

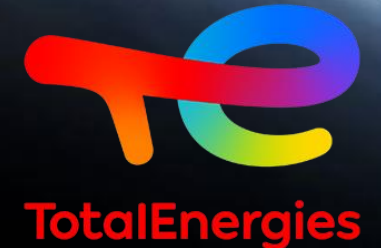


**SAFT**



# W3A Eurostar 3000 Li-Ion Battery In orbit Return of Experience

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Nasa Battery Workshop-Huntsville November 2024*



# Summary

1. Saft Li-Ion History : Stentor
2. Eurostar 3000 module design choices
3. W3A Eutelsat satellite In-Orbit Return of Experience
4. Conclusion

# Saft Satellite Li-Ion History

- 1993 : Saft Li-Ion cell technology development start
- 1996 : Beginning of development of Li-Ion satellite batteries for Stentor (Satellite de Télécommunication pour Expérimenter des Nouvelles Technologies en Orbite ) from CNES
- 1999 : VES140 cell qualification
- 2000 : Stentor battery qualification : First “industrial” application to qualify large batteries.
- 2002 : First orders of Li-Ion batteries for GEO and LEO satellites
- 2003 : Smart1 first ESA Moon Mission with full electric propulsion
- 2004 : W3A Launch
  
- 2024 : 393 satellites launched with Saft Li-Ion batteries



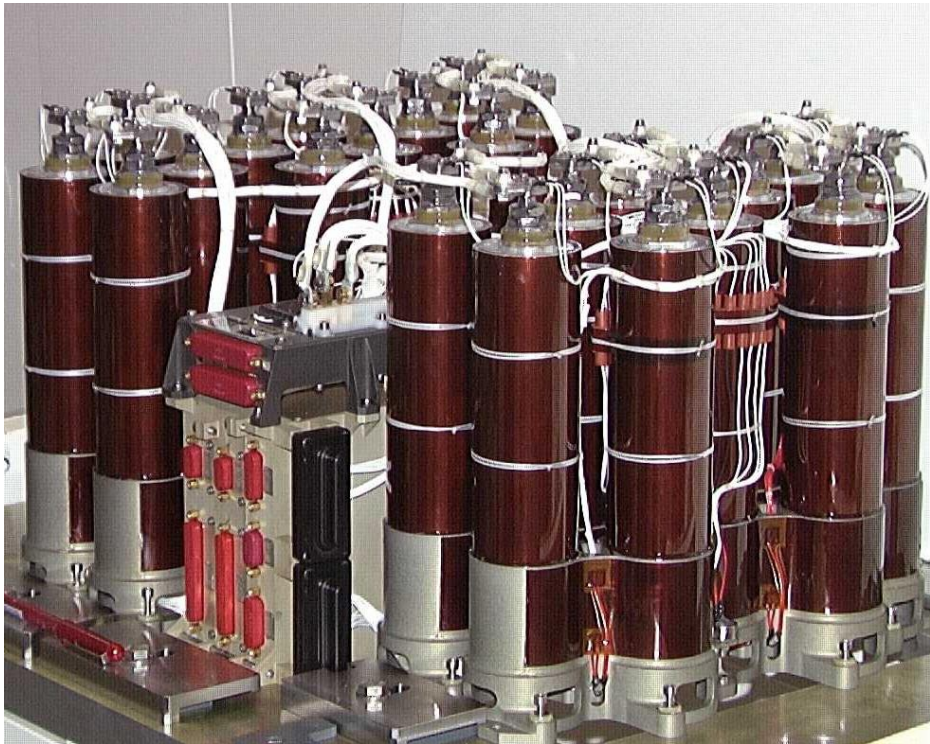
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# VES 140 S : Qualified cell on Stentor

- Energy > 140 Wh (39 Ah) @ 4.10 V
- Specific Energy 135 Wh/kg
- Dimensions : Diameter 54 mm  
Height 250 mm.
- Qualified in **October 1999**
- Product answer specification for the satellite range :
  - GEO 5 to 30 kW and LEO >1.5 kW
- Smart 1 ESA probe with electric propulsion to Moon Orbit
- Cell launched in orbit onboard 105 satellite programs for a total of 13,805 cells from 2002 to 2020. (6 remaining programs waiting for launch)
- Large range of users : Airbus, TAS, BSS, Northrop Grumman, OHB, CAST, IAI, ISRO, INVAP...

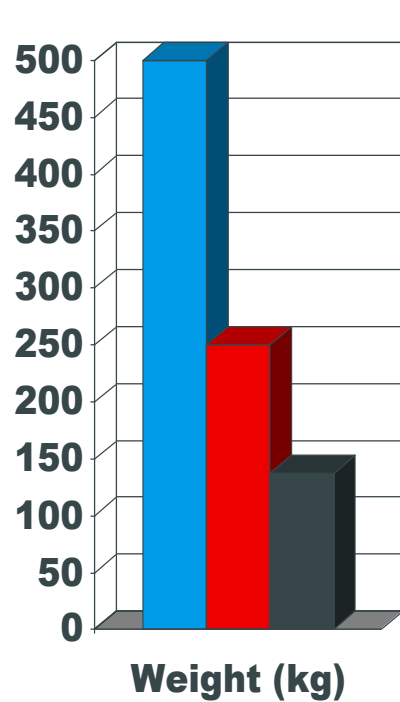


# The STENTOR program was supposed to give flight demonstration of Li-Ion battery

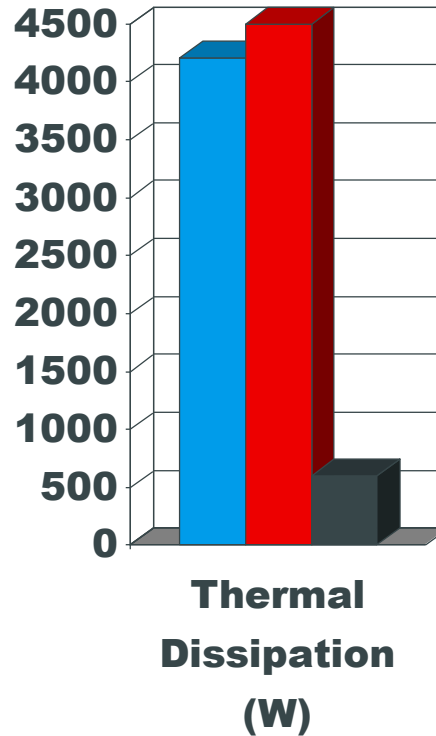


- The Stentor program was a CNES funded GEO Satellite embarking innovative technologies and products (Li-Ion Battery, Electric Propulsion, Active antennas...).
- SAFT has developed the STENTOR Li-Ion battery system (Cells, Battery assembly, By-pass system, Balancing Electronics, Management software)
- The STENTOR Li-Ion battery system included two flight batteries of 80 Ah (11 cell packages in series per battery) delivered for AIT in May 2000
- Unfortunately, 11 Dec 2002, Ariane V ECA V157 launch failed.

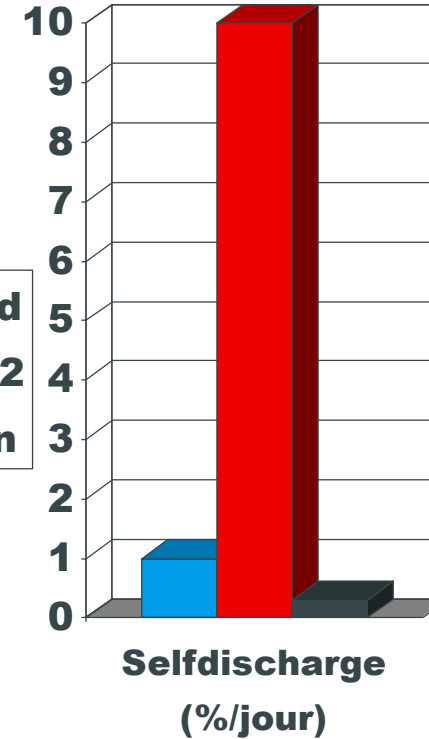
# Lithium-Ion Advantages for satellites applications



50 % weight saving % NiH<sub>2</sub>



Radiator size Reduction



Simplified launch operations

*8 kW GEO satellite*

Other advantages :

- Energetic efficiency : solar panels size reduction
- No memory effect
- Energy gauge using Capacity = f(voltage)
- Modularity and flexibility based on parallel assembly

# SMART1 : First in Orbit

- SMART1 : First of ESA's small missions to test new technologies, and first European spacecraft to go to the Moon.
- Its main objective was to test solar-electric primary propulsion for future deep-space missions.
- 5P 1S Battery with VES140 cell
- Launched the 27th Sept 2003 by Ariane 5
- 14 months elliptic route toward the moon...using PPS1350
- 15 novembre 2004 : First Moon Orbit
- End of Mission 3 September 2006 : The probe crashed on the Moon

