



National Aeronautics and
Space Administration

NASA earth

Sachidananda Babu

ESTO/Technology Validation Program Manager
NASA Earth Science Division





EARTH FLEET

Key

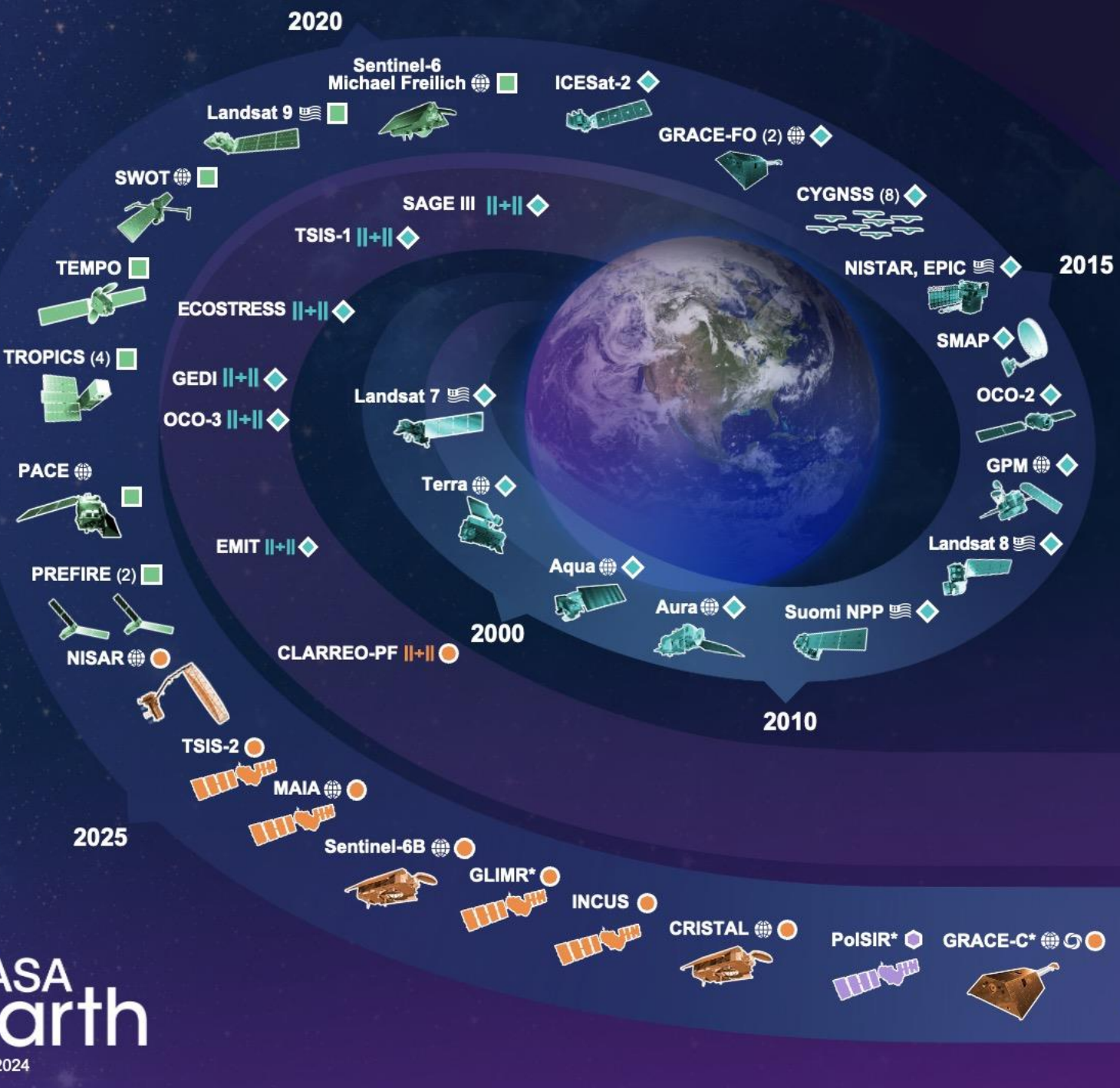
- International Partners
- U.S. Partner
- ISS Instrument
- JPSS Instrument
- Cubesat
- Launch Date TBD
- Earth System
- Observatory Mission
- (Pre) Formulation
- Implementation
- Operating
- Extended

