.036 deg LONG = -23.418 deg R = 937.3643 nm

The trajectory was terminated at 142:12:15.78 GET (approximately 26 seconds after conclusion of the trim maneuver). The total thrust velocity accumulated by the PGNCs (uncorrected) between 142:03:45.78 and 142:11:49.78 is given below in platform coordinates:

$\Delta V_x = 1326.776 \text{ ft/sec}$ $\Delta V_y = -1.064 \text{ ft/sec}$ $\Delta V_z = -5590.449 \text{ ft/sec}$

Selected orbit insertion parameters are listed below:

Perilune Altitude wrt LLS = 8.867 nm

Apolune Altitude wrt LLS = 47.52 nm

Selenographic Orbit Inclination = -14.568 deg

5.3 FINAL NAT

The final NAT was produced by updating the preliminary NAT to include reconstructions of critical maneuvers for which telemetered acceleration data was available and to reflect the results of the trajectory reconstruction efforts performed on the descent and rendezvous periods of the mission. These reconstructions will be discussed in detail later. Note that the preliminary NAT serves as the final NAT for those periods where no update was made.

5.3.1 CSM LOI-1 and TEI Burn Trajectory Reconstructions

Postflight trajectory reconstruction of the LOI-1 and TEI burns were delivered to MSC on 13 December 1969. These trajectories were generated to satisfy the special request from North American Rockwell to the MPSO Post-flight Trajectory Office, and were not generally distributed.

5.3.1.1 LOI-1 Burn

For the pre-burn comparison, the RTCC vector (MILX 473) was propagated in the HOPE Program using the L1 potential model. The RTCC propagated vector, time tagged 18 November 1969, 03:47:23.40 GMT (83:25:23.40 GET), was compared to the BET vector which was reconstructed, using the HOPE Program, from MSFN data covering a time interval of 870 minutes starting at 17 November 1969, 13:00:00 GMT.

For the post-burn comparisons, the RTCC vector (HSKX 497) was propagated in the HOPE Program using the L1 potential model. The RTCC propagated vector, time tagged 18 November 1969, 03:53:19.40 GMT (83:31:19.40 GET), was compared to the BET vector which was reconstructed from PIPA acceleration data using the pre-burn BET vector for initial conditions.

Likewise, a rev 1 trajectory was reconstructed with the HOPE Program using MSFN data covering a time interval of 85 minutes starting at 18 November 1969, 04:06:00 GMT. This was compared to the BET vector shortly after burn cutoff.

The results of these comparisons are tabulated in Table 5.10.

5.3.1.2 TEI-Burn

No pre-burn comparison was made.

For the post-burn comparison, the RTCC vector (HSKX 866) was propagated in the HOPE Program using the L1 potential model. The RTCC propa-

Table 5.10. LOI-1 AND TEI POWERED FLIGHT TRAJECTORIES

	Selenographic			Altitude Ft	Moon Inertial		Inclination Deg	Right Ascension Node Deg	Argument Perigee Deg	Radius	
	Latitude Deg	Longitude Deg	Path Angle Deg		Heading Deg	Velocity fps				Apolune N. Mi.	Perilune N. Mi.
PRE-LOI-1											
BET MSFN FIT	5.7566	175.6237	-8.4476	502963.1	229.3447	8173.587	---	---	---	---	1002.2976
RTCC	5.7438	175.6053	-8.4365	501438.0	229.3475	8174.511	---	---	---	---	1002.1005
Δ	+ .0128	+ .0184	- .0111	+ 1525.1	- .0028	- .924	---	---	---	---	+ .1971
POST LOI-1											
BET PIPA	-1.6900	153.8481	- .5918	374824.8	239.3104	5469.997	145.1314	163.6389	223.8094	1107.0664	999.0792
RTCC	-1.6702	153.8502	- .6203	375030.6	239.3549	5470.321	145.1770	163.6594	224.3505	1107.5601	999.0073
Δ	- .0198	- .0021	+ .0285	- 205.8	- .0445	- .324	- .0456	- .0205	- .5411	- .4937	+ .0719
BET PIPA	-1.6900	153.8481	- .5918	374824.8	239.3104	5469.997	145.1314	163.6389	223.8094	1107.0664	999.0792
MSFN	-1.6803	153.8493	- .6205	375122.2	239.3416	5470.263	145.1612	163.6620	224.3606	1107.5635	999.0217
Δ	- .0097	- .0012	+ .0287	- 297.4	- .0312	- .266	- .0298	- .0231	- .5512	- .4971	+ .0575
POST TEI											
BET PIPA	7.1977	176.4330	5.6417	441203.7	242.3095	8327.470	---	---	---	---	1001.7405
RTCC	7.1780	176.4092	5.6630	439891.5	242.2076	8328.854	---	---	---	---	1001.4603
Δ	+ .0197	+ .0238	- .0213	+ 1312.2	+ .1019	- 1.384	---	---	---	---	.2802

NOTE: All RTCC vectors referred to here are those used in generation of the Preliminary Free Flight BET.
Landing Site Radius is 937.3643 n.mi.

gated vector, time tagged 21 November 1969, 20:52:30.86 GMT (172:52:30.86 GET) was compared to the BET vector which was reconstructed from PIPA acceleration data using the propagated RTCC vector (BDAX 800) for initial conditions. The results are tabulated in Table 5.10.

5.3.2 LM Powered Descent

The "Final LM Powered Descent BET" update was delivered earlier than planned in response to requests for an improved surface relative trajectory during the visibility phase. The descent trajectory was initialized from the PIRX 159 vector determined by the RTCC and downrange/crossrange position errors are evident. Rev 14 trajectory determination subsequent to delivery of the BET improved the accuracy of the local solutions near PDI, causing the position errors at landing to decrease significantly. The relative trajectory, however, changed only slightly, and no further update to the BET was made.

The final version of the LM Powered Descent BET was initialized at 110:20:10.19 GET (2 seconds after Average G On). Initial conditions were taken from the PIRX 159 trajectory determined by the RTCC. The trajectory was reconstructed by integrating two-second IMU acceleration measurements with alignment corrections needed to satisfy the landing constraints.

5.3.2.1 IMU Corrections and Trajectory Constraints

In order to satisfy the velocity constraint at landing, the following set of alignment errors were used:

<u>Error</u>	<u>Magnitude</u>
PHIX (platform misalignment about X)	157.2 $\widehat{\text{sec}}$
PHIY (platform misalignment about Y)	-110.0 $\widehat{\text{sec}}$
XZMSL (X PIPA misalignment toward Z)	-53.8 $\widehat{\text{sec}}$
ZXMSL (Z PIPA misalignment toward X)	-59.5 $\widehat{\text{sec}}$

The "Best Estimate" of the coordinates of the Lunar Landing Site is derived from rev 15 rendezvous radar tracking of the CSM and rev 16 SXT sightings on the LM. The LM Ascent BET and the postflight processing of lunar surface alignment (P57) data are in close agreement with the "Best Estimate".

The coordinates of the Lunar Landing Site indicated by the BET are compared to the "Best Estimate" as follows:

LLS COORDINATES

<u>Source</u>	<u>Latitude</u>	<u>Longitude</u>	<u>Altitude</u> (With respect to MLR)
Powered Descent BET	-3.027 deg	-23.426 deg	-6354. ft
Best Estimate	-3.043 deg	-23.416 deg	-6861. ft

The unit error sensitivities of the state vector (in platform coordinates) at landing with respect to PHIX and PHIY (in sec) are as follows: Units are feet and feet per second.

	$\frac{\Delta X_{LLS}}{\Delta X_{LLS}}$	$\frac{\Delta Y_{LLS}}{\Delta Y_{LLS}}$	$\frac{\Delta Z_{LLS}}{\Delta Z_{LLS}}$	$\frac{\Delta \dot{X}_{LLS}}{\Delta \dot{X}_{LLS}}$	$\frac{\Delta \dot{Y}_{LLS}}{\Delta \dot{Y}_{LLS}}$	$\frac{\Delta \dot{Z}_{LLS}}{\Delta \dot{Z}_{LLS}}$
PHIX	0	-11.4	0	0	-0.025	0
PHIY	12.8	0	2.1	0.033	0	0.011

The unit error sensitivities of the state vector (platform coordinates) at landing with respect to the initial state (near PDI) are as follows: Units are feet and feet per second.

	$\frac{\Delta X_{LLS}}{\Delta X_{LLS}}$	$\frac{\Delta Y_{LLS}}{\Delta Y_{LLS}}$	$\frac{\Delta Z_{LLS}}{\Delta Z_{LLS}}$	$\frac{\Delta \dot{X}_{LLS}}{\Delta \dot{X}_{LLS}}$	$\frac{\Delta \dot{Y}_{LLS}}{\Delta \dot{Y}_{LLS}}$	$\frac{\Delta \dot{Z}_{LLS}}{\Delta \dot{Z}_{LLS}}$
ΔX_{PDI}	1.54	0	-0.09	1.60 E-3	0	0
ΔY_{PDI}	0	0.75	0	0	-0.63 E-3	0
ΔZ_{PDI}	-0.11	0	0.77	0	0	-0.61 E-3
$\Delta \dot{X}_{PDI}$	885.1	0	-11.5	1.57	0	-0.02
$\Delta \dot{Y}_{PDI}$	0	686.7	0	0	0.75	0
$\Delta \dot{Z}_{PDI}$	-12.9	0	688.3	-0.03	0	0.75

The position error in the LLS indicated by the BET is thus seen to be within reasonably expected error bounds for the PIRX 159 trajectory at PDI.

Two-second PIPA counts in the vicinity of landing are shown in Figure 5-2. The impulse caused by impact is evident in the interval 110:32:36.19 - 110:32:38.19 GET, indicating that the vehicle became stationary within this period. Altitude and descent rate from low gate to landing are shown in Figure 5-3, and the groundtrack is shown in Figure 5-4.

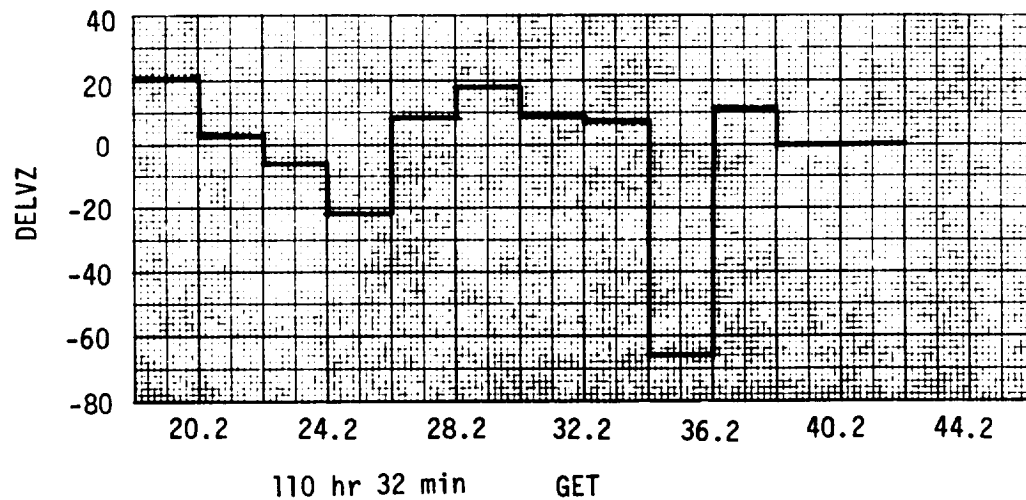
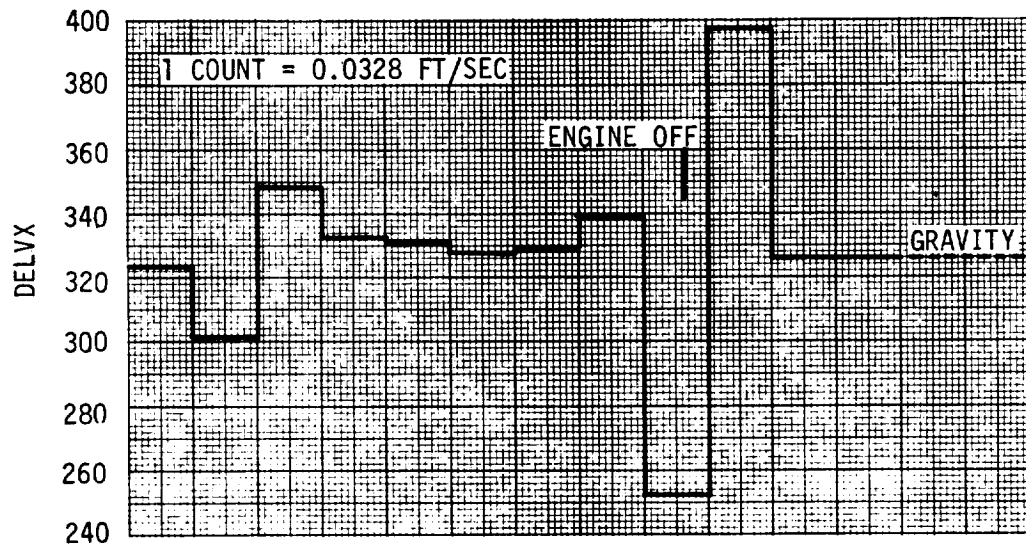


Figure 5-2. Two-Second PIPA Counts Near Landing Time

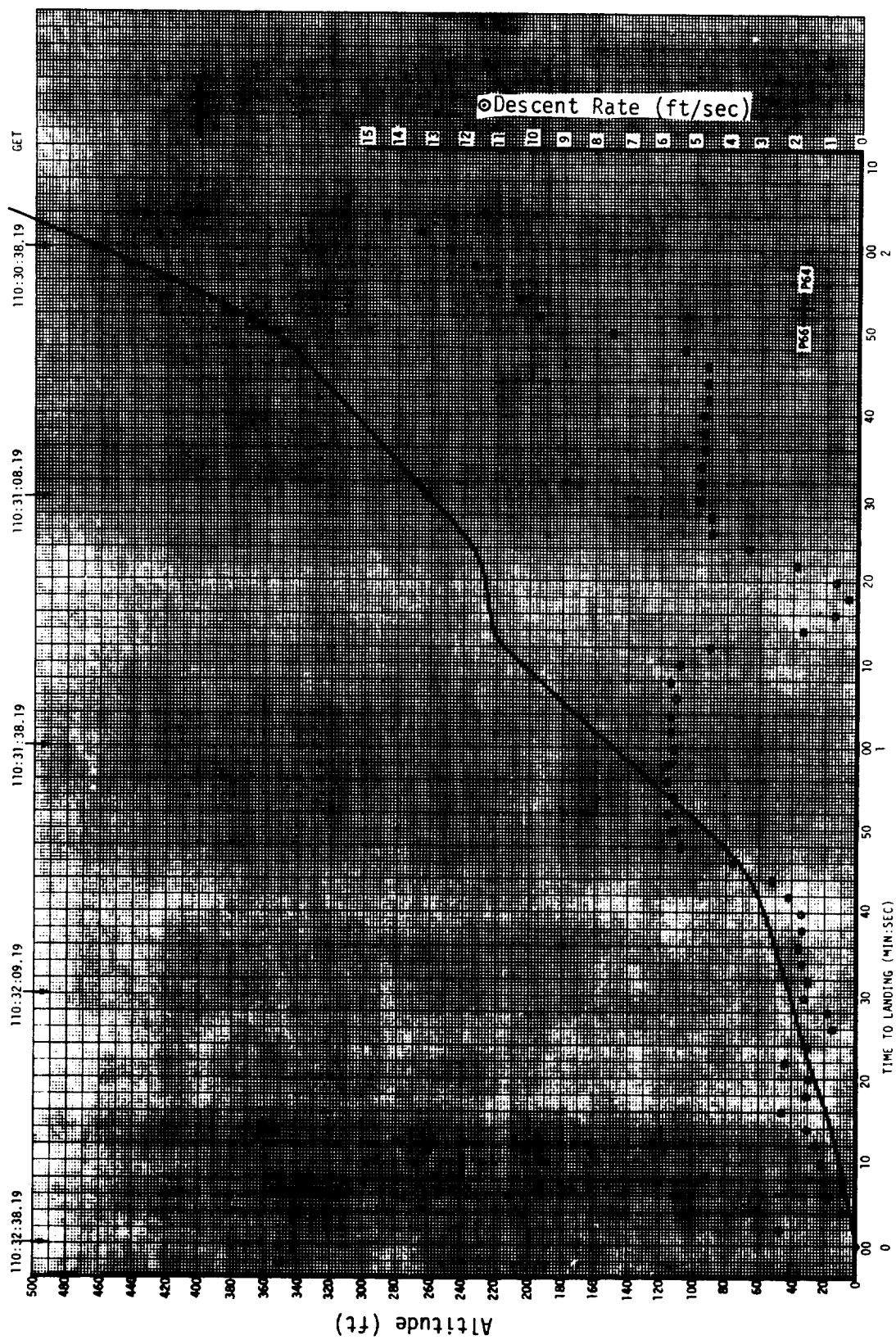


Figure 5-3. Altitude and Descent Rate: Low Gate to Landing

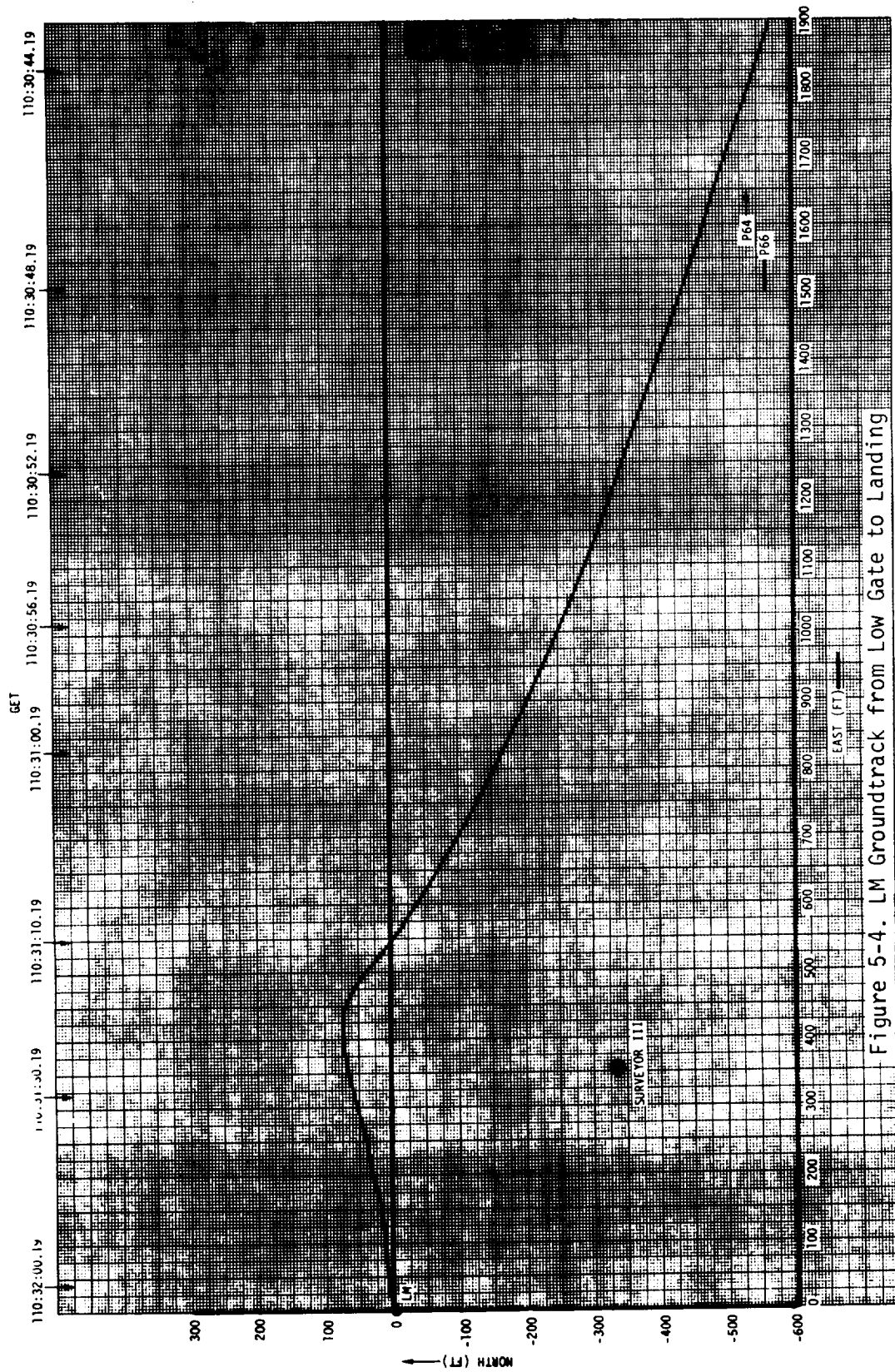


Figure 5-4. LM Groundtrack from Low Gate to Landing

5.3.2.2 Rev 14 Orbit Determination

Subsequent to the delivery of the BET, landing radar and high speed doppler data have been included in the HOPE Program orbit determination. The results are believed to indicate an improved local solution near PDI for the following reasons: 1) lower high speed doppler residuals; 2) LLS coordinates are closer to the "Best Estimate". The results of this further analysis of the LM powered descent trajectory is presented in Section 5.4.

5.3.3 LM Ascent and Insertion to CSI Trajectories

The postflight reconstruction of the LM trajectory from liftoff to CSI was delivered to MSC on 13 January 1969. This iteration superceeds the earlier (real time) version described in Section 5.2.8.

5.3.3.1 Ascent Trajectory

The powered flight trajectory (liftoff to orbit insertion) was reconstructed by integrating IMU acceleration data with corrections for known errors. The initial conditions were taken from the best available estimate of the LLS coordinates.*

LAT = -3.036 deg LONG = -23.418 deg R = 937.3643 nm

The powered flight reconstruction begins at 142:03:23.78 GET (APS ignition is 142:03:47.68) and ends at 142:11:51.78 (about 6 seconds after completion of RCS trim). Accumulated thrust velocities in platform coordinates (obtained from corrected IMU data) are as follows:

<u>GET</u>	<u>ΔV_X</u> <u>ft/sec</u>	<u>ΔV_Y</u> <u>ft/sec</u>	<u>ΔV_Z</u> <u>ft/sec</u>
142:03:47.78 (APS Ignition)	0	0	0
142:11:01.78 (APS Off)	1306.53	-5.66	5625.49
142:11:51.78 (End RCS Trim)	1313.12	-4.48	5595.18

*At the time of writing, the coordinates of the LLS were revised to
LAT = -3.043 deg, LONG = -23.416 deg, R = 937.365 nm (Reference 1).