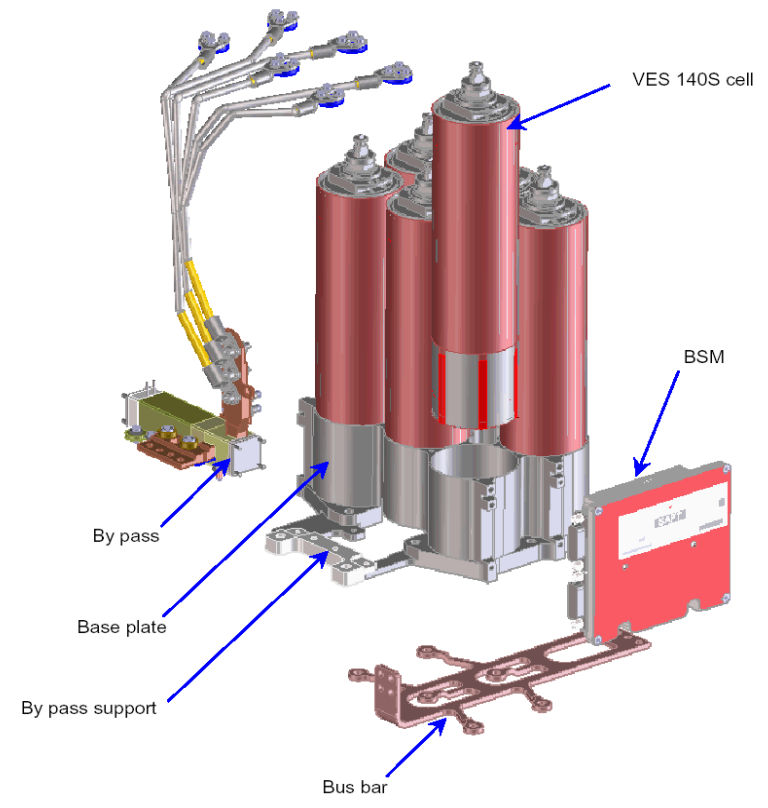


# E3000 Battery Development : Cell modules

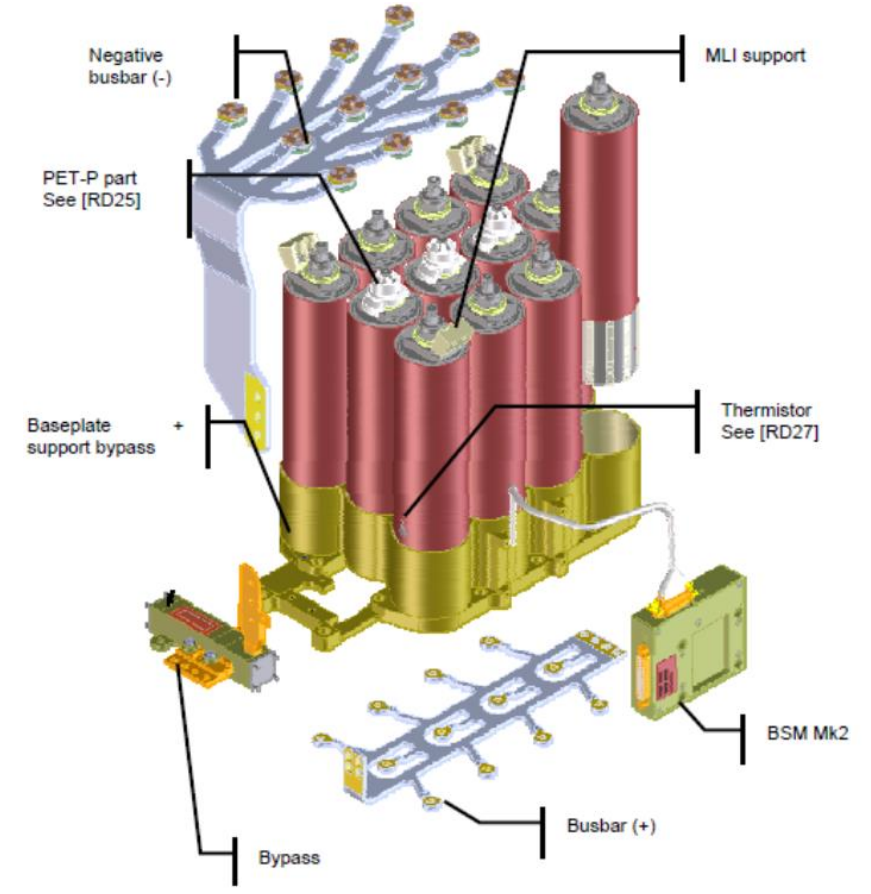


- Starting 2001 : Development of the Eurostar 3000 Li-Ion battery system:
  - cell and cell module including BSM and By-Pass
  - battery design, EPS and battery management software : Airbus
- Qualification : January 2003

2001 to 2013



Since 2012



VES 140 Cells: from 5P (190Ah BOL) to 12P (460 Ah BOL)

VES 180 Cells: from 5P (225Ah BOL) to 12P (540 Ah BOL)



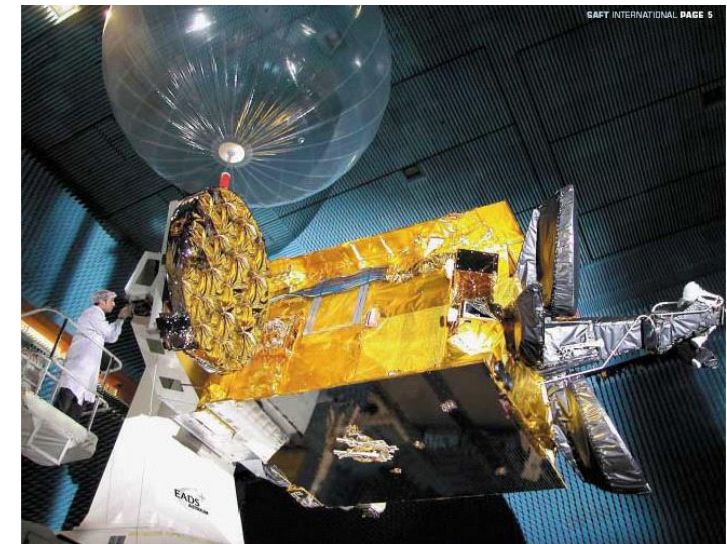


# W3A : First ever launched GEO Communication Satellite powered with Li-Ion Battery

- W3A Eutelsat's telecommunication Satellite with 55 transponders Ku and Ka bands
- First GEO satellite to be powered with Li-Ion Battery (9.3 kW)
- Eurostar 3000 Plate-form from Airbus (previously EADS Astrium)
- Launched the **16th of March 2004**
- Configuration with 2 Li-Ion batteries "6P11S" VES140

