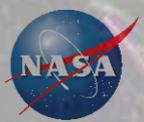




Probe of Extreme Multi-Messenger Astrophysics (POEMMA) Probe Mission

**Roy Young/ES34
Marshall Space Flight Center
roy.young@nasa.gov**

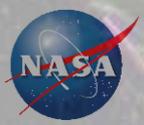


NASA Probe Studies for 2020 Decadal Survey



- NASA funding 10 Probe Class (below 1B\$) Mission (18 mos) Studies in Preparation for the 2020 Decadal Survey - POEMMA: Probe of Extreme Multi-Messenger Astrophysics
- PI responsible for the final report (due NLT Dec 2018)
- NASA will submit these studies to the Decadal Survey
- Decadal Survey Committee will have the option to prioritize any of these mission concepts, or recommend a competed line of Probes (similar to Explorers)
- Selection based on Science Merit (cost, schedule)

PI	Affiliation	Short title	Design Lab/Prog Office
Camp, J.	NASA's GSFC	Transient Astrophysics Probe	IDC/PCOS-COR
Cooray, A.	Univ. California, Irvine	Cosmic Dawn Intensity Mapper	TeamX/ExEP
Danchi, W.	GSFC	Cosmic Evolution through UV spectroscopy	IDC/PCOS-COR
Glenn, J.	Univ. of Colorado	Galaxy Evolution Probe	TeamX/ExEP
Hanany, S.	Univ. of Minnesota	Inflation Probe Mission Concept Study	TeamX/ExEP
Mushotzky, R.	Univ. of Maryland	High Spatial Resolution X-ray Probe	IDC/PCOS-COR
Olinto, A.	Univ. of Chicago	Multi-Messenger Astrophysics	IDC/PCOS-COR
Plavchan, P.	Missouri State Univ.	Precise Radial Velocity Observatory	No design lab funded/HQ grant
Ray, P.	Naval Research Lab	X-ray Timing and Spectroscopy	IDC/PCOS-COR
Seager, S.	MIT	Starshade Rendezvous	TeamX/ExEP

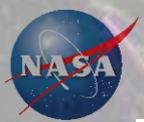


Starting Point Design

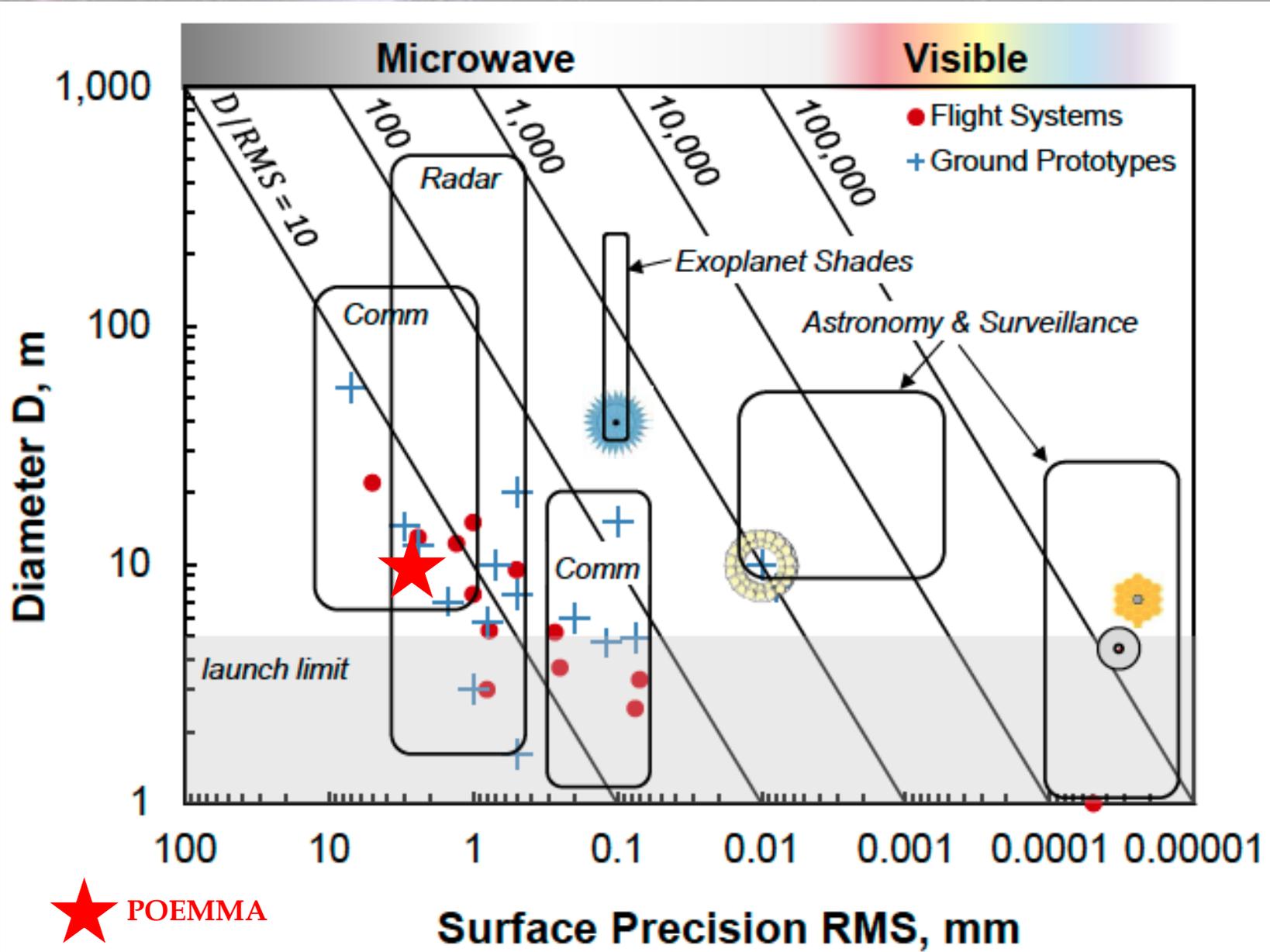


- 2 formation-flying spacecraft (OWL concept)
- f/1 Schmidt camera, 45° FoV,
- 10 m diam. primary mirror deployable (OWL: 7m)
- 4.3 m diam. corrector lens
- 8 m² focal plane with ~500,000 pixels
- (~ 0.06° pixel ~1 km² projected on the ground at 1,000km altitude)
- 14 m² effective aperture (OWL: 7.07 m²)
- weight TBD ~2400 kg;
- power consumption TBD ~600 W
- **Corrector:**
 - **Corrector Mirror Distance: +/- 1cm**
 - **Tilt to axis: +/- 1 degree**
 - **Decenter: < | +/- 0.5cm**
 - **Fabrication: Segmented aspheric fabrication concerns**
 - **Radius: Low sensitivity (Corrector thickness is important)**
- **Mirror**
 - **Fab: (more like microwave antenna than imaging mirror)**
 - **¼ to 1 wave in visible is more than sufficient quality.**
 - **Radius: = +/- 0.2cm (refocus to correct fab error)**
 - **Thickness – Focal Plane Distance: +/- 0.1cm**
 - **Decenter: +/- 0.1cm**
 - **Tilt to axis & petals: +/- 0.02 degrees, 1.2 arcmin**
- **Focal Plane**
 - **Tilt to axis: +/- 0.1 degree**