IMU Errors: The following corrections were made to the telemetered IMU acceleration data:

Accelerometer	BX	=	0.15 cm/sec ²	(153 μ g)
Bias	BY	=	0.20 cm/sec ²	(204 μ g)
	BZ	=	-0.29 cm/sec ²	(-296 μ g)
Platform	PHIY	=	-21.6 $\frac{\text{sec}}{\text{sec}}$	
Misalignment	PHIZ	=	-43.2 $\frac{\text{sec}}{\text{sec}}$	

The PIPA bias changes are computed from free orbit data on rev 14 (prior to PDI) and rev 30 (after insertion). The bias change is believed to be a result of removing power to the IMU.

Platform misalignments prior to liftoff have been estimated from lunar surface alignment (P57) data (Reference 14). The values obtained are as follows:

<u>Misalignment Angle</u>	<u>Mean</u>	<u>RMS Uncertainty</u>
ϕ_X	-0.013 deg	0.025 deg
ϕ_Y	-0.010 deg	0.025 deg
ϕ_Z	-0.007 deg	0.025 deg

The misalignments used in the BET were chosen so as to produce best agreement of insertion conditions with the free flight trajectory. They are well within the 1 σ uncertainties of the P57 estimates.

5.3.3.2 Comparison of Insertion Conditions

Comparison of the powered flight trajectory reconstruction and the free flight trajectory near orbit insertion is given in Table 5.11. The time of the comparison is 142:11:51.77 GET.

The inertial platform is aligned such that X \approx vertical, Y \approx cross-range, Z \approx downrange at liftoff. The sensitivities of the state vector at orbit insertion to the coordinates of the LLS are given in Table 5.12. All quantities are in feet and feet per second in platform coordinates.

It can be seen that the crossrange position difference will become insignificant if the LLS is moved approximately 1000 feet (.01 deg) to the South. This is consistent with the revised estimates of the LLS coordinates.

Table 5.11 COMPARISON OF POWERED FLIGHT AND FREE FLIGHT
TRAJECTORY CONDITIONS NEAR INSERTION (142:11:51.77 GET)

Trajectory Parameter	Powered Flight Trajectory	Free Flight Trajectory	Delta (Powered Flight-F.F.)
X	5604697.1 ft	5604377.1 ft	320 ft
Y	736.1 ft	-293.3 ft	1029 ft
Z	1317309.3 ft	1316757.5 ft	552 ft
V_X	-1227.864 ft/sec	-1227.995 ft/sec	0.13 ft/sec
V_Y	-1.051 ft/sec	-0.878 ft/sec	-0.17 ft/sec
V_Z	5401.836 ft/sec	5403.272 ft/sec	-1.44 ft/sec
Perilune Altitude (above LLS)	8.9513 nm	8.9183 nm	199 ft
Apolune Altitude (above LLS)	50.9877 nm	51.7589 nm	-4696 ft
Selenographic Orbit Inclination	165.4679 deg	165.4698 deg	-.0019 deg
Inertial Flight Path	0.4205 deg	0.4179 deg	.0026 deg
Inertial Heading	-55.3418 deg	-55.3403 deg	-.0015 deg

Table 5.12 SENSITIVITIES OF THE STATE VECTOR AT ORBIT
INSERTION TO THE COORDINATES OF THE LLS

	ΔX_{INS}	ΔY_{INS}	ΔZ_{INS}	$\Delta \dot{X}_{INS}$	$\Delta \dot{Y}_{INS}$	$\Delta \dot{Z}_{INS}$
ΔX_{LLS}	1.250	0	0.012	0.0010	0	0.0001
ΔY_{LLS}	0	0.882	0	0	-0.0005	0
ΔZ_{LLS}	0.010	0	0.883	0	0	-0.0004

Units are in ft and ft/sec

5.3.3.3 Free Flight Trajectory Insertion to CSI

The free flight LM trajectory from insertion to CSI was determined from 557 MSFN doppler observations, 22 SXT sightings, 14 VHF ranging observations, and 2 rendezvous radar marks. The converged residual statistics are summarized as follows:

<u>Station</u>	<u>Type</u>	<u>No. Obs.</u>	<u>Mean</u>	<u>Sigma</u>
HSK	MSFN	83	.070 cps	.477 cps
GWM	MSFN	195	-.027 cps	.558 cps
NBE	MSFN	195	-.018 cps	.555 cps
CRO	MSFN	84	.044 cps	.540 cps
CSM	SXT Shaft	22	-.002 deg	.022 deg
CSM	SXT Trunnion	22	-.006 deg	.007 deg
CSM	VHF Ranging	14	-232. ft	200. ft
LM	RR Range	2	323. ft	13. ft
LM	RR Range Rate	2	-.380 ft/sec	.157 ft/sec
LM	RR Shaft	2	.015 deg	.020 deg
LM	RR Trunnion	2	-.043 deg	.020 deg

Plots of the SXT and VHF ranging residuals are shown in Figures 5-5, 5-6, and 5-7.

The purpose of reconstructing the free flight trajectory is to obtain insertion conditions as accurately as possible. Since the bulk of the relative observations occur after rev 30 LOS, the MSFN data dominates the fit. This is reflected in the VHF range residuals which show a clearly defined trend. As a check on the quality of the onboard data, a fit was made without the MSFN. The residual statistics show substantial improvement:

<u>Data Type</u>	<u>Mean</u>	<u>Sigma</u>
SXT Shaft	-.002 deg	.022 deg
SXT Trunnion	.0003 deg	.004 deg
VHF Range	-37.8 ft	58.3 ft

The two trajectories agree very well just before LOS, since this region is included in both the MSFN and onboard data spans. Comparison of MCI state vectors (Relative - BET) at 142:25:00 GET yields:

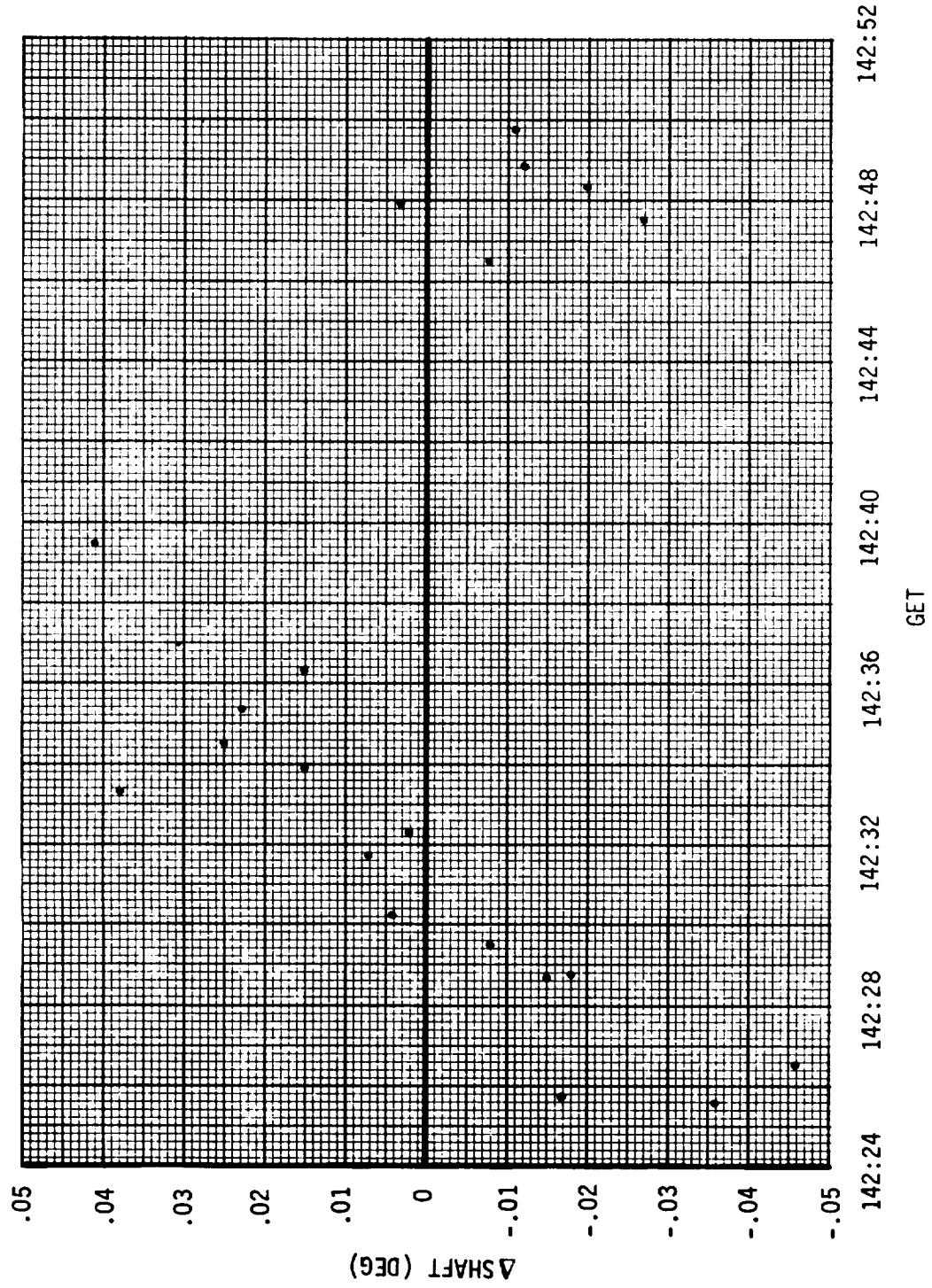


Figure 5-5. Converged SXT Shaft Residuals - Insertion to CSI

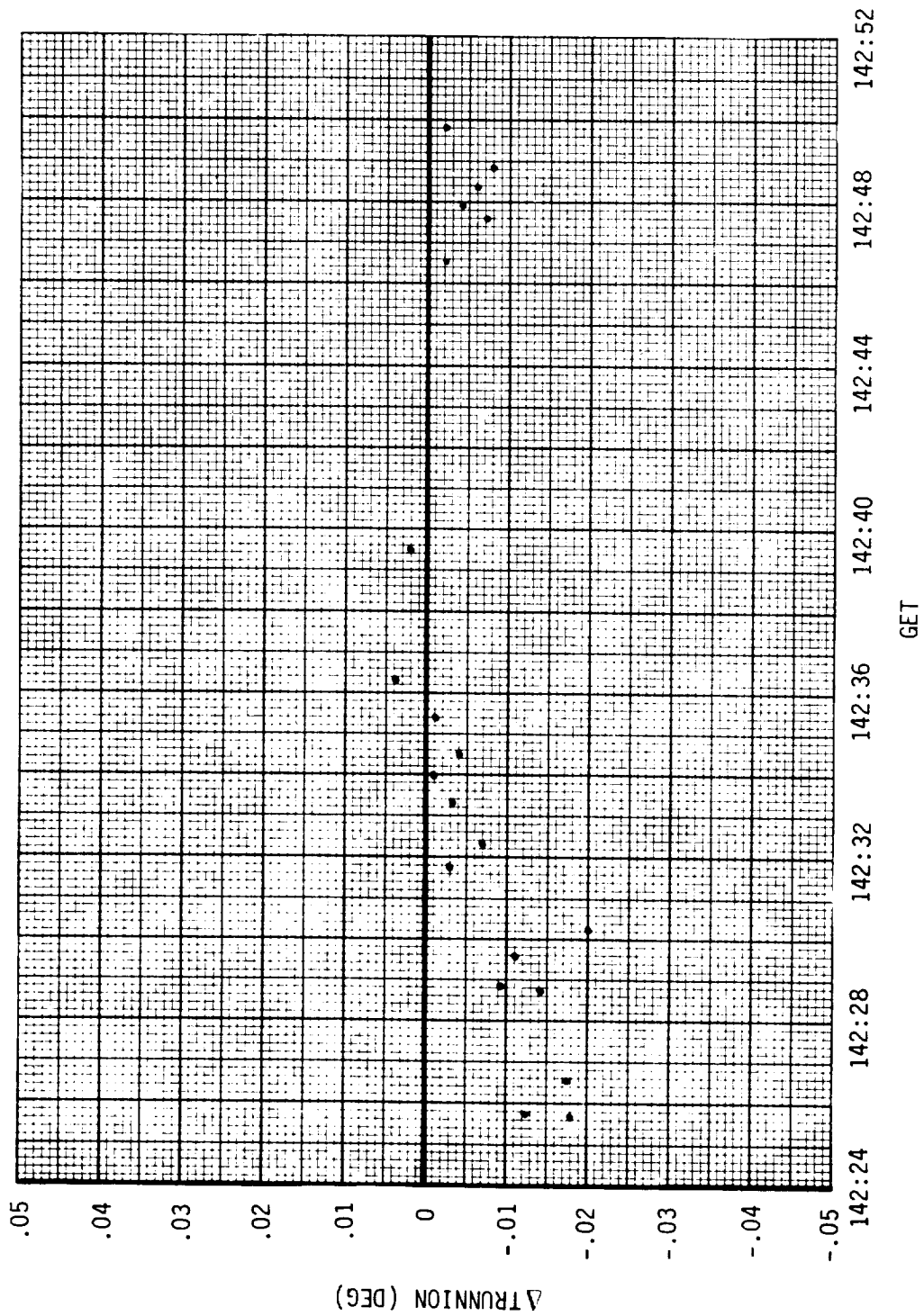


Figure 5-6. Converged SXT Trunnion Residuals - Insertion to CSI

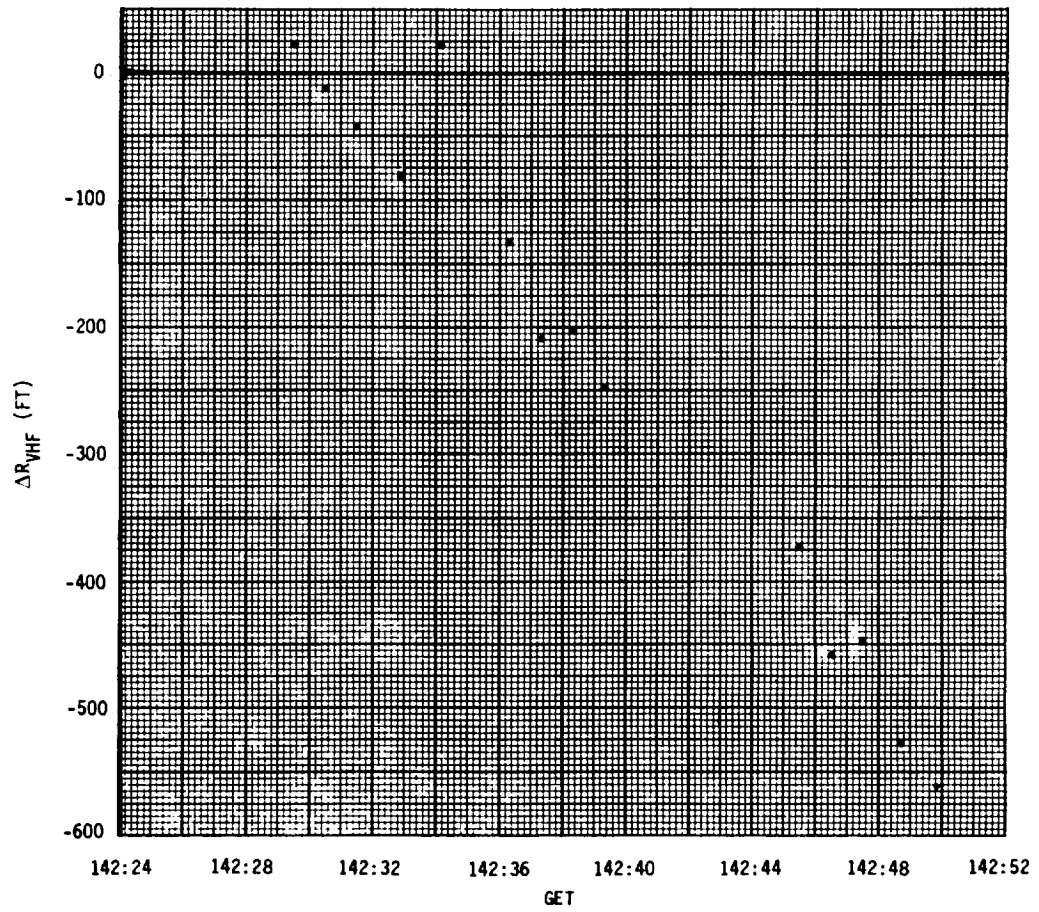


Figure 5-7. Converged VHF Ranging Residuals - Insertion to CSI

ΔX	ΔY	ΔZ	$\Delta \dot{X}$	$\Delta \dot{Y}$	$\Delta \dot{Z}$
-255 ft	334	202	0.50 $\frac{\text{ft}}{\text{sec}}$	-0.62	-0.39

5.3.4 LM Ascent Stage Impact Trajectory

The postflight trajectory for the LM on rev 34 (including the impact burn) is the final version of the rev 34 LM trajectory and superceeds the preliminary version of 25 November 1969.

The rev 34 (deorbit) trajectory for the LM was reconstructed from MSFN doppler data from RID (Madrid, 2-way), MIL, ACN (100 observations pre-burn and 159 observations post-burn), 5 SXT shaft angles, 2 SXT trunnion angles, 7 VHF ranging points, and the thrust profile of the deorbit burn obtained from IMU accelerometer data.

The converged residual statistics for all of the observations used in the fit are as follows:

<u>No. OBS.</u>	<u>Station</u>	<u>Type</u>	<u>Mean</u>	<u>Sigma</u>
86	RID	MSFN	.348 cps	1.404 cps
87	MIL	MSFN	.235 cps	1.416 cps
86	ACN	MSFN	.342 cps	1.356 cps
5	CSM	SXT Shaft	.001 deg	.019 deg
2	CSM	SXT Trunnion	.033 deg	.128 deg
7	CSM	VHF Range	-470 ft	506 ft

The accumulated thrust velocities in IMU platform coordinates due to the deorbit burn are:

$$\Delta V_X = -188.57 \text{ ft/sec}$$

$$\Delta V_Y = 54.15 \text{ ft/sec}$$

$$\Delta V_Z = -6.21 \text{ ft/sec}$$

The time of impact is estimated to be 149:55:16.46 GET. The selenographic coordinates of the impact point are:

$$\text{LATITUDE} = -3.944 \text{ deg}$$

$$\text{LONGITUDE} = -21.196 \text{ deg}$$

$$\text{RADIUS} = 5697847 \text{ ft}$$

$$\text{Selenographic Orbit Inclination} = -14.531 \text{ deg}$$

$$\text{Relative Velocity Magnitude} = 5517.2 \text{ ft/sec}$$

$$\text{Relative Flight Path Angle} = 3.717 \text{ deg}$$

5.4 POWERED DESCENT TRAJECTORY ANALYSIS

5.4.1 Rev. 14 Orbit Determination

Attempts to reconstruct the Rev. 14 trajectory from lo-speed doppler data yielded results not significantly different from PIRX-159 which was used to generate the Final BET (Section 5.3.2). The option for including Landing Radar velocity data in the HOPE orbit determination became available during the Apollo XII postflight period. All available trajectory data sources have now been used to determine the LM trajectory from AOS to Landing. These data include:

MSFN Lo-Speed (6 sec): AOS to PDI

MSFN Hi-Speed (compacted to 2 sec): PDI to Landing
Landing Radar Velocity

Thrust Acceleration Profile from LGC Telemetry.

The principal discrepancy in the Final BET is the large downrange position error at time of landing. ("Error" is defined as difference from the Best Estimate of the LLS.) As illustrated in Figure 5-8, the trajectory determination employing hi-speed doppler and landing radar reduces both the downrange and crossrange errors by about half. No further attempts were made to refine the crossrange error. Examination of the doppler residuals (2-way and 3-way) revealed a pattern similar to that of Figure 5-9. The time interval 110:04 - 110:13, containing the large "wiggle," is just the period in which the LM is passing over the Mare Nectaris. (The relative position of the LM during this period is listed in Appendix F.)

The nominal coordinates of the Mascon in the Mare Nectaris are: 15 deg S, 34 deg E, 100 km below MLR. The Mascon term with these coordinates was added to the L1 potential model. The size of the Mascon was varied in several trajectory fits in order to find the effects on: (a) the doppler residual pattern; and, (b) the state vector at PDI. These results are summarized in Figure 5-11.

- (a) The peak-to-peak "wiggle" amplitude (i.e., the difference in the residuals at 110:10:09 and 110:06:57) was chosen as an indicator of the effectiveness of this fit technique.