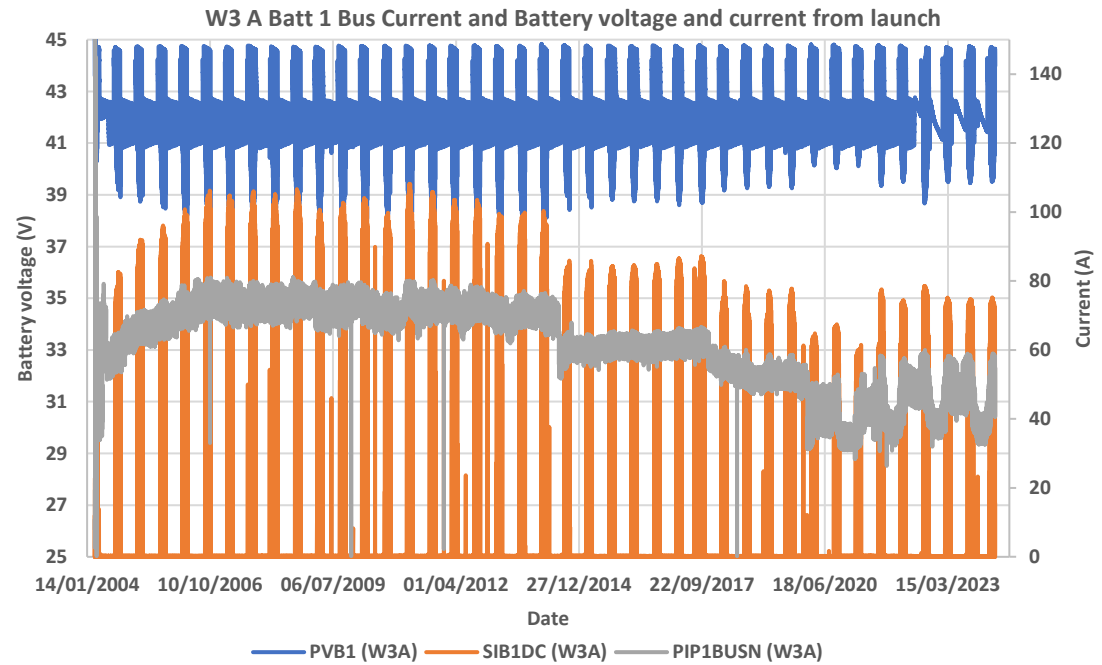


6P Module : 840 Wh- 105 Wh/kg

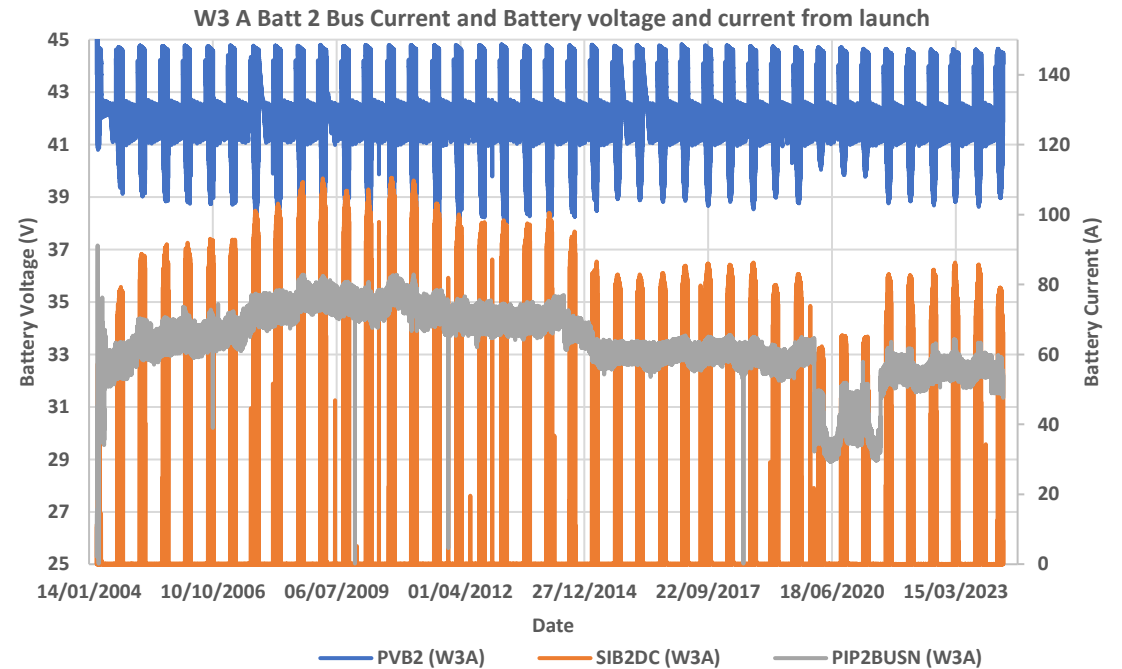


Airbus Credits

- The two Lithium batteries in-orbit on **W3A** have **very similar behavior in 2024** and continue to meet **the expected performances after twenty years in orbit.**
- Nominal and homogeneous behaviour of the batteries
- Behaviour in solstice and performances during eclipse seasons are fully nominal
