

VL10ES High Energy Space Cell Development

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Agenda

1. VL10ES main objectives
2. Key R&D orientations
3. Feasibility & Qualification test plan
4. EM0 design steps
5. EM0 cycling tests results
6. EM0 preliminary safety tests
7. Conclusions & Perspectives

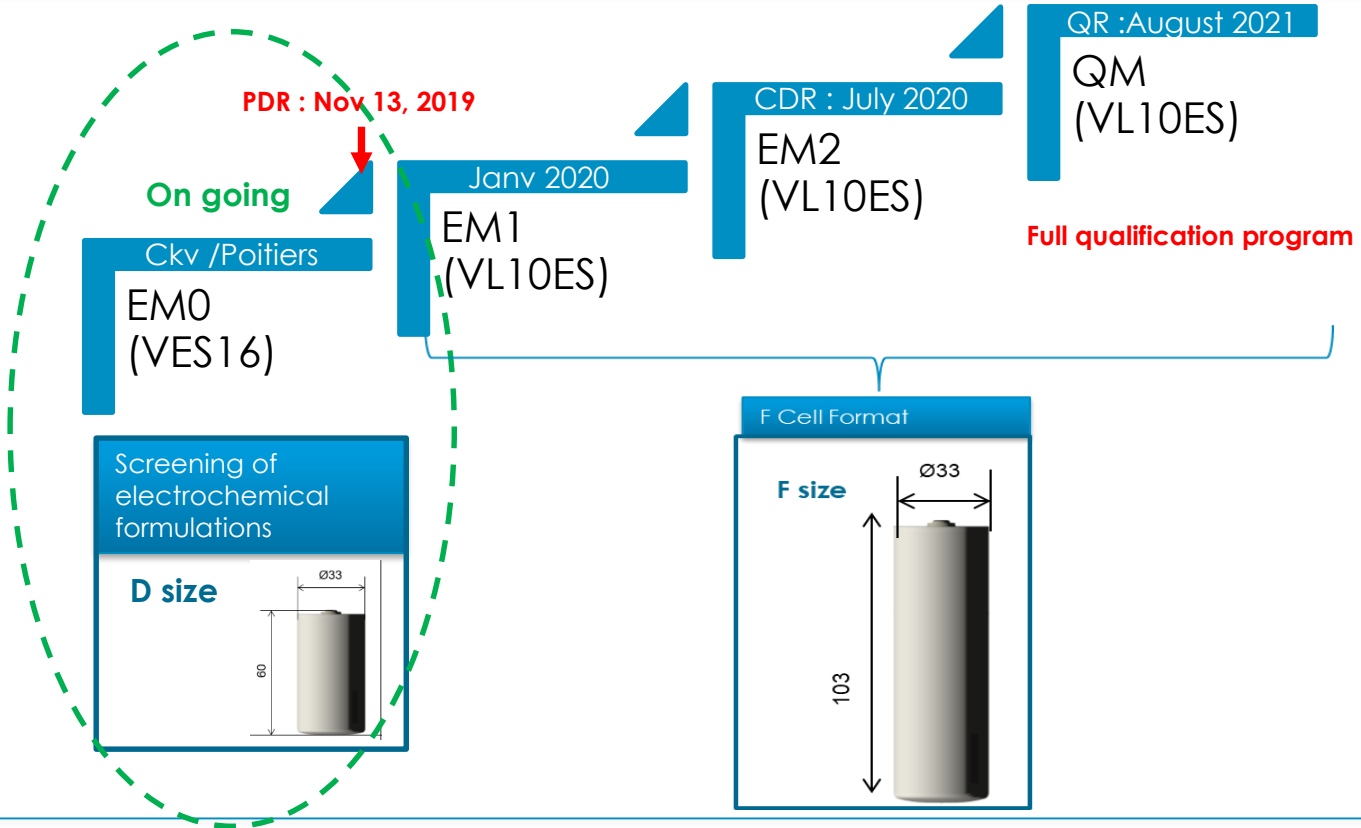
A satellite constellation is shown orbiting Earth. The satellites are connected by a network of green lines, representing data links. The Earth's surface is visible in the background, showing continents and oceans. A large, semi-transparent blue trapezoidal shape is overlaid on the right side of the image, containing the main title.

