

ST-R3-09 Multi-resolution Scanning – ISS / Astrobee



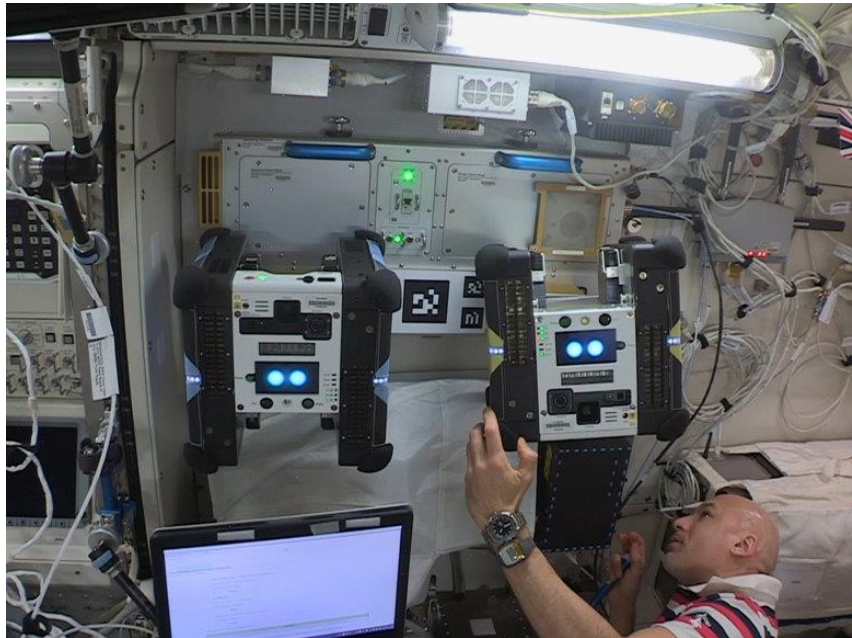
Astrobee Working Group
November 2023 Update

Summary

- This slide deck presents details on the CSIRO Space Technology Future Science Platform (FSP) Round 3/4 project : **ST-R3-09 Multi-resolution Scanning – ISS / Astrobee**
- The goal is to develop a sensor payload for the Astrobee robotic platform onboard the International Space Station (ISS) to support multi-resolution 3D scanning
- Future applications of the technology include extra-vehicular scanning and off-world scanning

Outline

- Introductions and background
- Technical progress
- Next steps



Introductions and background

Project Team

- **CSIRO:**

- Marc Elmoultie - (Project leader) stereo-depth fusion algorithm
- Ross Dungavell – lidar SLAM implementation
- Paul Flick – payload design
- Peter Dean – stereo-depth fusion implementation
- Tea Molnar – software engineer ROS integration
- Lauren Hanson – mechanical engineering payload
- David Haddon – systems integration
- Matt van de Werken – electronics engineer
- Michael Lofgren – software engineering advisor
- Rosie Attwell – flightplan scheduling
- Anna Campbell - logistics



- **Boeing :**

- Connie Miller – (Payload Developer) Space & Launch Focal
- Leighton Carr – Associate Technical Fellow

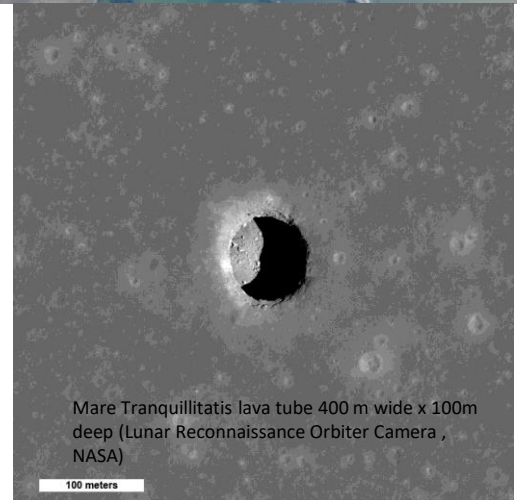


Motivation

- Collaboration with Boeing to develop a fused sensor system to support space based autonomous multi-resolution scanning from multiple platforms and payloads, including:
- Interior ISS/Gateway using autonomous scanning, supported by Astrobees & sensor payload
- External hull integrity scanning using systems such as CanadARM2 and Seeker CubeSAT
- Off-world surface scanning on lunar or mars rovers or potentially multi-legged robots