Invest/CubeSats

- MURI-FD 2023
- SNOOPI 2024
- ARGOS* 2024
- ARCSTONE* 2025
- GRITSS* 2025
- GRATTIS* 2026

JPSS Instruments

- OMPS-LIMB 2022
- LIBERA 2027
- OMPS-LIMB 2027
- OMPS-LIMB 2032



ISS INSTRUMENTS

MISSIONS

Recent and Upcoming Earth Science Launches

*Agency LRD Aug 2024
**Launch Date NET Oct 2024
***Agency LRD Oct 2024

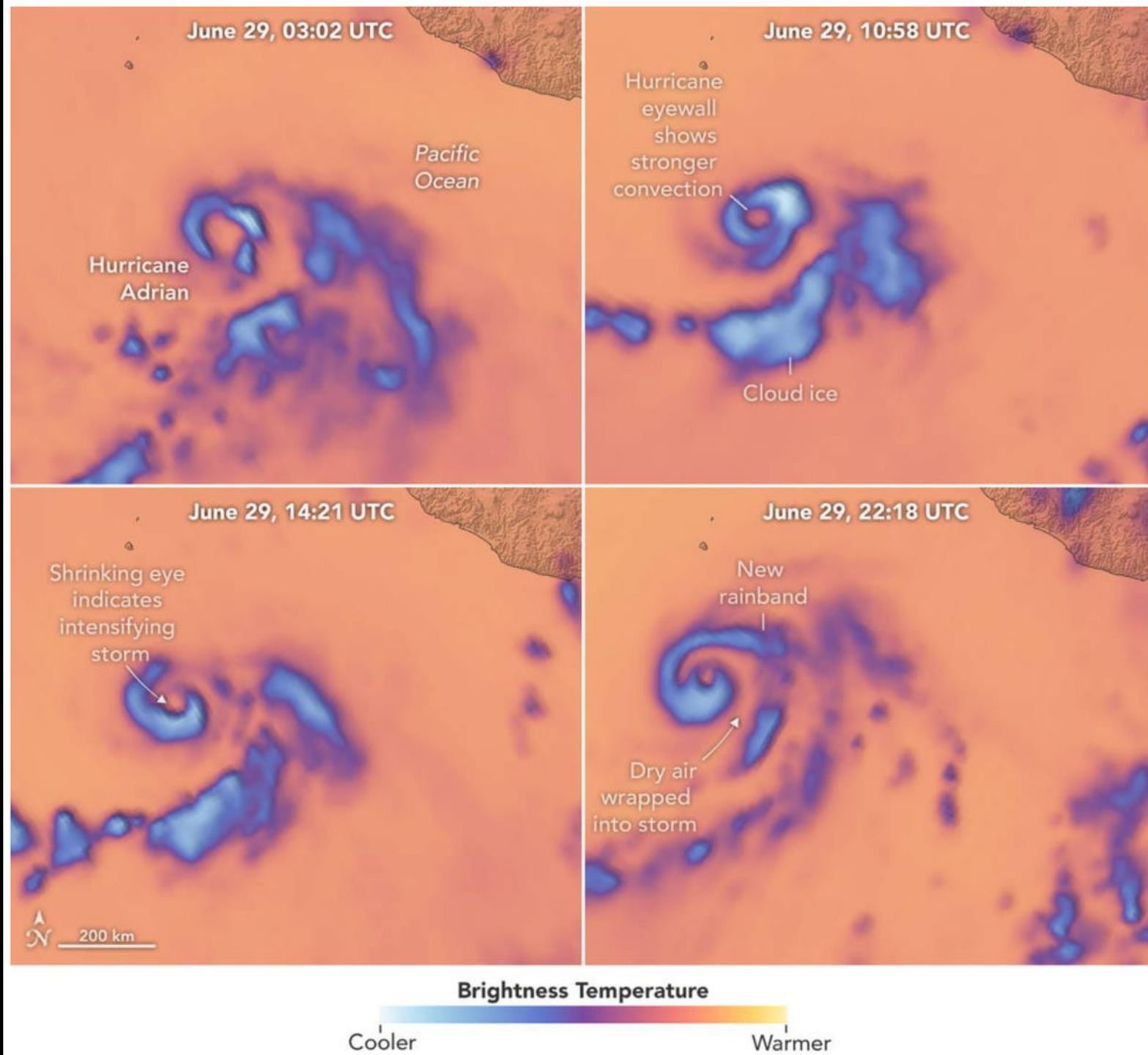
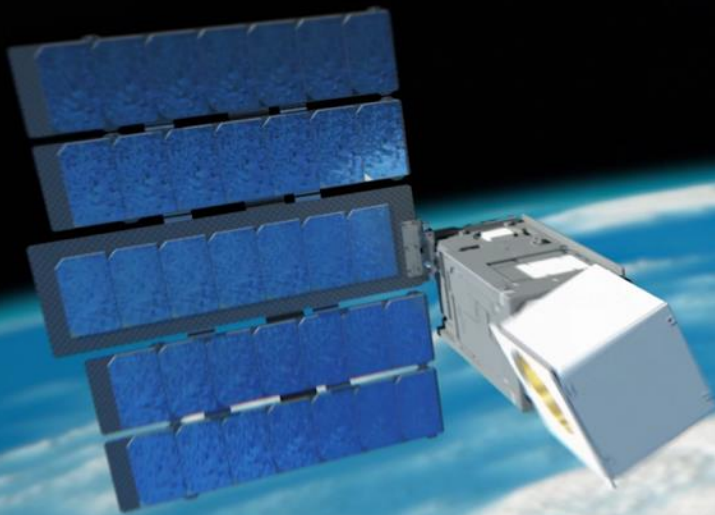




NASA TROPICS

CubeSat Constellation Mission Releases First-Light Data

Four 3U CubeSats, 30° inclination, 550 km
12-channel microwave sounder payload
60-min-revisit tropical cyclone obs



First Data Returns from SNOOPI CubeSat

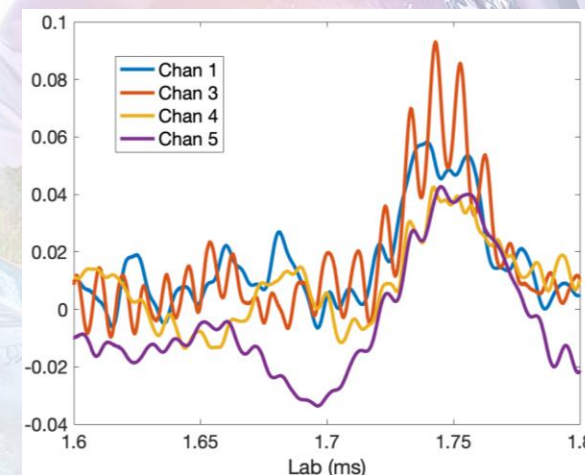
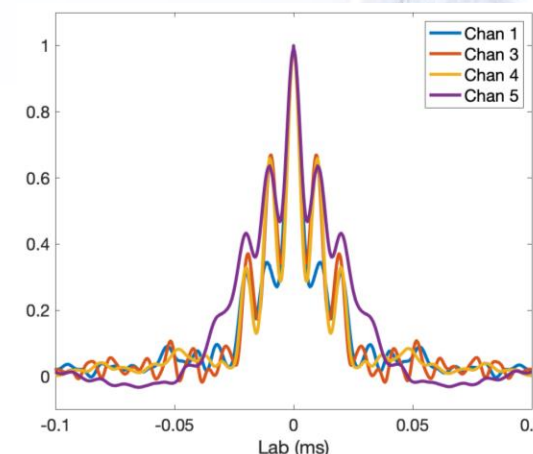
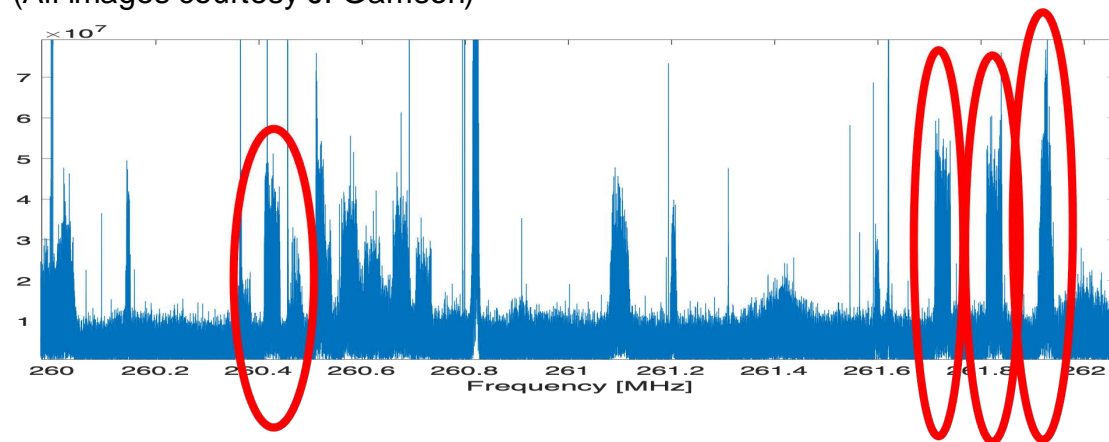
Currently on orbit, the Signals of Opportunity P-band Investigation (SNOOPI) tech demo is using P-band ***signals of opportunity*** to demonstrate measurements of root zone soil moisture (RZSM) and snow water equivalent (SWE). SNOOPI leverages signals from existing telecommunications satellites and therefore does not require a transmitter, making this technology very cost effective.

The 6U CubeSat was launched on March 21 onboard NASA's SpaceX 30th Commercial Resupply Services mission to the ISS, and subsequently deployed in mid-April. In late May and June, SNOOPI took its first data in both bands: low band (240-270 MHz) and high band (360-380 MHz). The team continues work filtering RFI and conducting data quality analysis.