1. VL10ES OBJECTIVES

VL10ES development : our goals

- Address LEO & GEO markets, including Constellations, with a new Safe High Energy density cell, able to sustain long range space cycling.
- Develop an innovating electrochemistry able to reach more than 220 Wh/Kg at cell level and able to sustain long range cycling in LEO (20% DOD - 30%DOD) & GEO (70% DOD targeted).
- Use innovating & industrial active materials/components for electrode densification in synergy with Saft Corporate Solid State program.
- Use a similar stainless steel casing than VES16, to keep advantages regarding robustness & safety, even for a High Energy density cell.
- Reduce cell number in large batteries, reduce overall cost at battery system, that will answer the satellite builder targets, and propose an alternative with low cost approach for constellation systems
- BoL VL10ES Qualification Review for mid-2021.

VL10ES global Development Plan



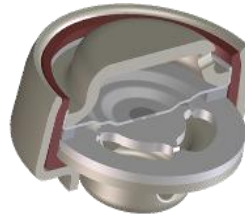
A satellite view of Earth at night, showing city lights and the curvature of the planet. A large, semi-transparent blue trapezoidal shape is overlaid on the right side of the image, containing the section header text.

2. KEY R&D ORIENTATIONS

VL10ES : robust casing & safety devices ; VES16 continuity

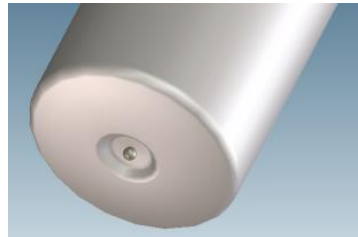
Symmetric double vent safety device

- Similar to VES16
- Pressure release in case of local significant gas generation



Circuit breaker

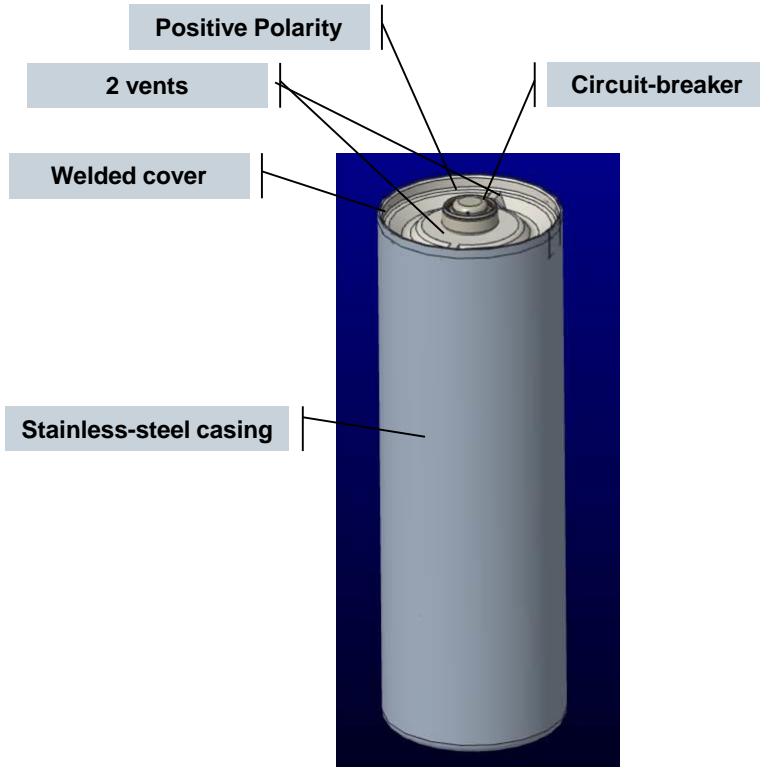
- Similar to VES16
- New plastic parts for T° robustness