- **50 Eurostar 3000 S/C with Li-Ion batteries in flight** are performing perfectly well: they deliver the expected performances, and have very similar and stable behavior



Battery 1

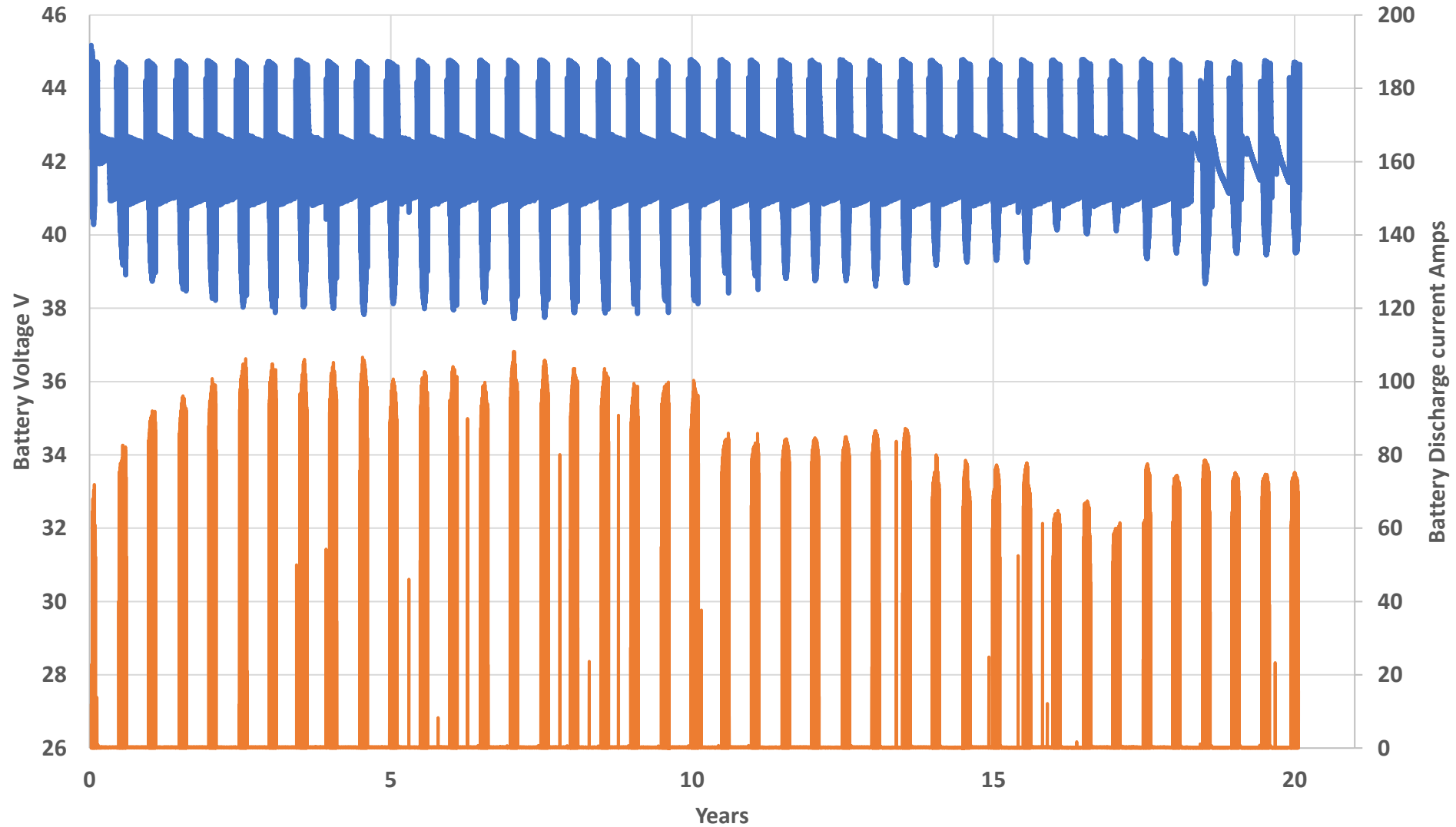


Battery 2

# W3A Battery Voltage and Current

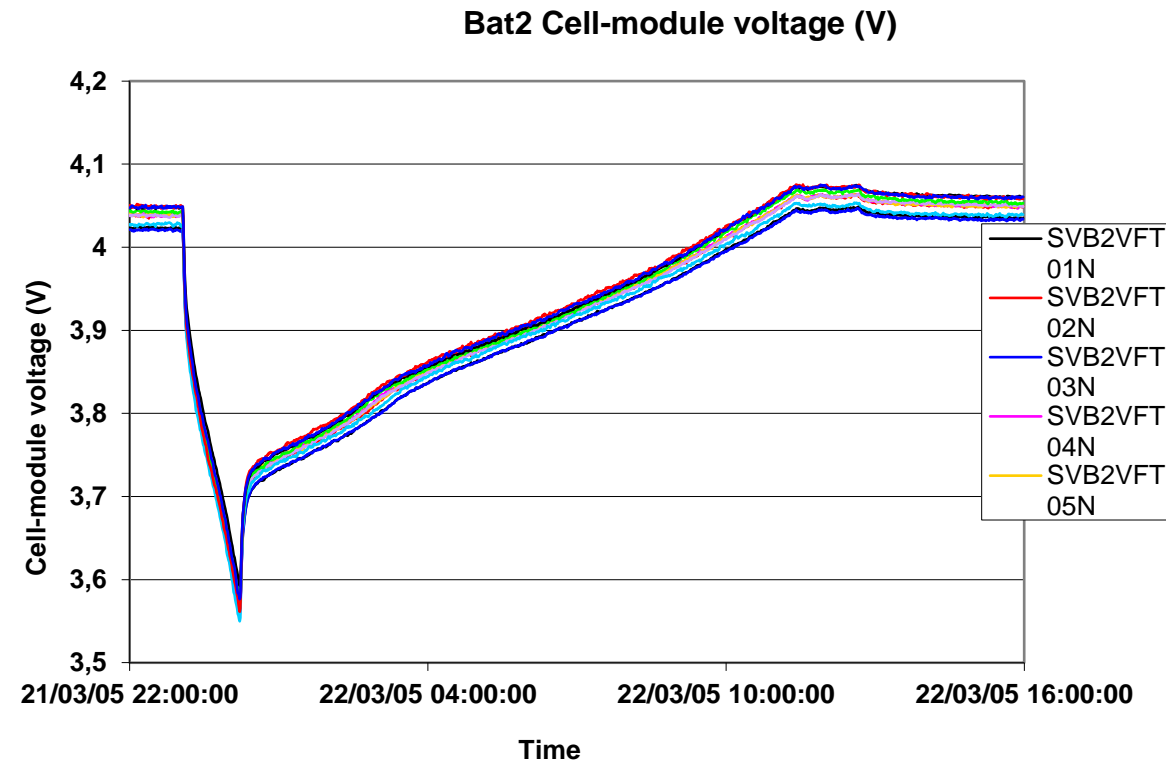
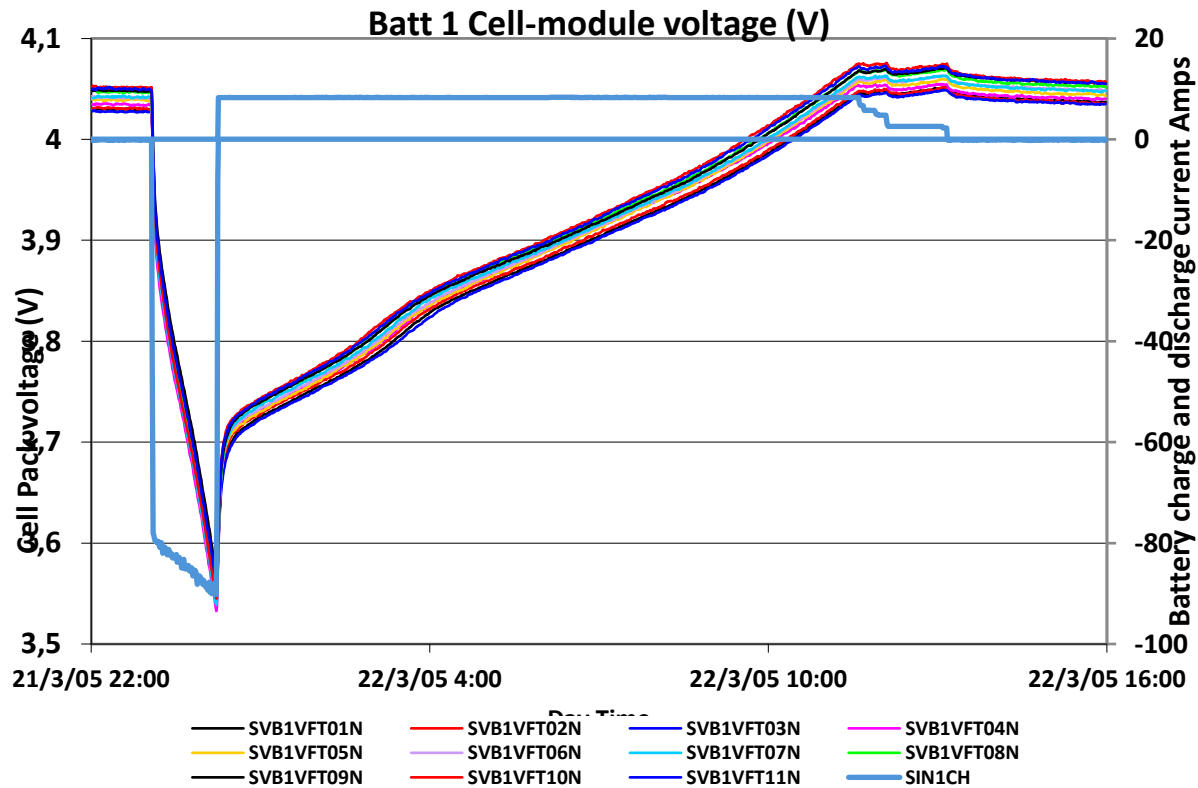