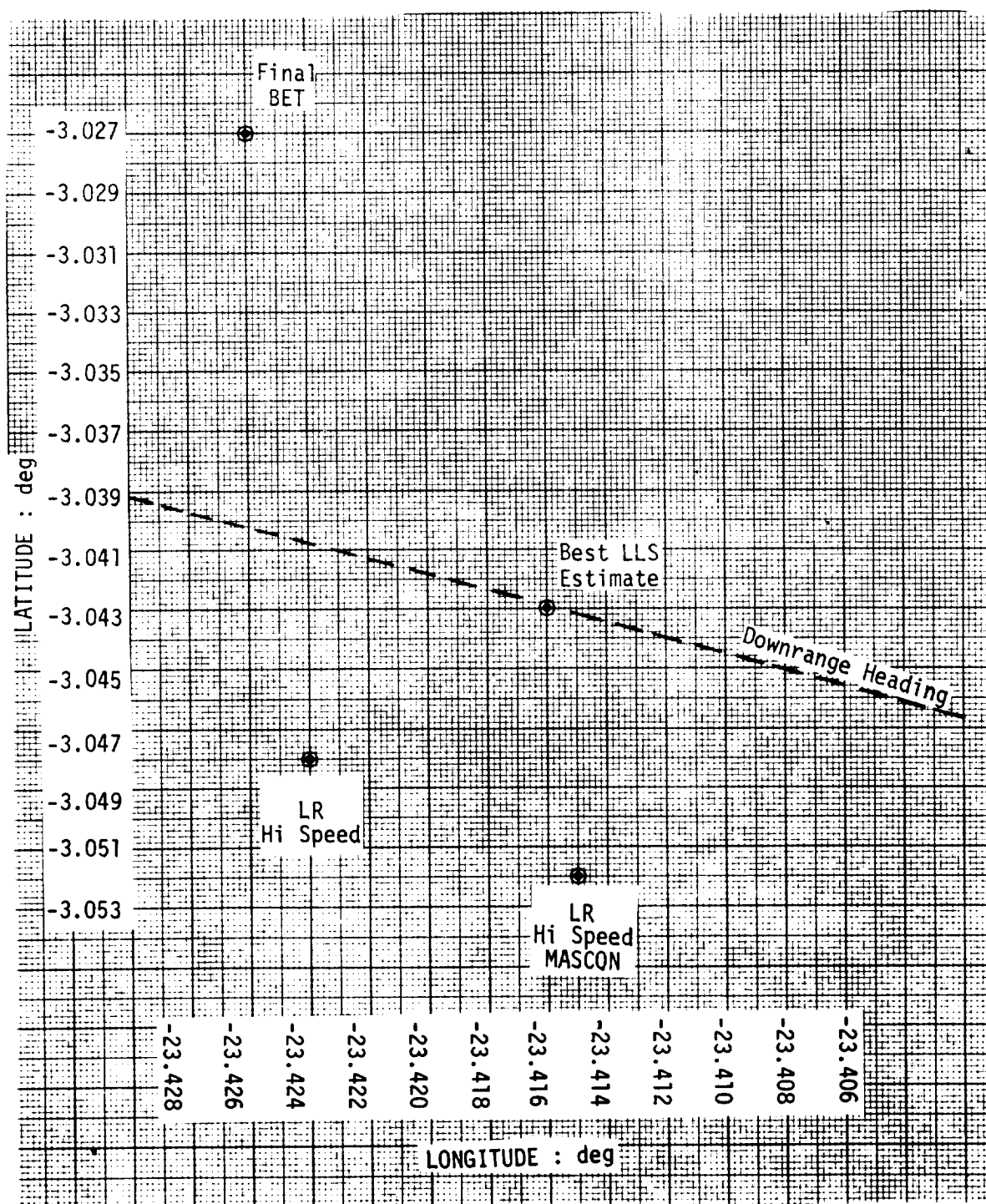


Figure 5-8 LLS COORDINATES

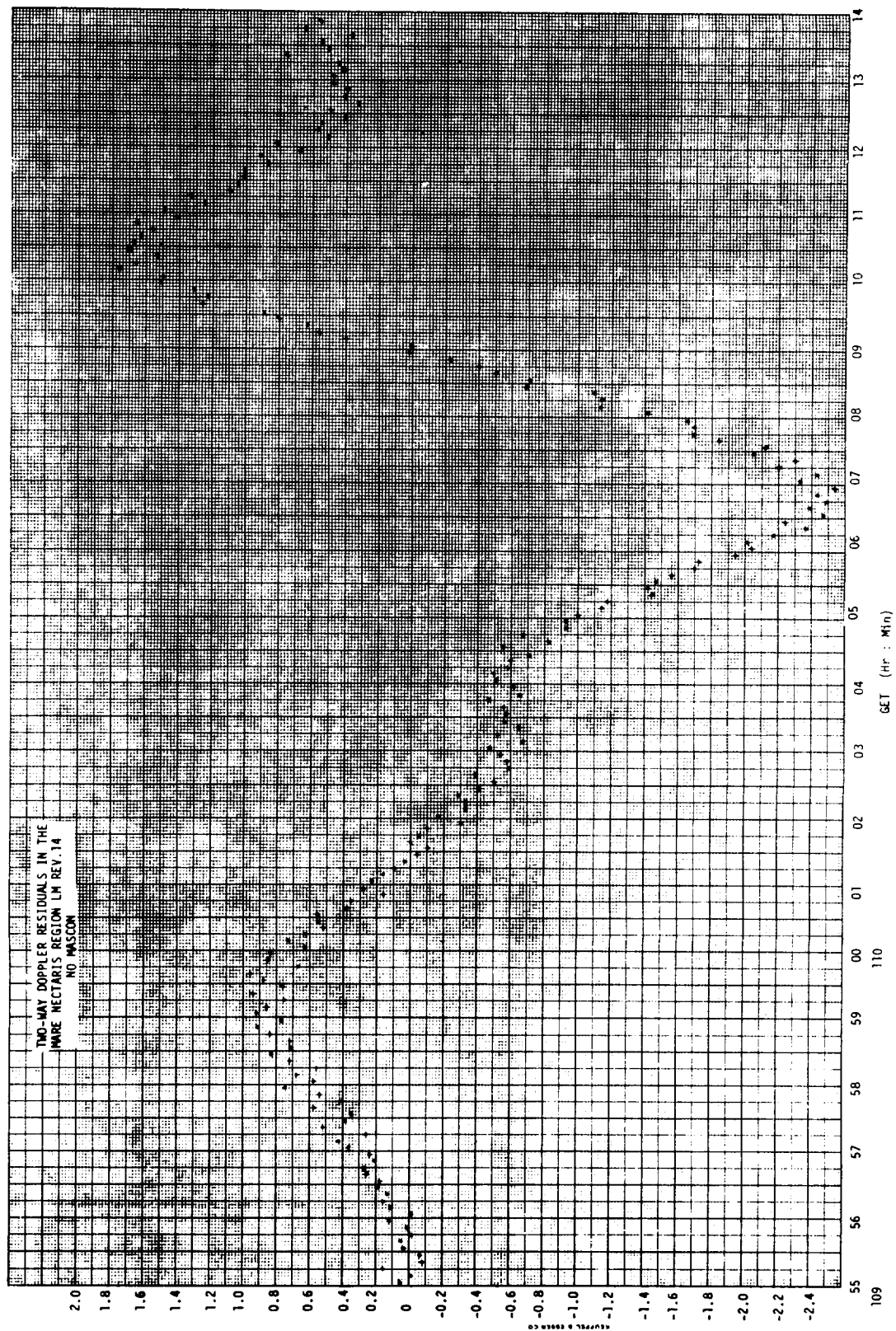


Figure 5-9 TWO-WAY DOPPLER RESIDUALS IN THE
MARE NECTARIS REGION LM REV. 14
NO MASCON

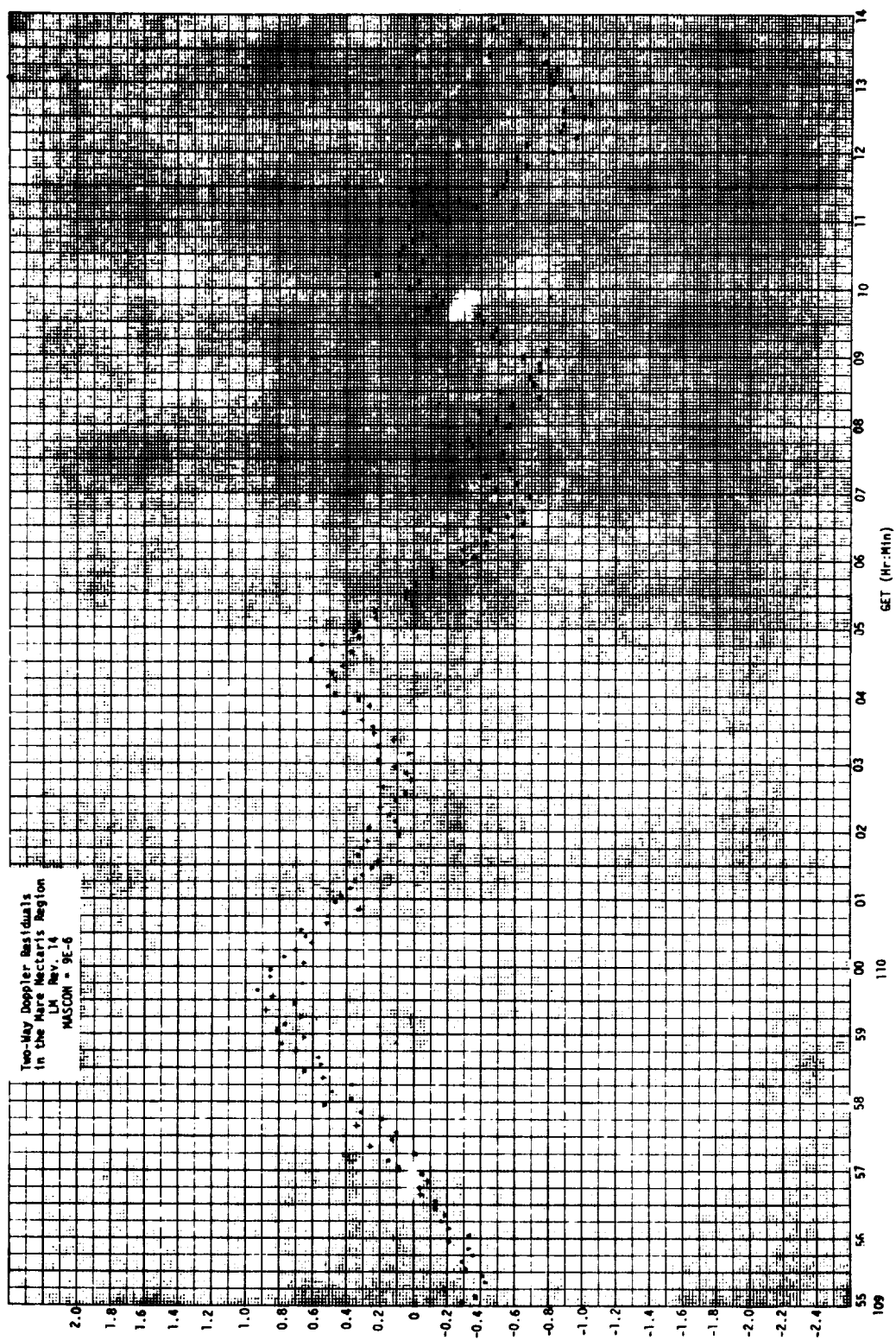


Figure 5-10 TWO-WAY DOPPLER RESIDUALS
IN THE MARE NECTARIS REGION
LM REV.14
MASCON = 9E-6

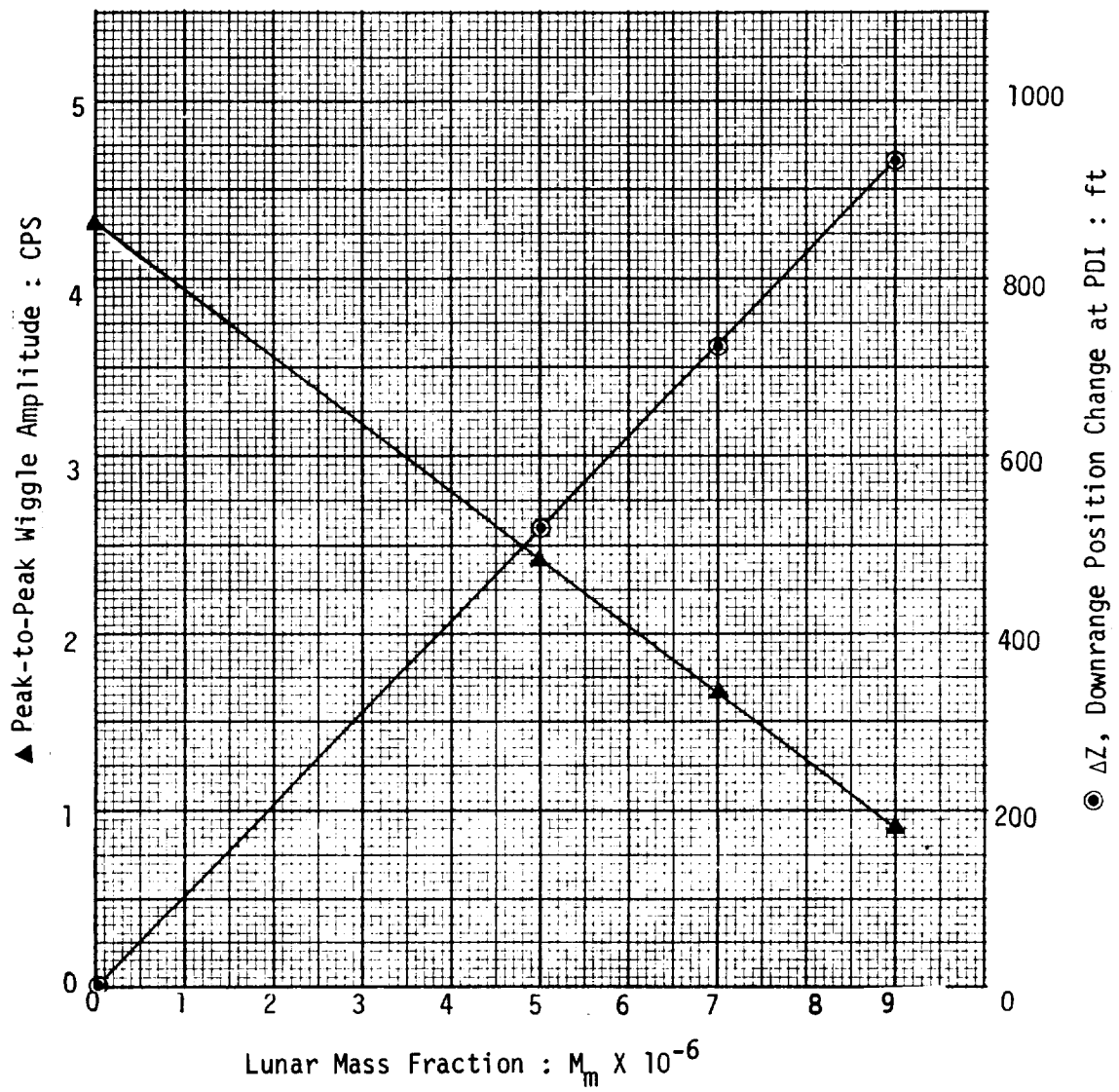


Figure 5-11 EFFECT OF MASCON MAGNITUDE ON TRAJECTORY DETERMINATION

Figure 5-11

The amplitude is seen to decrease linearly with Mascon size over the range of values used. The best fit was achieved with a value of $9\text{E-}6$ lunar mass units. The 2-way doppler residuals from this fit are plotted in Figure 5-10.

- (b) The only significant effect on the state vector at PDI is in the downrange position component. This quantity is seen to decrease linearly with Mascon size (Figure 5-11). The landing site resulting from the trajectory fit using a Mascon size of $9\text{E-}6$ is shown in Figure 5-8. The downrange position component is in very close agreement with the Best LLS Estimate. The IMU errors required to null the relative velocity at landing are:

PHIX	=	85.9 sec
PHIY	=	-98.4 sec
ZXMSL	=	-40.7 sec
XZMSL	=	-30.0 sec

5.4.2 Analysis of Landing Radar Velocity Data

The analysis of the landing radar velocity data consisted of inspection of residuals (difference between observed measurement and computed measurement) obtained from selected LM trajectories.

The landing radar data were obtained by processing the downlink telemetry data with a special purpose computer program which outputs on-board observations on punched cards in a HOPE-compatible format.

The HOPE Program was used to compute simulated landing radar observables from the LM trajectories and from auxiliary information such as REFSMAT, gimbal angles, and radar operating mode. The LM trajectories were generated by the HOPE Program utilizing telemetered acceleration data in the IGS burn option to model the descent burn. Residuals were then formed by subtracting the computed from the actual observable value.

The trajectory used as an independent reference to evaluate these data was obtained from the PIRX159 state vector and associated platform misalignment corrections published as the final A-50 NAT (Section 5.3). This state vector was obtained in the RTCC from lo-speed MSFN doppler data obtained from AOS (acquisition of signal) to PDI on rev. 14. The

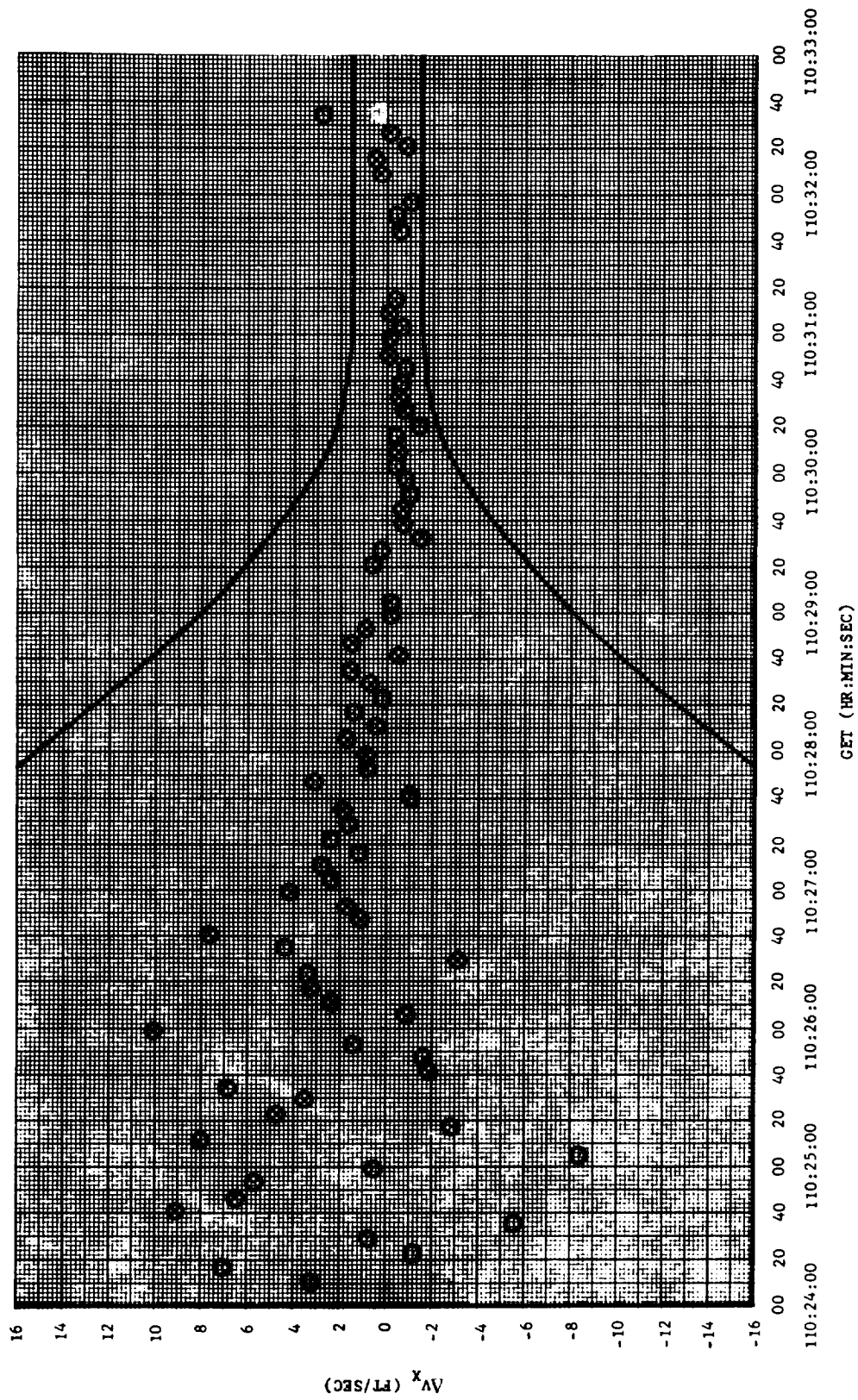


Figure 5-12 LANDING RADAR X-ANTENNA VELOCITY RESIDUALS (NOMINAL ANTENNA ORIENTATION)