Low band retrievals from June 6 (a confirmed reflected signal capture)

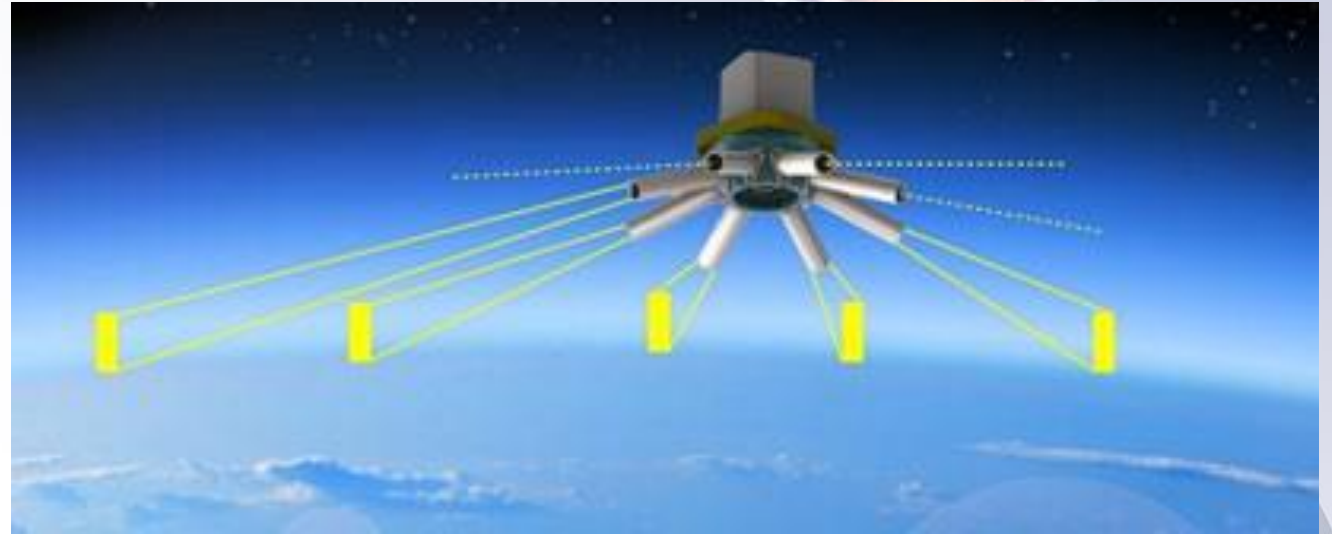
Below, the red ellipses show the data channels transmitted in the low-band, each tracked independently to produce the lines shown on the autocorrelations at right. (All images courtesy J. Garrison)



ARGOS Instrument Overview



- Eight simultaneous viewing directions to atmospheric limb
 - Forward and backward along orbit track
 - Perpendicular to orbit
 - 45° azimuth between each of these directions
- Each viewing direction measures simultaneous radiance profiles at 870 nm and 1550 nm
- All measurements captured on single focal plane
- No moving parts

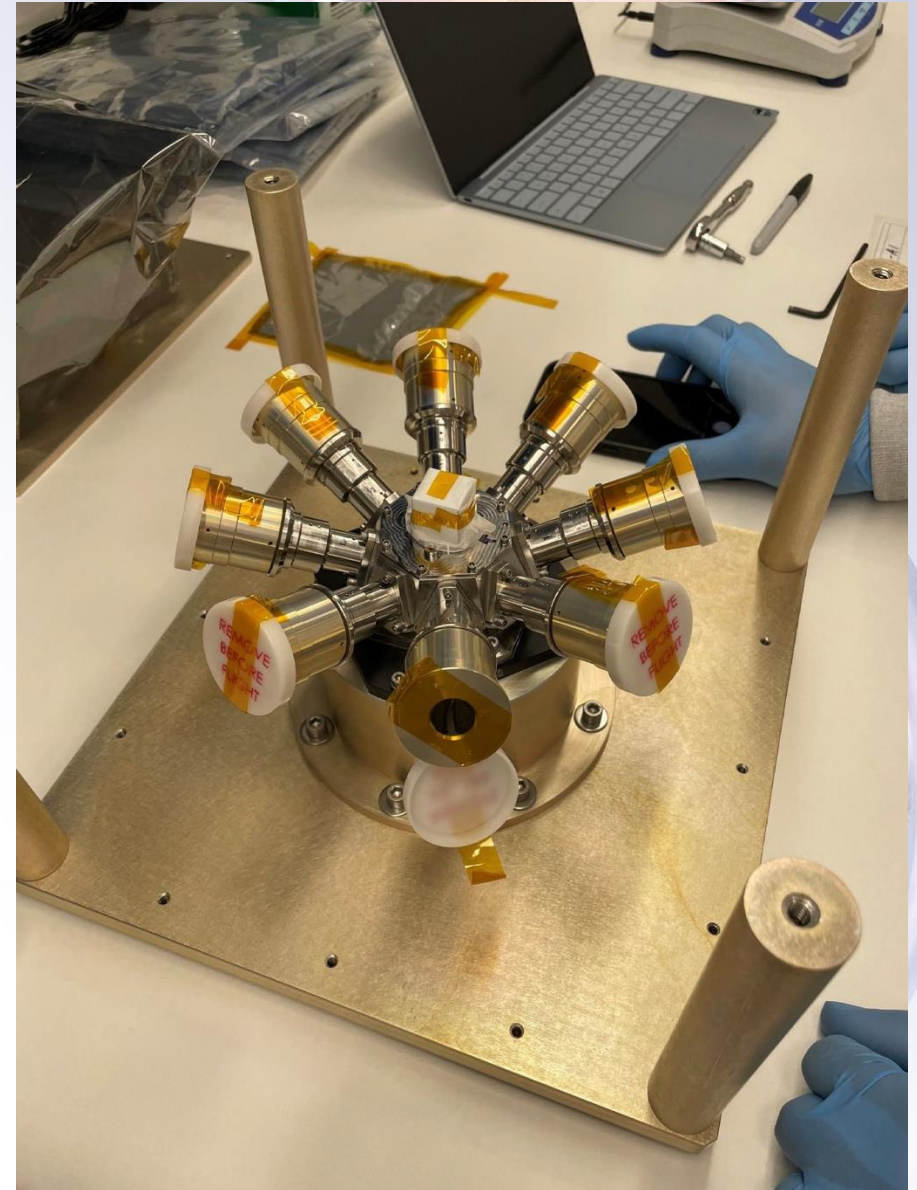


- 550 km altitude, Sun-synchronous orbit
- Vertical slits cover 0-60 km altitude at Earth limb with 0.5 km sampling
- Collect short individual images (< 25 msec) to prevent saturation due to clouds
- Nominal along-track profile sampling is 45 km (6 second averaging)