Cell Bottom

- Similar to VES16
- Fill-in hole / closing ball
- Stainless-steel Bottom Cover

VL10ES : summary of targeted performances versus VES16



Cell type	VES16 (D-size)	VL10ES (F-size)
Dimensions (Ø x H)	33 x 60 mm (D-size)	103 x 60 mm (F-size)
Weight	≤ 115 g	≤ 215 g
Volume	0.051 dm ³	0.086 dm ³
Voltage range	2.7 V - 4.1 V	2.7 V - 4.2 V
Nominal capacity	4.5 Ah @ 4.1V, 20°C	> 12 Ah @ 4.2V, 20°C
Nominal energy	16 Wh @ 4.1V, 20°C	> 47 Wh @ 4.2V, 20°C
Specific energy	> 140 Wh/kg	> 220 Wh/kg
Internal resistance	≤ 35 mΩ @ 20% DoD	≤ 25 mΩ @ 20% DoD / TBC
Operating temperature	+10°C / +40°C	+10°C / +40°C
Mechanical design margins	EWR & ECSS compliant	EWR & ECSS compliant

A background image showing a view of Earth from space, with a bright light source (likely the sun) creating a lens flare effect. A large blue trapezoidal shape is overlaid on the right side of the image, containing the main title.

3. VL10ES CELL QUALIFICATION PLAN

VL10ES Cell Qualification plan content

EM0 selection in D VES16 format

Electrical	Mechanical	Thermal	Space life Tests	Safety
Accelerated 80-100% DOD			LEO real time	Overcharge
			GEO real time	Over temperature

VES10ES cell qualification (QM) F final format

Electrical	Mechanical	Thermal	Space life Tests	Safety
Accelerated 80-100% DOD	Vibration	T/V	LEO real time	Overcharge
DST	Shock	Dissipation	GEO real time	Overdischarge
Dch vs T°	DPA	Capacity	LEO accelerated	Ext. short
Impedance, Ri	Strains Measures/Model	Thermoneutral potential	GEO accelerated	Over temperature
EMF vs SoC			Storage vs T° & SOC	ARC test
			Radiation Test	Safety test vs VES16

EM1



4. EMO DESIGNS STRATEGY

EM0 prototypes : 2 development phases



EM0	EM01, EM02	EM03	EM05	EM07 (a,b,c)	EM08	EM012	EM013
Design	Positive step 1 Negative A & B Separators Electrolytes	Positive step 2 Negative A Separators Electrolytes	Positive step 3 Negative C	Positive step 4 Negative C Porosity range Σ Electrolytes	Positive step 4 Negative D (with SiOx)	Positive step 5 Negative C	Positive step 5 Negative D (with SiOx)
Targeted Energy in final F format (C/2)	190-195 Wh/Kg	205-210 Wh/kg	200-205 Wh/kg	215-220 Wh/Kg	220-225 Wh/Kg	220-225 Wh/kg	≥ 225 Wh/Kg
Cycling 100% DOD C/3 - C/2 (criteria : -20% capacity at 1000 cycles)	1000 cycles (-11%)	830 cycles (-16%)	720 cycles (-7%)	720 cycles (-12%) Stabilization	Similar trend to EM07 at 350 cycles	Better trend than EM07 240 cycles (-2%)	Better trend than EM07 240 cycles (-5%)