W3A Battery 1 Voltage and Discharge Current



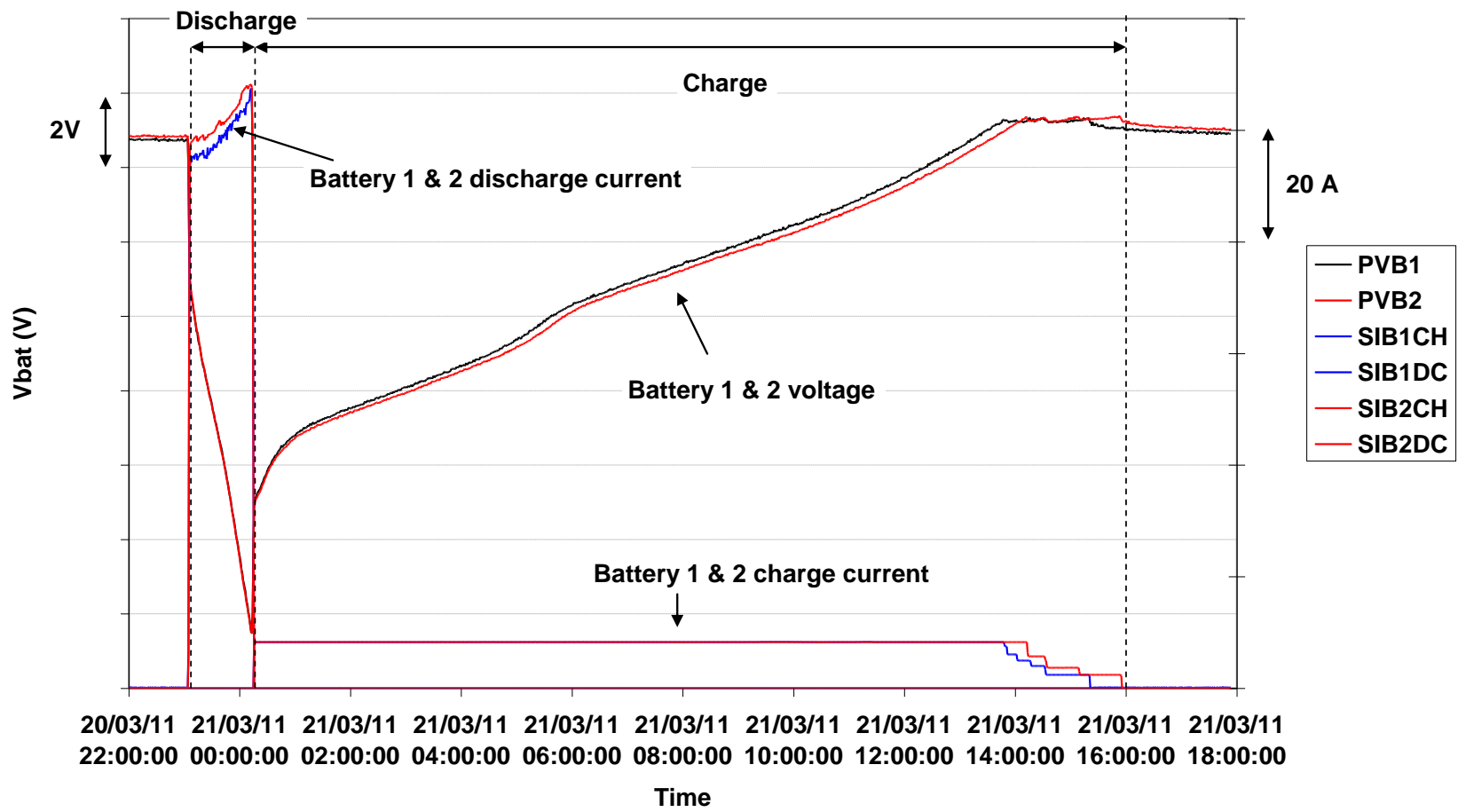
— U bat — I Bat

- Same behaviour for the two batteries after 1 year in orbit

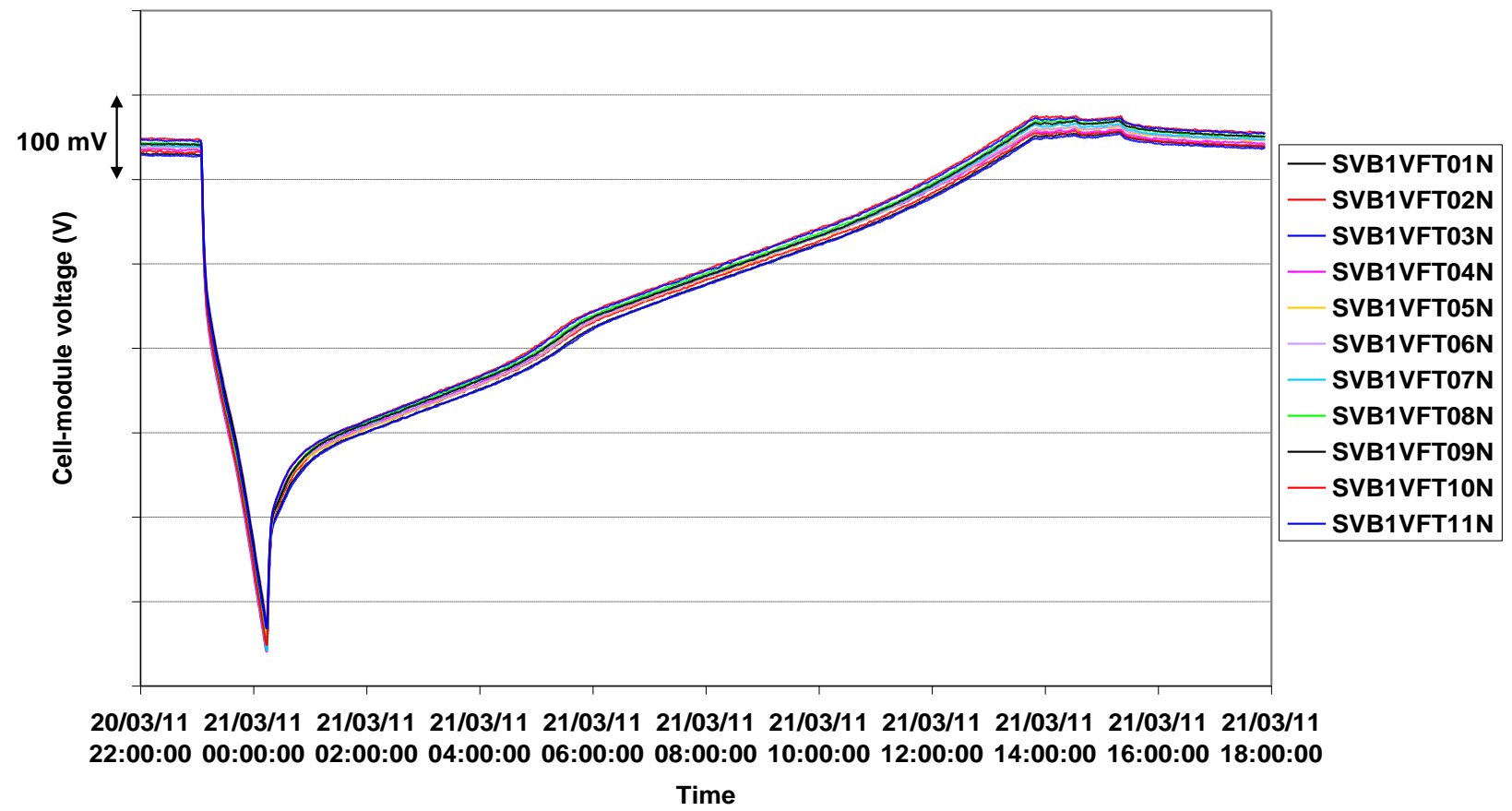


- Homogeneous performances of the two batteries after 7 years

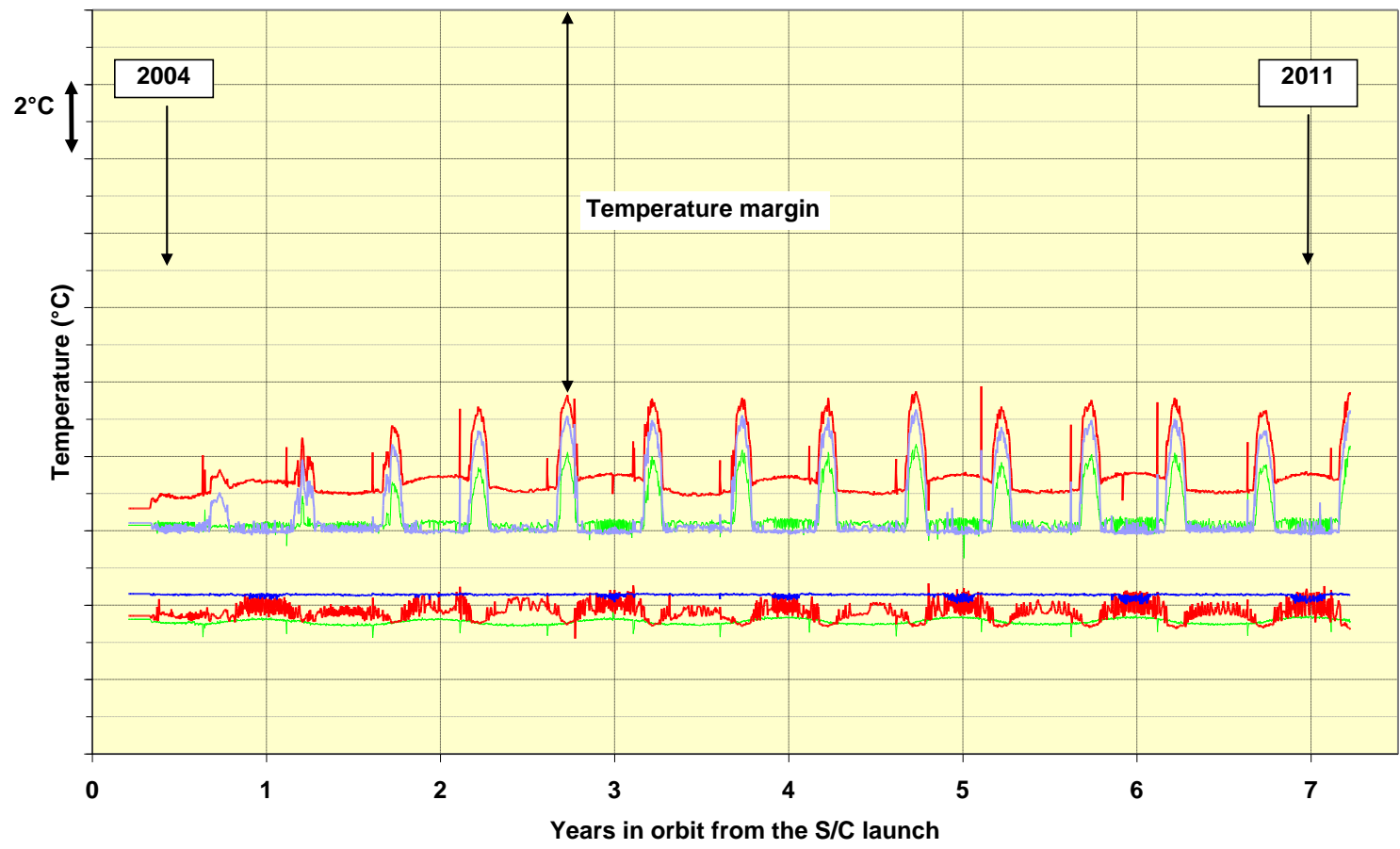
Battery 1 and 2 electrical behaviour eclipse of the 21/03/2011: nominal



- Homogeneous behaviour of the 11 cell-modules after seven years : performing balancing system

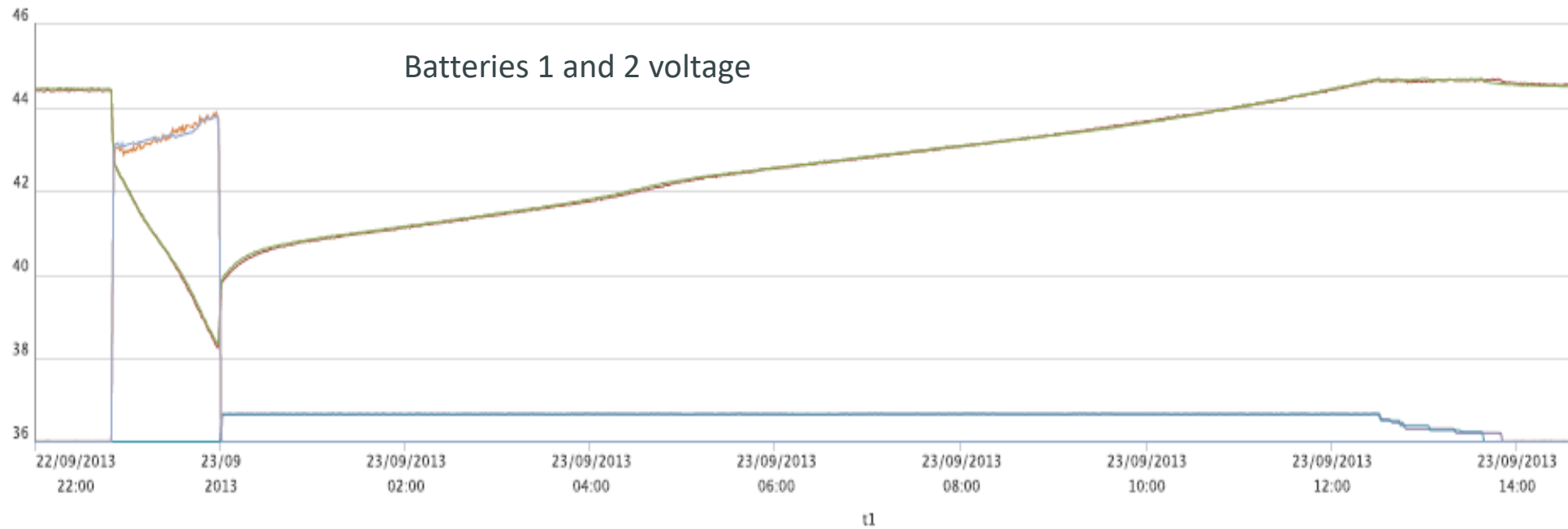


- Temperature increase in eclipse in the expected range
- Nominal behaviour, with margin