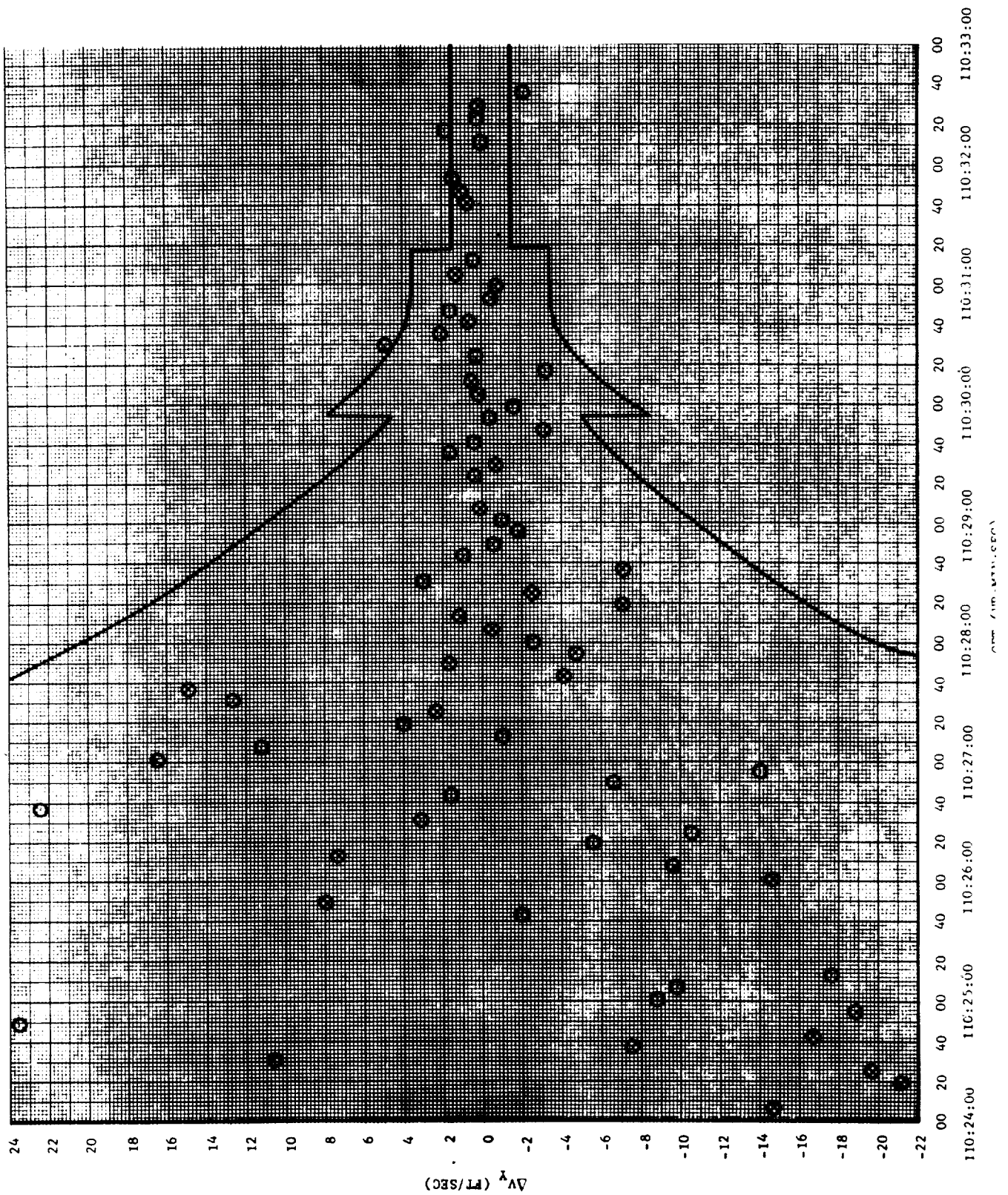


Figure 5-13 LANDING RADAR Y-ANTENNA VELOCITY RESIDUALS

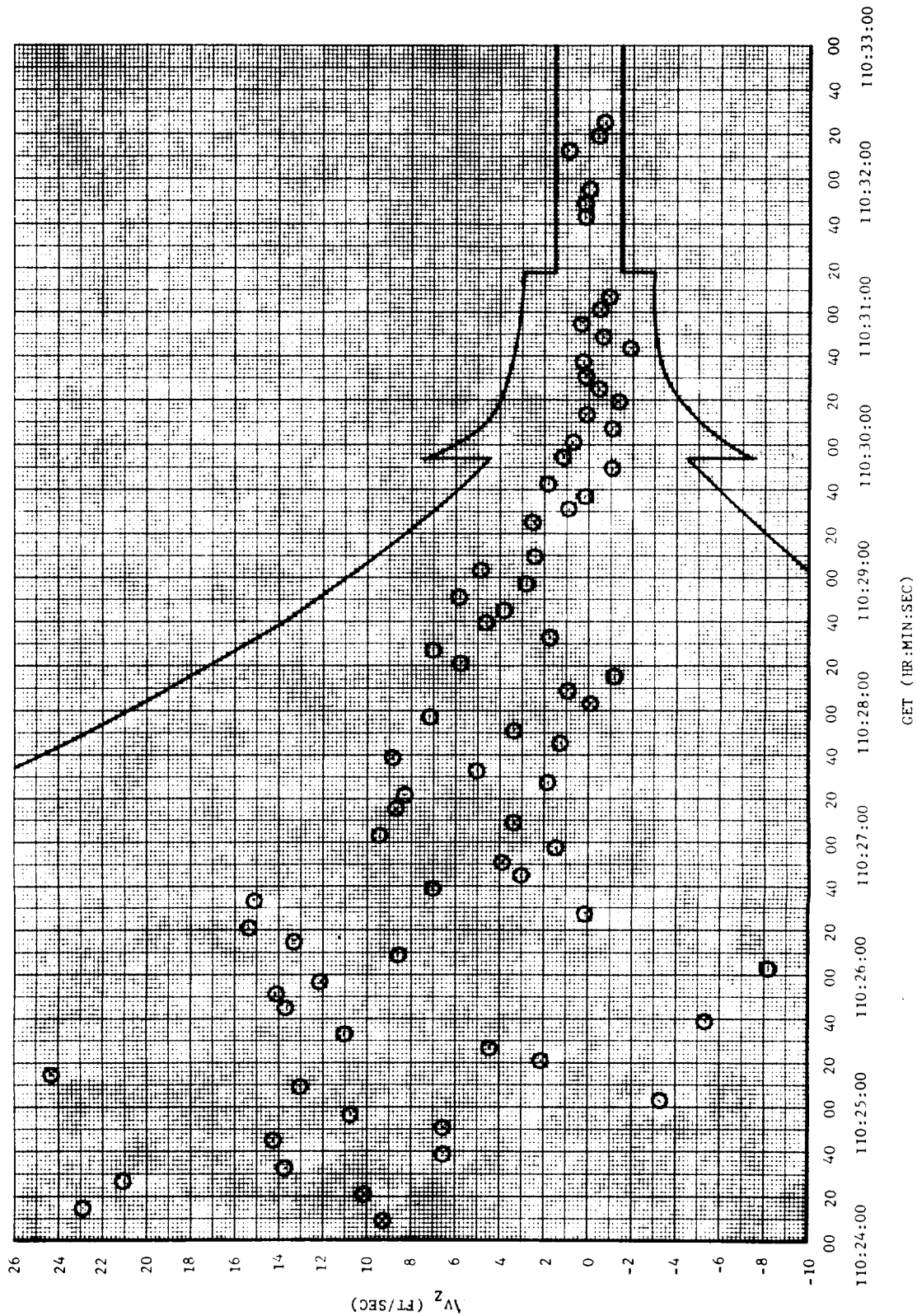


Figure 5-14 LANDING RADAR Z-ANTENNA VELOCITY RESIDUALS (NOMINAL ANTENNA ORIENTATION)

TABLE 5.13 LANDING RADAR VELOCITY RESIDUAL STATISTICS

	V_{XA} (fps)	V_{YA} (fps)	V_{ZA} (fps)	
Mean	1.09	-1.81	4.77	} Nominal Antenna Orientation
St. Dev.	3.04	11.52	6.36	
Noise	3.13	11.32	5.22	
Mean	-1.74	-1.81	1.11	} Y-Antenna Axis Misalignment Corrected
St. Dev.	3.11	11.52	4.91	
Noise	3.11	11.32	5.20	

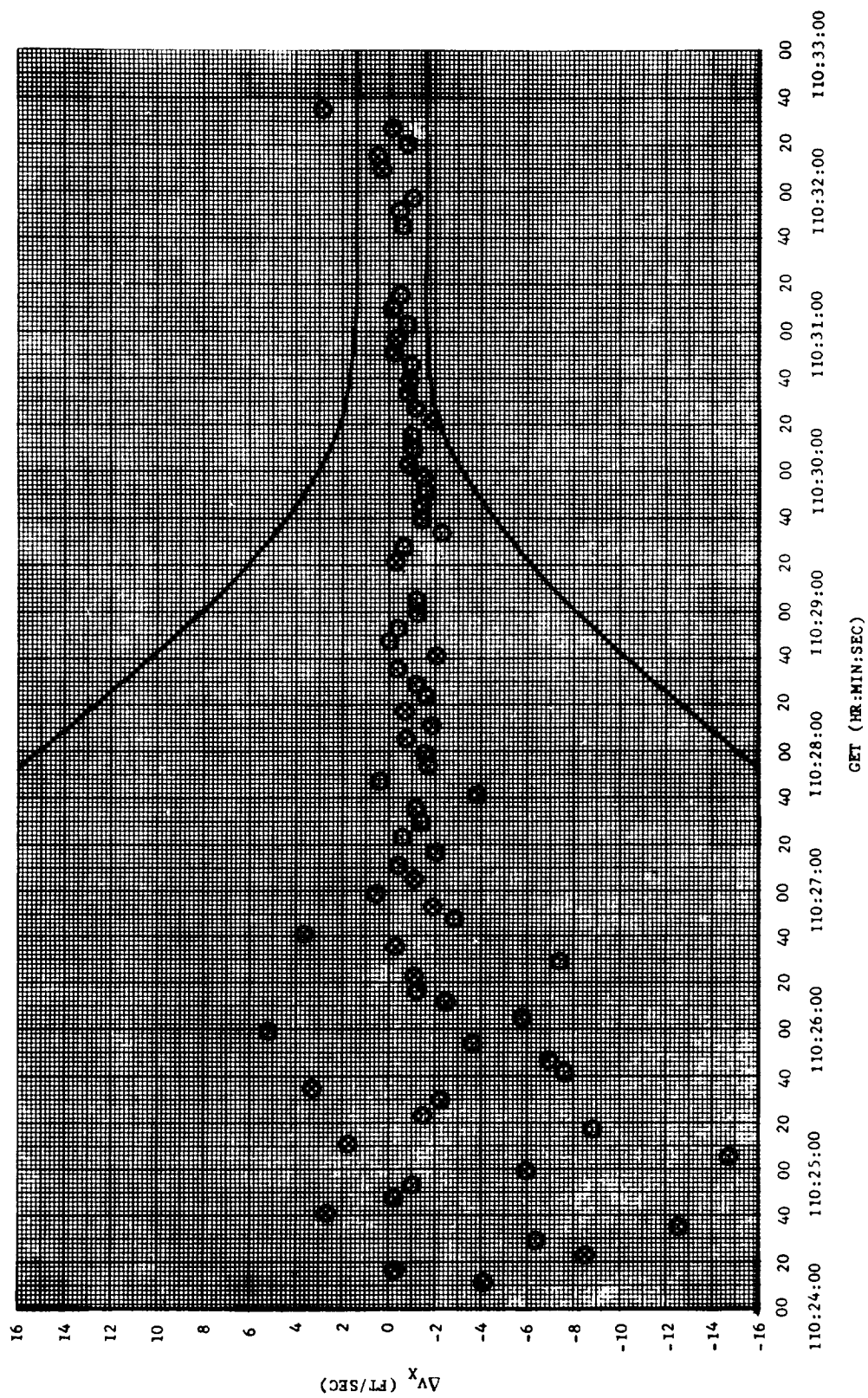


Figure 5-15 X ANTENNA VELOCITY RESIDUALS (CORRECTED ANTENNA ORIENTATION)

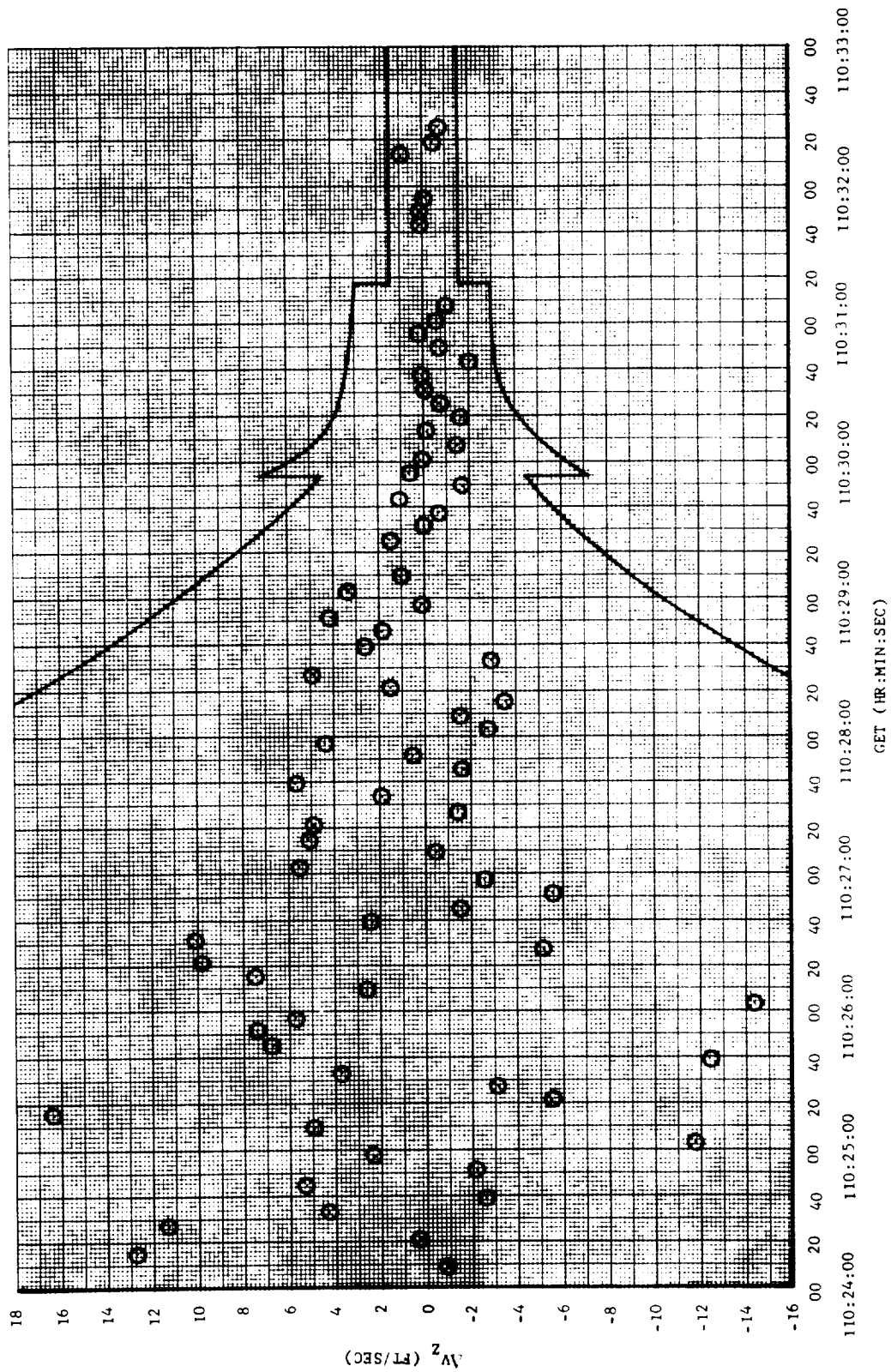


Figure 5-16 Z-ANTENNA VELOCITY RESIDUALS (CORRECTED ANTENNA ORIENTATION)

postflight reconstructions discussed in paragraph 5.4.3 were obtained using hi-speed MSFN doppler and mode 2 landing radar observations.

Figures 5-12, 5-13 and 5-14 show the velocity residuals computed from the PIRX159 trajectory. Note that a few values fell outside the specification limits plotted. The V_{YA} residuals are considerably more erratic than V_{XA} or V_{ZA} . This can be seen in the residual plot as well as in the high standard deviation listed in Table 5.13. The noise estimates listed in Table 5.13 indicate that this erratic residual pattern in V_{YA} (and to a lesser extent in V_{ZA}) is caused more probably by random noise than by a systematic measurement error.

Postflight analysis performed by other TRW tasks posed the possibility that the landing radar may have been misaligned by .2 degrees about the Y-antenna axis. When the nominal orientation figures are changed to reflect this misalignment, the residuals plotted in Figures 5-15 and 5-16 are produced. Notice that the V_{XA} and V_{ZA} mean values are significantly altered (Table 5.13). The resulting statistics are more desirable than those obtained with the nominal orientation values.

In conclusion, the landing radar velocity observation obtained from the Apollo 12 mission do not appear to be as good qualitatively as the Apollo 11 data (Reference 15). The residual statistics are significantly higher than corresponding Apollo 11 statistics, even when the apparent antenna misalignment is corrected.

5.4.3 Lunar Surface Altitude from LR Range

Landing Radar range residuals were used to compute surface altitude along the ground-track of the pierce point. The results are plotted in Figure 5-12. The time tags on the data points are in the LGC clock time (TLGC = GET - 0.68 sec).

The ground-track of the range beam pierce point is plotted on Lunar Map ORB-I-7 in Figure 5-18. The time ticks correspond to the 2-sec range measurement times indicated in Figure 5-17 and the time tags are in min:sec after 110 hours LGC time. The small ellipses show the approximate size

of the range beam. The endpoint of the ground-track is located relative to the "Snowman" formation as shown in Reference 1, and the ground-track is plotted relative to this point. A full listing of surface altitude relative to LLS, latitude and longitude of range beam pierce point, central angle from LLS, and time may be obtained from the Task A-50 Monitor. These data are based on the MASCON = $9E-6$ trajectory fit which yields the best absolute estimates of latitude and longitude. Note that these coordinates do not correspond to the grid markings on the ORB-I-7 map.

A very flat terrain over the final 2 deg. of the approach to the LLS is indicated by the contour map and verified by the Landing Radar data. The downhill slope between the final 2-3 deg. of the approach is also indicated by the radar, but it is difficult to make any quantitative comparisons. A very rough terrain is indicated over the first 3 deg. after range beam lock-on. Unfortunately, no suitable contour maps of this region are currently available.

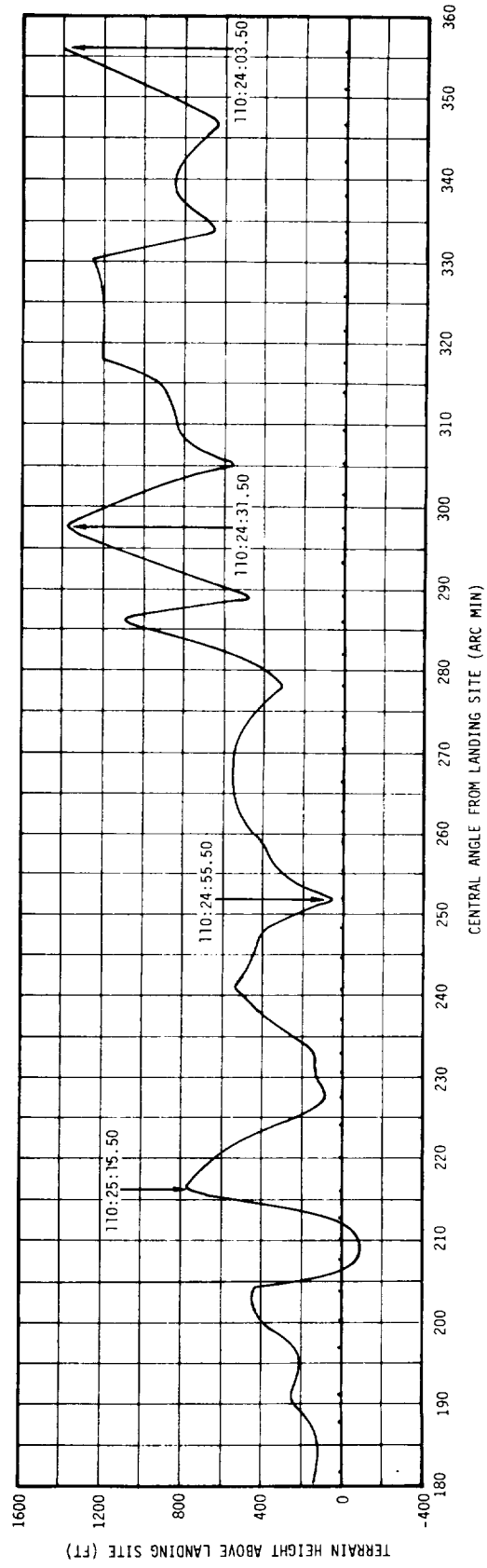
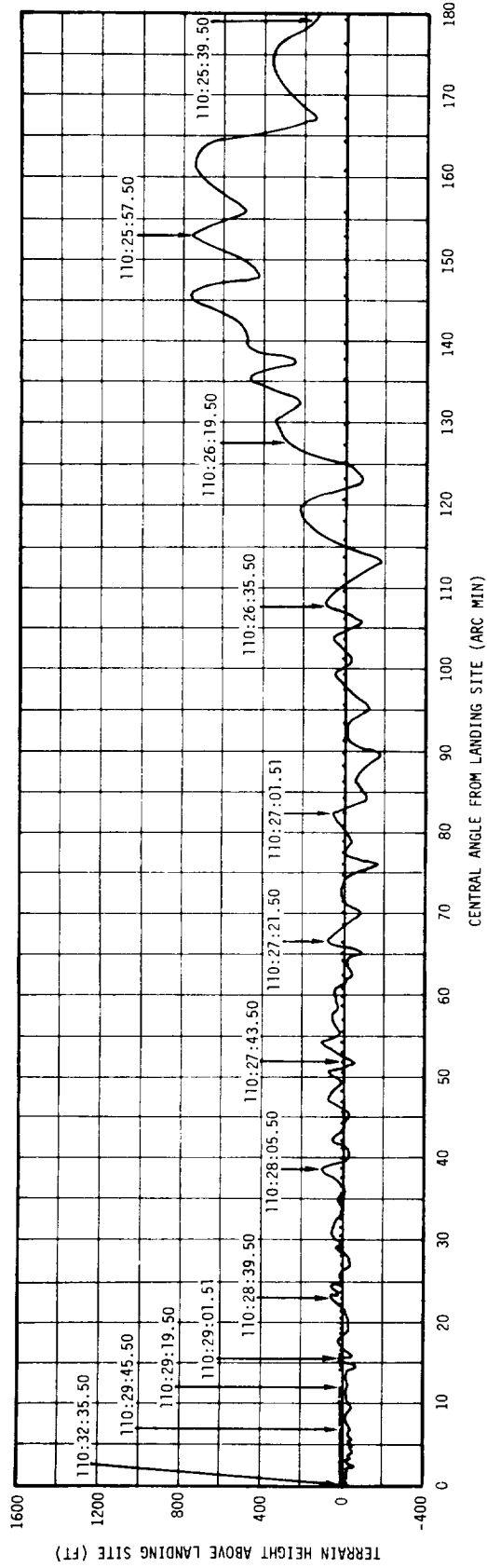


Figure 5-17 LUNAR SURFACE ALTITUDE
5-49

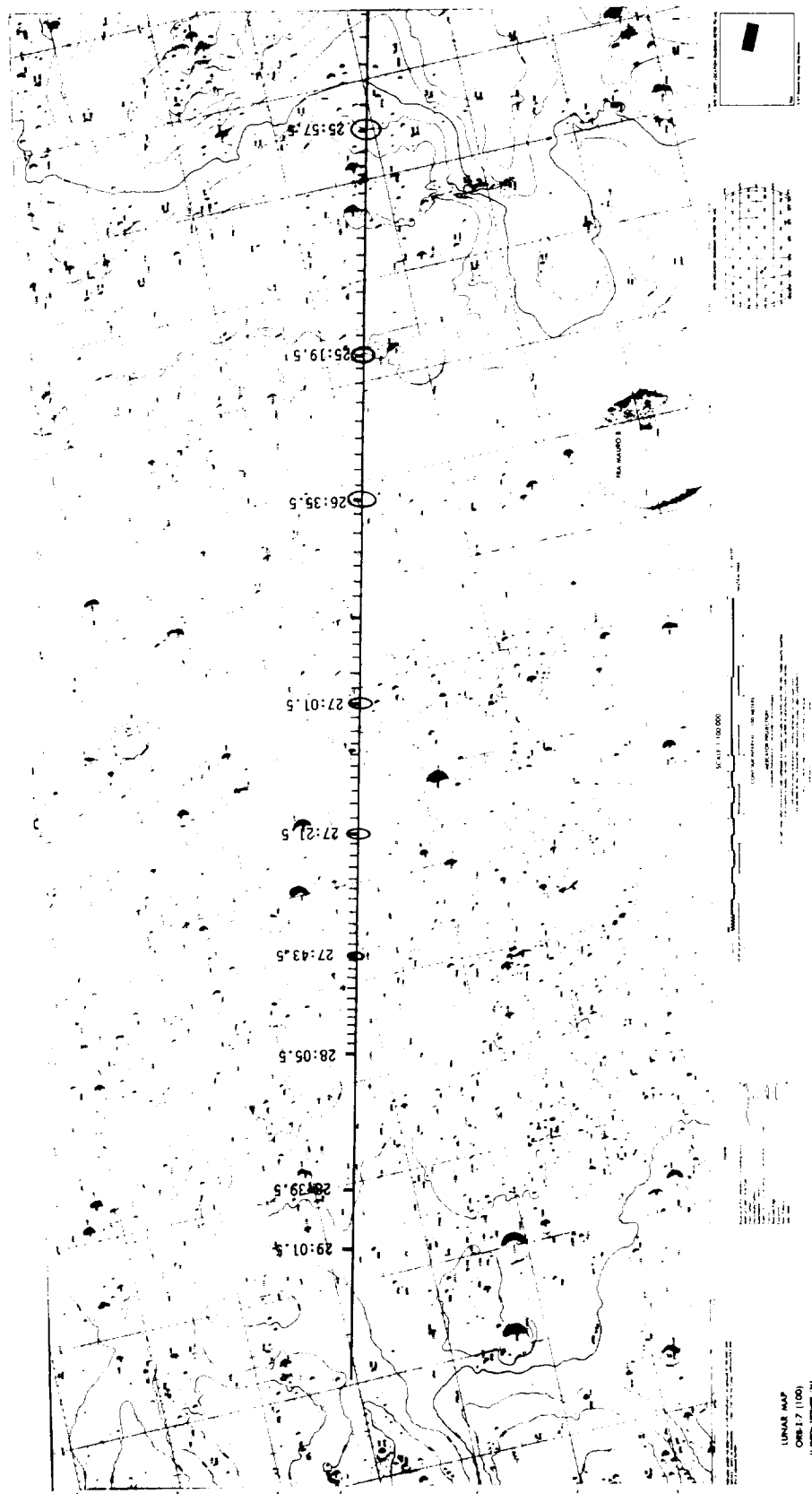


Figure 5-18 GROUNDTRACK OF LR RANGE BEAM

APPENDIX A

SUPPLEMENTARY DATA

Appendix A contains supplementary information which did not appear in the main body of the report. This information includes a summary of ground based and onboard data weights used in the HOPE Program, a summary of the components used in the L1 lunar potential model, and a summary of the USBS station locations.