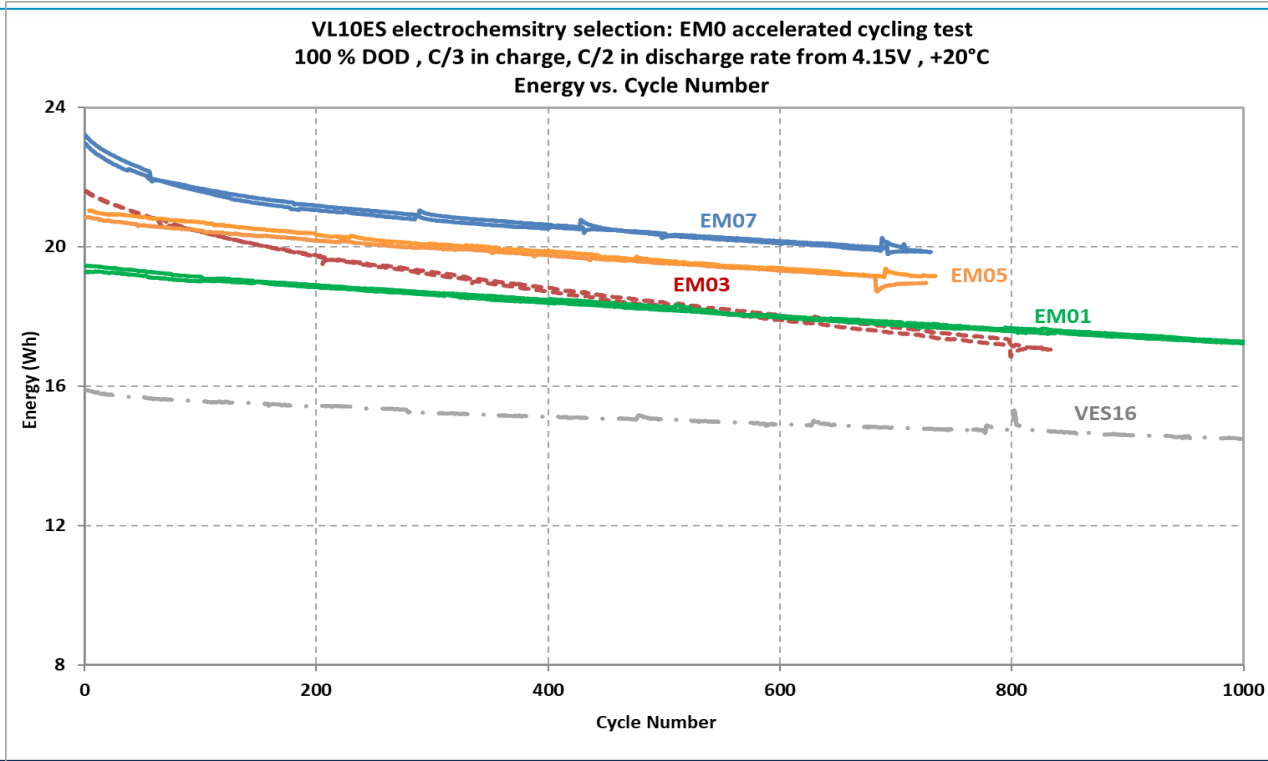
* VES16 /155 Wh/Kg [100% DOD 4.1V] = 1000 cycles (-10%)



5. EMO CYCLING TESTS RESULTS

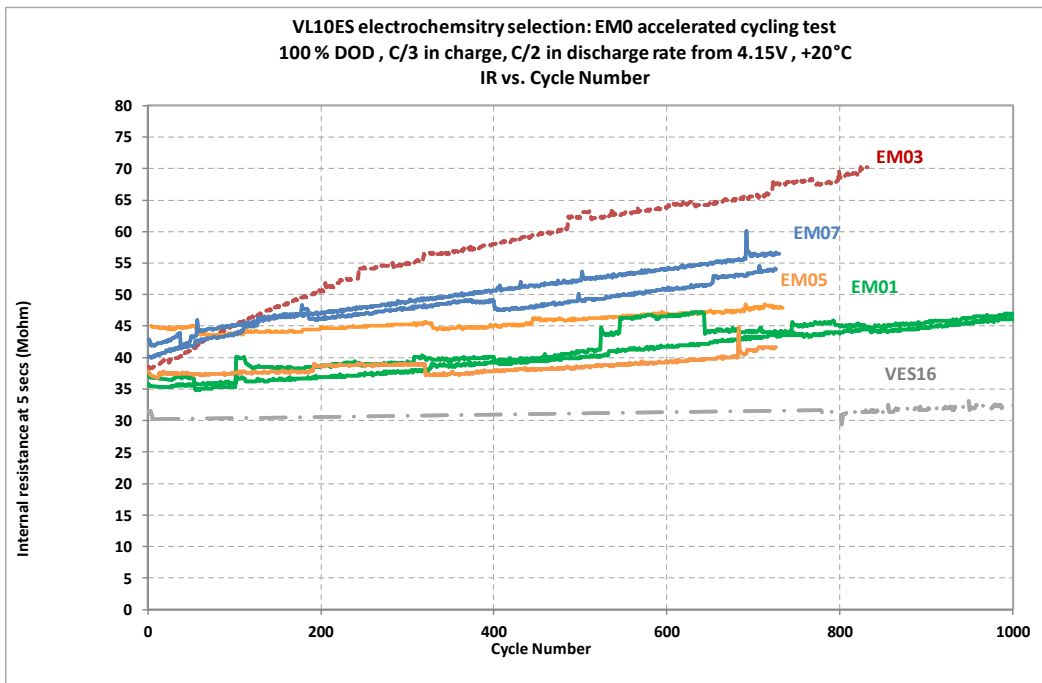
EM0 second first step selection : 100% DOD accelerated cycling