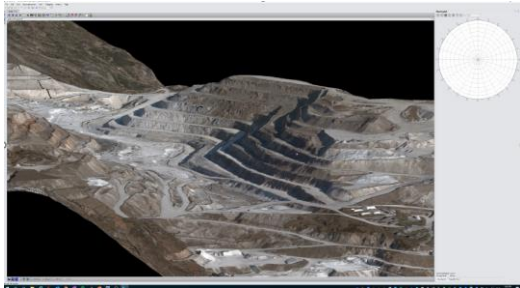


STS114 image of ISS (NASA)



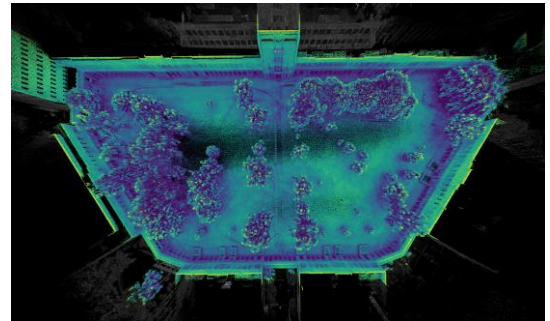
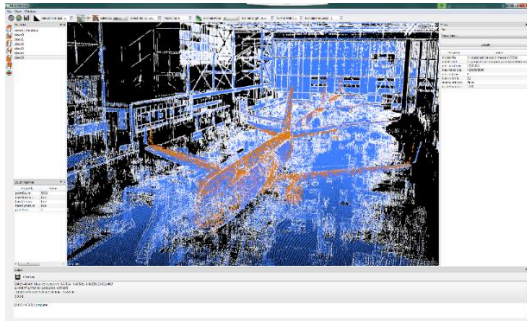
Mare Tranquillitatis lava tube 400 m wide x 100m deep (Lunar Reconnaissance Orbiter Camera, NASA)

Multi resolution scanning – key technologies



CSIRO Stereo-Depth Fusion (SDF) – Hi Resolution stereo-depth fusion scanning

+



CSIRO WildCat SLAM (Simultaneous Localisation and Mapping) – large area positioning and mapping

Target performance metrics – ISS internal scanning

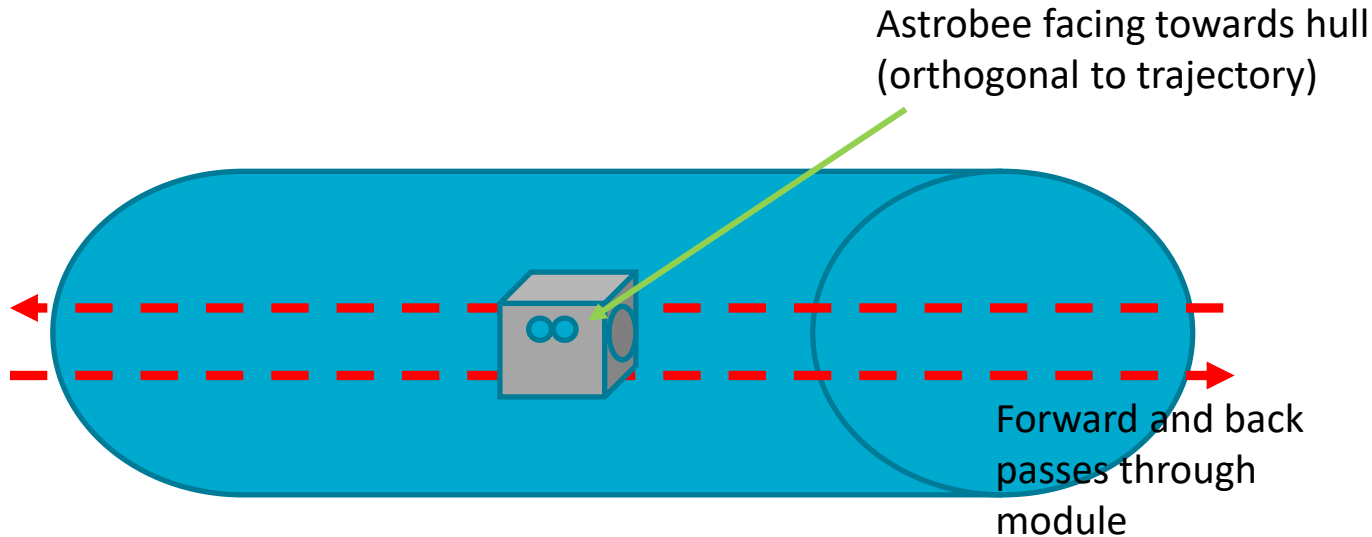
- Scan speeds up to 0.3 m/s
- Scanning range up to 2.5m
- Point cloud resolutions down to sub-millimeter to mm