Table A.1 lists the data weights used in the HOPE Program for ground based radar data and Table A.2 lists the data weights used in the HOPE Program for onboard data by type and observable.

Table A.3 lists the terms of the L1 potential model.

Table A.4 lists the S-band tracking stations and their locations as used in the Apollo 12 postflight analysis. All locations are referenced to the Fischer Ellipsoid of 1960. The surface refractivity for the month of November is also listed.

Table A.1 GROUND BASED RADAR DATA WEIGHTING

DATA TYPE	RADAR	WEIGHTING
Range	USB: 30-ft antenna 85-ft antenna	600 ft.
Doppler (2-way)	USB: 30-ft antenna 85-ft antenna	0.1 cycle/sec.
Doppler (3-way)	USB: 30-ft antenna 85-ft antenna	0.1 cycle/sec.

Table A.2 NOMINAL ONBOARD RADAR DATA WEIGHTING

DATA TYPE	SHAFT	TRUNNION	RANGE	RANGE RATE	V _X	V _Y	V _Z
Rendezvous Radar	.01	.01	30.	1.			
Sextant	.001	.001					
VHF Ranging			30.				
Landing Radar			50.		.1	.1	.1

Table A.3 L1 LUNAR POTENTIAL MODEL

TERM	VALUE
J2	2.07108×10^{-4}
J3	-2.1×10^{-5}
C22	2.0716×10^{-5}
C31	3.4×10^{-5}
C33	2.583×10^{-6}
	All other harmonics are zero

Table A.4 USBS Station Locations

Station	Antenna	Identification	Latitude* (deg)	Longitude* (deg)	Altitude* (ft)	Surface Refractivity
Antigua	30 ft	ANG	17.01692	298.24715	141.08	372
Ascension	30 ft	ACN	-7.95479	345.67287	1784.78	355
Bermuda	30 ft	BDA	32.35129	295.34182	68.90	345
Canary Island	30 ft	CYI	27.76454	344.36519	567.59	343
Honeysuckle Creek	85 ft	HSK	-35.58349	148.97829	3755.25	300
Carnarvon	30 ft	CRO	-24.90658	113.72546	65.62	329
Goldstone	85 ft	GDS	35.34159	243.12680	2976.05	276
Grand Bahama	30 ft	GBM	26.63286	281.76234	16.40	364
Guam	30 ft	GWM	13.31058	144.73692	301.84	375
Guaymas	30 ft	GYM	27.96321	249.27915	62.34	328
Hawaii	30 ft	HAW	22.12631	200.33433	3776.25	304
Madrid	85 ft	MAD	40.45499	355.83201	2553.81	299
Merritt Island	30 ft	MIL	28.50827	279.30658	32.81	349
Texas	30 ft	TEX	27.65375	262.62153	32.81	336
Honeysuckle Creek Wing	85 ft	NBE	-35.40099	148.98176	2196.85	300
Goldstone Wing	85 ft	PIR	35.38957	243.15103	3186.02	276
Madrid Wing	85 ft	RID	40.42829	355.75147	2527.56	299

*All quantities are referenced to the Fischer Ellipsoid of 1960.

APPENDIX B
ORBITAL PARAMETERS AT CERTAIN EVENT TIMES

Appendix B contains selected orbital parameters which were obtained by propagating RTCC state vector solutions or Task A-50 BET solutions to desired event times.

Table B.1 contains conditions for the desired events which were within the earth's SOI (sphere of influence). Table B.1 contains the following information: vector source, event, event time (GET), latitude, longitude, altitude, apogee, perigee, velocity, flight path angle, and heading angle. Apogee and perigee distances are referenced to the center of the earth.

Table B.2 contains the same information as Table B.1 with the exception that the desired events were within the lunar SOI. Altitude, apolune, and perilune distances are referenced to the Apollo 12 landing site radius (937.3643 N. Mi.).

The conditions at T_{CA} (time of closest approach to the moon) for a free-return circumlunar mission were obtained from an RTCC vector at 4 hours 13 minutes 09.96 seconds GET (TLI cutoff vector HSR001 propagated through approximately 10 minutes of venting) and are given below:

TIME	83 hours 42 minutes 05.51 seconds GET
LATITUDE	25.33 S
LONGITUDE	172.51 E
ALTITUDE	470.7 N. Mi.
APOLUNE	N/A
PERILUNE	470.7 N. Mi.
VELOCITY	7172.3 FPS
FLIGHT PATH	0.000 DEG
HEADING	-112.10 DEG

In Table B.1 the listed values are referenced to an earth centered inertial (ECI), mean of nearest Besselian year coordinate system and in Table B.2 the values are referenced to a selenocentric, mean of nearest Besselian year coordinate system.

Table B.1 APOLLO 12 EARTH SOI ORBITAL PARAMETERS

SOURCE	EVENT	TIME GET	LATITUDE		LONGITUDE		ALTITUDE		APO*		PER*		VELOCITY FPS	FLIGHT PATH		HEADING ANGLE DEG
			DEG	N,S	DEG	E,W	N.M.	N.M.	N.M.	N.M.	ANGLE DEG					
HSRC001	TLI C/O	02:53:14	16.19N		154.28W		199.5	233313.5	3559.8			35390.1	8.603		63.90	
HSRC001	S-IVB/CM SEP	03:18:04.9	28.91N		79.76W		3820.1	228963.2	3559.9			24862.9	45.096		100.13	
HSRC001	First Docking	03:26:53.3	26.70N		71.01W		5337.7	228894.9	3560.0			22535.1	49.896		105.29	
BDAX074	S/C Ejection	04:13:00.9	18.59N		58.77W		12506.3	229584.5	3559.5			16451.1	60.941		114.52	
MILX163	MCC1 IGN	30:52:44.4	1.11S		63.05W		116929.1	229852.2	3545.9			4317.4	75.833		120.80	
HSKX236	MCC1 C/O	30:52:53.6	1.11S		63.09W		116935.4	228278.0	3146.6			4297.5	76.597		120.05	
HSKX366	MCC2 IGN	188:27:15.8	15.97N		137.71E		180031.2	286180.0	2712.3			3035.6	-78.444		91.35	
HAUX881	MCC2 C/O	188:27:20.2	15.97N		137.70E		180029.0	286198.5	2731.8			3036.0	-78.404		91.36	
NBEX127	MCC3 IGN	241:21:59.7	14.86N		92.17E		25059.0	276203.7	3469.1			12082.9	-68.540		96.00	
GWMS147	MCC3 C/O	241:22:05.4	14.87N		92.15E		25048.3	275925.5	3465.5			12084.7	-68.547		96.01	
GWMS147	CM/SM SEP	244:07:20.1	0.32N		117.25E		1949.5	279020.0	3465.1			29029.1	-36.454		105.92	
GWMS147	Entry Interface	244:22:19.1	13.80S		173.52E		65.8	285902.1	3463.8			36116.2	-6.483		98.17	

* Referenced to earth center

Table B.2 APOLLO 12 LUNAR SOI ORBITAL PARAMETERS

SOURCE	EVENT	LATITUDE		LONGITUDE	ALTITUDE*		APO* PER*		VELOCITY	FLIGHT PATH		HEADING
		DEG N,S	DEG E,W		N.M.	N.M.	N.M.	N.M.		ANGLE	ANGLE	
									FPS	DEG	DEG	DEG
BET	LOI-1 IGN	83:25:23.4	5.76N	175.62E	83.91	NA	64.94		8173.6	-8.447	-130.66	
BET	LOI-1 C/O	83:31:15.7	1.62S	154.03E	62.91	170.20	61.66		5470.1	-0.630	-120.70	
GWMX504	LOI-2 IGN	97:48:48.1	1.67S	151.67E	62.79	170.37	61.42		5470.6	-0.662	-120.72	
HANX519	LOI-2 C/O	87:49:05	1.88S	150.85E	62.74	66.10	54.59		5331.4	0.301	-120.50	
NBEX128	Undocking	107:54:02.3	13.52S	86.96E	63.02	63.08	56.91		5329.0	-0.034	-92.75	
NBEX128	CSM SEP IGN	108:24:36.8	6.61S	7.44W	59.22	63.91	56.99		5350.0	-0.177	-54.83	
CROX609	CSM SEP C/O	108:24:51.2	6.45S	8.14W	59.15	64.06	56.58		5350.5	-0.204	-54.85	
BET	DOI IGN	109:23:39.9	6.64N	172.21E	60.52	63.27	57.25		5343.0	0.172	-125.19	
PIRX159	DOI C/O	109:24:08.9	6.29N	170.76E	61.52	61.53	8.70		5268.0	-0.024	-125.15	
PIRX159	PDI IGN	110:20:38.1	6.76S	7.82W	7.96	62.30	7.96		5566.4	-0.025	-54.86	
HSKX624	CSM LOPC-1	119:47:13.2	14.01S	77.68E	62.20	62.50	57.61		5333.5	-0.068	-90.73	
BET	Insertion	142:10:50.9	0.59S	33.05W	9.97	51.93	9.21		5542.5	0.336	-54.94	
BET	CSI IGN	143:01:51	5.16N	164.68E	51.46	52.51	9.94		5310.3	0.055	-125.57	
BET	CSI C/O	143:02:32.1	4.65N	162.64E	51.48	51.49	41.76		5354.9	0.015	-125.71	
BET	TPI IGN	144:36:26	14.57N	128.99W	44.50	44.73	40.91		5382.5	0.052	-102.07	
WADX700	Lunar Docking	145:36:20.2	14.53S	46.98E	58.14	63.43	58.04		5357.1	-0.040	-75.71	
BET	CSM SEP	148:04:30.9	1.40N	43.34W	59.94	64.66	59.08		5347.4	0.153	-55.81	
BET	LM Deorbit IGN	149:28:14.8	14.32S	62.86E	57.62	63.52	57.94		5361.8	-0.116	-87.63	
BET	LM Deorbit C/O	149:29:36.9	14.47S	58.62E	57.42	57.59	-63.15		5176.8	-0.274	-84.10	
BET	LM Lunar Impact	149:53:16.4	3.95S	21.20W	0.38	57.89	-63.79		5502.6	-3.728	-54.14	
GWMX731	CSM LOPC-2 IGN	159:04:45.5	6.65S	110.34E	58.70	64.23	56.58		5353.2	-0.196	-118.68	
GWMX747	CSM LOPC-2 C/O	159:05:04.8	6.82S	109.40E	58.90	64.66	56.81		5353.0	-0.199	-114.18	
BDAX800	TEI IGN	172:27:16.0	8.73N	170.26W	64.60	66.00	55.68		5322.9	-0.202	-115.73	
HSKX866	TEI C/O	172:29:27.1	7.77N	178.60W	66.00	NA	64.10		8350.4	2.718	-116.45	

* Referenced to landing site radius (937.3643 N.M.)

APPENDIX C

APOLLO 12 RELATIVE TRAJECTORY LISTING

A HOPE listing of the trajectories for both the LM and the CSM during periods of separated free flight was delivered to the Task Monitor on 5 February 1970. The trajectories used in generating this print were obtained from the best available fits using all available data types.

Table C.1 lists the trajectories by segment, the propagation interval, vector source, and RSS total position and velocity comparisons. The position and velocity deltas listed are comparisons between adjacent segments. Where a maneuver was performed between the segments, the ΔV produced by that maneuver has been removed from the RSS velocity differences (DOI = 72.4 ft/sec, CSI = 45 ft/sec). Note that all the comparisons are reasonable with the largest difference being between the CSM fits made on revs 13 and 14. This difference is partially attributable to the limited tracking coverage on rev 14 (see Figure 5-1, page 5-5). The residual statistics obtained from the CSM and LM trajectories are summarized in Tables C.2 and C.3.

CSM Trajectories

Note that the rev 13 CSM trajectory was obtained from a fit using low speed MSFN data and using the IGS burn option in HOPE to model the separation burn. No explanation was readily apparent for the high MSFN residual statistics. Since the requirement for a NAT update did not exist and task requirements were more immediate, an in-depth analysis was not performed. It was felt, however, that the trajectory obtained still represented the best overall fit of the data. The rev 14 CSM trajectory was obtained from a free flight fit using low speed MSFN data.

The trajectories used for the CSM on revs 30 and 31 were those obtained in real time by the RTCC. The rev 30 trajectory was from an SS2 type fit (solution constrained to the input inclination). It was necessary to use this fit technique because of the extremely poor tracking coverage (see Figure 5-1, page 5-5). The RTCC trajectories were used because no significant improvements were made in postflight fits (note the residual statistics in Table C.2).

LM Trajectories

The LM trajectory for the undock to DOI period was of good quality due to the good MSFN tracking. Table C.3 lists residual statistics obtained with the trajectory.

The trajectory from DOI to PDI was obtained from a fit which used CSM sextant, VHF ranging, high speed MSFN, and landing radar data. The fit also used telemetered accelerometer data to model the powered descent. The trajectory generated by this fit in the free flight segment (DOI to PDI) was considered the best available and, therefore, was chosen for the relative trajectory print. Although the relative trajectory extended only through free flight periods, the residual statistics listed in Table C.3 were taken from the entire data arc and do include observations taken during powered descent. (This trajectory was not published as a BET.)

The trajectories used for the insertion to CSI and CSI to TPI segments used both onboard relative (CSM sextant, VHF ranging, LM rendezvous radar) and ground based MSFN tracking data. As was stated in Reference 2, the presence of MSFN data caused the relatively large residual statistics for the onboard data types (especially in the CSI to TPI period where the onboard and ground based data arcs are coincident). Statistics obtained from free flight fits using only onboard data are listed in Table C.4 as a gross indication of data quality.

The insertion to CSI trajectory was obtained from a free flight fit and the CSI to TPI trajectory was obtained from a fit which modeled the

CDH maneuver with the HOPE IGS burn option. Since no tracking data were available after TPI, no trajectories after that point were included in the deliverable.

Note that the above-mentioned trajectories were not necessarily published as a BET. However, they represent the best trajectories obtained before Task A-50 was directed by MSC to de-emphasize rendezvous analysis.

TABLE C.1 RELATIVE TRAJECTORY SEGMENT SUMMARY

Segment	Propagation Interval		Vector Source	RSS Comparisons ΔR (feet)	ΔV (fps)
	Start (d:h:m:s GMT)	Stop (d:h:m:s GMT)			
Pre Sep to DOI	19:04:45:00	19:05:45:39.9	CSM - Rev 13 IGS fit LM - Rev 13 Free Flight Fit	8524	2.3
DOI to PDI	19:05:46:08.9	19:06:42:38.1	CSM - Rev 14 Free Flight Fit LM - Rev 14 IGS Fit	3812	1.4
Insertion to CSI	20:14:32:50.9	20:15:23:51.0	CSM - RTCC Rev 30 (CROX 680) LM - Rev 30 Free Flight Fit	1765	2.4
CSI to TPI	20:15:24:32.1	20:16:58:26.0	CSM - RTCC Rev 31 (ACNX 687) LM - Rev 31 IGS Fit	2287	.04

TABLE C.2 CSM FIT RESIDUAL STATISTICS

CSM Rev 13 Fit

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
2 way doppler (GDS)	133	-.0233 cps	1.09 cps
Range (GDS)	17	-345. feet	8.26 feet
3 way doppler (HAW)	110	-.0074 cps	1.08 cps
3 way doppler (MIL)	98	-.0387	1.04

CSM Rev 14 Fit

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
2 way doppler (HSK)	143	.0251 cps	.560 cps
3 way doppler (GDS)	142	.0056	.564
3 way doppler (CRO)	29	.1051	.259

CSM Rev 30 (RTCC CROX 680)

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
2 way doppler (HSK)	242	.134 cps	.415 cps
3 way doppler (CRO)	76	.065	.186

CSM Rev 31 (RTCC ACNX 687)

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
2 and 3 way doppler (HSK)	66	.088 cps	.582 cps
2 way doppler (MAD)	73	.015	.545
3 way doppler (GWM)	118	-.033	.505
3 way doppler (CRO)	111	.035	.536
3 way doppler (ACN)	21	-.003	.188

TABLE C.3 LM FIT RESIDUAL STATISTICS

LM Undock to DOI (Rev 13)

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
2 way doppler (PIR)	116	.001 cps	.616 cps
3 way doppler (BDA)	115	-.015	.617
3 way doppler (HAW)	91	.071	.629
3 way doppler (MIL)	94	-.005	.622
3 way doppler (NBE)	43	.088	.376
3 way doppler (GWM)	45	-.118	.502

LM DOI to PDI (Rev 14 - DOI to Landing Fit)

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
2 way doppler (PIR)	695	.004 cps	.889 cps
3 way doppler (GWM)	702	-.047	.862
3 way doppler (NBE)	695	.043	.854
3 way doppler (MIL)	288	-.004	.852
3 way doppler (HAW)	700	-.014	.855
CSM sextant shaft	3	-.002 deg.	.062 deg.
CSM sextant trunnion	3	.443	.035
CSM VHF Ranging	6	687. feet	248. feet

LM Insertion to CSI (Rev 30)

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
2 way doppler (HSK)	83	.070 cps	.477 cps
3 way doppler (GWM)	195	-.027	.558
3 way doppler (NBE)	195	-.018	.555
3 way doppler (CRO)	84	.044	.540
CSM sextant shaft	22	-.002 deg.	.022 deg.
CSM sextant trunnion	22	-.006	.007
CSM VHF Ranging	14	-232. feet	200. feet
LM RR shaft	2	.015 deg.	.020 deg.
LM RR trunnion	2	-.043	.020
LM RR range	2	323. feet	13. feet
LM RR range rate	2	-.380 fps.	.157 fps.

LM CSI to TPI (Rev 31)

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
2 way doppler (RID)	134	-.061 cps	.950 cps
3 way doppler (NBE)	64	-.641	.788
3 way doppler (CRO)	112	-.034	.914
3 way doppler (GWM)	76	.309	.845
3 way doppler (ACN)	20	.699	.377
CSM sextant shaft	31	-.078 deg	.095 deg
CSM sextant trunnion	31	-.040	.028
CSM VHF ranging	32	-277. feet	63.2 feet
LM RR shaft	53	.011 deg	.026 deg
LM RR trunnion	53	-.120	.072
LM RR range	53	-122. feet	97.6 feet
LM RR range rate	53	-.331 fps	.285 fps

TABLE C.4 ONBOARD DATA FREE FLIGHT FIT RESIDUAL STATISTICS

Insertion to CSI

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
CSM sextant shaft	22	-.002 deg	.022 deg
CSM sextant trunnion	22	.0003	.004
CSM VHF ranging	14	-37.8 feet	53.3 feet
LM RR shaft	2	.030 deg	.019 deg
LM RR trunnion	2	-.041	.017
LM RR range	2	251.7 feet	8.81 feet
LM RR range rate	2	-.476 fps	.227 fps

CSI to CDH

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
CSM sextant shaft	35	-.007 deg	.034 deg
CSM sextant trunnion	35	-.007	.010
CSM VHF ranging	34	-25.2 feet	177. feet
LM RR shaft	30	.050 deg	.017 deg
LM RR trunnion	30	-.075	.051
LL RR range	30	10.8 feet	251. feet
LM RR range rate	30	-.672 fps	.390 fps

CDH to TPI

<u>Data Type (station)</u>	<u>Number of Obs.</u>	<u>Mean</u>	<u>Std. Dev.</u>
CSM sextant shaft	8	.001 deg	.007 deg
CSM sextant trunnion	8	-.003	.006
CSM VHF ranging	10	-85.5 feet	30.2 feet
LM RR shaft	23	.083 deg	.016 deg
LM RR trunnion	23	-.082	.028
LM RR range	23	29.9 feet	53.5 feet
LM RR range rate	23	-.434 fps	.292 fps

APPENDIX D

LM RENDEZVOUS RADAR DATA, CSM VHF RANGING DATA AND CSM SEXTANT (APOLLO 12)

The LM rendezvous radar data that was used in the analysis are listed in the two card format of the HOPE orbit determination program. The first card specifies the vehicle taking the observation, the vehicle that is being observed, the time of the observation (year (mod 1900), month, day, hour, minute, and second (GMT)), three code numbers, shaft observable, trunnion observable, range observable, and range rate observable. The second card specifies the inner, middle, and outer gimbal angles. The units are feet, degrees, and seconds.