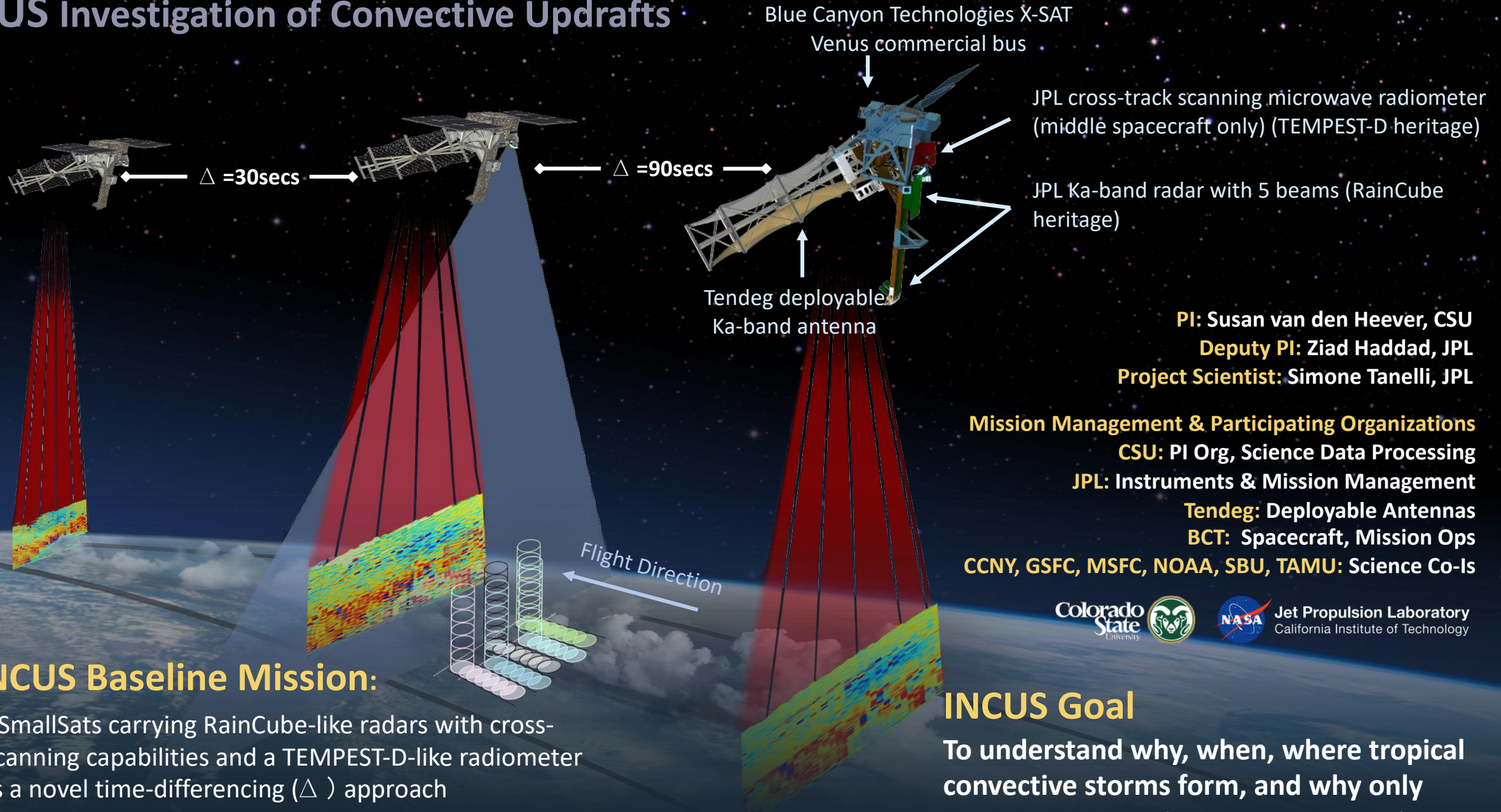
Where is ARGOS



- ARGOS instrument delivered to Loft Orbital facility in Colorado on April 18
- Upcoming key dates:
 - Integration to spacecraft hub: August 2024
 - Spacecraft-level vibration testing: early December 2024
 - Spacecraft delivery to SpaceX: early January 2025
- Target launch date: February 1, 2025
- Hope for 2-4 week commissioning period to get first data
- Adapt OMPS LP retrieval algorithm to derive aerosol products



INCUS Investigation of Convective Updrafts

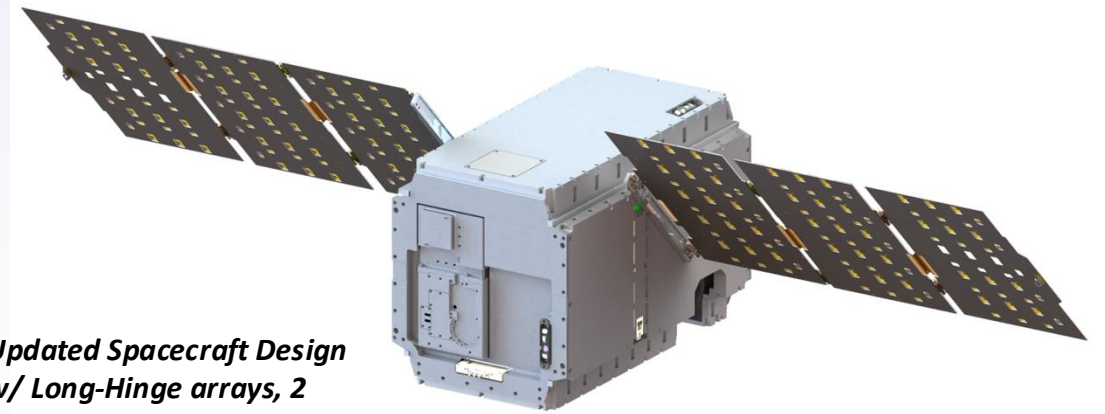


POLARIZED SUBMILLIMETER ICE-CLOUD RADIOMETER (PoISIR)

Project Overview and Update

Significant Accomplishments

- Project PDR Set for 9/10-9/12/2024; KDP-C set for 10/29/2024
- Updated Spacecraft design to get a power-positive power budget
- PDR Data Product generation underway
- First pre-PDR EPR held with VDI and TK. Other subsystem EPRs scheduled and to be complete by the end of July.
- Significant progress on budget and schedule updates to prepare for PDR, with projected improvement in reserves since PPBE26
- Updated contract documents for BCT. RFP will be issued soon with award planned for Fall 2024.



*Updated Spacecraft Design
w/ Long-Hinge arrays, 2
SADAs, 16U size*

Key Project Information

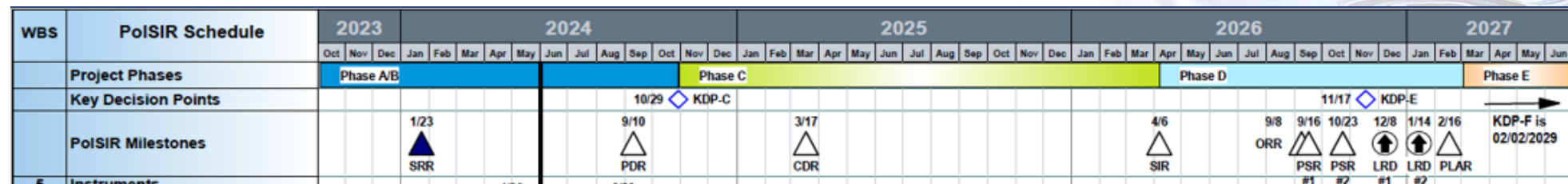
- | | |
|--|--|
| <ul style="list-style-type: none"> • PI-Led Earth Venture Instrument Mission • Data products: IWP, Deff • Two 16U CubeSats to launch separately • RAAN separation: ~6h • Orbit inclination: ~51° • Orbit altitude: ~550 km | <p>Mission Phase: Phase A/B (10/2023 -)</p> <p>Launch Date: <u>12/2026 / 1/2027</u></p> <p>Mission Life: 2-yr coincident science</p> <p>Category: 3</p> <p>Class: D</p> <p>Launch Vehicle: TBD</p> <p>Program Office: ESSP</p> <p>Technical Authority: GSFC</p> |
|--|--|