- No degradation measured on the two batteries after ten years



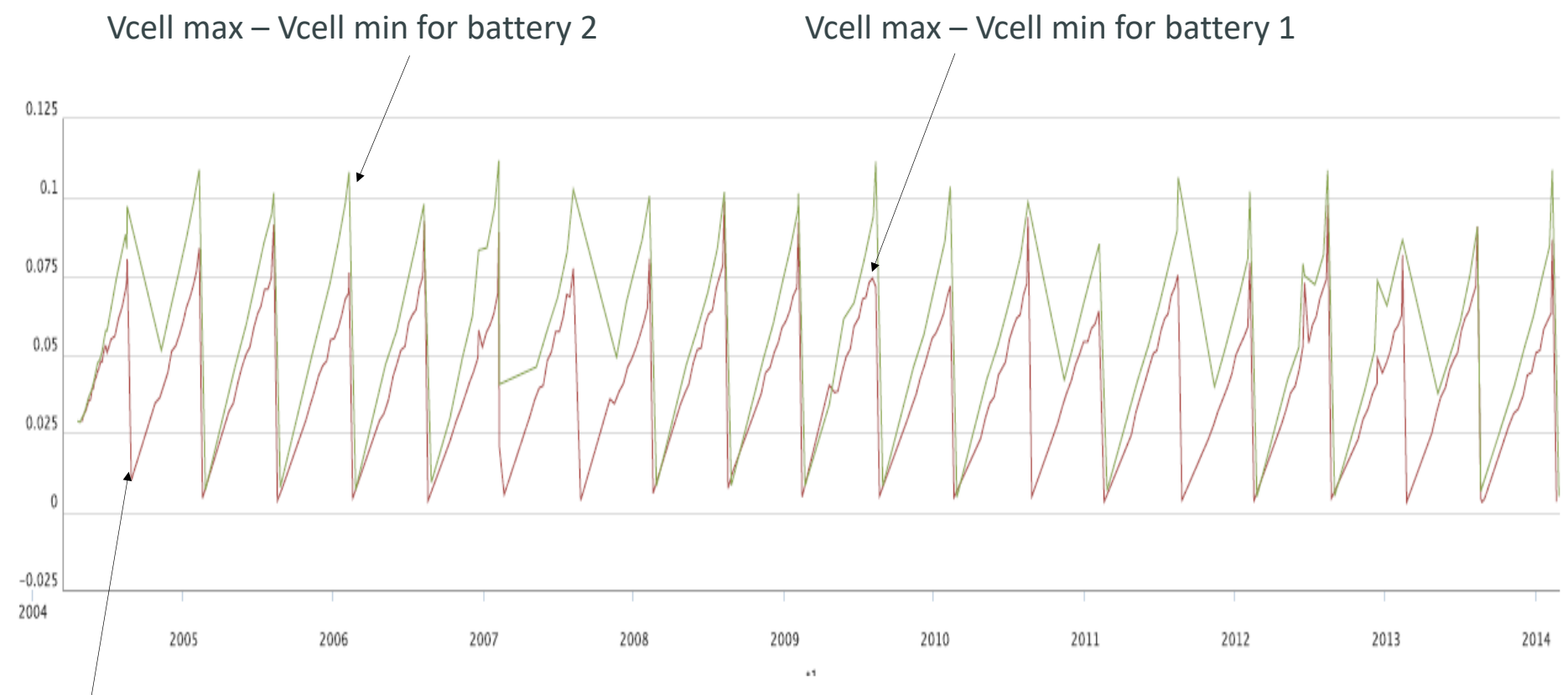
Discharge

Charge: 13 hours



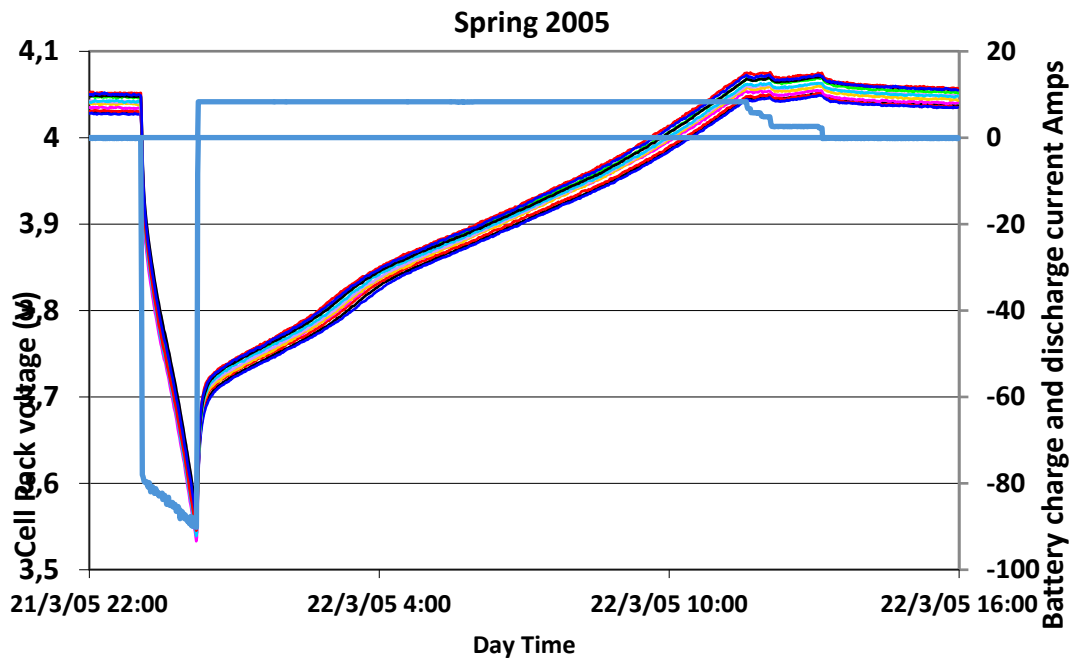
# Balancing performance review after 10 years (F.Mallet ESPC 2014)

- Balancing operation : Spread between max and min cell module voltage increases according to time.
- **Spread is below 10 mV before equinox season**

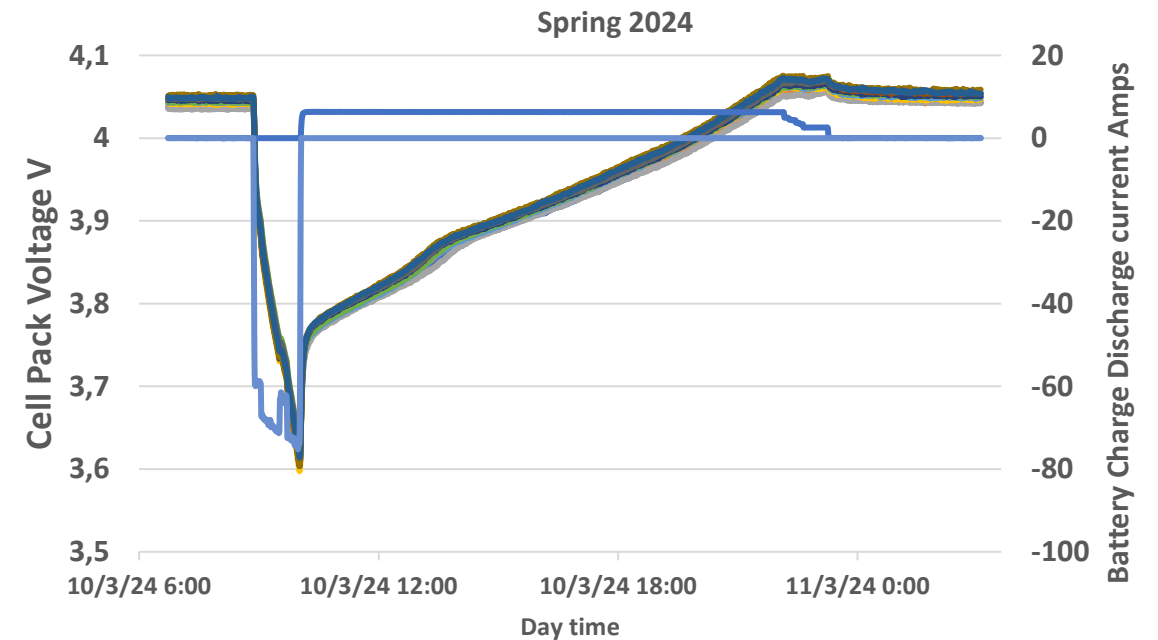


Balancing before equinox season

- The two Lithium batteries in-orbit on W3A have exactly same similar behavior from Spring Season 2005 to 2024.
- Cell to cell spread is constant during the 20 years with less than 15 mV at EOC



— SVB1VFT01N	— SVB1VFT02N	— SVB1VFT03N	— SVB1VFT04N
— SVB1VFT05N	— SVB1VFT06N	— SVB1VFT07N	— SVB1VFT08N
— SVB1VFT09N	— SVB1VFT10N	— SVB1VFT11N	— SIN1CH



— SVB1VFT01N	— SVB1VFT02N	— SVB1VFT03N	— SVB1VFT04N	— SVB1VFT05N
— SVB1VFT06N	— SVB1VFT07N	— SVB1VFT08N	— SVB1VFT09N	— SVB1VFT10N
— SVB1VFT11N	— SIB1CH	— SIB1DC		