First Order Properties	POEMMA	OWL
Aperture	3.70 m	3.0 m
Focal Length	2.86 m	3.0 m
F/#	0.78	1
Stop	4.4 m shifted from corrector plate	corrector plate
RMS Spot Radius (.36um)	~1.5mm	1.5mm
Primary Mirror diameter	6.7 m (segmented)	3.8 m (monolith)
Corrector diameter	3.8 m (segmented)	3 m (monolith)
Corrector Thickness	12.6 - 47.5 mm (segmented)*	100 mm (monolith)



Large Reflectors - Current State of the Art

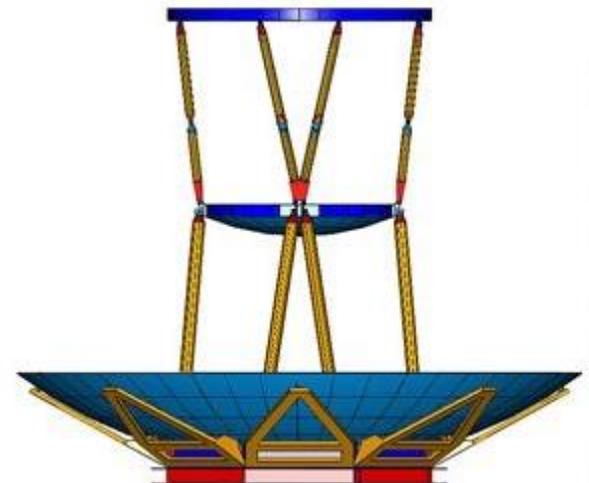
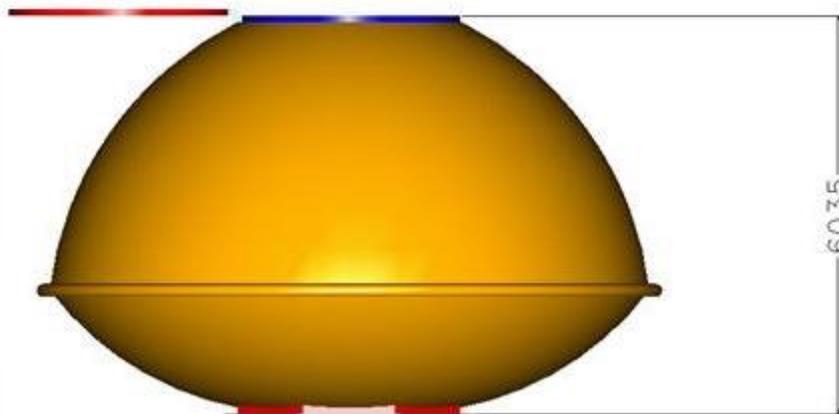


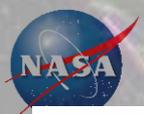
★ POEMMA

CAD Views of OWL

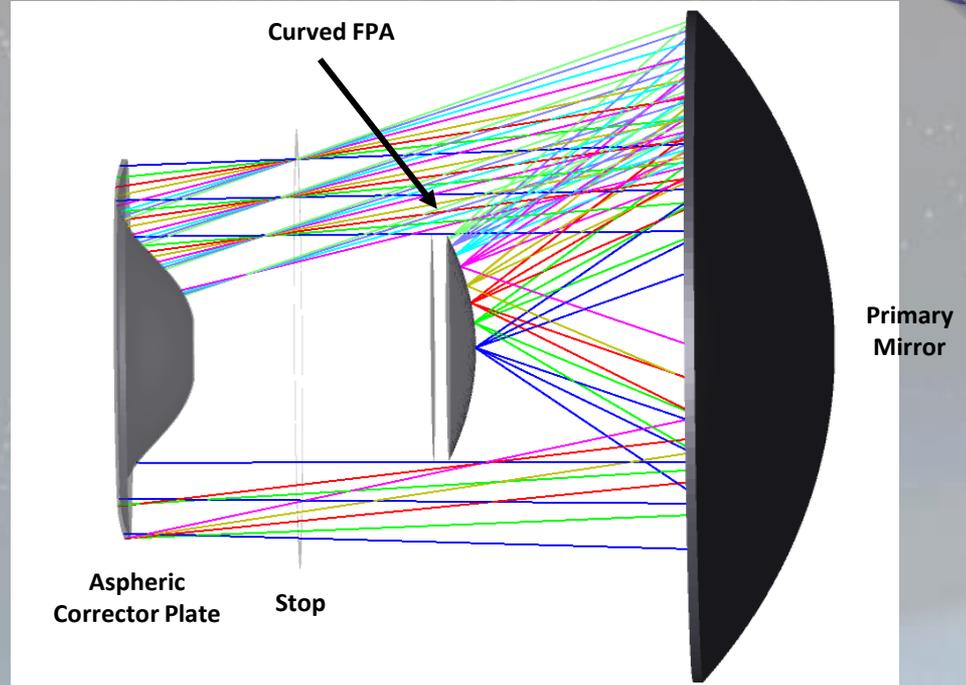
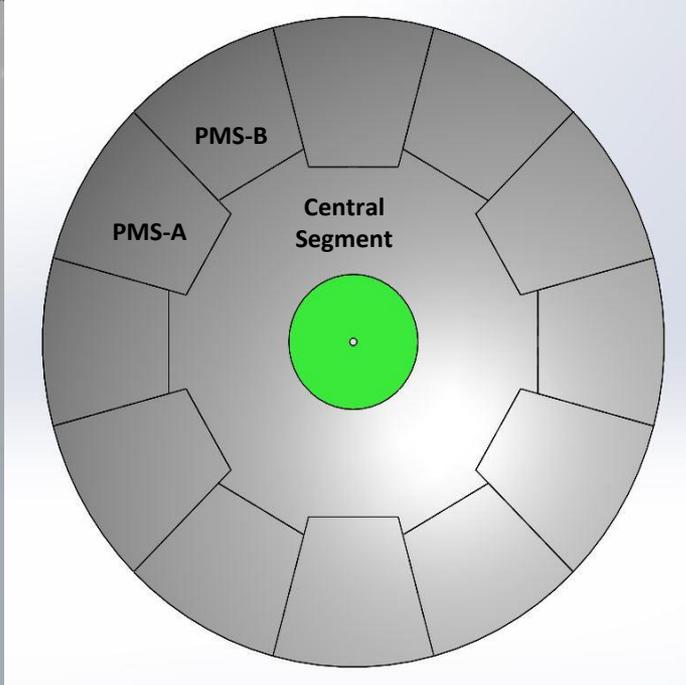
Instrument Synthesis and Analysis Laboratory