Instrument Development

- GSFC: IFAs, IECS, and Instrument I&T
- Newton Engineering (contract awarded 4/2024): Payload mechanical
- VDI and TKI (contract awarded 2/2024): Optical Assemblies and SWAs

Spacecraft and Ground System

- Spacecraft Bus: BCT
- Mission Operations Center (MOC) and Ground Station: BCT with KSAT S-band services
- Science Data Center: University of Wisconsin
- Science Instrument Team: GSFC



PoISIR (EVI-6) Science



Science Objectives

- Constrain seasonally influenced diurnal cycle of tropical ice water path (IWP) and particle diameter (Deff)
- Determine the diurnal variability of ice clouds in the convective outflow areas
- Determine the relationship between shortwave and longwave radiative fluxes and the diurnal variability of ice clouds

Benefits

- Enable development of better cloud parameterizations
- Reduce uncertainty in 10-40 year climate model predictions

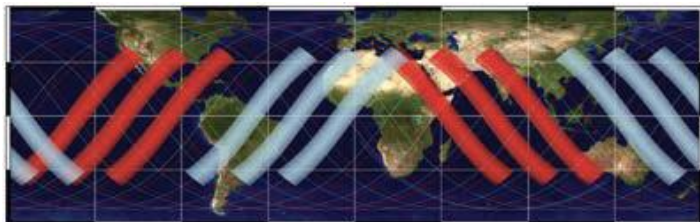
Team

- **PI:** Ralf Bennartz (Vanderbilt U)
- **D-PI:** Dong Wu (GSFC)
- **Science Team:**

Adams, Ian	NASA-GSFC	Johnson, Ben	UCAR
Barahona, Donifan	NASA-GSFC	Karpowicz, Bryan	NASA GMAO
Berndt, Emily	NASA-MSFC	Kroodsma, Rachael	NASA-GSFC
Brogniez, Helene	U Paris Saclay	Loveless, Daniel	Indiana Univ
Braun, Jessica	UW-Madison	Merrelli, Aronne	U-Michigan
Ehsan, Negar	NASA-GSFC	Pettersen, Claire	U-Michigan
Elsaesser, Greg	GISS/Columbia U	Rapp, Anita D.	Texas A&M
Gong, Jie	NASA-GSFC	Vanags, Chris	Vanderbilt

Mission Overview

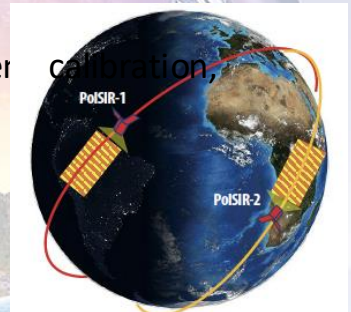
- \$37M (FY24) Class-D mission selected in 2023
- Two CubeSats: Each has two radiometers:
 - 683-GHz (QV & QH)
 - 325/9.5-GHz (QV & QH)
 - 325/1.5-GHz (QV)
 - 325/3.5-GHz (QV)
- Circular precessing low Earth orbit (LEO) for full diurnal sampling at 35°S-35°N latitudes



Ground tracks for PoISIR-1 (red) and PoISIR-2 (blue) for one full day. The blue and red semi transparent areas show the swath coverage for three consecutive orbits.

Mission Roles, Partnerships, Contributions

- Vanderbilt U: Mission science, cloud ice retrieval
- GSFC: Mission management, instrument development, data distribution
- U Wisconsin: Science data processing
- Blue Canyon Technologies (BCT): Spacecraft
- Virginia Diodes Inc (VDI): Submm-wave receivers
- NOAA: Operational enhancement options (TBD)
- U Paris Saclay (Franch Contribution): cloud retrievals



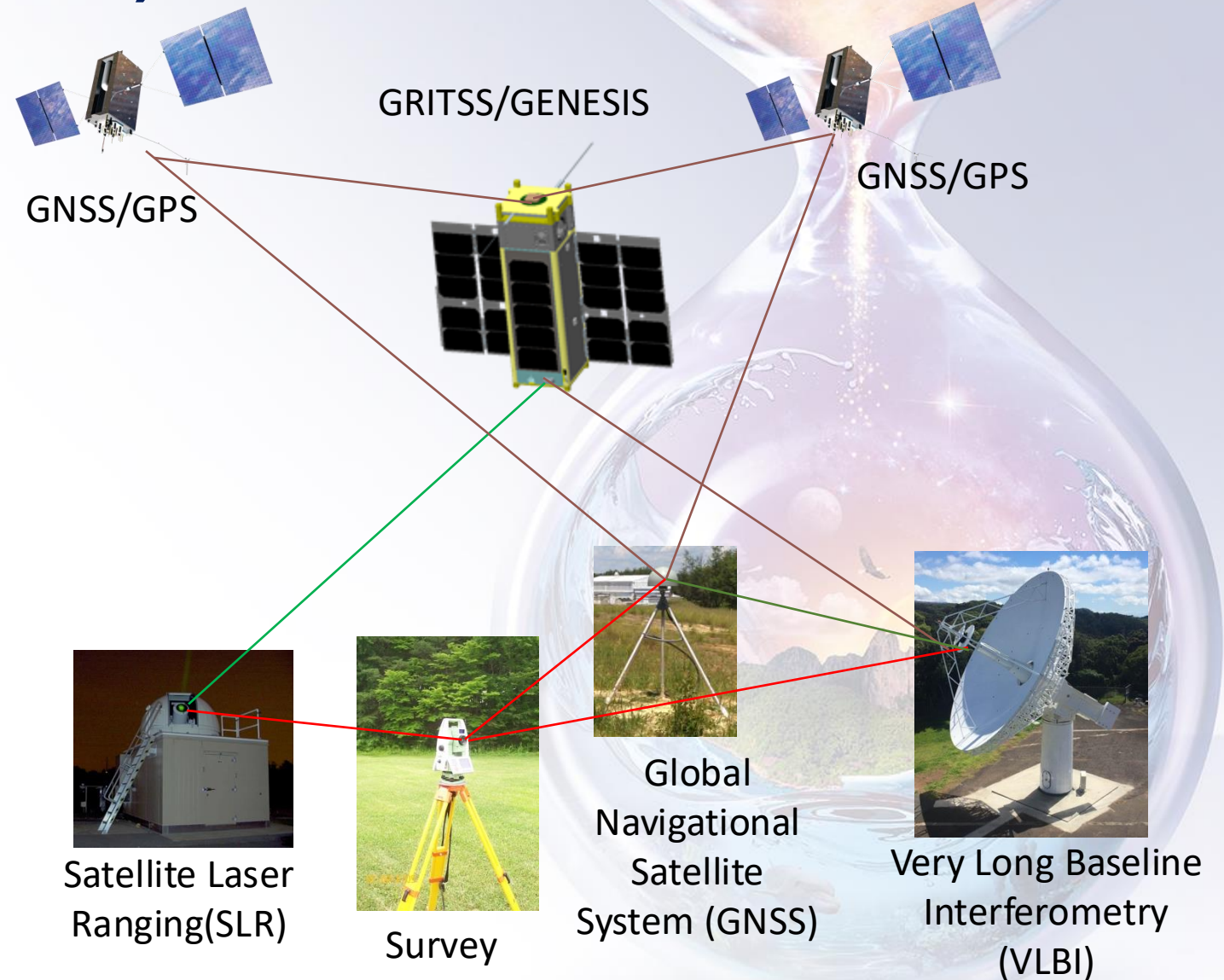
Science Enhancement Options

- GSFC, Columbia U, UCAR/JCSDA, GMAO, UMich, TAMU

Geodetic Reference Instrument Transponder for Small Satellites (GRITSS)