ISS Japanese Experiment Module (NASA)

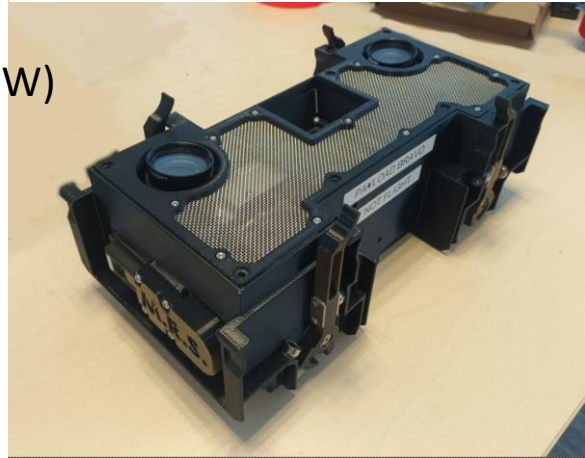
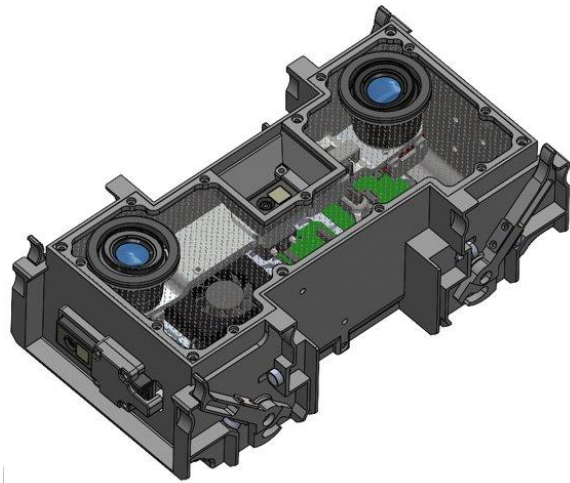
Proposed experiments on ISS

- Full internal scan, initial testing in Japanese experimental module
- Scanning of other modules to follow
- Trajectory to be finalised with NASA