The CSM VHF ranging data are also listed in the same format. The card format differences are the following: 1) vehicle ID's are reversed, 2) code numbers are different, 3) range is the only observable, and 4) gimbal angles are not needed to process the ranging data.

The CSM sextant data are also listed. The card format is also similar to the rendezvous radar cards.

DOI TO POI

VEH1	VEH2	YMMDDHHMMSS	SSS	XF	SHAFT(DEG) INNER(DEG)	TRUN(DEG) MIDDLE(DEG)	RANGE(FT) OUTER(DEG)	RRATE(FPS)
CSM	LEM	691119	55424	539	62	.977783	20.376901	1
						328.3703613	3.6584473	2
CSM	LEM	691119	55427	398	62	.780029	20.511484	1
						328.2714844	3.6694336	2
CSM	LEM	691119	55513	129	62	.252686	22.648325	1
						326.7114258	3.9111328	2
CSM	LEM	691119	55543	867	114		5.2514648	1
							50978.608	2
CSM	LEM	691119	55646	348	114		55049.606	1
								2
CSM	LEM	691119	55752	566	114		59485.170	1
								2
CSM	LEM	691119	55853	609	114		63616.929	1
								2
CSM	LEM	691119	6	5	6.078114		95091.206055	1
								2
CSM	LEM	691119	6	7	3.359114		108701.7	1
								2

INS TO CSI

VEH1	VEH2	YYMMDDHHMMSS.SSS	XF	SHAFT(DEC) INNER(DEC)	TRUN(DEC) MIDDLE(DEC)	RANGE(FT) OUTER(DEC)	RRATE(FPS)
LEM	CSM	691120144847.130	11	358.7036 304.9036	359.4177 16.7322	1205892.8 .0330	-347.8011 2
LEM	CSM	691120145013.727	11	359.0881 299.8169	359.7363 2.8784	1176552.2 .0989	-331.4781 2
CSM	LEM	691120145040.914	62	359.066162 62.8198242	22.044077 .0983770	359.8242187 1151849.203	1 2
CSM	LEM	691120145128.984	114				1
CSM	LEM	691120145042.953	62	359.011230 62.7429199	22.016611 .0988770		2
CSM	LEM	691120145127.680	62	359.978027 60.3918457	21.934213 .2307129	359.7912598 .5822754 1133013.250	2 1 2
CSM	LEM	691120145229.437	114				1
CSM	LEM	691120145212.633	62	359.670410 57.2387695	22.631845 .3955078		2
CSM	LEM	691120145330.516	114			.6042480 1114724.141	2 1
CSM	LEM	691120145345.875	62	.087891 52.1411133	22.645578 .7470703	1.3623047 1089751.297	1 2 1
CSM	LEM	691120145413.984	62	359.615479 50.7897949	22.447824 .9008789	1.2524414	2
CSM	LEM	691120145515.219	62	.461426 49.9658203	19.918222 .5822754	1.1865234 1070550.781	1 2 1
CSM	LEM	691120145610.414	114				2
CSM	LEM	691120145558.977	62	1.153564 45.5273437	21.950693 .5822754	1.7358398	1 2
CSM	LEM	691120145625.648	62	.780029 43.9672852	22.033090 .4943848	1.3623047 1039137.266	1 2 1
CSM	LEM	691120145816.805	114				2
CSM	LEM	691120145721.773	62	.296631	23.005380		1

CSM	LEM	691120145918.023114	39.8913574	.4064941	.9448242 1025101.437	2 1
CSM	LEM	69112015 019.258114			1011916.266	2 1
CSM	LEM	69112015 120.703114			999399.469	2 1
CSM	LEM	691120145815.906 62	.285645	23.063059		2
CSM	LEM	69112015 124.430 62	36.8371582	.3295898	.8129883	1
CSM	LEM	69112015 124.430 62	359.439697	22.590647		2
CSM	LEM	69112015 720.234114	26.8066406	.5932617	.5163574 940339.625	2 1
CSM	LEM	69112015 828.758114			932622.961	2 1
CSM	LEM	69112015 831.000 62	359.549561	23.499765		2
CSM	LEM	69112015 930.328114	2.0654297	1.5820313	1.9226074 925756.945	1 2
CSM	LEM	69112015 832.273 62	359.527588	23.508005		1
CSM	LEM	69112015 933.789 62	1.9885254	1.5820313	1.9006348	2
CSM	LEM	69112015 953.695 62	.878906	22.730722		1
CSM	LEM	69112015 953.695 62	359.3627930	1.6259766	2.8234863	2
CSM	LEM	691120151045.430114	.714111	22.406626		1
CSM	LEM	691120151045.430114	358.571773	1.6259766	2.6477051 918101.039	2 1
CSM	LEM	691120151023.086 62	.109863	22.173166		2
CSM	LEM	691120151146.125114	357.1765137	1.6259766	2.2302246 912389.492	1 2
CSM	LEM	691120151048.984 62	359.494629	22.442331		1
CSM	LEM	691120151148.656 62	355.4736328	1.6918945	1.9226074	2
CSM	LEM	691120151148.656 62	359.143066	22.901010		1
CSM	LEM	691120144736.453 62	351.7272949	1.9226074	2.0214844	2
CSM	LEM	691120144744.875 62	.043945	22.005625		1
CSM	LEM	691120144830.258 62	72.7734375	359.2419434	359.3078613	2
CSM	LEM	691120144830.258 62	359.835205	21.967172		1
CSM	LEM	691120144830.258 62	72.3669434	359.2529297	359.1760254	2
CSM	LEM	691120144830.258 62	358.593750	22.222605		1
CSM	LEM	691120144830.258 62	69.6862793	359.4396973	358.6267090	2

CSI TO CDH

VEH1	VEH2	YMMUDDHMMSS.SSS	XF	SHAFT(DEG) INNER(DEG)	TRUN(DEG) MIDDLE(DEG)	RANGE(FT) OUTER(DEG)	RRATE(FPS)
LEM	CSM	691120154257.062	11	.1648	.7910	797900.3	-75.3361
LEM	CSM	691120154449.758	11	130.8801	357.1545	359.1650	2
LEM	CSM	691120154558.156	11	359.0332	359.4727	789270.7	-77.8471
LEM	CSM	6911201547 6.156	11	126.4197	358.4399	.5164	2
LEM	CSM	691120154813.977	11	.2087	.5713	784017.9	-80.3581
LEM	CSM	691120154921.914	11	121.8823	357.2754	359.3738	2
LEM	CSM	691120155139.508	11	.2856	.8020	778465.0	-81.6141
LEM	CSM	691120155243.75	11	118.4436	358.0884	359.1211	2
LEM	CSM	691120155350.523	11	359.8352	.0549	772837.0	-83.4971
LEM	CSM	691120155459.148	11	115.5542	358.1433	359.8792	2
LEM	CSM	6911201555715.109	11	359.8572	359.3079	767133.9	-86.0091
LEM	CSM	6911201556 7.109	11	112.2144	359.5386	.6152	2
LEM	CSM	691120155715.109	11	359.1541	359.0112	754977.4	-90.4031
LEM	CSM	691120155826.539	11	106.1829	359.8242	.9009	2
LEM	CSM	691120155934.453	11	.8789	359.2090	749124.3	-92.9141
LEM	CSM	691120155934.453	11	101.3049	.8130	.7031	2
LEM	CSM	691120155934.453	11	.1208	.3516	742821.0	-95.4261
LEM	CSM	691120155934.453	11	98.8220	1.5820	359.5386	2
LEM	CSM	691120155934.453	11	.0769	359.3188	736292.5	-97.3091
LEM	CSM	691120155934.453	11	95.5042	.8130	.5603	2
LEM	CSM	691120155934.453	11	.0220	.1208	729463.8	-100.4481
LEM	CSM	691120155934.453	11	92.2742	3.2410	359.7363	2
LEM	CSM	691120155934.453	11	359.3958	359.4946	722710.2	-102.9591
LEM	CSM	691120155934.453	11	89.6045	4.4934	.3186	2
LEM	CSM	691120155934.453	11	359.8572	.3296	715281.3	-106.0981
LEM	CSM	691120155934.453	11	85.6934	359.6265	359.5166	2
LEM	CSM	691120155934.453	11	359.6375	359.4617	708002.4	-108.6091
LEM	CSM	691120155934.453	11	82.6172	1.3403	.3625	2
LEM	CSM	691120155934.453	11	359.7253	.7581	700498.4	-111.1211
CSM	LEM	691120155938.195114		79.2773	359.4177	359.0771	2
CSM	LEM	6911201530 2.727 62		.560303	22.368173	853268.891	1
CSM	LEM	6911201530 2.727 62		294.9060059	359.4177246	359.2309570	2

CSM	LEM	691120153038.289114				849197.891	1
CSM	LEM	6911201530 6.680 62	.538330	22.403879			2
CSM	LEM	691120153039.117 62	294.6752930	359.4177246	359.2089844		1
CSM	LEM	691120153039.117 62	.351563	22.640085			2
CSM	LEM	691120153139.508114	292.7966309	359.3957520	359.0551758		1
CSM	LEM	691120153139.508114			845066.133		2
CSM	LEM	691120153111.516 62	.626221	21.439829			1
CSM	LEM	691120153136.094 62	292.3461914	359.1979980	358.9782715		2
CSM	LEM	691120153239.687114	.417480	22.758188			1
CSM	LEM	691120153239.687114	289.7973633	359.4067383	359.1320801		2
CSM	LEM	6911201532 5.750 62	359.912109	22.749948	841055.898		1
CSM	LEM	691120153244.727 62	288.3032227	359.4506836	358.8464355		2
CSM	LEM	691120153341.703114	.582275	22.500009			1
CSM	LEM	691120153341.703114	286.5893555	359.5385742	359.4396973		2
CSM	LEM	691120153441.906114			836863.383		1
CSM	LEM	691120153542.250114					2
CSM	LEM	691120153611.453 62	.549316	22.392893			1
CSM	LEM	691120153613.406 62	276.3061523	.1867676	.3735352		2
CSM	LEM	691120153745.219114			824468.102		1
CSM	LEM	691120153745.219114	.494385	22.368173			2
CSM	LEM	691120153745.219114	276.2292480	.1867676	.3295898		1
CSM	LEM	6911201537 8.953 62	358.901367	21.920480	820154.062		2
CSM	LEM	691120153751.758 62	273.9001465	.1538086			1
CSM	LEM	691120153821.656 62	359.692383	22.071543	359.2639160		2
CSM	LEM	691120154226.484 62	271.6040039	.1977539	359.8352051		1
CSM	LEM	691120154226.484 62	359.187012	22.129221			2
CSM	LEM	691120154228.906 62	270.0549316	.2087402	359.5275879		1
CSM	LEM	691120154228.906 62	.032959	21.854563			2
CSM	LEM	691120154228.906 62	258.1237793	359.1979980	358.7475586		1
CSM	LEM	691120154228.906 62	359.978027	21.857309			2

CSM	LEM	691120154311.713	62	258.0229297	359.2389844	358.7255859	2
				358.945312	22.208372		1
CSM	LEM	691120154342.375	62	255.5200195	359.4067383	358.3190918	2
				358.846436	22.068796		1
CSM	LEM	6911201545	62	254.1577148	359.4946289	358.4069824	2
				359.439697	22.140207		1
CSM	LEM	691120155311.883	62	249.8950195	358.8244629	357.8576660	2
				359.725342	22.206125		1
CSM	LEM	691120155314.523	62	226.0327148	359.0991211	358.6047363	2
				359.736328	22.200632		1
CSM	LEM	691120155339.063	62	225.9008789	359.0991211	358.6157227	2
				359.802246	22.153940		1
CSM	LEM	6911201554	62	224.7583008	359.1210937	358.7036133	2
				359.802246	22.090769		1
CSM	LEM	69112015544.312	62	223.7655312	359.1430664	358.7585449	2
				359.725342	22.678537		1
CSM	LEM	6911201555	62	221.0668945	359.3298340	358.9562988	2
				359.934082	22.785654		1
CSM	LEM	691120155522.258	62	219.7265625	359.4506836	359.2749023	2
				.560303	22.464304		1
CSM	LEM	6911201557	62	216.5295410	359.7473145	.1647949	2
				1.186523	22.129221		1
CSM	LEM	691120155845.258	62	214.6838379	359.8901367	.7800293	2
				359.454629	22.491770		1
CSM	LEM	691120155916.258	62	209.5751953	.2416992	.2087402	2
				359.725342	22.763681		1
CSM	LEM	691120155943.305	62	207.8063965	.3735352	.5603027	2
				359.714355	22.634592		1
CSM	LEM	69112016	62	206.6308594	.4614258	.6811523	2
				359.626465	22.398386		1
CSM	LEM	69112016	62	205.0598145	.6042480	.8569336	2
				.834961	21.914987		1
CSM	LEM	69112016	62	191.4367676	1.2304688	2.5927734	2
				359.670410	21.923227		1
CSM	LEM	69112016	62	189.8876953	1.5051270	2.2302246	2
				359.923096	21.846323		1
CSM	LEM	69112016	62	188.3715820	1.7797852	2.8015137	2
				358.923340	22.178659		1
CSM	LEM	69112016	62	185.6909180	2.0764160	2.5927734	2
				.900879	22.637338		1
CSM	LEM	69112016	62	182.8894043	1.8237305	3.6145020	2

CSM	LEM	691120161023.766114	629364.039	2
CSM	LEM	691120161124.219114	621282.805	1
CSM	LEM	691120153848.539114	815657.734	2
CSM	LEM	691120153952.227114	811161.414	1
CSM	LEM	691120154055.703114	806604.320	2
CSM	LEM	691120154159.312114	801925.719	1
CSM	LEM	6911201543 2.594114	797247.109	2
CSM	LEM	6911201544 5.766114	792386.211	1
CSM	LEM	6911201545 5.953114	787768.367	2
CSM	LEM	6911201546 8.094114	782907.477	1
CSM	LEM	691120155149.227114	753863.641	2
CSM	LEM	691120155250.023114	748273.617	1
CSM	LEM	691120155358.336114	741893.695	2
CSM	LEM	6911201555 2.383114	735696.055	1
CSM	LEM	6911201556 4.484114	729559.180	2
CSM	LEM	6911201557 7.406114	723240.023	1
CSM	LEM	6911201558 9.836114	716738.578	2
CSM	LEM	691120155912.156114	709994.086	1
CSM	LEM	69112016 016.039114	703188.836	2
CSM	LEM	69112016 119.930114	696140.547	1

CDH TO TPI

VEH1	VEH2	YMMDDHHMMSS	SSS	XF	SHAFT(DEG) INNER(DEG)	TRUN(DEG) MIDDLE(DEG)	RANGE(FT) OUTER(DEG)	RRATE(FPS)
CSM	LEM	691120162813	.687	62	359.637451	22.642832		1
					125.0244141	3.1091309	4.9218750	2
CSM	LEM	691120162816	.266	62	359.626465	22.653818		1
					124.8815918	3.1420898	4.9658203	2
CSM	LEM	691120162833	.239	62	359.461670	22.741709		1
					123.6730957	3.4497070	5.3063965	2
CSM	LEM	691120162920	.258	62	359.846191	22.467050		1
					121.9921875	3.7133789	5.8996582	2
CSM	LEM	691120163453	.703	62	.834961	22.236337		1
					106.7321777	2.5488281	4.8339844	2
CSM	LEM	691120163454	.688	62	.845947	22.247324		1
					106.6662598	2.5378418	4.8339844	2
CSM	LEM	691120163658	.812	62	.340576	22.291269		1
					100.9094238	2.4719238	4.4055176	2
CSM	LEM	691120163720	.227	62	.131836	22.093515		1
					100.1293945	2.4938965	4.2626953	2
LEM	CSM	691120162423	.500	11	357.890625	359.055176	509371.520	-143.1381
					13.4912109	1.4062500	.5493164	2
LEM	CSM	691120162525	.055	11	359.439697	.527344	500591.836	-142.5111
					9.0087891	1.6040039	359.1210937	2
LEM	CSM	691120162633	.000	11	359.549561	359.780273	490911.680	-142.5111
					5.6689453	3.3618164	359.8571777	2
LEM	CSM	691120162746	.703	11	359.417725	.186768	480481.117	-141.8831
					2.2851563	2.6806641	359.4506836	2
LEM	CSM	691120162852	.859	11	.307617	359.593506	471026.078	-142.5111
					358.2531738	1.1645508	.0659180	2
LEM	CSM	6911201630	.758	11	359.143066	359.318848	461495.996	-141.8831
					356.2426758	2.7795410	.2966309	2
LEM	CSM	6911201631	8.758	11	359.846191	.164795	451815.836	-141.2551
					352.3205566	2.9015137	359.4836426	2
LEM	CSM	691120163251	.203	11	.274658	359.703369	437408.156	-141.2551
					347.0800781	358.6267090	359.9340820	2
LEM	CSM	691120163326	.563	11	358.363037	359.516602	432380.477	-140.6271
					347.3327637	358.3300781	.1867676	2

LEM	CSM	691120163345.945	11	344.2675781	.527344	359.263916	429679.039	-141.2551	2
LEM	CSM	6911201634	5.250	359.890137		356.7810059	.3625488	-140.6271	2
LEM	CSM	691120163510.656	11	344.0258789	.120850	358.0224609	359.5385742	-140.6271	2
LEM	CSM	691120163618.516	11	340.7519531		.285645	417747.680	-140.6271	2
LEM	CSM	691120163725.406	11	358.714600		357.8466797	359.3847656	-140.6271	2
LEM	CSM	691120163833.680	11	339.0380859		357.3962402	408367.680	-139.9991	2
LEM	CSM	691120163941.680	11	358.945312		359.033203	398987.680	-139.3721	2
LEM	CSM	691120164049.594	11	335.7531738		2.0983887	.5932617	-138.7441	2
LEM	CSM	691120164157.578	11	359.022217		1.9116211	389382.559	-138.1161	2
LEM	CSM	69112016429.688	11	326.4257812		.384521	370547.520	-138.1161	2
LEM	CSM	6911201643	5.547	321.6796875		2.1643066	359.2639160	-137.4881	2
LEM	CSM	691120164413.797	11	359.472656		359.714355	361242.559	-136.8601	2
LEM	CSM	691120164521.367	11	319.9108887		3.5156250	.0109863	-136.2331	2
LEM	CSM	691120164629.688	11	317.2521973		.351563	351862.559	-136.2331	2
LEM	CSM	691120164737.750	11	359.132080		2.2302246	359.3298340	-135.6051	2
CSM	LEM	691120162621.656114		313.2641602		359.252930	342482.559	-136.2331	2
CSM	LEM	691120162940.383114		.164795		2.4279785	.4174805	-136.2331	2
CSM	LEM	691120163318.148114		313.2641602		.483398	333252.637	-136.2331	2
CSM	LEM	691120163421.930114		359.956055		1.6259766	359.2309570	-136.2331	2
CSM	LEM	691120163740.617114		310.5065918		359.417725	324022.719	-135.6051	2
CSM	LEM	691120163841.805114		359.318848		3.0541992	.3076172	-135.6051	2
CSM	LEM	691120164246.789114		308.2543945		359.000244	314792.797	-135.6051	2
CSM	LEM	691120164246.789114				1.9445801	.7031250	-135.6051	2
CSM	LEM	691120164246.789114					492408.395	-135.6051	2
CSM	LEM	691120164246.789114					464215.219	-135.6051	2
CSM	LEM	691120164246.789114					433470.074	-135.6051	2
CSM	LEM	691120164246.789114					424538.187	-135.6051	2
CSM	LEM	691120164246.789114					396648.816	-135.6051	2
CSM	LEM	691120164246.789114					388203.016	-135.6051	2
CSM	LEM	691120164246.789114					354359.051	-135.6051	2

CSM	LEM	691120164348.023114	345913.250	2
				1
				2
CSM	LEM	691120164449.297114	337588.973	1
				2
CSM	LEM	691120164550.555114	329264.695	1
				2

LM DEORBIT

VEH1	VEH2	Y Y M M D D H H M M S S . S S S	X F	SHAFT(DEG) INNER(DEG)	TRUN(DEG) MIDDLE(DEG)	RANGE(FT) OUTER(DEG)	RRATE(FPS)
CSM	LEM	691120215126.867114				25094.357	1
							2
CSM	LEM	691120215356.12 114				53287.532227	1
							2
CSM	LEM	691120215441.26 62		356.132812	22.192392		1
							2
CSM	LEM	6911202155 .016114		211.9812012	14.3481445	357.4401855	1
						65318.241	2
CSM	LEM	6911202156 3.83 114				77105.904297	1
							2
CSM	LEM	691120215610.12 62		357.604980	21.541452		1
				212.2668457	14.8974609	357.4182129	2
CSM	LEM	691120215615.48 62		357.868652	21.519480		1
				212.2778320	14.8974609	357.4951172	2
CSM	LEM	691120215636.40 62		357.121582	21.464548		1
				212.3657227	15.1831055	357.1325684	2
CSM	LEM	691120215714.91 114				89987.269531	1
							2
CSM	LEM	691120215830.547 62		356.528320	22.428598		1
				211.6735840	16.3146973	357.2204590	2
CSM	LEM	69112022 018.797114				122129.920	1
							2
CSM	LEM	69112022 420.445114				162839.893	1
							2

APPENDIX E

APOLLO 12 LANDING RADAR DATA

The LM landing radar data that was used in the analysis is listed in the two card format of the HOPE orbit determination program. The first card specifies the vehicle, the time of the observation (year (mod 1900), month, day, hour, minute, and second), three code numbers, V_{XA} measurement, V_{YA} measurement, V_{ZA} measurement, and the slant range measurement (ρ). The second card specifies the inner, middle, and the outer gimbal angles. The units are feet and feet per second.