- ◆ Observations of a common space-based reference has the potential International Terrestrial Reference Frame.
- ◆ The NASA GRITSS flight demonstration will have lessons learned and applications for ESA's GENESIS mission
- ◆ GRITSS uses a novel approach that upconverts GPS signals as a VLBI source
- ◆ Targeting launch Fall 2025



(ACMES) Overview



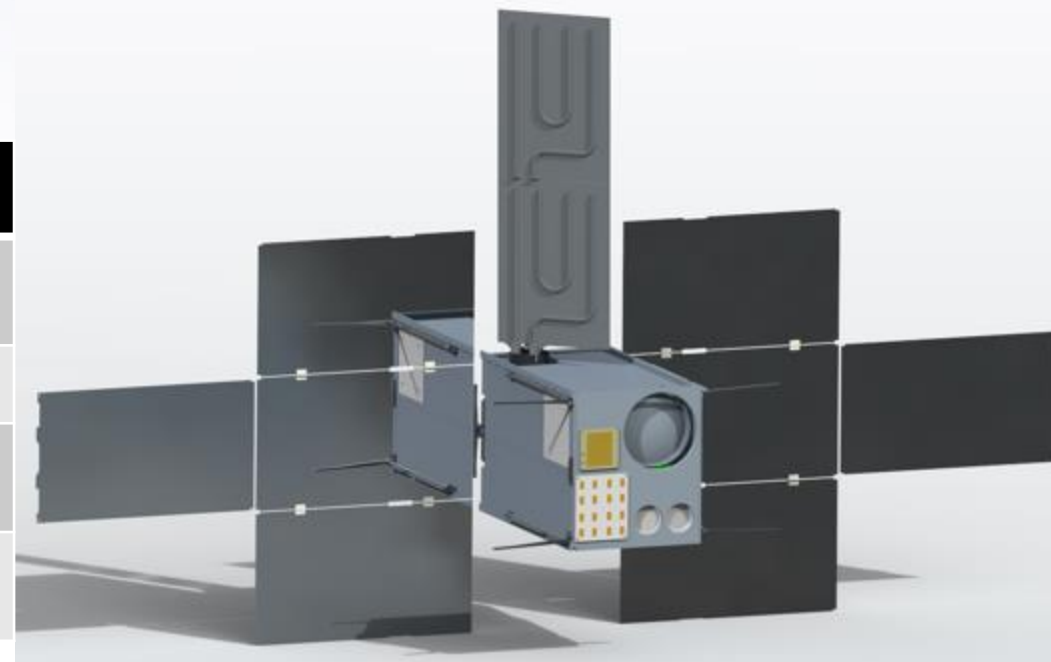
Principle Investigator: Charles Swenson

Partners: Center for Space Engineering at Utah State University, Hawaii Space Flight Laboratory, Orion Space Solutions.

Funding: NASA Earth Science Technology Office under an In-Space Validation of Earth Science (InVEST) grant

Key Parameters (current best estimate)

Form Factor	# Spacecrafts	Orbit	Altitude	Inclination	Launch Date
+16U	1	SSO	550 km	>85°	2025
Launch Interface	Mission Duration	Onboard Processing	Active Thermal Control	Power (Peak)	Telemetry
ISI M3S	1 to 4 years	Unibap IX10	+100 W MPFL	~230 W	+180 GB/day



ACMES Scientific Objectives

ACMES is a two-part mission

A 1-year technology demonstration followed by up to 4 years of scientific return

- **HyTI 2.0**

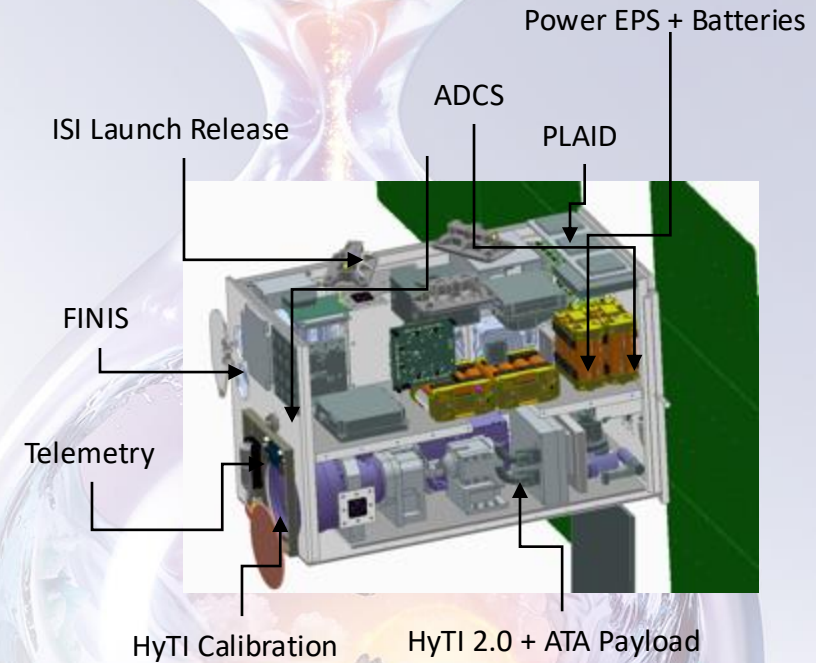
- LandSAT quality TIRS imaging from a CubeSat platform.
- LWIR ground mapping for geology/mineralogy, Volcanic activity, surface moisture, and hydrology

- **FINIS**

- Low SWAP Methane detection
- CH₄ Total Column abundance, point leak detection, and plum geometry

- **PLAID**

- Planar-style space weather plasma probe
- Electron temperature, density, and potential



ECE 5230 Space Mission Design



Nicholas Wallace	Joseph Dewsnup	Nathan Brinkerhoff	Travis Blaylock	Kelly Burch	Andrew Nelson	Dalton Yerke	Grable Hart	Tyler Rose	Weston Seegmiller	Charles Swenson
Zachary Hall	David Allen	Zachary Clarke	Michael Kirk	Jason Powell	Rowan Antonuccio	Benjamin Lewis	Oliver Parkinson	Nathan Dickson	Alessa Love	Bruno Mattos

Not Pictured: Aubrey Hjorth, David Pipkin, Isaiah Olsen

ARCSTONE InVEST Overview



Project management
Engineering coordination
ü Instrument electronics
Flight and ground software
Mechanical, Thermal & Structural
Instrument I&T
Science and data products
Operations
Outreach



**CubeSat Launch Initiative:
SpaceX/Maverick**



Lunar calibration
approach and
validation analysis



✓ 6U CubeSat Bus:
✓ Mechanical
✓ Power/Electric ü
Electronics/Data ü
Avionics
✓ Bus Storage & Maintenance
System I&T
Operations