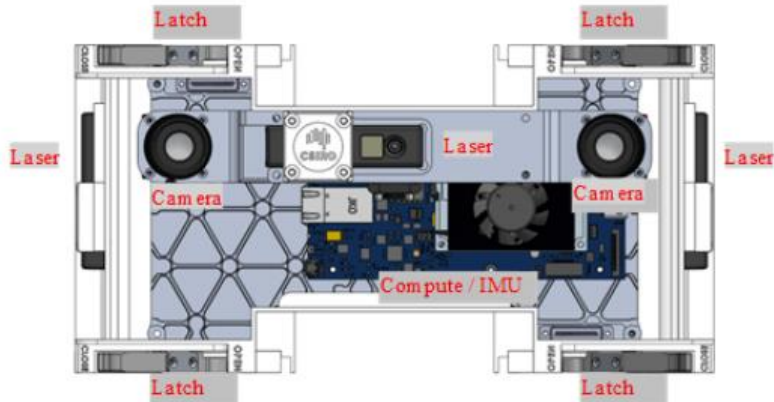
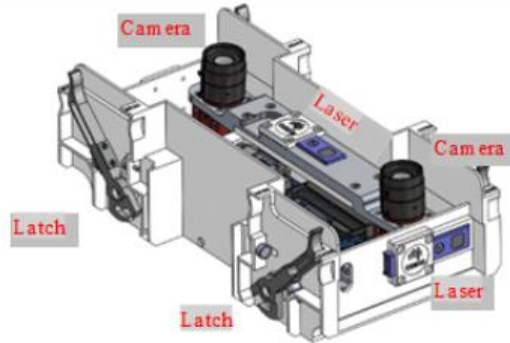


Specifications

- Housing
 - ULTEM ~500g
- Compute
 - Xavier NX
 - Mass 100gms / Power 15W +
- Stereo camera
 - Machine vision or similar units
 - Mass ~400gms / Power consumption (~5W)
- Additional sensors
 - Picoflexx (x 3) & IMU
 - Mass 30 gms / Power consumption ~2W
- Total mass: ~1500gms
- Note: Astrobee payload nominals
 - power supply (14V/2A) / ~1900gms