(C/3, D/2 , 20°C and 4.15 V)



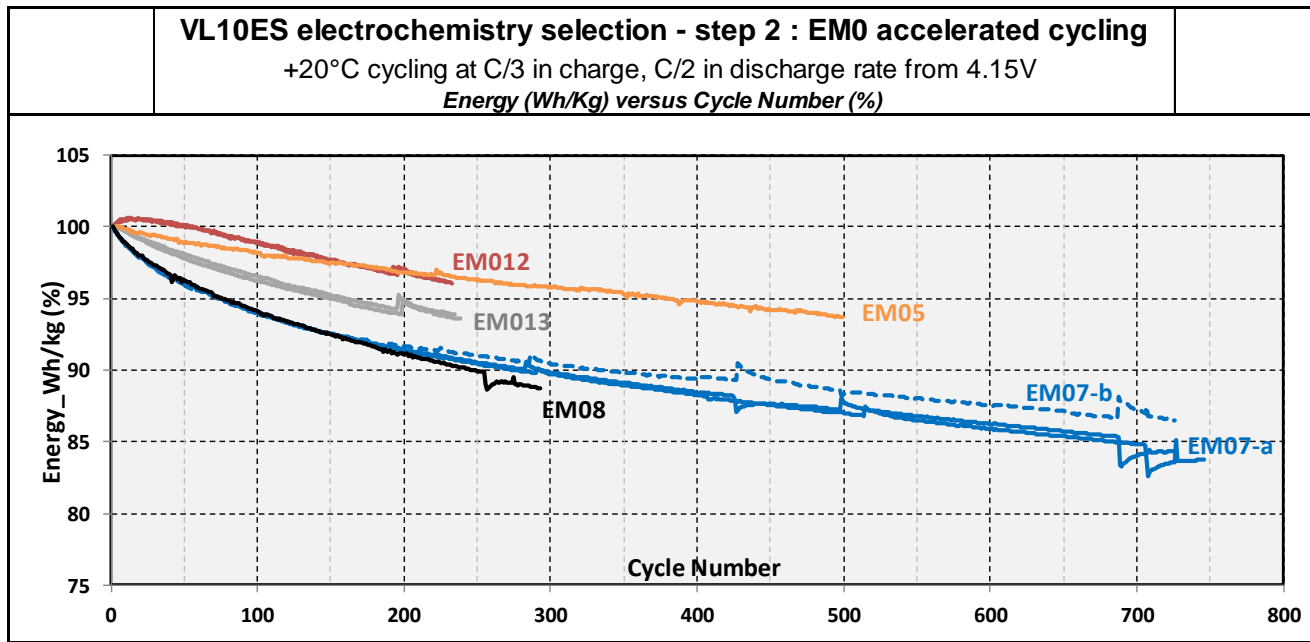
- **Fading ranking : EM03 > EM07 > EM01 ~ EM05 ~ VES16 ; EM07 stabilization after 100 cycles**

EM0 second first step selection : 100% DOD accelerated cycling (C/3, D/2 , 20°C and 4.15 V)



- Internal resistance stability over 1000 cycles u
- Same ranking as per fading : EM03 > EM07 > EM01 ~ EM05 ~ VES16