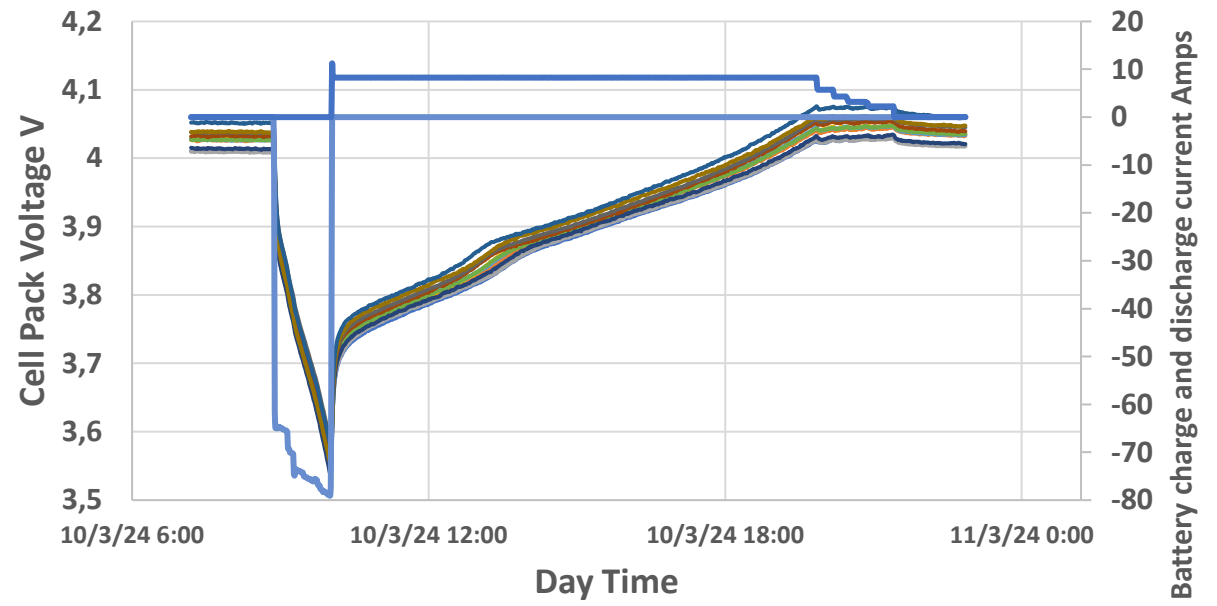
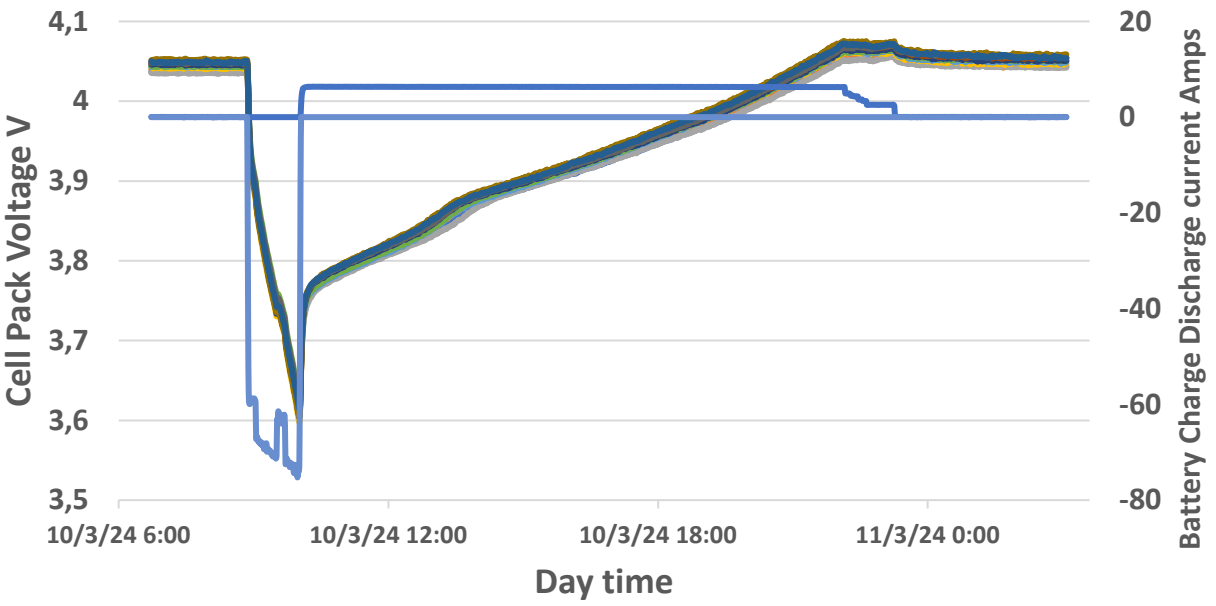
# Longest Eclipse Day March 2024



- The two Lithium batteries in-orbit on W3A are showing very similar behavior (39.5 and 39 V).
- Cell Pack Voltage
  - Batt 1 from 3.597 V to 3.627 V (75.2 amps discharge ) : 30 mV spread
  - Batt 2 from 3.543 V to 3.581 V (79 Amps discharge) : 38 mV spread

W3 A Batt 1 : Longest eclipse day Spring 2024

W3 A Batt 2 : Longest eclipse day Spring 2024



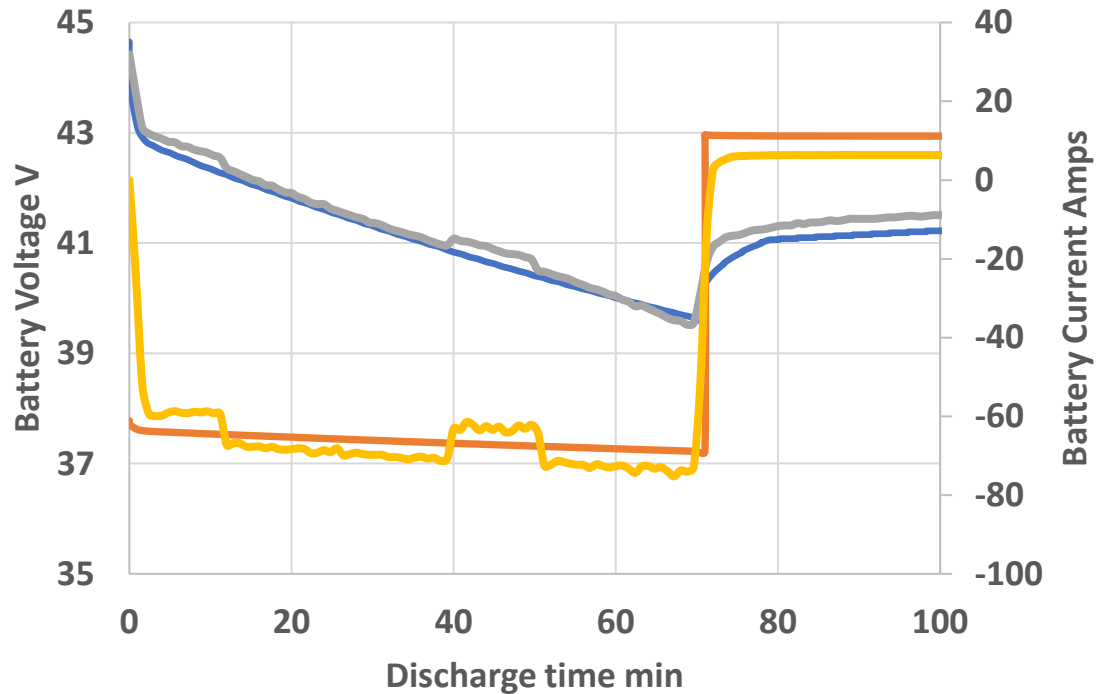
SVB1VFT01N SVB1VFT02N SVB1VFT03N SVB1VFT04N SVB1VFT05N  
 SVB1VFT06N SVB1VFT07N SVB1VFT08N SVB1VFT09N SVB1VFT10N  
 SVB1VFT11N SIB1CH SIB1DC

SVB2VFT01N SVB2VFT02N SVB2VFT03N SVB2VFT04N SVB2VFT05N  
 SVB2VFT06N SVB2VFT07N SVB2VFT08N SVB2VFT09N SVB2VFT10N  
 SVB2VFT11N SIB2DC SIB2CH



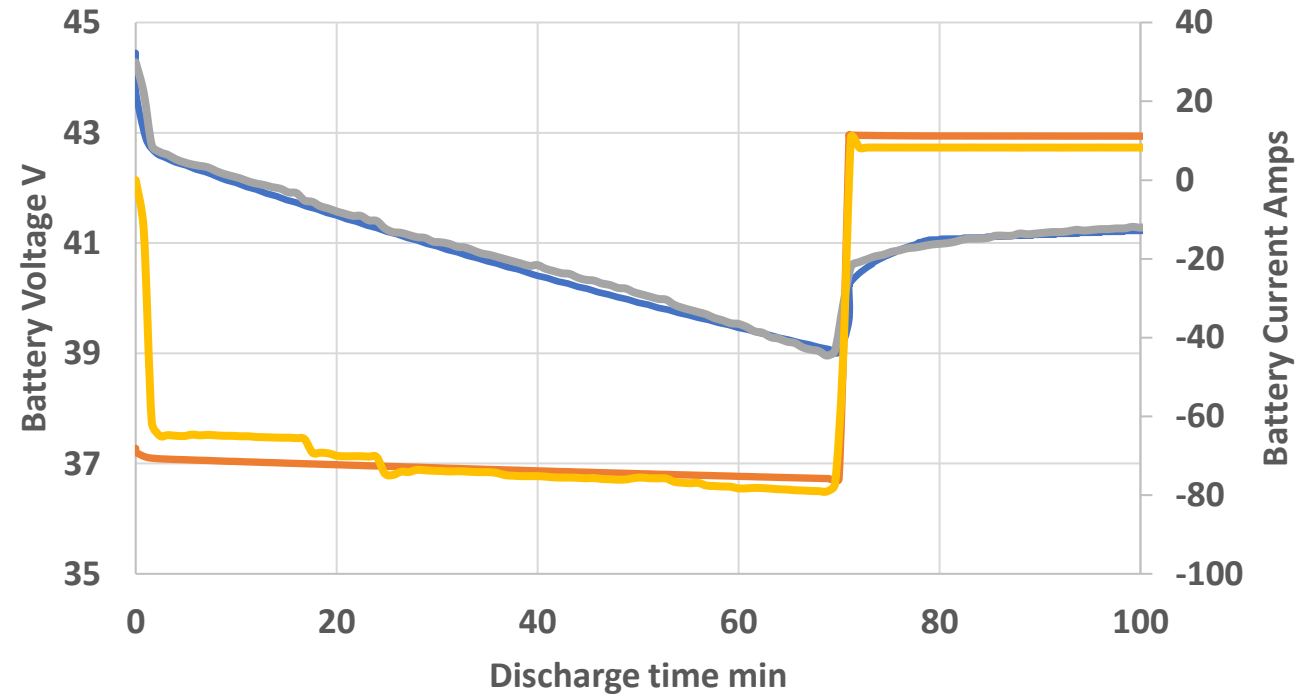
- Good fitting with SLIM 20 years simulation on longest day of March 2024 eclipse