✓ Xiphos sub-contract Management
✓ Publication



ANALYTICAL MECHANICS ASSOCIATES

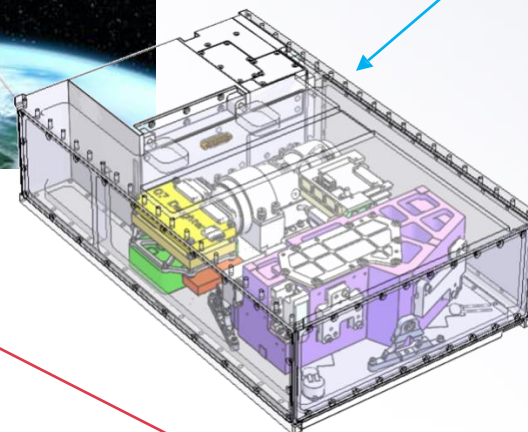
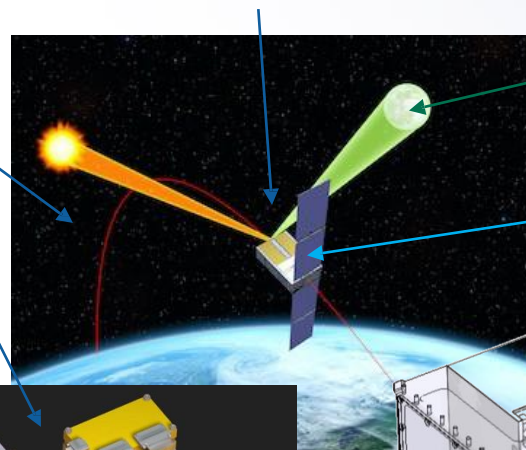
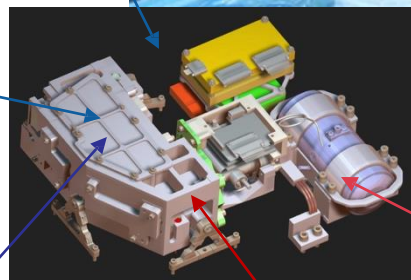
SMCE Cloud support
SPS development and Operations



✓ Optical black coating



✓ Flight Calibration System
ü IDCA characterization
Instrument characterization
Uncertainty budget
L1 data validation



✓ Opto-mechanical design
✓ Radiometric modeling ü
Instrument fabrication ü
Instrument assembly ü
Functional testing
Support for inst. modeling



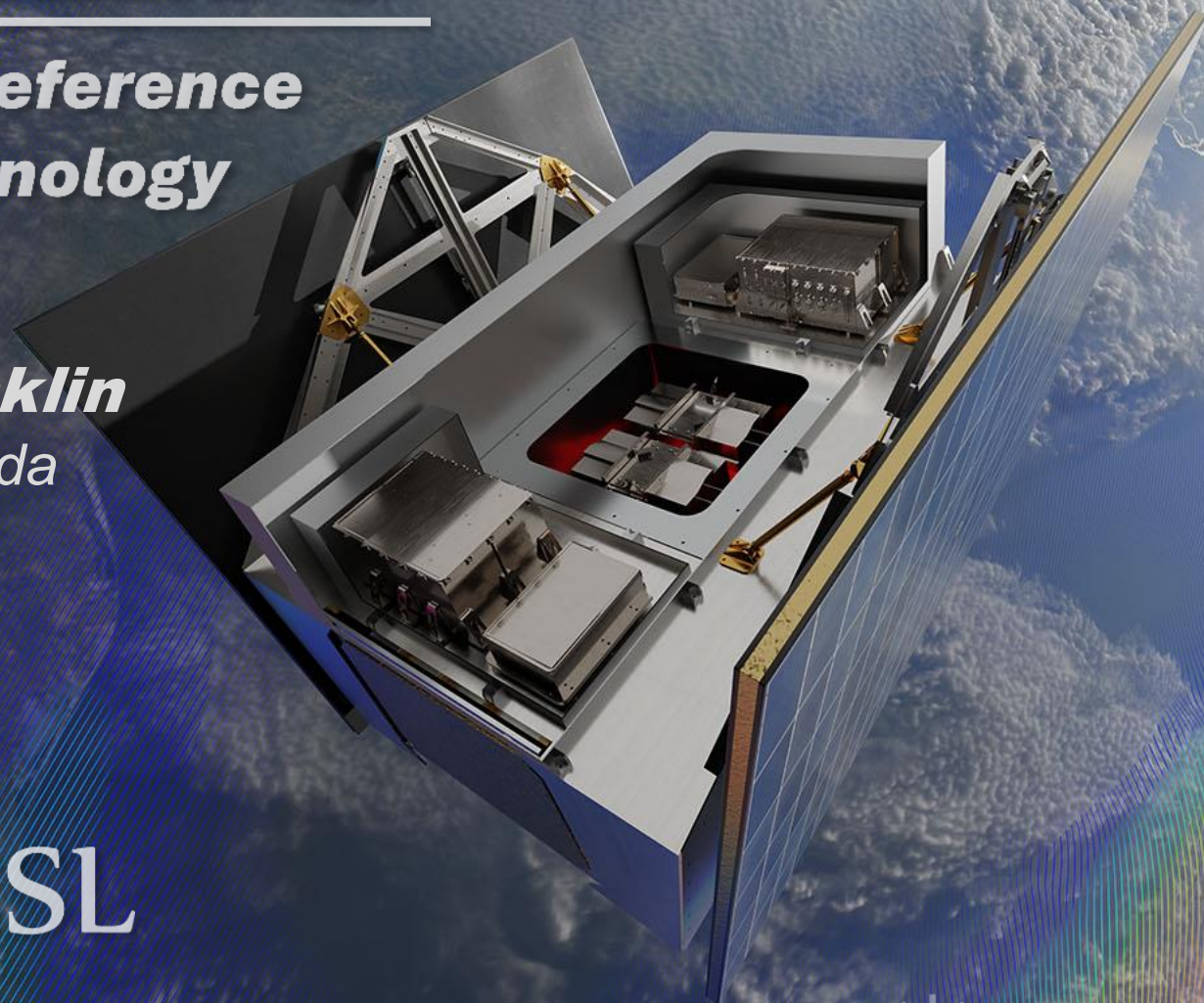
✓ Payload Analysis
✓ Input to payload design
✓ Flexure design

GRATLLIS

**Gravitational Reference
Advanced Technology
Test In Space**

PI: John W. Conklin
for University of Florida
grattis.ufl.edu

ESTO
EARTH SCIENCE TECHNOLOGY OFFICE



UF  **PSSL**



EMBRY-RIDDLE
Aeronautical University

CROSSX**TRAC**

BAE SYSTEMS

FIBERTEK, INC.