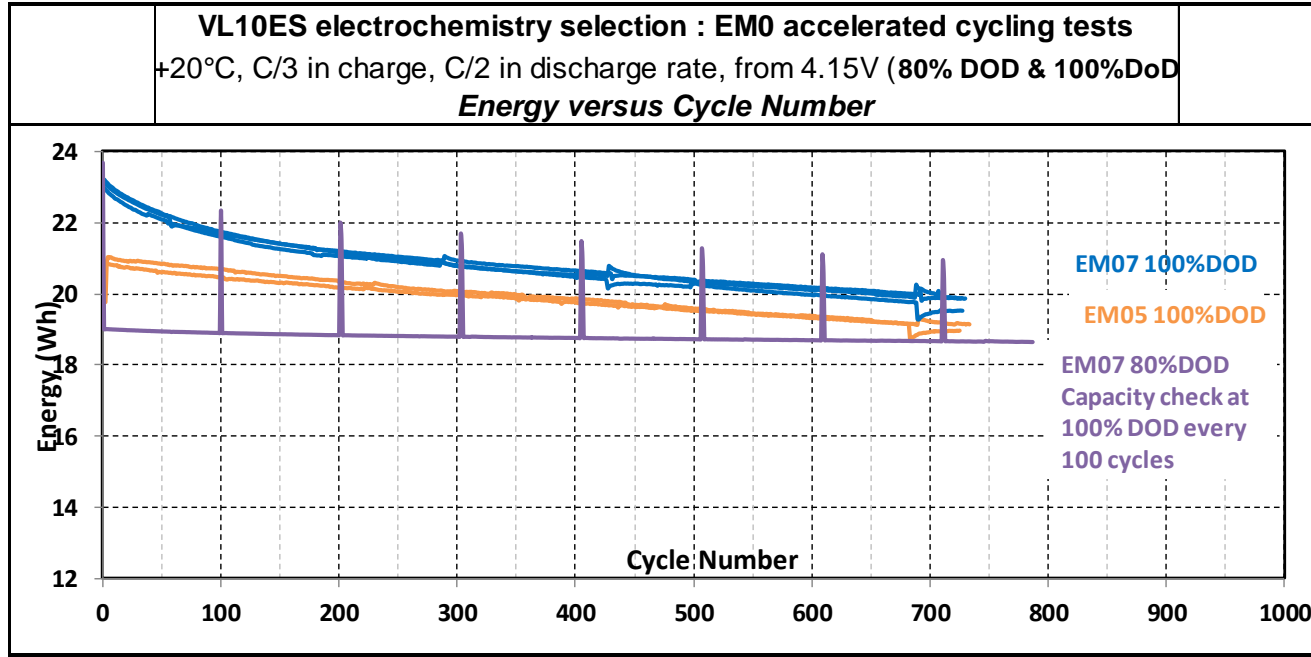
EM0 second step selection : 100% DOD accelerated cycling (C/3, D/2 , 20°C and 4.15 V)



- Low fading in 100% DOD accelerated cycling for last EM012 & EM013 designs
- Slight fading increase using SiOx additive in EM013 versus EM012 (wo SiOx) / TBC on cycling

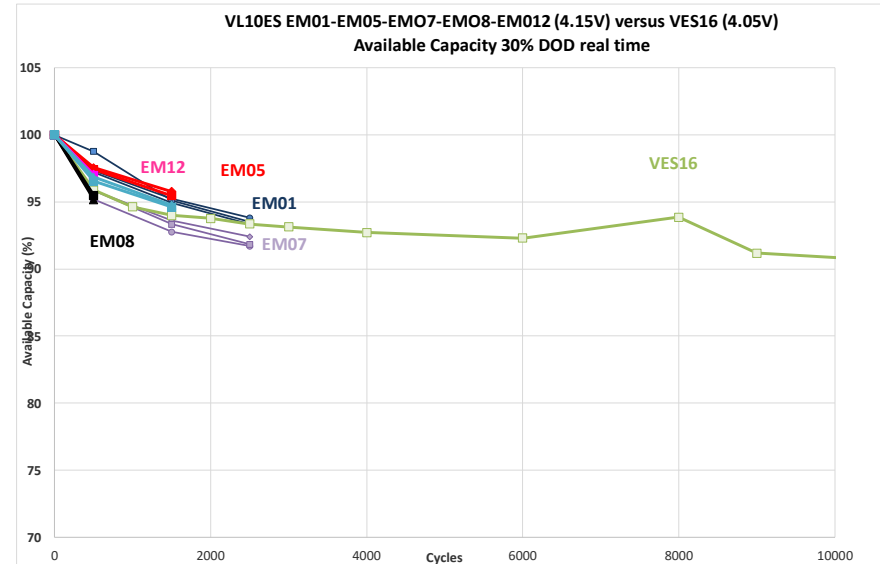
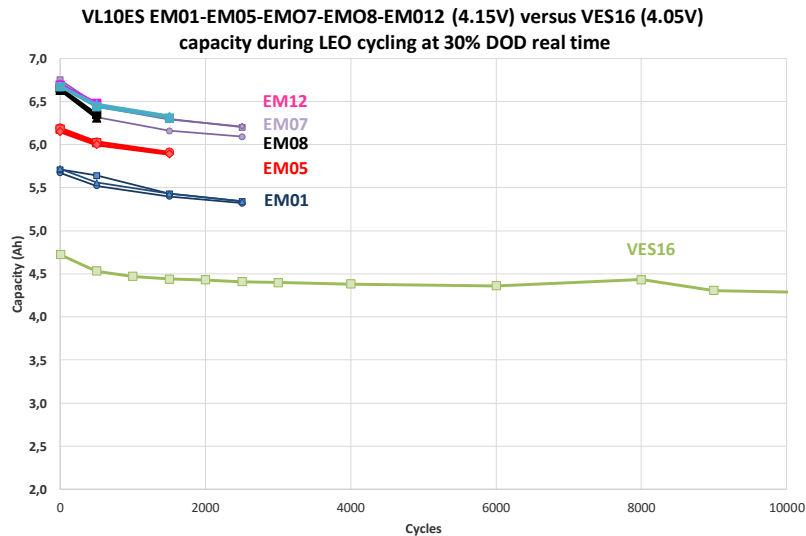
EM0 selection : 80% DOD accelerated cycling