W3A-Bat 1 Season 40 day 23



UI Bat Slim U Batt 1 I Bat Slim I Batt 1

W3A-Bat 2 Season 40 day 23



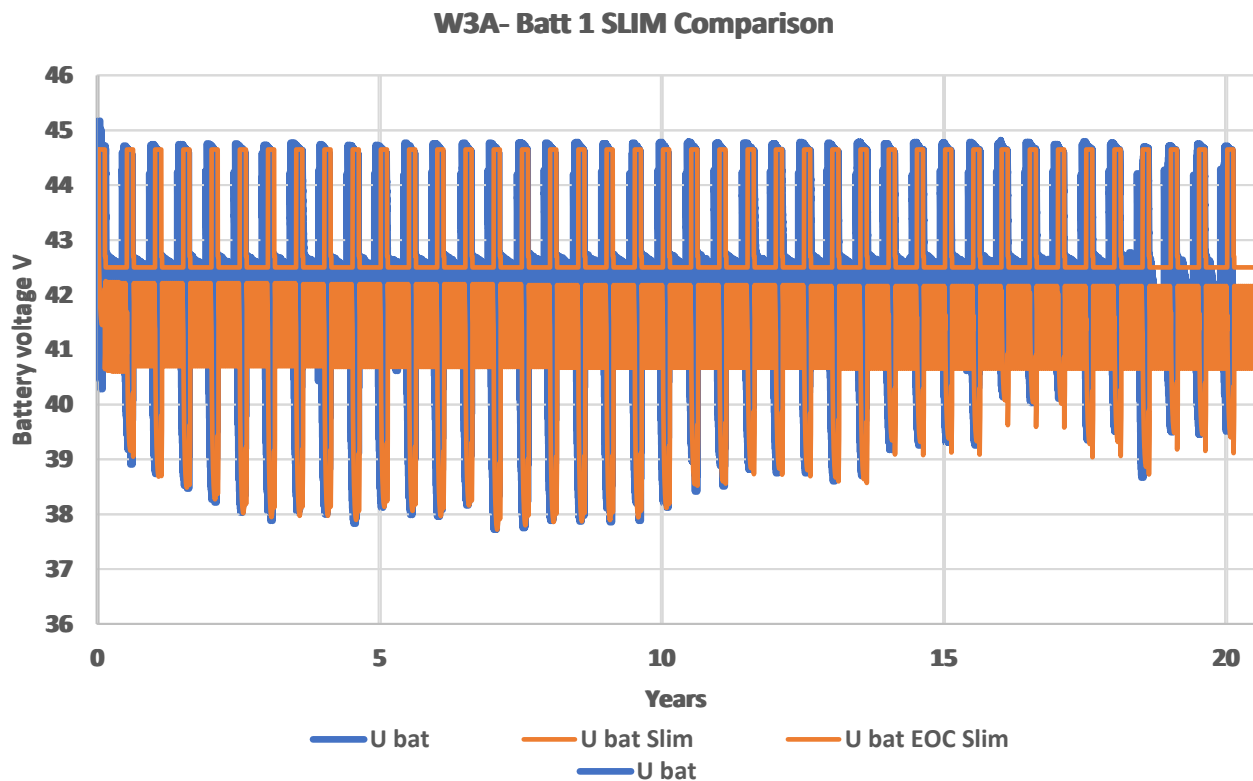
U Bat Slim U Batt 1 I Bat Slim I Batt 1



# NASA Aerospace Battery Workshop W3 A Battery degradation evaluation



- Battery health status have been done after 7 years in 2011 and 10 years in 2017
- Articles presented at ESPC 2011 and 2017
- Analysis with Slim Saft Life model tool after 20 years : **W3A battery energy loss 2.5 %**



- The **two Lithium batteries in-orbit on W3A have very similar behavior in 2024** and continue to meet the expected performances after **twenty years in orbit (40 GEO seasons)**.
- Nominal and homogeneous behaviour of the batteries
- Behaviour in solstice and performances during eclipse seasons are fully nominal
- Temperatures and balancing managements are as expected
- Battery Min EODV along the mission is 37,8 V obtained from 2009 and 2014 corresponding to the highest discharge battery current 108-109 Amps.
- **Battery Energy fading after 20 years in orbit calculated at 2.5 % losses** (Slim Model calculation )
- Battery EODV in March 2024 = 40,2V corresponding to 3.64 V/cell package

## **393 satellites in-orbit with Li-ion (GEO, MEO & LEO) : 372 operational**

More than 3 Billion of cell hours in orbit with **no failure or deviations**

Total over 4.2 MWh in-orbit with 700 batteries and more than 50 000 cells in orbit

- **213 GEO satellites Launched + 1 Moon Mission :**

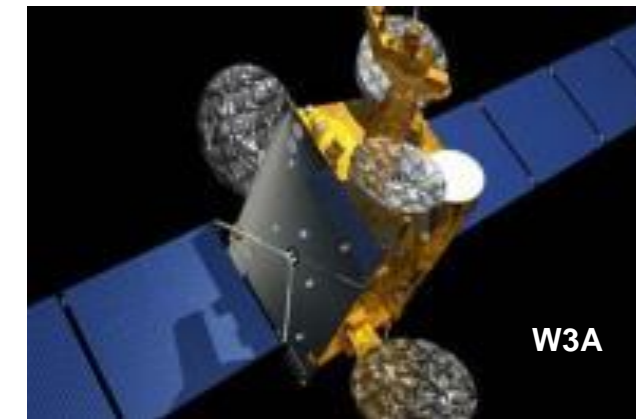
2003 : Smart 1 has been able to reach Moon orbit thanks to ion thrusters' engines powered with Li-Ion battery

1<sup>st</sup> GEO Telecommunication satellite W3A launched 20 years ago (March 2004) with VES140 batteries

- **5 MEO satellite** flying with VES technology:

- **175 LEO satellites including :**

80 first Iridium Next satellites with VES16 batteries



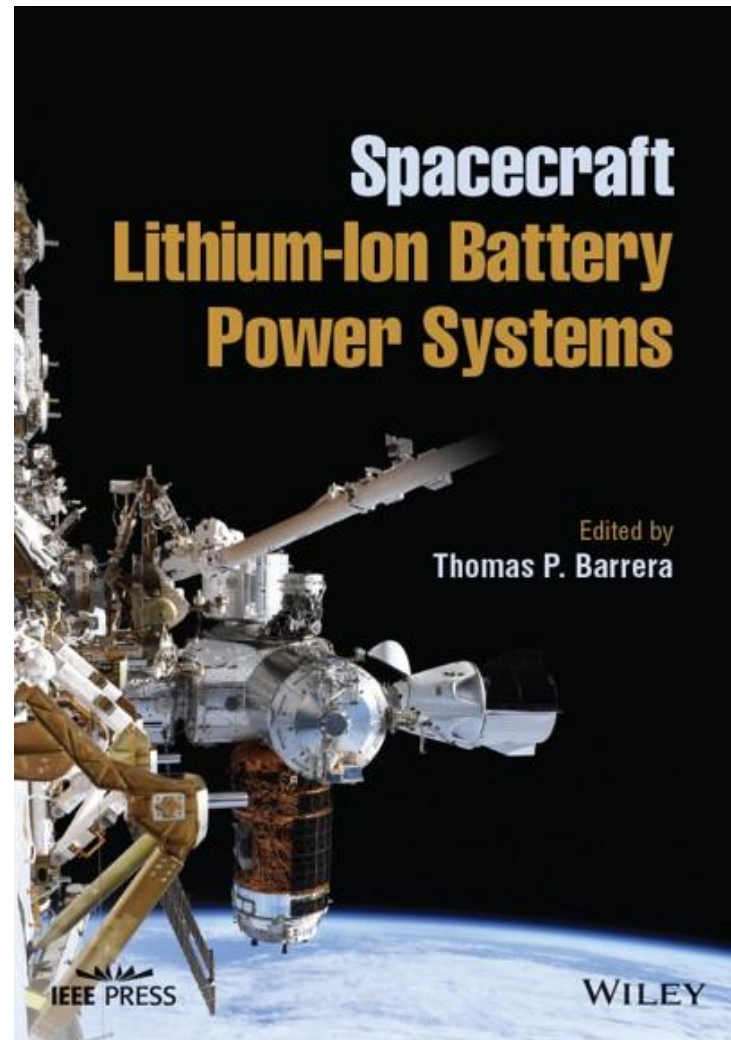


# Conclusion and Acknowledgements



- W3A Battery Energy fading after **20 years is demonstrating very impressive performances**
- **105 satellites powered with VES140 batteries** have been launched are always in operation with limited degradation and no performance deviation
- All Saft cell technology evolutions are exhibiting the same trend with very low degradation as per VES140 : VES16, VES180, VL51ES and the new VL10ES (220 Wh/kg) that is now qualified
- New technology step will be with the Solid State batteries

***Saft and Airbus thank Eutelsat, CNES, and ESA, for their contribution and support all over the Eurostar 3000 battery development.***



Thank you