APEX

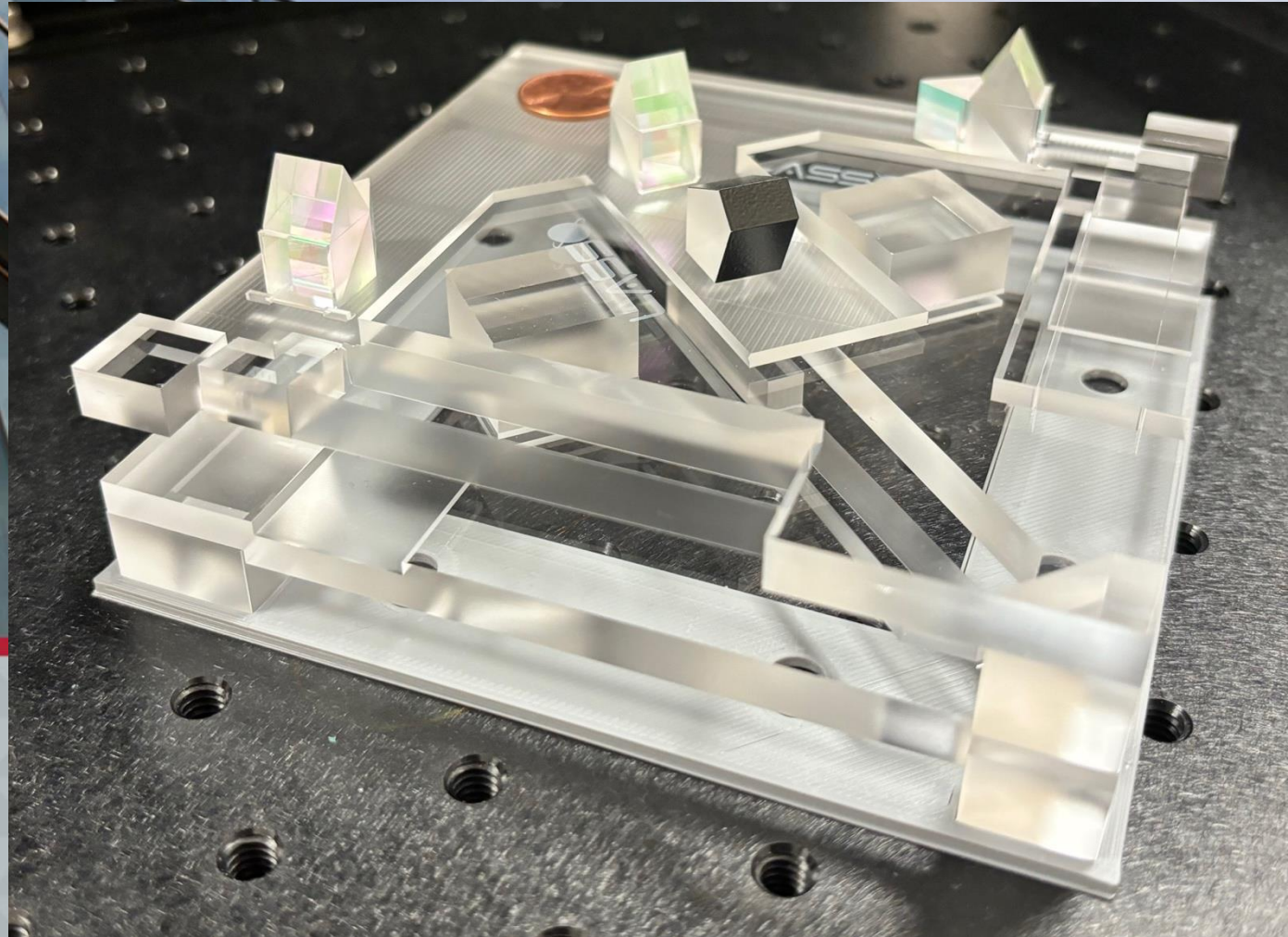
SPACEX

Optomechanical accelerometers for space geodesy



THE UNIVERSITY OF ARIZONA

Wyant College
of Optical Sciences



Principal Investigator: Felipe Guzman

Team: Jose Sanjuan, Moritz Mehmet,
Andrea Nelson, Ian Harley, Lee Ann
Capistran, Jackson Dahn

GRATTIS Payload: Simplified Gravitational Reference Sensors



GRATTIS will fly two Gravitational Reference Sensors (GRS) on 170 kg satellite in Earth orbit

Using LISA tech to improve to improve GRACE accelerometers

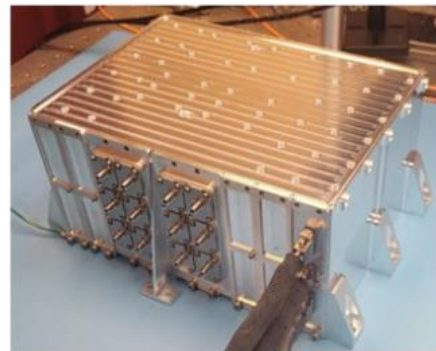
By comparing measurements from two sensors, we can determine noise floor

Under development since 2019 via NASA ESTO IIP program

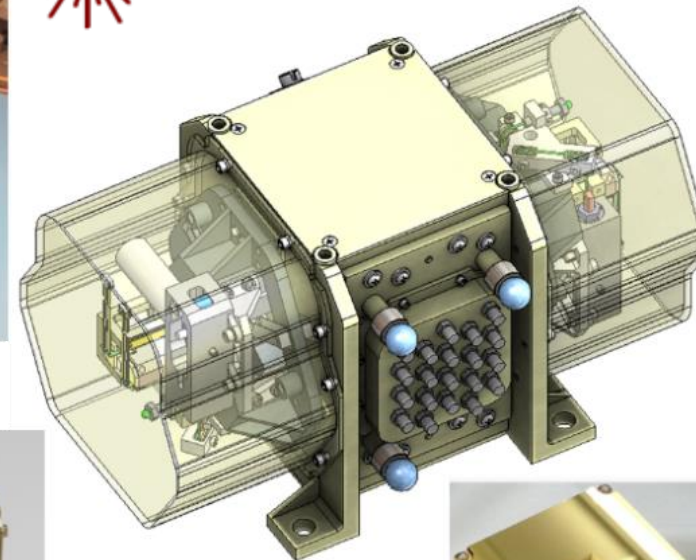
Currently in TRL 6 phase

Launch planned for early 2027

FIBERTEK, INC.



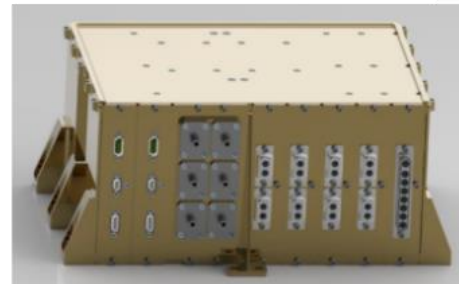
TRL 6 LISA Charge Management Device



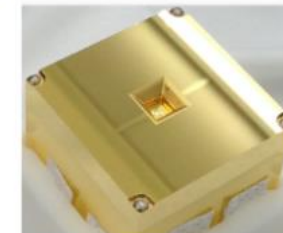
TRL 6 S-GRS Head



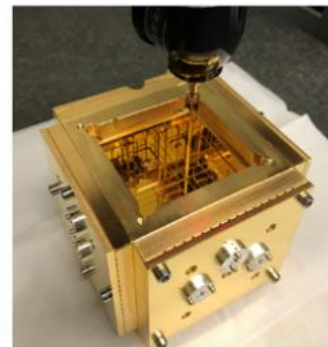
TRL 6 S-GRS Caging Mechanism



TRL 6 S-GRS Sensing, Actuation and Charge-control Unit



LISA Pathfinder flight Test Mass



UF prototype Electrode Housing

BAE SYSTEMS

CROSS X TRAC

UF PSSSL



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esto.nasa.gov