LANDING RADAR OBSERVATIONS

VEH1	YMMDDHHMMSS.SSS	XF	VX(FPS) INNER(DEG)	VY(FPS) MIDDLE(DEG)	VZ(FPS) OUTER(DEG)	RANGE(FT)
LEM	691119	646 4.187125	83.3422852	359.0222168	358.1433105	47438.2341
LEM	691119	646 6.687125		-244.823999		2
LEM	691119	646 6.187125	83.1335449	359.0881348	358.0554199	1
LEM	691119	646 8.797125	83.1555176	359.0441895	358.0773926	2
LEM	691119	646 8.187125	83.1005859	359.2309570	358.0334473	47718.7741
LEM	691119	64610.797125-2947.201569	83.0017090	359.1760254	358.0334473	2
LEM	691119	64610.187125	83.1005859	359.1430664	357.9895020	47956.1551
LEM	691119	64612.797125	83.0017090	359.1320801	357.9785156	2
LEM	691119	64612.187125	82.9467773	359.0661621	357.9895020	1
LEM	691119	64614.797125	82.9138184	359.0332031	357.9895020	2
LEM	691119	64614.187125	82.7160645	359.0991211	357.9455566	47432.8391
LEM	691119	64616.797125-2865.928772	82.8149414	359.0551758	357.9565430	1
LEM	691119	64616.187125	82.2326660	359.3737793	357.9125977	2
LEM	691119	64618.797125	82.4414062	359.3737793	357.9345703	46445.5551
LEM	691119	64618.187125	82.0349121	359.0222168	357.8906250	2
LEM	691119	64620.867125	82.2106934	359.2199707	357.9016113	46256.7291
LEM	691119	64620.187125	81.8151855	358.7805176	357.9125977	2
LEM	691119	64622.797125-2805.779144	82.0129395	358.8684082	357.9235840	46024.7451
LEM	691119					2
LEM	691119					1

LEM	691119	64622.187125	81.5844727	358.8574219	357.9125977	45754.9951	2
LEM	691119	64624.887125	81.7163086	358.8134766	357.9125977	45754.9951	2
LEM	691119	64624.187125	81.5625000	-251.126400	357.8906250	45868.2901	2
LEM	691119	64626.859125	81.4855957	358.9782715	357.9016113	45868.2901	2
LEM	691119	64626.187125	81.3977051	359.3078613	357.8027344	45781.9701	2
LEM	691119	64628.879125-2735.067563	81.3757324	359.1870117	357.8137207	45781.9701	2
LEM	691119	64628.187125	80.9912109	359.4726562	357.7697754	45760.3901	2
LEM	691119	64630.887125	80.9582520	359.3627930	357.7587891	45760.3901	2
LEM	691119	64630.187125	80.9582520	-203.373598	357.8686523	45037.4591	2
LEM	691119	64632.879125	80.8374023	359.4396973	357.8247070	45037.4591	2
LEM	691119	64632.187125	81.1560059	359.0441895	358.0554199	44584.2801	2
LEM	691119	64634.906125-2694.882355	81.1120605	359.2529297	358.0114746	44951.1401	2
LEM	691119	64634.187125	80.9912109	358.9782715	358.2092285	44951.1401	2
LEM	691119	64636.879125	81.2658691	359.0222168	358.1982422	45226.2851	2
LEM	691119	64636.187125	80.5187988	-225.674400	358.3410645	45226.2851	2
LEM	691119	64638.879125	80.8703613	359.2199707	358.3190918	44109.5201	2
LEM	691119	64638.187125	80.0573730	359.9450684	358.4399414	44109.5201	2
LEM	691119	64640.187125	80.3100586	359.6923828	358.4179687	44406.2451	2
LEM	691119	64640.879125-2587.591949	79.7607422	.2636719	358.4948730	44406.2451	2
LEM	691119	64642.887125	79.8266602	.1977539	358.4729004	44406.2451	2
LEM	691119	64642.887125	79.7058105	-199.495199	358.4948730	44406.2451	2

LEM	691119	64642.188125	79.5080566	359.9780273	358.4729004	44368.4791
LEM	691119	64644.879125	79.7607422	359.5605469	1975.957260	2
LEM	691119	64644.187125	79.5520020	359.5495605	358.4619141	1
LEM	691119	64646.879125-2546.891174	79.8156738	359.5385742	358.4399414	2
LEM	691119	64646.187125	79.7277832	359.5275879	358.4179687	2
LEM	691119	64648.879125	-179.860798	358.3959961	43920.6951	2
LEM	691119	64648.187125	79.7058105	359.4506836	358.3410645	1
LEM	691119	64650.906125	79.8046875	359.5715332	358.3630371	2
LEM	691119	64650.187125	79.6289062	359.1430664	1916.668152	1
LEM	691119	64652.887125-2476.952759	79.8486328	359.3518066	358.2861328	2
LEM	691119	64652.187125	79.0905762	358.6596680	358.3081055	2
LEM	691119	64654.887125	79.4750977	358.8024902	358.2421875	1
LEM	691119	64654.187125	-249.429600	358.2861328	43505.2801	2
LEM	691119	64650.879125	79.1235352	358.9453125	358.3410645	1
LEM	691119	64656.187125	79.1894531	358.8354492	358.3190918	2
LEM	691119	64658.879125-2428.523956	78.8928223	359.3078613	1900.545670	1
LEM	691119	64658.187125	78.6840820	358.3740234	358.3850098	2
LEM	691119	647.879125	78.9587402	359.5935059	358.3959961	2
LEM	691119	647.188125	78.7170410	359.5275879	358.3850098	2
LEM	691119	647.188125	-195.131998	358.3850098	42744.5841	1
LEM	691119	647.188125	79.1015625	359.7143555	358.3850098	2
LEM	691119	647.188125	78.9038086	359.6923828	42620.5001	2
LEM	691119	647.2.879125	79.2993164	359.7583008	358.3959961	2
LEM	691119	647.2.188125			1818.719727	1
LEM	691119	647.2.188125			358.2861328	2
LEM	691119	647.2.188125			42571.9451	2

LEM	691119	647	4.906125-	79.3542480	359.8571777	358.3410645	2
				2390.656769			1
LEM	691119	647	4.137125	79.0136719	359.7143555	358.2092285	2
							42625.8951
LEM	691119	647	6.887125	79.4311523	359.7692871	358.2312012	2
				-196.343998			1
LEM	691119	647	6.187125	78.3764648	359.7912598	358.1323242	2
							42496.4151
LEM	691119	647	8.879125	78.8269043	359.7473145	358.1652832	2
					1844.897095		1
LEM	691119	647	8.187125	77.7502441	359.9890137	358.0773926	2
							41967.7051
LEM	691119	64710.879125-	77.9370117	359.8132324	358.0773926		2
			2275.251953				1
LEM	691119	64710.187125	77.4536133	.1538086	358.0114746		2
							41595.4501
LEM	691119	64712.379125	77.2558594	.0329590	358.0114746		2
			-193.192799				1
LEM	691119	64712.187125	77.9589844	.0329590	357.9455566		2
							41212.4051
LEM	691119	64714.879125	77.4206543	.0988770	357.9345703		2
				1747.815506			1
LEM	691119	64714.187125	79.1015625	359.8901367	357.8576660		2
							41401.2291
LEM	691119	64716.879125-	78.6071777	.0549316	357.8906250		2
			2304.489563				1
LEM	691119	64716.187125	79.6398926	359.8571777	357.8137207		2
							41536.1041
LEM	691119	64718.937125	80.0683594	359.9560547	357.8576660		2
			-209.191198				1
LEM	691119	64718.187125	78.5192871	359.8681641	357.7697754		2
							41854.4101
LEM	691119	64720.879125	79.3981934	359.6594238	357.8027344		2
					1754.229828		1
LEM	691119	64720.187125	77.0361328	.5053711	357.7697754		2
							41136.8751
LEM	691119	64722.379125-	77.5964355	.1977539	357.7917480		2
			2160.362366				1
LEM	691119	64722.187125	76.7944336	.4064941	357.8906250		2
							40424.7341
LEM	691119	64722.187125	76.1462402	.4504395	357.7478027		2

LEM	691119	64724.887125	77.9479980	-150.530399	359.7143555	358.1982422	40041.6891	1
LEM	691119	64724.187125	77.2998047	.0549316	358.0773926	1647.960144	40041.6891	2
LEM	691119	64726.879125	78.3654785	359.5495605	358.4509277		40322.2291	1
LEM	691119	64726.187125	78.3544922	359.6154785	358.3740234		40322.2291	2
LEM	691119	64728.879125-2131.857552	77.4865723	.0109863	358.6596680		40246.7001	1
LEM	691119	64728.187125	78.1347656	359.9780273	358.6157227		40246.7001	2
LEM	691119	64732.918125	77.2448730	.0659180	358.9672852		39361.9191	1
LEM	691119	64732.187125	76.9152832	359.9011230	358.9013672		39361.9191	2
LEM	691119	64734.879125-2086.044769	77.8161621	.0219727	359.0332031		39475.2151	1
LEM	691119	64734.187125	77.3657227	359.9890137	358.9892578		39475.2151	2
LEM	691119	64738.969125	76.3439941	.2856445	358.9672852		39011.2451	1
LEM	691119	64738.187125	77.0141602	.4724121	359.0222168		39011.2451	2
LEM	691119	64740.887125-1969.351974	75.1794434	.1428223	358.8134766		38363.8451	1
LEM	691119	64740.187125	75.8386230	.1208496	358.8793945		38363.8451	2
LEM	691119	64742.887125	76.2231445	-155.620798	358.6596680		37975.4051	1
LEM	691119	64742.188125	75.8935547	359.9340820	358.7036133		37975.4051	2
LEM	691119	64744.879125	77.4096680	359.7033691	358.5058594		38142.6491	1
LEM	691119	64744.187125	76.8164062	359.6813965	358.5278320		38142.6491	2
LEM	691119	64746.906125-1963.040787	76.8164062	.0659180	358.4289551		38148.0441	1
LEM	691119	64746.187125	77.0141602	.0439453	358.4399414		38148.0441	2
LEM	691119	64748.887125	-144.470398					1

LEM	691119	64748.187125	75.3332520	.2526855	358.3520508	37678.6801	2
LEM	691119	64750.887125	75.9045410	.5163574	358.4179687		2
LEM	691119	64750.187125	76.7175293	.3076172	1502.164383		1
LEM	691119	64752.879125-1909.975159	76.3879395	.4064941	358.2202148	37058.2541	2
LEM	691119	64752.187125	76.9372559	.4943848	358.1762695		2
LEM	691119	64756.887125	77.3547363	359.8791504	358.1542969	37149.9701	2
LEM	691119	64756.187125	75.9704590	359.0332031	1460.904709		1
LEM	691119	64758.387125-1841.968781	75.7067871	359.7802734	358.0114746	36448.6201	2
LEM	691119	64758.187125	76.6625977	359.5166016	357.9235840		2
LEM	691119	648 .949125	76.7724609	359.1760254	357.9895020	36324.5351	2
LEM	691119	648 .188125	76.2780762	-110.776799	357.9235840		1
LEM	691119	648 2.887125	76.7504883	.7910156	357.9565430	36346.1151	2
LEM	691119	648 2.188125	74.9047852	1.1425781	1428.833115		1
LEM	691119	648 4.879125-1768.295166	75.2893066	1.2634277	357.8466797	35801.2201	2
LEM	691119	648 4.187125	75.4321289	.0109863	357.8686523		2
LEM	691119	648 6.887125	74.5971680	.0439453	357.8247070	34981.1801	2
LEM	691119	648 6.187125	76.5527344	-154.166399	357.7478027		1
LEM	691119	648 8.879125	76.2121582	.0439453	357.9455566	35612.3951	2
LEM	691119	648 8.187125	75.0915527	.0988770	357.9125977		2
LEM	691119	648 8.187125	76.1352539	.8239746	1382.199265	35412.7801	1
LEM	691119	64810.879125-1668.475174	76.0107422	358.0114746		2	2
LEM	691119	64810.879125-1668.475174	73.6962891	.4943848	357.9345703		2

LEM	691119	64810.187125	74.1137635	.4284668	357.9785156	34786.9591
LEM	691119	64812.887125		-96.232799		2
LEM	691119	64812.187125	75.0476074	1.2634277	357.9455566	1
LEM	691119	64814.918125	74.4104004	.7910156	357.9345703	2
LEM	691119	64814.187125	74.2565918	359.6704102	357.8576660	34436.2851
LEM	691119	64816.887125	74.3334961	.3295898	357.8356934	2
LEM	691119	64816.887125	74.6740723	359.5825195	357.7697754	34468.6551
LEM	691119	64816.187125	74.5052773	359.6484375	357.8137207	2
LEM	691119	64818.887125		-123.866399		1
LEM	691119	64818.187125	74.2016602	.9228516	357.7807617	2
LEM	691119	64820.687125	74.7399902	.6921387	357.8027344	34123.3751
LEM	691119	64820.187125	73.6413574	1.1096191	357.7697754	2
LEM	691119	64822.879125	73.8830566	.9667969	357.7587891	33972.3141
LEM	691119	64822.187125	73.3117676	.1977539	357.8686523	2
LEM	691119	64824.979125	72.8283691	.1428223	357.7917480	33524.5301
LEM	691119	64824.187125	73.7402344	359.8352051	357.9016113	2
LEM	691119	64826.887125	73.3886719	.0109863	357.8796387	33141.4851
LEM	691119	64826.187125	73.7622070	359.9340820	357.9016113	2
LEM	691119	64828.949125	74.1357422	.0659180	357.9235840	33146.8801
LEM	691119	64828.187125	73.5205078	.6921387	357.8796387	2
LEM	691119	64830.887125	73.9529199	.4614258	357.9345703	33065.9551
LEM	691119	64830.137125		-102.535200		2
LEM	691119	64830.137125	73.4436035	1.0656738	358.2641602	1
LEM	691119	64830.137125				2
LEM	691119	64830.137125				32925.6851

LEM	691119	64832.879125	73.3776855	.7910156	358.0224609	32413.1601	2
LEM	691119	64832.187125	73.3007812	.7360840	1214.733505		1
LEM	691119	64834.879125	72.8613281	.7580566	358.9343262		2
LEM	691119	64834.187125	72.6306152	.3295898	359.5166016	32008.5351	2
LEM	691119	64836.879125	72.8063965	.4614258	359.3298340		2
LEM	691119	64836.187125	70.9277344	-81.931199			1
LEM	691119	64835.187125	71.9604492	.7910156	.0329590	31501.4051	2
LEM	691119	64838.977125	70.3344727	.8239746	359.8791504		2
LEM	691119	64838.187125	70.2685547	1.5051270	1226.001907		1
LEM	691119	64840.887125	72.7404785	1.8896484	.5273438	31053.6201	2
LEM	691119	64840.187125	71.6308594	1.1096191	.3515625		2
LEM	691119	64842.949125	72.1472168	1.6589355	.6921387	31026.6451	2
LEM	691119	64842.188125	72.1362305	-95.990399			1
LEM	691119	64845.000125	71.1584473	.4504395	1.0437012	31032.0401	2
LEM	691119	64844.187125	71.5429687	.9997559	.9558105		2
LEM	691119	64846.898125	70.6091309	.7910156	1144.175980		1
LEM	691119	64846.187125	71.3122559	.6811523	1.1645508	30568.0701	2
LEM	691119	64848.898125	70.9497070	1.1206055	1.1755371		2
LEM	691119	64848.187125	70.4882812	1.1206055	1.2084961	30373.8501	2
LEM	691119	64850.879125	72.5537109	-90.415200			1
LEM	691119	64850.187125	71.6857910	.9558105	.9887695	30093.3101	2
LEM	691119	64850.879125	70.9277344	.5053711	1.0546875		2
LEM	691119	64850.187125	72.5537109	1.0107422	1058.882874		1
LEM	691119	64850.187125	71.6857910	.9887695	.7910156	30087.9151	2
LEM	691119	64850.187125	71.6857910	.9887695	.8569336		2

LEM	691119	64852.887125-1231.585587	71.7077637	1.4282227	.5273438	1
LEM	691119	64852.887125	72.1032715	1.7028809	.6262207	2
LEM	691119	64854.949125	69.5434570	-105.443999		2
LEM	691119	64854.187125	70.6420898	.9667969	.1208496	1
LEM	691119	64856.887125	69.5544434	1.2854004	.2966309	2
LEM	691119	64856.187125	69.5544434	1.5930176	1062.696793	1
LEM	691119	64858.879125-1153.919189	70.8178711	1.2854004	359.6923828	2
LEM	691119	64858.187125	70.8178711	.9228516	359.8352051	2
LEM	691119	649 .887125	70.1147461	2.2412109	359.4396973	1
LEM	691119	649 .188125	70.7519531	1.7907715	359.4067383	2
LEM	691119	649 2.867125	70.5541992	-66.660000		1
LEM	691119	649 2.188125	70.4663086	1.2194824	359.3957520	2
LEM	691119	649 4.859125-1110.127182	70.6201172	1.8566895	359.4177246	2
LEM	691119	649 4.187125	70.4882812	.3735352	359.3957520	1
LEM	691119	649 6.867125	70.5981445	.9667969	359.4396973	2
LEM	691119	649 6.187125	70.8508301	1.4062500	359.4177246	2
LEM	691119	649 8.957125	69.6572656	1.4721680	359.4287109	2
LEM	691119	649 8.187125	69.6862793	-76.113600		1
LEM	691119	64910.898125-1064.403183	69.4665527	.9228516	359.3737793	2
LEM	691119	64910.187125	69.1479492	1.8676758	359.3957520	2
LEM	691119	64912.854125	69.1479492	.0988770	359.3957520	2
LEM	691119	64912.854125	69.1479492	.4064941	359.3078613	2
LEM	691119	64912.854125	69.1479492	1.3842773	359.5056152	1
LEM	691119	64912.854125	69.1479492	1.0766602	359.4177246	2
LEM	691119	64912.854125	69.1479492	-84.355200		1

LEM	691119	64912.187125	68.1701660	1.1975098	359.5166016	27212.3801	2
LEM	691119	64914.969125	68.1042480	1.2414551	359.4946289		2
LEM	691119	64914.187125	68.0163574	1.7028809	956.253746		1
LEM	691119	64916.859125-1024.217575	67.2253418	1.1645508	359.4946289	20786.1751	2
LEM	691119	64916.187125	68.4997559	2.2302246	359.4836426		2
LEM	691119	64918.867125	67.9064941	1.9665527	359.4506836	26694.4601	2
LEM	691119	64918.187125	66.9946289	-97.929600			1
LEM	691119	64920.859125	67.1044922	.3625488	359.4836426		2
LEM	691119	64922.867125-1002.707985	67.4450684	.6701660	359.4171246	26396.9451	2
LEM	691119	64922.187125	66.6210937	.6042480	921.755104		1
LEM	691119	64924.867125	68.8293457	.6042480	359.5385742		2
LEM	691119	64924.187125	68.6315918	359.6923828	359.4506836	25971.5301	2
LEM	691119	64924.867125	66.7309570	1.8566895	359.5495605		1
LEM	691119	64924.187125	68.0053711	1.3842773	359.5275879	26025.4801	2
LEM	691119	64926.867125	65.1049805	-84.355200			1
LEM	691119	64926.187125	65.4565430	.9338379	359.4396973		2
LEM	691119	64928.898125	67.2802734	.8679199	359.4616099	25825.8651	2
LEM	691119	64928.187125	66.8627930	1.3732910	913.433830		1
LEM	691119	64930.867125	67.0056152	1.7141113	359.4726562	25065.1701	2
LEM	691119	64930.187125	68.0932617	1.4392090	359.7253414		1
LEM	691119	64932.859125	64.8522949	-60.357600		25054.3801	2
LEM	691119	64932.859125	64.8522949	1.1975098	359.9450684		1
LEM	691119	64932.859125	64.8522949	1.0766602	359.8911367	24903.3201	2
LEM	691119	64932.859125	64.8522949	878.588470			1
LEM	691119	64932.859125	64.8522949	1.2854004	.1208496		2

LEM	691119	64954.859125	63.3361816	1.4282227	359.8571777	20754.5651	2
LEM	691119	64954.187125	63.7426758	-75.628799			1
LEM	691119	64956.859125	63.5778809	1.0546875	359.8022461		2
LEM	691119	64956.187125	63.7866211	.8509336	359.8242187		2
LEM	691119	64958.859125	63.7866211	1.4282227	728.805435		1
LEM	691119	64958.187125	-772.542389	1.4282227	359.7143555		2
LEM	691119	64958.187125	63.5998535	.9118652	359.7583008	20447.0501	2
LEM	691119	650.859125	63.8305664	2.2521973	359.8461914		2
LEM	691119	650.188125	62.8637695	1.6369629	359.7473145	20128.7451	2
LEM	691119	650.188125	63.4240723	-32.481600			1
LEM	691119	650.188125	61.6992187	3.2080078	.1647949		2
LEM	691119	650.188125	62.3364258	2.6477051	.0439453	19724.1201	2
LEM	691119	650.188125	-725.401588	3.3178711	710.429268		1
LEM	691119	650.188125	61.9848633	3.3068848	.3955078		2
LEM	691119	650.188125	62.1826172	1.5380859	.4064941	19362.6551	2
LEM	691119	650.188125	62.1496582	2.1862793	.3076172		2
LEM	691119	650.188125	62.8088379	-62.296800	.4064941	18995.7951	2
LEM	691119	650.188125	62.2595215	1.0546875	.4284668		1
LEM	691119	650.188125	62.5671387	.9118652	.4284668	18574.9851	2
LEM	691119	650.188125	-691.784790	2.6257324	661.715111		1
LEM	691119	650.188125	61.2597656	2.6257324	.4833984	18305.2351	2
LEM	691119	650.188125	62.1386719	2.3400879	.4943848		2
LEM	691119	650.188125	60.1062012	2.1862793	.4394531		1
LEM	691119	650.188125	60.1062012	2.8894043	.5053711	17970.7451	2
LEM	691119	650.188125	60.4577637	-61.327199			1
LEM	691119	650.188125	60.4577637	1.0217285	.2966309		2
LEM	691119	650.188125	60.4577637	1.4172363	.3845215	17485.1951	2