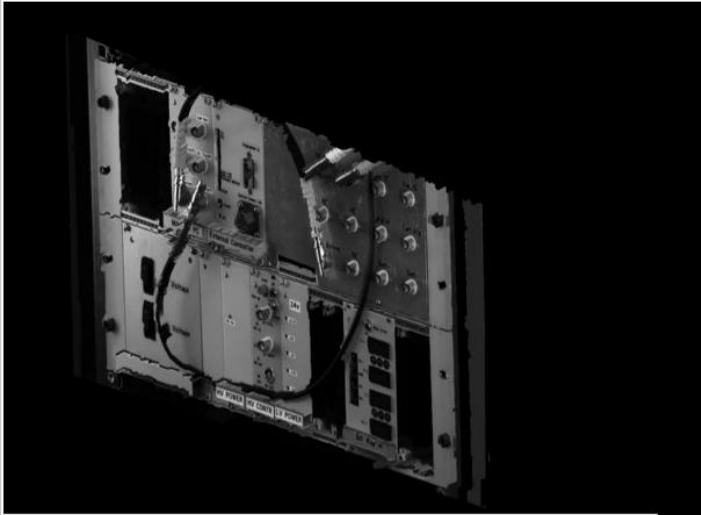
Sensor Layout



Payload testing

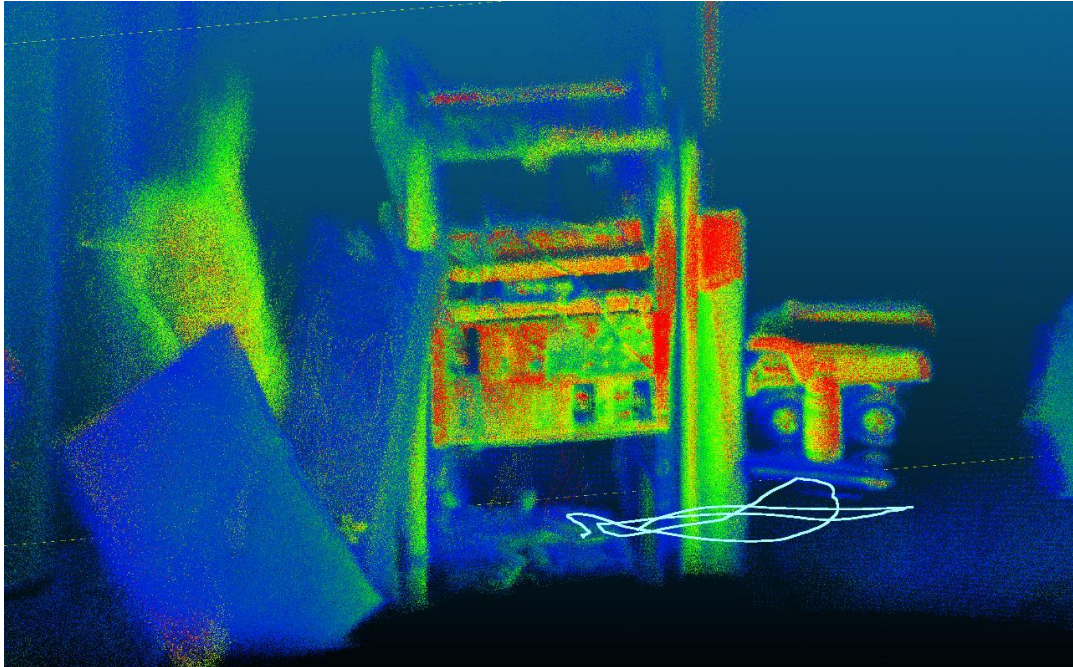
Testing & Validation

- High resolution 3D reconstructions from payload



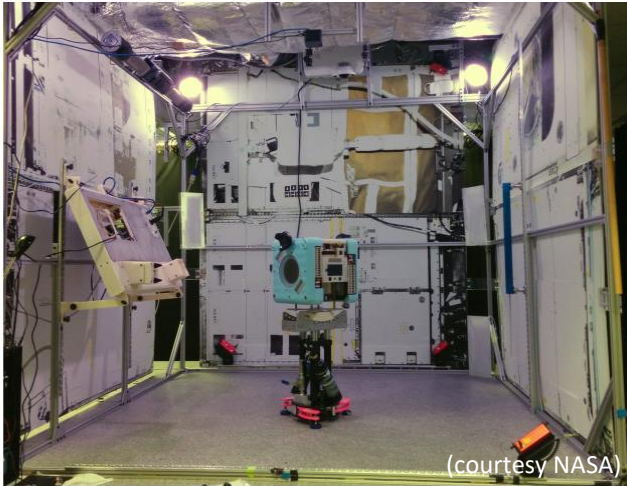
Testing & Validation