OWL - Orbiting Wide-Angle Light Collectors
 IDL Study - 2001





POEMMA Optical Layout



Corrector Lens

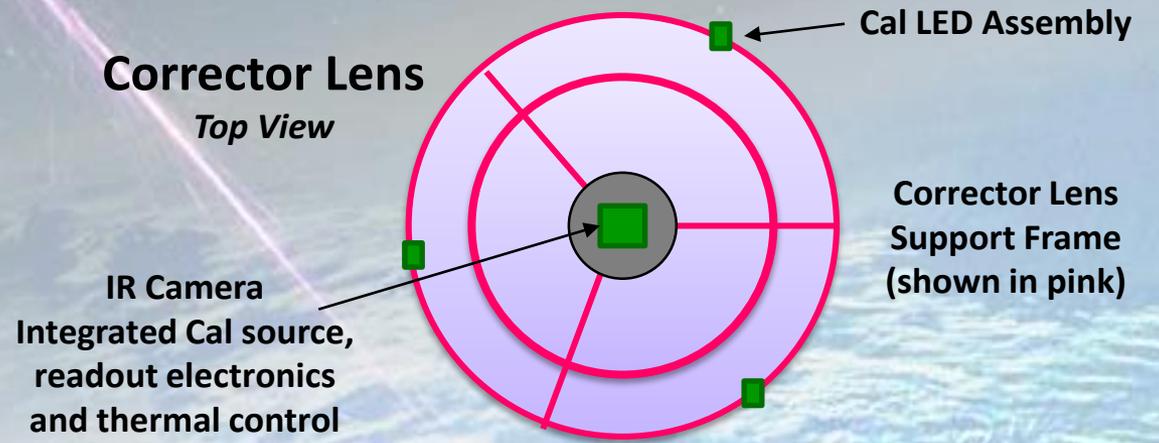
Side View



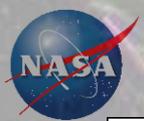
60mm edge thickness
 12.7mm center thickness
 ~465kg

Corrector Lens

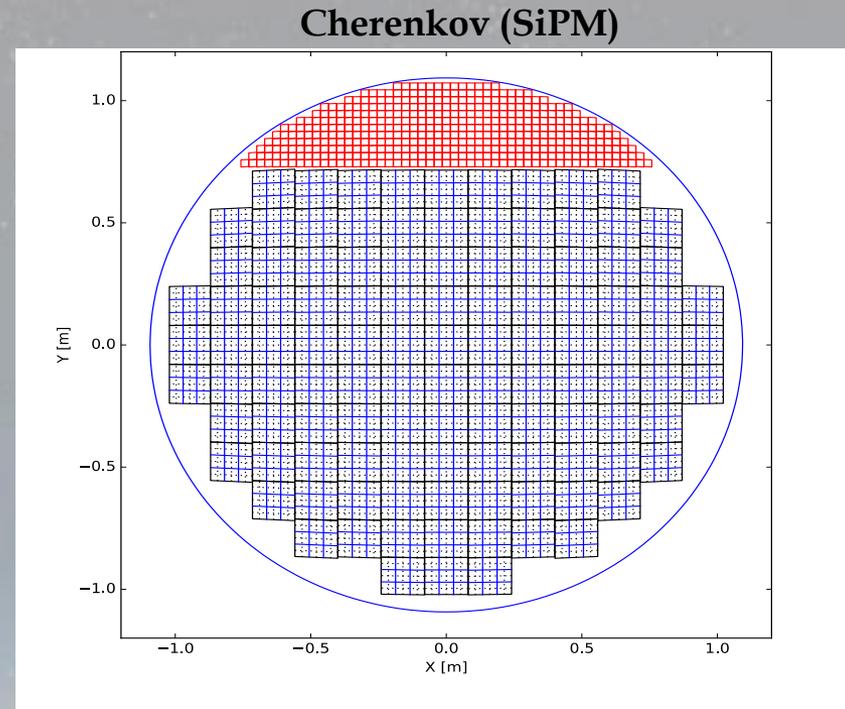
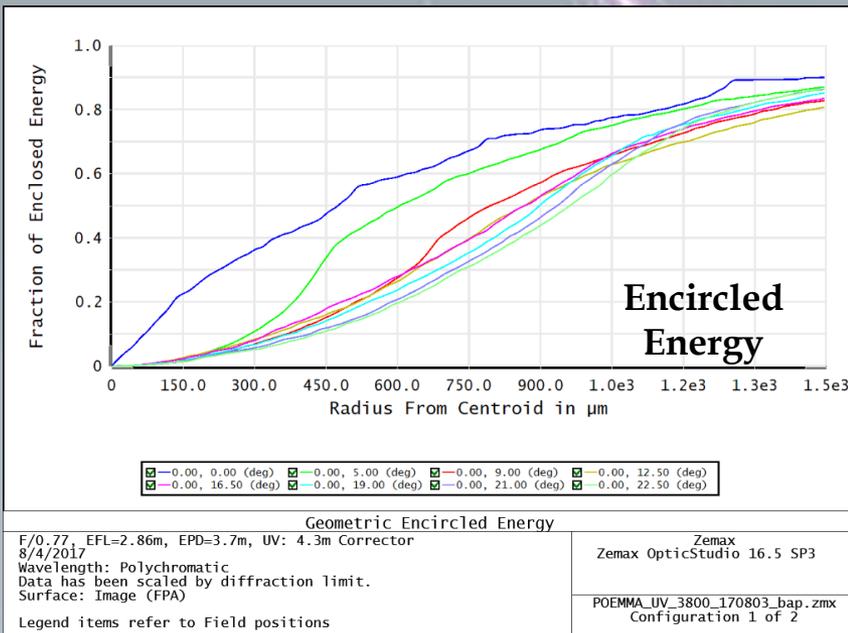
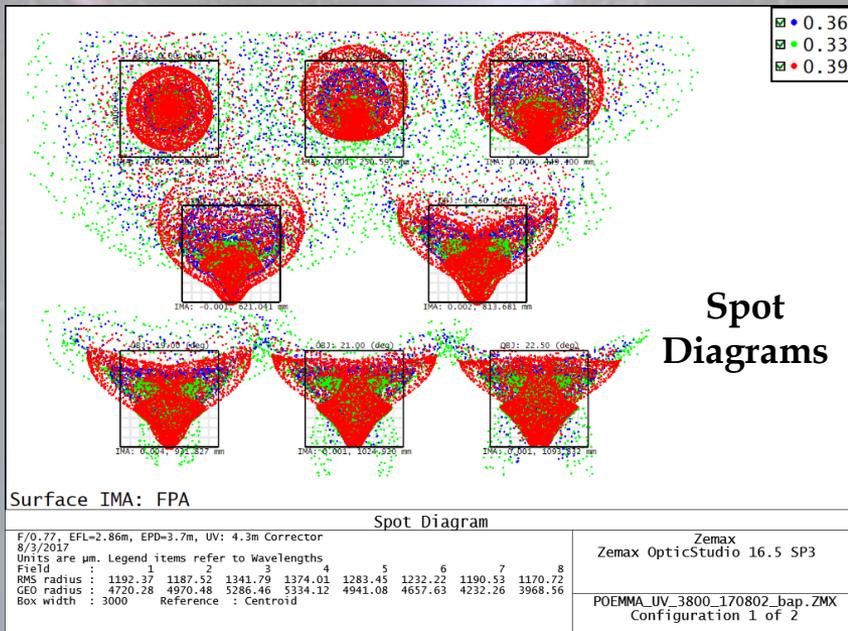
Top View