LEM	691119	65014.567125	61.1938477	1.9995117	629.816372 .1098633	1
LEM	691119	65014.187125	60.8203125	1.8347168	.1867676	2
LEM	691119	65016.867125	-672.851189	2.5438281	.1098633	2
LEM	691119	65016.187125	62.0068359	2.8564453	.1098633	2
LEM	691119	65018.867125	61.1718750	-62.054400	.1318359	1
LEM	691119	65018.187125	61.3916016	1.8996484	.1208496	2
LEM	691119	65020.859125	61.8200684	1.6918345	587.517036	1
LEM	691119	65020.187125	61.3037109	1.5380859	.1757813	2
LEM	691119	65022.898125	-650.311195	2.4389648	.1647949	2
LEM	691119	65022.187125	62.5122070	2.2741699	.3295898	2
LEM	691119	65024.867125	60.9741211	-23.755200	.2197266	1
LEM	691119	65024.187125	61.3146973	3.1311035	.7800293	2
LEM	691119	65026.867125	59.9853516	2.8234863	.5932617	2
LEM	691119	65026.187125	59.6777344	3.5156250	569.487595	1
LEM	691119	65028.867125	-602.783989	3.5485840	1.2194824	2
LEM	691119	65028.187125	60.5736133	2.7136230	1.0766602	2
LEM	691119	65030.867125	60.0073242	2.8234863	1.4611816	2
LEM	691119	65030.187125	61.0620117	-23.270400	1.3732910	2
LEM	691119	65032.867125	60.6665039	2.2412109	1.6589355	1
LEM	691119	65032.187125	60.6115723	2.0874023	1.5820313	2
LEM	691119	65034.867125	60.6994629	2.8234863	515.225914	1
LEM	691119	65034.867125	-565.174393	2.2851563	1.8457031	2
LEM	691119	65034.867125	-565.174393	2.2851563	1.7797852	2
LEM	691119	65034.867125	-565.174393	2.2851563	1.7797852	1

LEM	691119	65034.187125	59.6118164	3.5156250	2.0104980	13633.1651	2
LEM	691119	65036.867125	59.8425293	3.2080078	1.9775391		2
LEM	691119	65036.187125	59.8205566	-28.845600	1.9885254		1
LEM	691119	65036.187125	59.7985840	2.6586914		13217.7501	2
LEM	691119	65038.867125	59.9743652	1.4392090	2.0214844		2
LEM	691119	65038.187125	60.1940918	1.5930176	481.247353		1
LEM	691119	65040.906125	-537.095993		1.8896484		2
LEM	691119	65040.187125	59.3041992	1.8786621		12921.0251	2
LEM	691119	65040.187125	59.9523926	1.4172363	1.9116211		2
LEM	691119	65042.867125	58.8427734	-10.180800	1.7907715		1
LEM	691119	65042.188125	59.4360352	3.2409668	1.8237305	12511.0051	2
LEM	691119	65044.867125	59.5349121	2.9333496	1.7028809		2
LEM	691119	65044.187125	59.7546387	2.7795410		12187.3051	2
LEM	691119	65046.867125	-512.108788	1.8786621	1.5161133		1
LEM	691119	65046.187125	59.9853516	1.9885254	1.0327148		2
LEM	691119	65048.867125	60.3918457	-21.573600	1.1865234	11583.0651	2
LEM	691119	65048.187125	59.5458984	2.3291016			1
LEM	691119	65048.187125	60.1611328	2.2521973	.7250977	11210.8101	2
LEM	691119	65050.867125	58.9416504	2.6916504	.8569336		2
LEM	691119	65050.187125	59.3261719	2.7136230	408.089436		1
LEM	691119	65052.859125	-480.166393	3.1201172	.5053711	10822.3701	2
LEM	691119	65052.187125	59.5349121		.5712891		1
LEM	691119	65054.859125	59.1943359	3.0432129	.4174805	10455.5101	2
LEM	691119	65054.859125	60.6994629	-12.120000	.4614258		1
LEM	691119	65054.859125		2.9992676	.2966309		2

LEM	691119	65054.187125	60.1062012	3.1970215	.3405762	10110.2301
LEM	691119	65056.359125	61.9519043	2.4389648	337.531918	2
LEM	691119	65056.187125	61.1718750	2.6367188	.1208496	1
LEM	691119	65058.898125	-468.059196	2.5817871	.1867676	2
LEM	691119	65053.187125	62.2155762	2.5817871	359.9450684	9813.5051
LEM	691119	651	61.6592187	2.5488281	.0000000	2
LEM	691119	651	60.4687500	-18.180000		1
LEM	691119	651	60.6005859	2.0434570	359.7802734	2
LEM	691119	651	59.5129395	2.1862793	359.7912598	9128.3401
LEM	691119	651	59.1503906	2.5378418	315.861919	2
LEM	691119	651	-421.819996	2.1203613	359.8132324	1
LEM	691119	651	59.2712402	2.8564453	.0000000	2
LEM	691119	651	59.0295410	2.8344727	359.9780273	8297.5101
LEM	691119	651	58.9196777	-15.028800		2
LEM	691119	651	58.8647461	1.9555664	.0329590	1
LEM	691119	651	59.2932129	2.1972656	.0329590	2
LEM	691119	651	58.9526367	1.7578125	272.001835	1
LEM	691119	65120.750125	-301.391994	1.4062500	.0000000	2
LEM	691119	65120.187125	26.9824219	.8239746	3.1530762	7579.9751
LEM	691119	65122.867125	27.2680664	.8129883	3.1311035	2
LEM	691119	65122.187125	27.4658203	-21.573600		1
LEM	691119	65122.187125	27.4438477	1.1865234	2.9992676	2
LEM	691119	65124.867125	27.6306152	1.7578125	3.0981445	5049.7201
LEM	691119	65124.167125		1.5490723	256.572796	2
LEM	691119	65124.167125			2.8344727	1
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2.702	LEM	691119	65126.879125	27.4548340 -277.177597	1.7138672	2.9003906	2
2.230	LEM	691119	65126.187125	28.1799316	1.1645508	2.5598145	1
2.131	LEM	691119	65128.867125	27.8613281	1.8676758	2.7026367	2
18.42	LEM	691119	65128.867125	28.1909180	-25.694400	2.6257324	4504.8251
2.098	LEM	691119	65128.187125	27.9162598	1.1645508	2.6257324	2
2.010	LEM	691119	65130.867125	28.6303711	1.1535645	2.5488281	4240.4701
1.999	LEM	691119	65130.187125	28.1469727	1.1096191	2.8564453	2
2.340	LEM	691119	65132.867125	-252.834396	1.3293457	2.7465820	1
2.395	LEM	691119	65132.187125	28.5974121	1.1865234	2.9553223	2
1.757	LEM	691119	65134.867125	28.4106445	.7360840	2.8894043	3781.8951
25.20	LEM	691119	65134.867125	28.6633301	-21.089800	3.0212402	2
4.240	LEM	691119	65136.906125	28.5205078	.4943848	3.0102539	3555.3051
3.427	LEM	691119	65136.187125	28.7841797	1.2524414	2.9882813	2
3.581	LEM	691119	65138.867125	28.7841797	.9118652	3.0212402	3328.7151
3.933	LEM	691119	65138.867125	-225.915195	.5163574	2.8125000	1
3.175	LEM	691119	65138.187125	28.6523437	.7690430	2.8784180	2
3.240	LEM	691119	65140.879125	28.7841797	-25.936800	2.5598145	3134.4951
-18.66	LEM	691119	65140.879125	28.7841797	.2087402	2.6477051	1
3.625	LEM	691119	65142.867125	28.9599609	.3076172	2.6477051	2
3.383	LEM	691119	65142.867125	28.6853027	.8020020	2.4389648	1
3.394	LEM	691119	65144.758125	-202.602396	.8679199	2.4609375	2
3.515	LEM	691119	65142.188125	28.8061523	1.2524414	2.4279785	2773.0301
2.680	LEM	691119	65146.879125	28.7292480	-26.421600	.7800293	2
2.983	LEM	691119	65146.879125	29.2016602	2.2961426		1

54.2871094	346.4978027	268.6711	5146.187125	28.6962891	1.2744141	2.2851563	2461.1991
1.454400		2	5148.879125			190.002558	2
56.5283203	345.7067871	1	5148.187125	28.2348633	4.1857910	357.8027344	1
		2					2
55.5725098	345.9704590	257.8811	5150.867125	28.3117676	3.2629395	358.0444336	2310.1391
7.281120		2		-178.774397			2
54.6716309	346.8603516	1	5150.187125	28.3557129	3.7902832	358.0114746	1
		2					2
56.5612793	345.6518555	248.1701	5152.867125	27.9382324	3.6584473	357.7148437	2161.2371
		2			-25.452000		2
47.2119141	347.4755859	1	5152.137125	28.3447266	3.9001465	357.7587891	1
		2					2
50.4528809	346.2890625	237.3801	5154.867125	28.7072754	3.9550781	357.9235840	2039.3101
11.635200		2				179.427599	2
47.2119141	347.4755859	1	5154.187125	27.6745605	4.3725586	357.3413086	1
		2					2
46.7944336	347.0361328	231.9851	5156.898125	28.3776855	4.6472168	357.5939941	1917.3831
	8.321280	2		-156.234398			2
47.2119141	347.4755859	1	5156.187125	27.6833008	4.1967773	357.9675293	1
-7.514400		2					2
3.1311035	344.4323730	157.5341	5158.867125	27.8942871	4.3835449	357.8906250	1800.8511
		2			-23.270400		2
3.5046387	344.3994141		5158.187125	28.3117676	3.6474609	357.6928711	1
	7.974560	2					2
2.9492676	348.4533691	144.5861	52.867125	28.1030273	3.6364746	357.6818848	1678.9241
		2				163.478479	2
3.1530762	346.0363770		52.188125	27.9821777	3.0761719	357.4621582	1
		2					2
2.7136230	359.0441895	134.8751	52.2.867125	27.8942871	3.1860352	357.4182129	1578.5771
		2		-136.527998			2
2.5708008	355.0891113		52.2.188125	27.5976562	2.6037598	357.7917480	1
-1.939200		134.8751					2
2.1643066	7.4487305	2	52.4.859125	27.6306152	2.8784180	357.8027344	1471.7561
		1			-19.634400		2
2.6586914	2.2521973	123.0061	52.4.187125	27.4328613	2.6367188	358.1542969	1
	7.801200	2					2
1.7578125	9.7668457		52.6.867125	27.7734375	2.7246094	357.9895020	1383.2781
		1				152.730160	2
1.8896484	9.6459961	111.1371	52.6.137125	25.8837891	2.9003906	358.3410645	1
		2					2
		1					1296.9581

LEM	691119	65223.867125	23.8623047	-10.665600	1
LEM	691119	65223.137125	2.1972656	354.6276855	2
LEM	691119	65230.867125	24.2138672	2.2741699	633.3731
LEM	691119	65230.137125	24.2907715	1.9006348	2
LEM	691119	65232.867125	24.1259766	2.0544434	594.5291
LEM	691119	65232.137125	-64.271199	354.0893555	2
LEM	691119	65232.867125	24.4226074	1.3073730	1
LEM	691119	65232.137125	24.0380859	1.6918945	560.0011
LEM	691119	65234.879125	25.7629395	-11.392800	2
LEM	691119	65234.137125	25.2355957	.1538086	1
LEM	691119	65236.879125	27.8942871	.3295898	531.9471
LEM	691119	65236.187125	27.8283691	354.8034668	2
LEM	691119	65238.867125	-59.763199	90.667279	1
LEM	691119	65238.187125	27.6745605	359.9670410	500.6561
LEM	691119	65240.867125	27.5317383	359.8352051	2
LEM	691119	65240.187125	27.0593262	.0329590	471.5231
LEM	691119	65242.879125	27.2460938	-10.180800	2
LEM	691119	65242.188125	26.6857910	359.3957520	1
LEM	691119	65244.859125	27.2241211	355.1660156	2
LEM	691119	65244.187125	-48.042399	355.1000977	1
LEM	691119	65246.859125	25.8947754	75.584959	2
LEM	691119	65246.187125	26.0156250	359.5056152	433.7581
LEM	691119	65246.066125	25.4333496	353.3203125	2
LEM	691119	65246.187125	25.0156250	359.4946289	403.5461
LEM	691119	65246.187125	25.4333496	-10.423200	2
LEM	691119	65246.187125	25.0156250	.8569336	1
LEM	691119	65249.066125	25.0917871	351.3317871	2
LEM	691119	65249.066125	25.0917871	.8459473	388.4401
LEM	691119	65249.066125	25.0917871	351.2109375	2
LEM	691119	65249.066125	25.0917871	63.276400	1

LEM	691119	65248.187125	26.0375977	.7141113	351.1340332	361.4651	2
LEM	691119	65251.258125	26.0375977	.8239746	351.0351562	361.4651	2
LEM	691119	65250.187125	-35.291200	1.2524414	350.9252930	351.7541	2
LEM	691119	65253.289125	26.8066406	.7690430	351.2109375	351.7541	2
LEM	691119	65252.187125	26.9604492	-6.544800	351.1010742	345.2801	2
LEM	691119	65255.289125	27.3669434	2.9992676	351.1010742	345.2801	2
LEM	691119	65254.187125	27.0593262	2.2521973	351.0461426	335.5691	2
LEM	691119	65257.297125	27.5097656	3.9221191	350.7934570	335.5691	2
LEM	691119	65256.187125	27.4987793	3.7463379	350.8813477	335.5691	2
LEM	691119	65259.309125	-27.563200	3.9660645	349.6948242	323.7001	2
LEM	691119	65258.187125	27.3229980	3.8452148	350.9033203	323.7001	2
LEM	691119	653 1.289125	27.5317383	-3.151200	350.9033203	315.0681	2
LEM	691119	653 1.289125	27.0153809	3.7243652	347.9919434	315.0681	2
LEM	691119	653 1.289125	27.1582031	3.9770508	349.4091797	309.6731	2
LEM	691119	653 1.289125	26.6967773	3.4277344	346.5856934	309.6731	2
LEM	691119	653 1.289125	26.8505859	3.7792969	346.8054199	302.1201	2
LEM	691119	653 1.289125	-20.736799	.3955078	346.4648437	302.1201	2
LEM	691119	653 1.289125	25.9716797	.8898926	346.4538574	289.1721	2
LEM	691119	653 1.289125	26.1254883	1.454400	346.5527344	289.1721	2
LEM	691119	653 1.289125	25.6420898	359.6484375	346.5527344	279.4611	2
LEM	691119	653 1.289125	25.7849121	.1208496	346.5307617	279.4611	2
LEM	691119	653 1.289125	27.0263672	355.4846191	346.5527344	279.4611	2
LEM	691119	653 1.289125	26.7956543	358.8354492	346.6625977	279.4611	2
LEM	691119	653 1.289125	-13.524000	346.6625977	346.6625977	279.4611	2
LEM	691119	653 1.289125	23.2580566	354.8583984	346.1901855	279.4611	2

LEM	691119	653	8.187125	26.6748047	354.2871094	346.4978027	268.6711
LEM	691119	65311	.297125		1.454400		2
LEM	691119	65310	.187125	16.3256836	356.5283203	345.7067871	1
LEM	691119	65313	.289125				2
LEM	691119	65312	.187125	20.2587891	355.5725098	345.9704590	257.8811
LEM	691119	65315	.297125			7.281120	2
LEM	691119	65314	.187125	14.3920898	354.6716309	346.8603516	1
LEM	691119	65315	.297125	15.2270508	356.5612793	345.6518555	2
LEM	691119	65314	.187125	-6.182400			1
LEM	691119	65315	.297125	12.3925781	347.2119141	347.4755859	2
LEM	691119	65316	.187125	12.6342773	350.4528809	346.2890625	237.3801
LEM	691119	65315	.297125		11.635200		2
LEM	691119	65316	.187125	12.3925781	347.2119141	347.4755859	1
LEM	691119	65315	.297125	10.8764648	346.7944336	347.0361328	2
LEM	691119	65314	.187125			8.321280	1
LEM	691119	65315	.297125	12.3925781	347.2119141	347.4755859	2
LEM	691119	65314	.187125		-7.514400		1
LEM	691119	65315	.297125	3.2189941	3.1311035	344.4323730	2
LEM	691119	65316	.187125	3.5375977	3.5046387	344.3994141	157.5341
LEM	691119	65342	.879125			7.974560	2
LEM	691119	65345	.047125	3.8122559	2.9992676	348.4533691	1
LEM	691119	65344	.187125	3.9660645	3.1530762	346.0363770	2
LEM	691119	65347	.199125	-6.955200			1
LEM	691119	65344	.187125	3.6254883	2.7136230	359.0441895	2
LEM	691119	65347	.199125	3.3618164	2.5708008	355.0891113	134.8751
LEM	691119	65346	.187125		-1.939200		2
LEM	691119	65349	.289125	3.7023926	2.1643066	7.4487305	1
LEM	691119	65348	.187125	3.6694336	2.6586914	2.2521973	2
LEM	691119	65348	.187125	4.4055176	1.7578125	7.801200	1
LEM	691119	65351	.359125	4.1967773	1.8896484	9.7668457	2
LEM	691119	65351	.359125	-5.409600			1

LEM	691119	65350.187125	4.2736816	1.4941406	9.4592285	96.0311	2
LEM	691119	65353.289125	4.3725586	1.5820313	9.8657227		2
LEM	691119	65352.187125	4.4055176	1.6809082	9.4702148		1
LEM	691119	65355.309125	4.4165039	1.6149902	8.9428711	86.3201	2
LEM	691119	65354.187125	5.5920410	1.9006348	5.200800		1
LEM	691119	65356.187125	4.3945313	1.6589355	8.6352539		2
LEM	691119	65357.289125	5.9765625	1.9006348	9.1845703	77.6881	2
LEM	691119	65357.289125	-3.348800	1.7687988	9.0087391	71.2141	2
LEM	691119	65357.289125	5.9106445	1.7687988	9.3383789		1
LEM	691119	65357.289125	5.9106445	1.7687988	9.3383789		2
LEM	691119	654 8.859125	5.9106445	1.7687988	4.507360		1
LEM	691119	654 8.187125	-1.030400	1.7687988	9.3383789		2
LEM	691119	65410.957125	2.1862793	.7031250	351.7712402	56.1081	2
LEM	691119	65410.957125	2.2961426	.7580566	351.7932129		2
LEM	691119	65410.187125	2.2082520	-.969600	351.8371582		1
LEM	691119	65410.187125	2.2082520	.8020020	351.8371582	55.0291	2
LEM	691119	65413.090125	2.2082520	.8020020	351.3647461		2
LEM	691119	65412.187125	2.2192383	.6042480	.520080		1
LEM	691119	65415.199125	2.1643006	.5822754	351.6284180	45.3181	2
LEM	691119	65414.187125	-1.2889000	.6701660	351.7932129		1
LEM	691119	65417.297125	2.5488281	.7360840	351.3867187	43.1601	2
LEM	691119	65416.187125	2.4279785	1.454400			1
LEM	691119	65419.309125	356.1437988	1.5490723	351.9639941	43.1601	2
LEM	691119	65419.309125	.0769043	.9118652	351.8811035		2
LEM	691119	65419.309125	354.4738770	1.8566895	352.3095703		1

APPENDIX F
LM FREE FLIGHT TRAJECTORY, REV. 14
REGION OF MARE NECTARIS

ALTITUDE ABOVE MLR (FEET)	SELENOGRAPHIC LATITUDE (DEGREES)	SELENOGRAPHIC LONGITUDE (DEGREES)	GET (HR-MN- SEC)
158129.0	-15.1012	67.9194	109 58 .008
153844.2	-15.1773	66.2583	109 58 30.008
149594.1	-15.2414	64.5937	109 59 .008
145382.2	-15.2934	62.9257	109 59 30.008
141211.8	-15.3331	61.2546	110 0 .008
137086.3	-15.3604	59.5807	110 0 30.008
133009.0	-15.3754	57.9040	110 1 .008
128983.2	-15.3779	56.2249	110 1 30.008
125012.1	-15.3678	54.5435	110 2 .008
121099.0	-15.3452	52.8601	110 2 30.008
117247.1	-15.3099	51.1749	110 3 .008
113459.5	-15.2621	49.4882	110 3 30.008
109739.3	-15.2016	47.8002	110 4 .008
106089.5	-15.1285	46.1111	110 4 30.008
102513.3	-15.0428	44.4212	110 5 .008
99013.5	-14.9446	42.7306	110 5 30.008
95593.0	-14.8339	41.0397	110 6 .008
92254.7	-14.7107	39.3486	110 6 30.008
89001.4	-14.5752	37.6576	110 7 .008
85835.8	-14.4275	35.9669	110 7 30.008
82760.5	-14.2677	34.2767	110 8 .008
79778.2	-14.0958	32.5871	110 8 30.008
76891.3	-13.9120	30.8985	110 9 .008
74102.2	-13.7166	29.2110	110 9 30.008
71413.4	-13.5095	27.5248	110 10 .008
68827.1	-13.2911	25.8400	110 10 30.008
66345.6	-13.0614	24.1569	110 11 .008
63970.8	-12.8207	22.4755	110 11 30.008
61705.0	-12.5693	20.7960	110 12 .008
59549.9	-12.3072	19.1186	110 12 30.008
57507.5	-12.0348	17.4434	110 13 .008
55579.4	-11.7522	15.7704	110 13 30.008
53767.5	-11.4599	14.0999	110 14 .008
52073.2	-11.1579	12.4318	110 14 30.008
50497.9	-10.8465	10.7663	110 15 .008
49043.2	-10.5262	9.1034	110 15 30.008
47710.2	-10.1970	7.4432	110 16 .008
46500.0	-9.8594	5.7857	110 16 30.008
45413.9	-9.5136	4.1310	110 17 .008
44452.6	-9.1600	2.4791	110 17 30.008
43617.1	-8.7988	.8299	110 18 .008
42908.1	-8.4304	-.8164	110 18 30.008
42326.2	-8.0551	-2.4600	110 19 .008
41872.0	-7.6733	-4.1008	110 19 30.008
41464.3	-7.1521	-6.2940	110 20 9.500

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