- SLAM trajectory and Point Cloud



NASA AMES Astrobee Test Facilities

- Payload to undergo full suite of ground-based testing

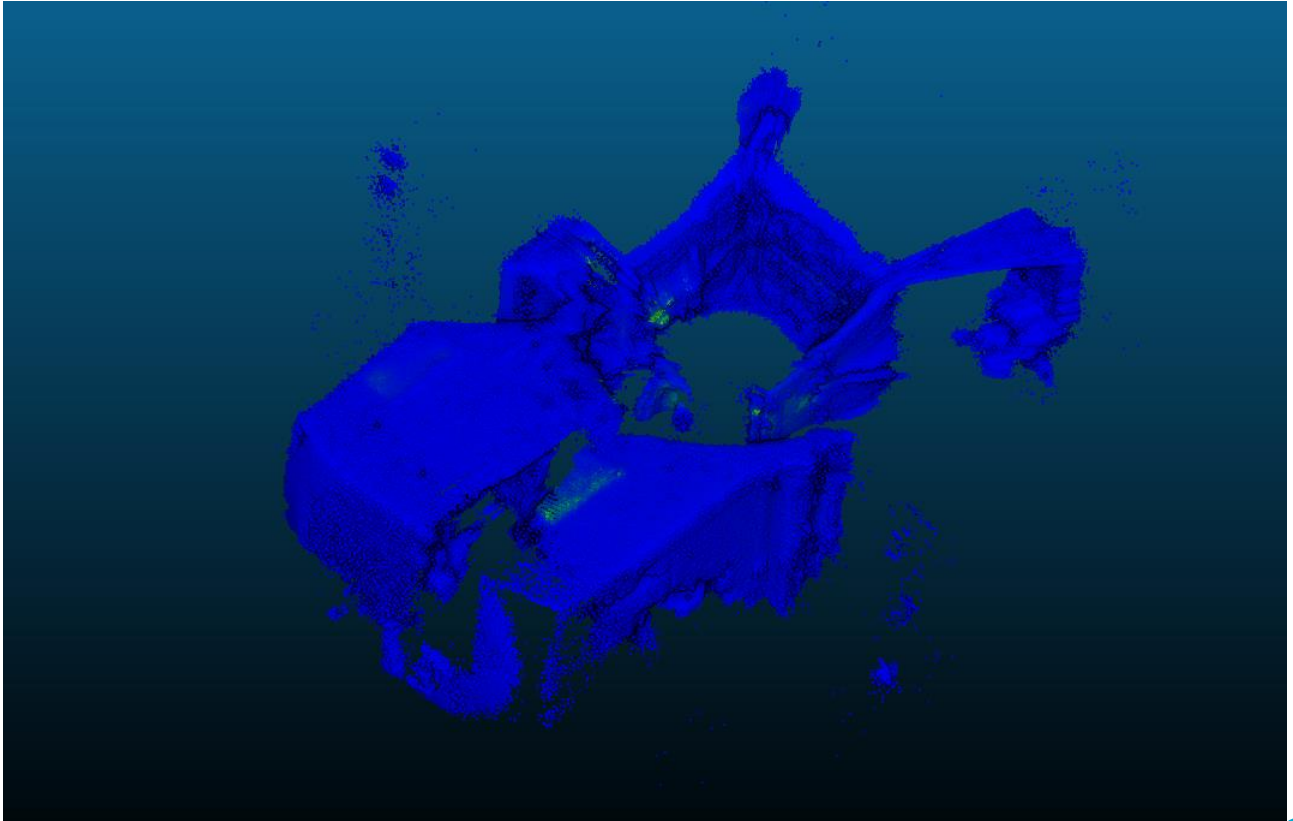


Granite Lab.

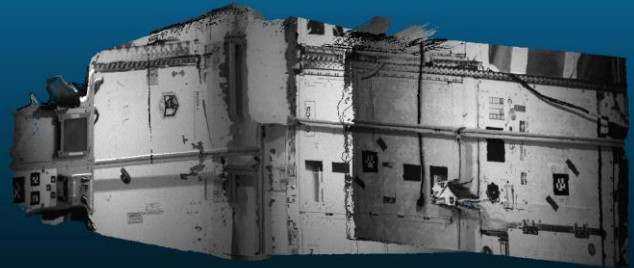
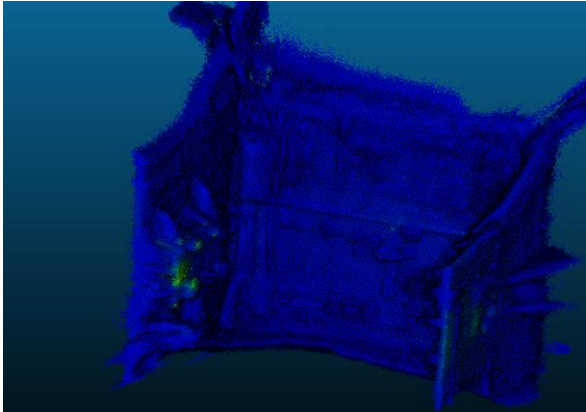