3 pie segments, 3 annulus segments
 Division at inflection point
 Spines align with tripod



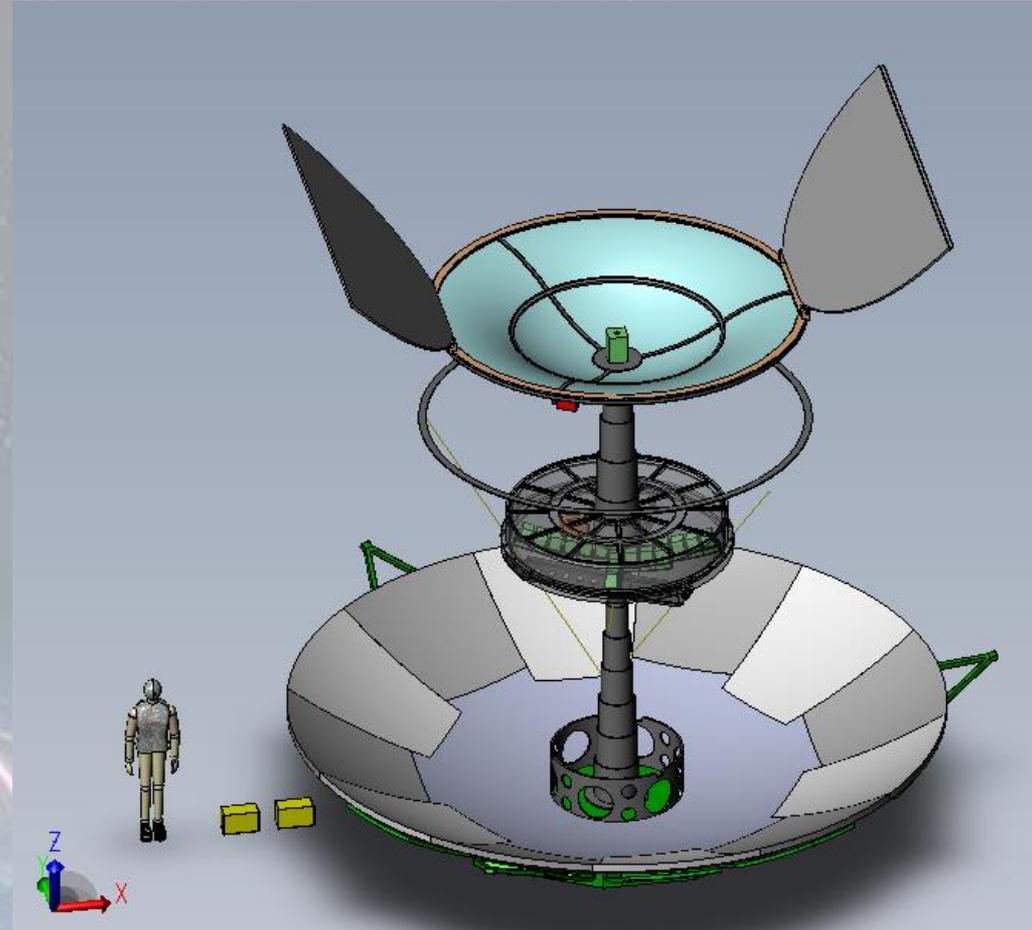
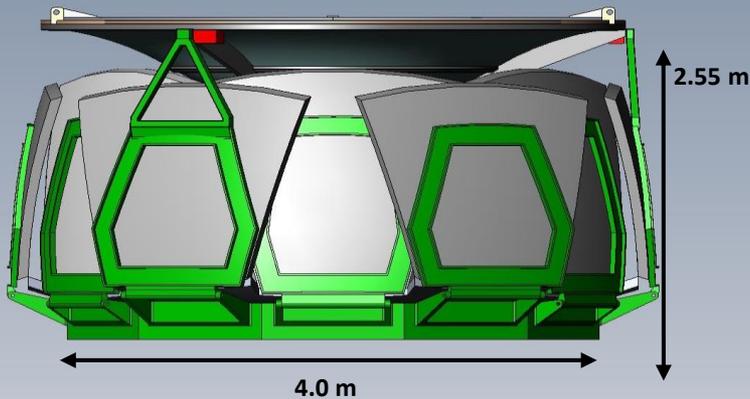
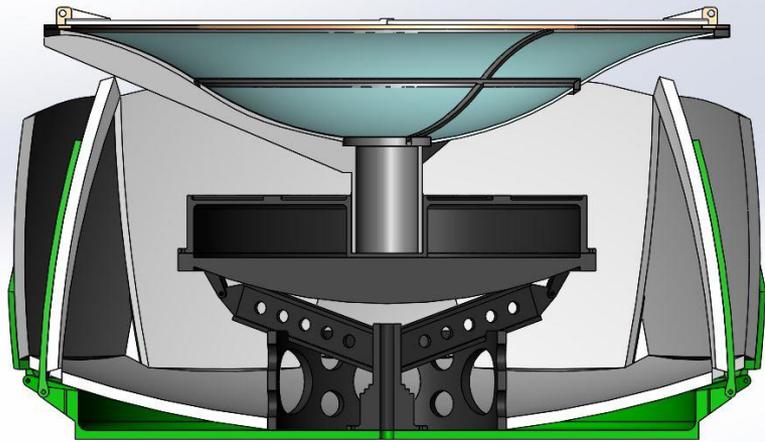
Optical Performance