(C/3, D/2 , 20°C and 4.15 V)

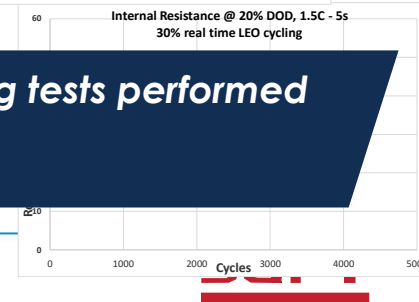


- EM07 capacity loss improvement in cycling at 80% DOD versus 100% DOD

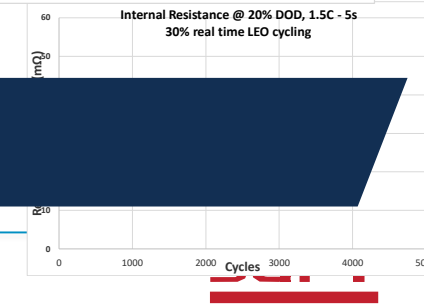
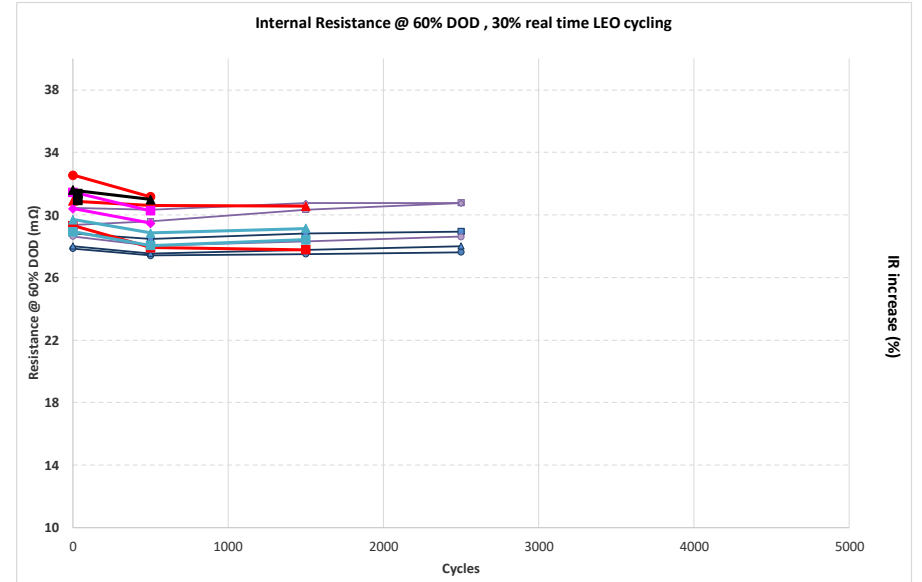
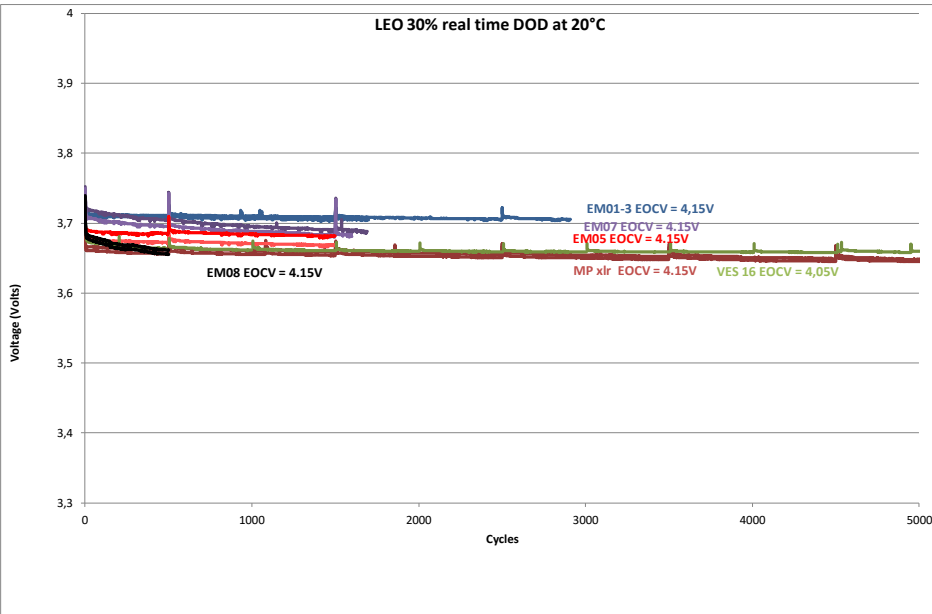
EM0 : LEO 30% DOD cycling trend (C/3, D/2)