Micro-Gravity Test Facility

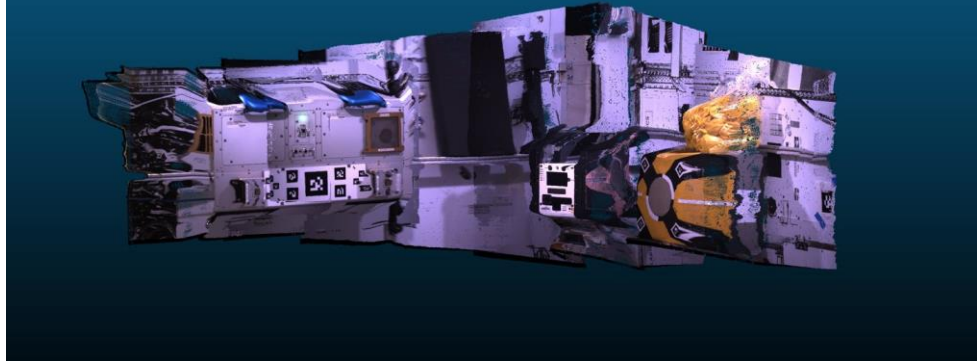
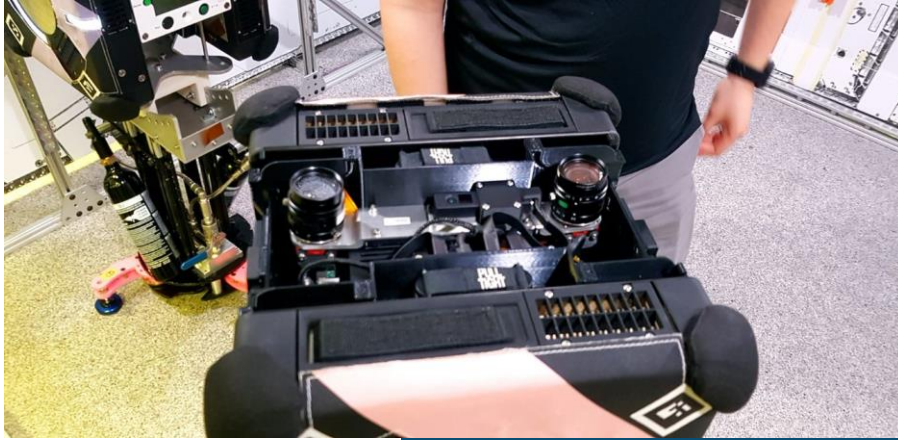
NASA AMES Granite Lab Testing AUG 2022