Fluorescence (PDM)

The POEMMA focal surface is composed of a hybrid of two types of cameras:

- 90% of the focal surface is dedicated to the POEMMA fluorescence camera
 - multi-anode photomultiplier tubes (MAPMTs) with typical time between images is 1 msec
- Cherenkov camera occupies the crescent moon shaped edge of the focal surface which images the limb of the Earth
 - composed of Silicon photo-multipliers (SiPMs)



Design of light shield not considered in IDL study

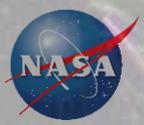
Trade between inflatable (hemispheric deployment) and conical shape

Issue with sequential PM Petal deployment if using conical shape shield

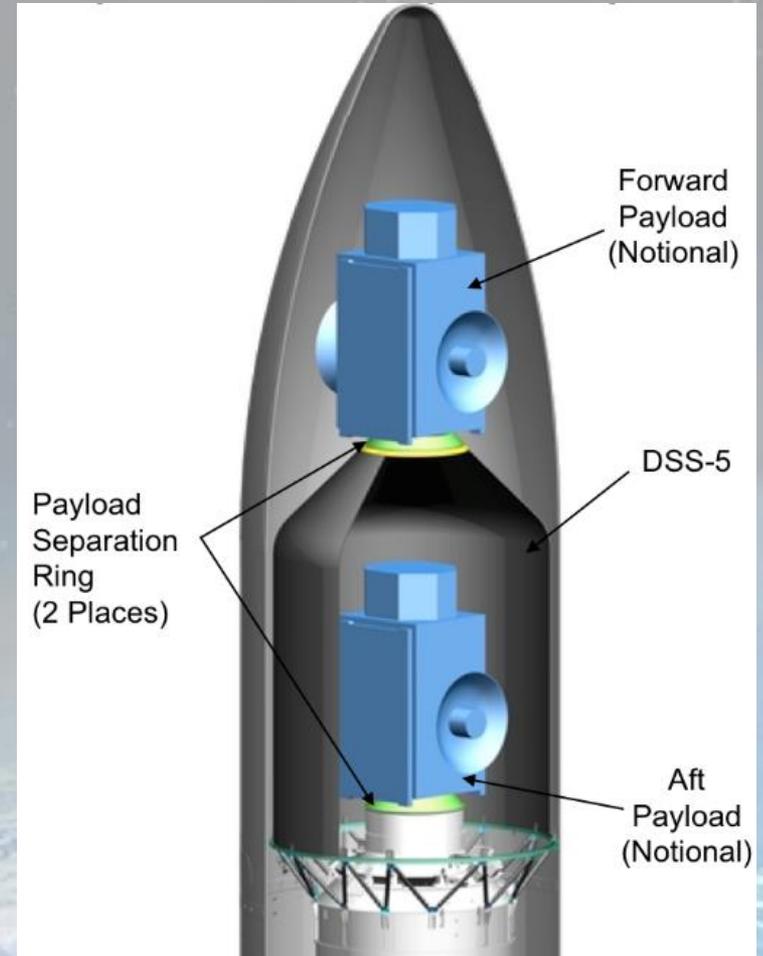
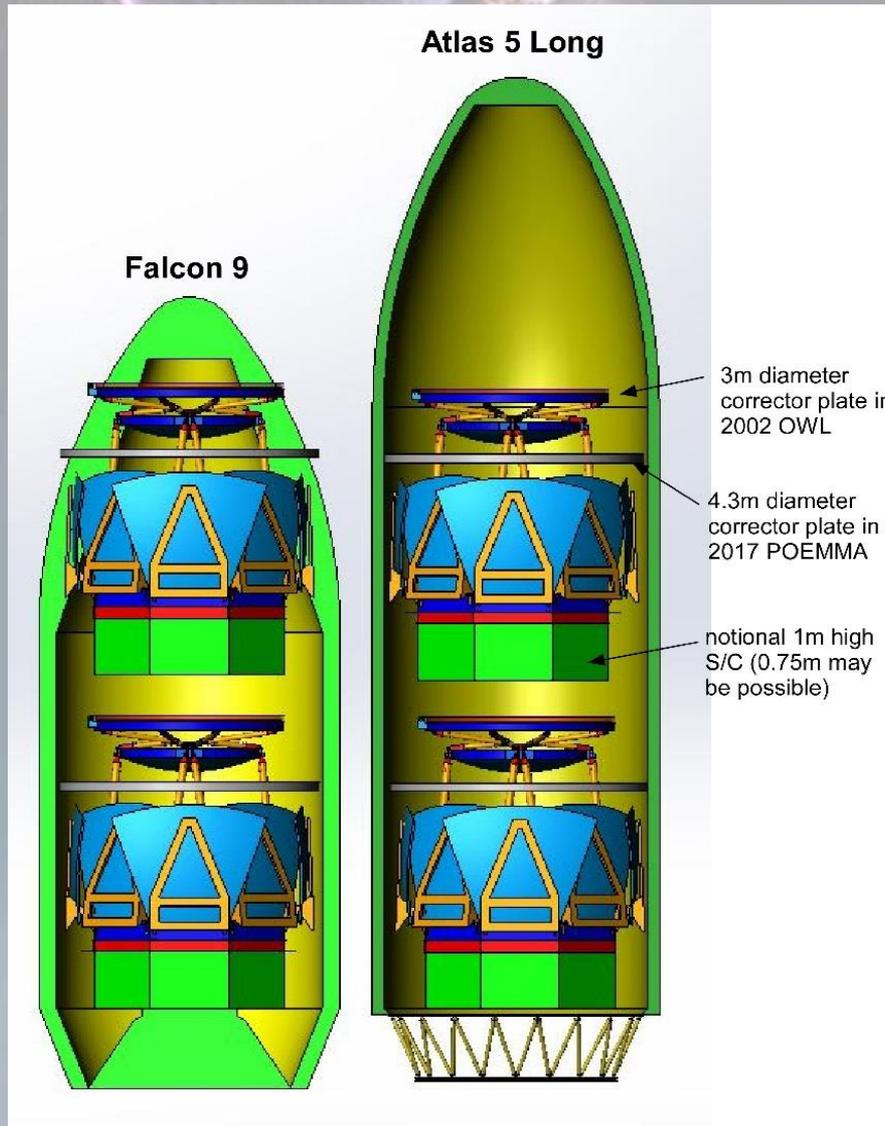
Complexity and reliability of hemispherical deployment