- Similar fading ranking in LEO real time (3000 cycles) versus accelerated cycling tests performed at 80-100% DOD
- Similar trend as per VES16



EM0 : LEO 30% DOD cycling trend



- EODV stabilization in 2000 cycles as per VES16
- Good trend in energy and internal resistance in LEO 30% at 3000 cycles

A satellite view of Earth at night, showing city lights and the curvature of the planet. A large, semi-transparent blue trapezoidal shape is overlaid on the right side of the image, containing the title text.

6. EMO PRELIMINARY SAFETY TESTS RESULTS

Cell safety status - (VES16 size- EM07)

Pin test @ 4.2V

6 cells tested

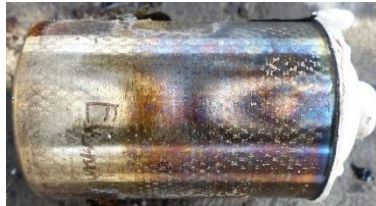
No rupture, no explosion, no flying part



Over-heating test @ 4.2V

6 cells tested

No rupture, no explosion, no flying part



- **Robustness of stainless-steel casing / Symmetric opening of the 2 vent devices / cell depressurization**

A background image of Earth from space, showing the horizon and clouds. A large blue trapezoidal shape is overlaid on the right side of the image, containing the section title.

7. CONCLUSIONS & PERSPECTIVES

CONCLUSIONS

- 12 months R&D works and EM0 prototyping-testing has permitted to select three VL10ES-EM1 series within disruptive electrochemical concepts for High Energy / Long Life & Safety trade-off.
- Innovating formulation & design concepts are associated to dense electrodes industrial process and a new cell assembly line is under qualification for VL10ES production.
- EM0 testing has shown very promising results in terms of capacity loss, stability of internal resistance and safety. The VL10ES cell will deliver the lifetime and power requirements for either GEO or LEO applications (including constellations).
- From preliminary results, the VL10ES mission usable specific energy will be 60-70 Wh/Kg in LEO and 150-155 Wh/Kg in GEO
- VL10ES will allow to reduce battery weight and cost, still maintaining long mission time. VL10ES has been designed to be used at higher DOD than VL18650 cells both in LEO and GEO cycling.
- The BoL Qualification Review of the VL10ES cell is targeted for mid-2021.

Acknowledgement

The authors would like to thank ESA and CNES for their support in VL10ES development.