Use MMOD resistant materials (i.e. Kevlar29) sparingly – mass and packaging considerations

Thermally transmissive

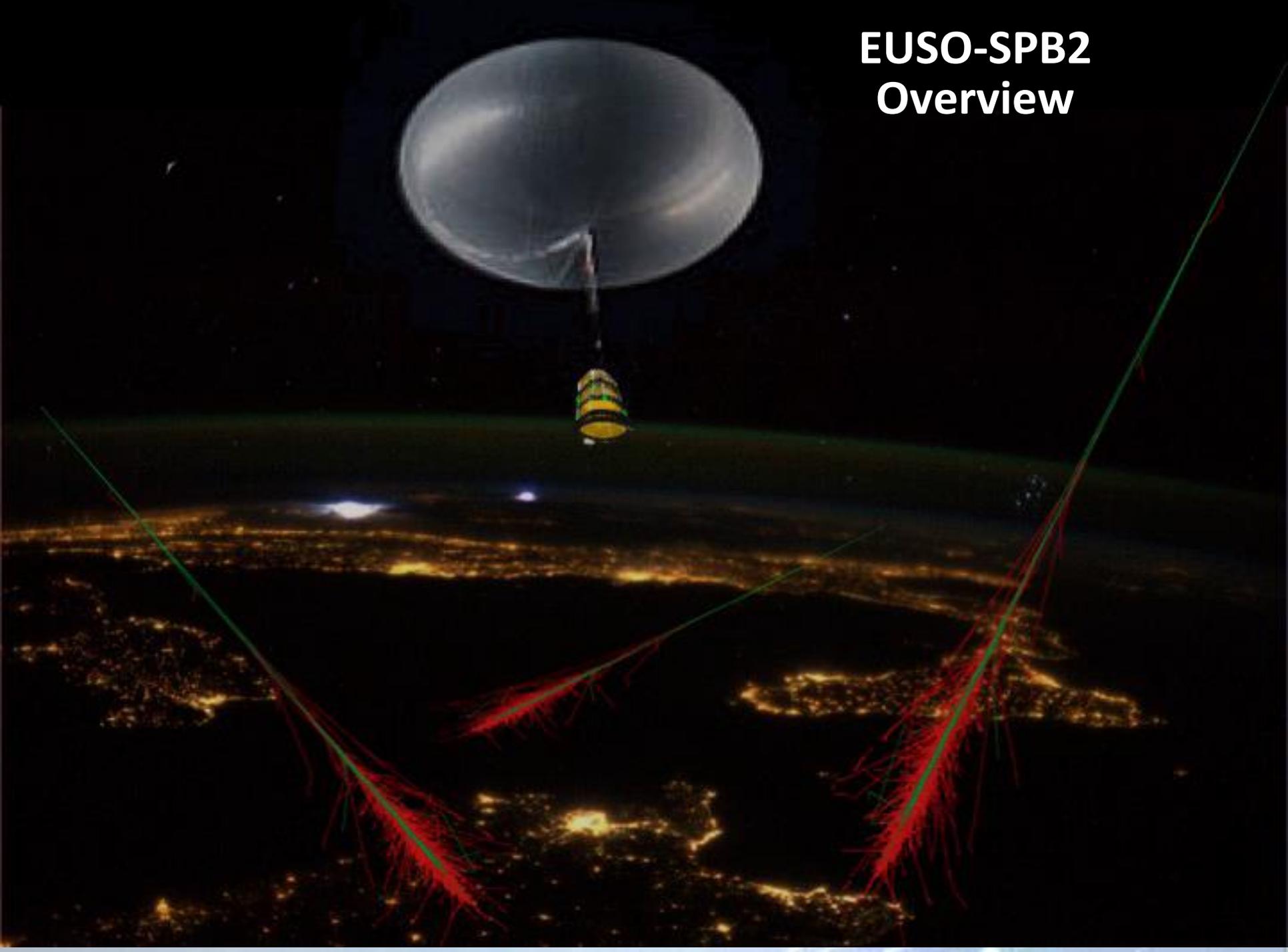


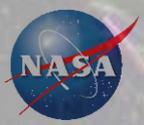
Launch Configuration



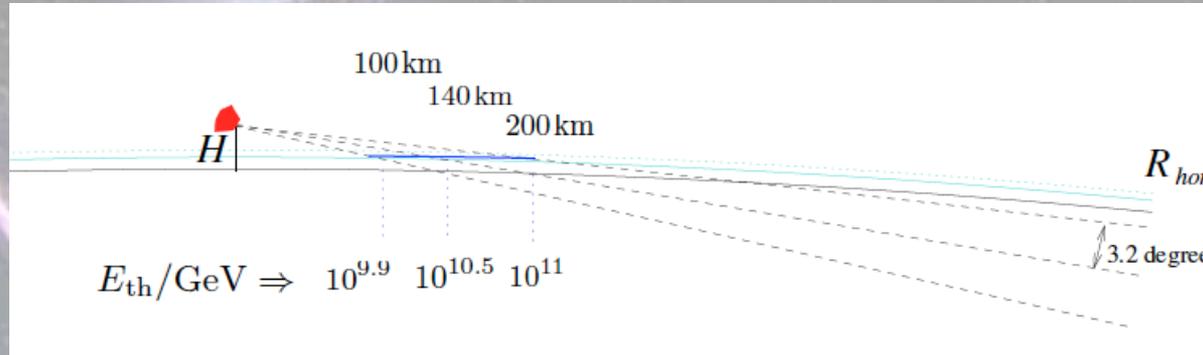
Atlas 5 with DSS-5

EUSO-SPB2 Overview

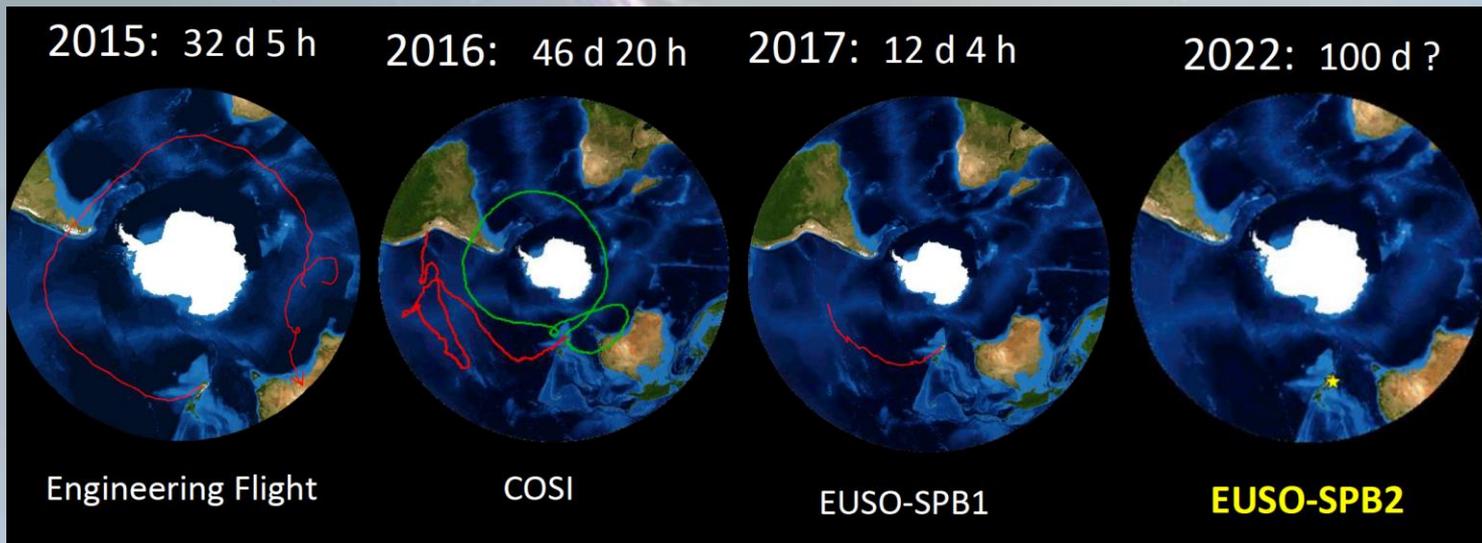




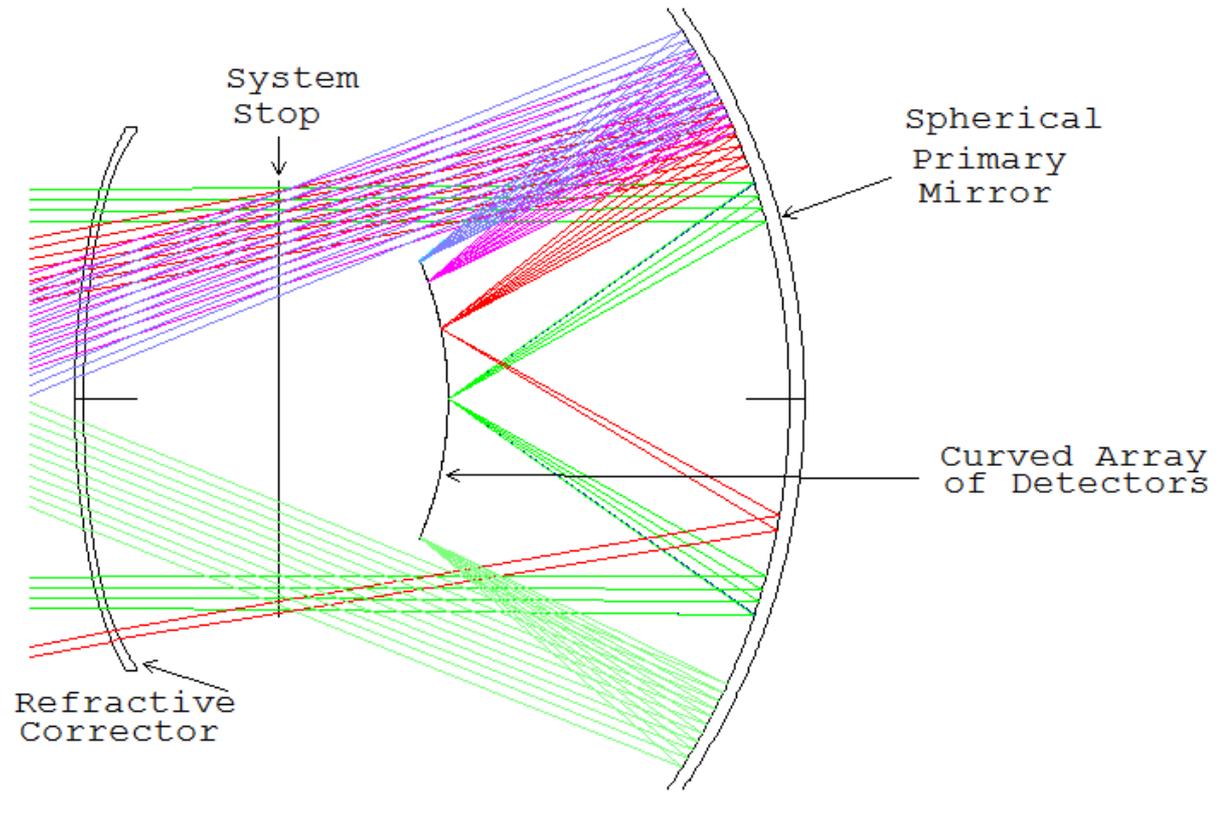
EUSO-SPB2 Observation Approach



- Detect high altitude UHECR and UHEv through Cherenkov signal
- Pathfinder for POEMMA Instrumentation:
 - Schmidt camera: corrector, stop, mirror, coatings
 - Detector technology (MAPMT, SiPM, others), filters
 - Corrector material: PMMA vs glass
 - Limb field of view
- Background measurement for upward going EAS shower
- Detect EAS showers through fluorescence



SPB-2 -Optics



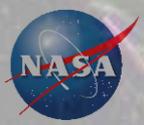
3D Layout

Scale: 0.0667

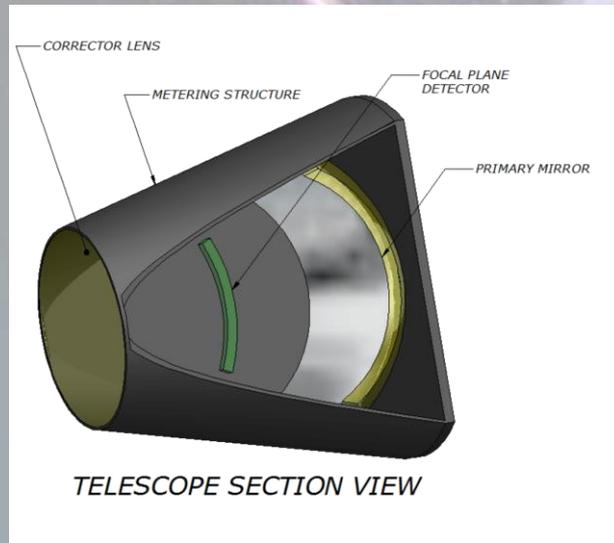
300.00 Millimeters

Center for Applied Optics
University of Alabama in Huntsville
Patrick J. Reardon

EUSO-SPB2 Chr.zmx
Configuration 1 of 1



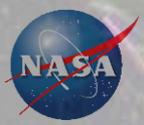
SP2 Telescope



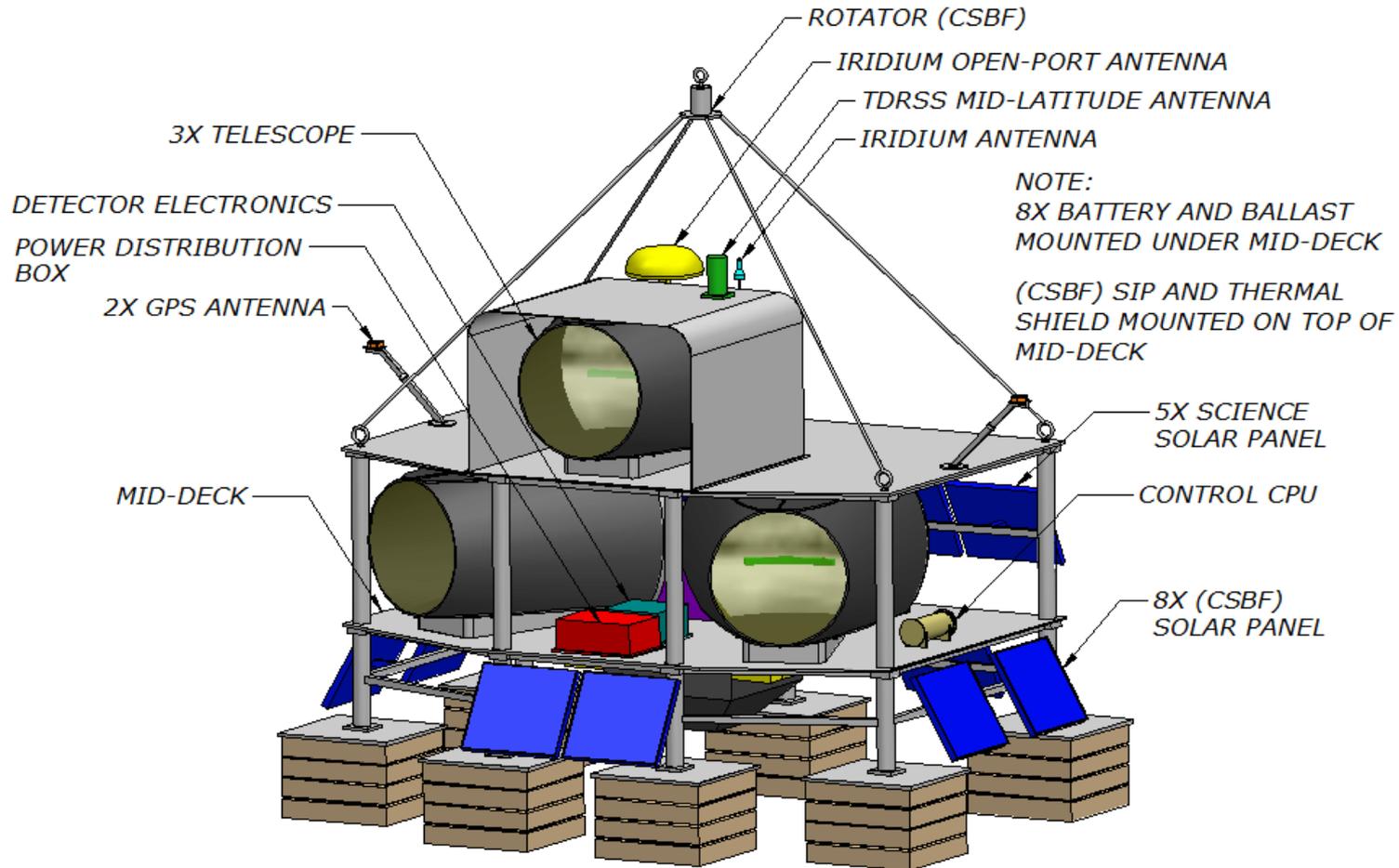
- Optics FoV: 29°x3.2°
- Corrector Plate: PMMA ~1m²
 - Image resolution: 1mm
 - Pixel size: 3mm square
 - Mirror: 1.8(H) x 1.1(V) split

Ring Imaging Cherenkov, RICH, mirror for the Alpha Magnetic Spectrometer, AMS-02, which is operating on the International Space Station since May, 2011. Composite Mirror Applications, Inc. (CMA), in Tucson, AZ produced a conical mirror 1.3m diameter 0.5m in height, from high modulus carbon fiber, flight qualified composite materials, having an optical surface on the inside of the cone. The flight model mirror was completed to specification, yielding nearly 2m² of replicated optical surface area and weighs 8 kg.

Replicated carbon fiber RICH mirror for AMS-02, Robert C. Romeo, 2006



EUSO-SPB2 Gondola



GONDOLA ASSEMBLY