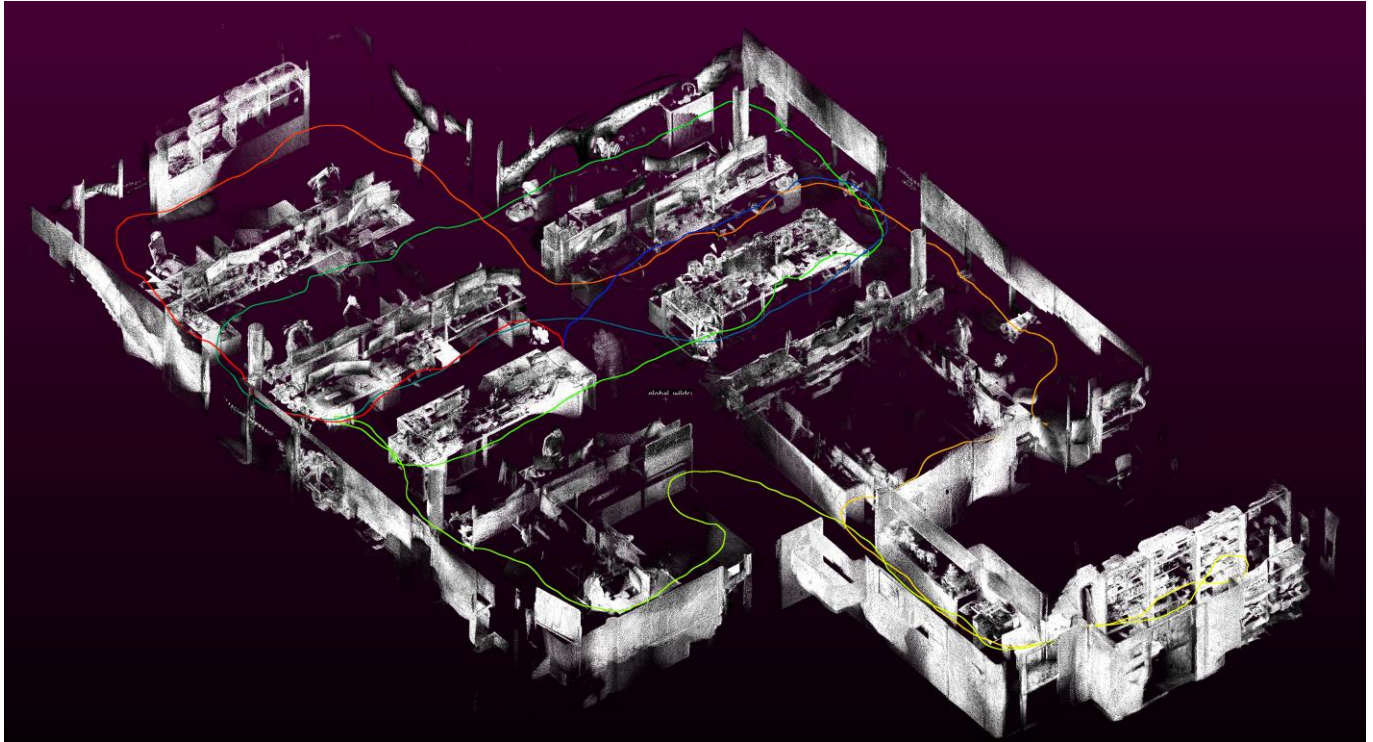
NASA AMES Granite Lab Testing AUG 2022



NASA AMES Granite Lab Testing MAR 2023



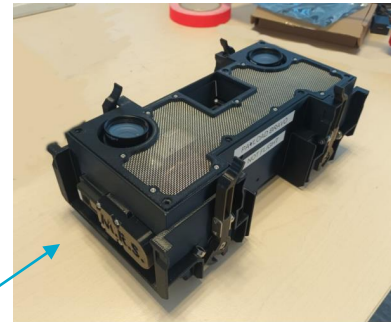
Testing & Validation



Payload integration process

Safety Assessment Process

- Phase 0 – identification of applicable safety requirements, known hazards and causes
- Phase 1 – identify hazards with planned and/or implemented controls
- Phase 2 – hazard controls implemented and incorporated into design and verifications document
- Phase 3 – final safety review and acceptance

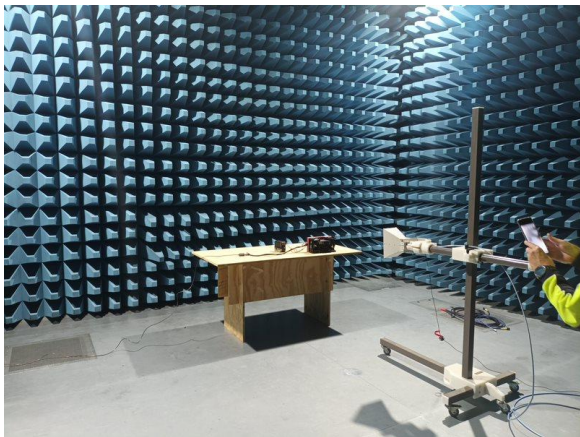
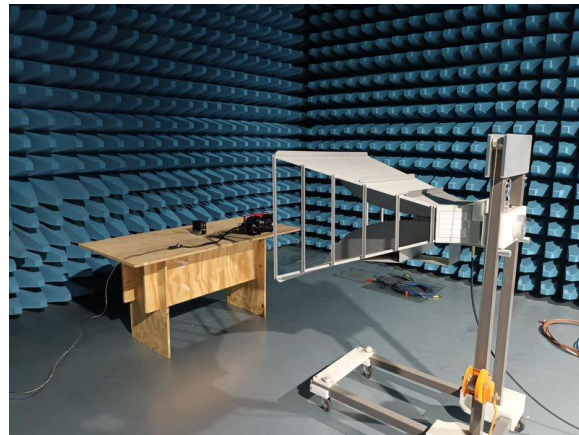


Mesh – design update

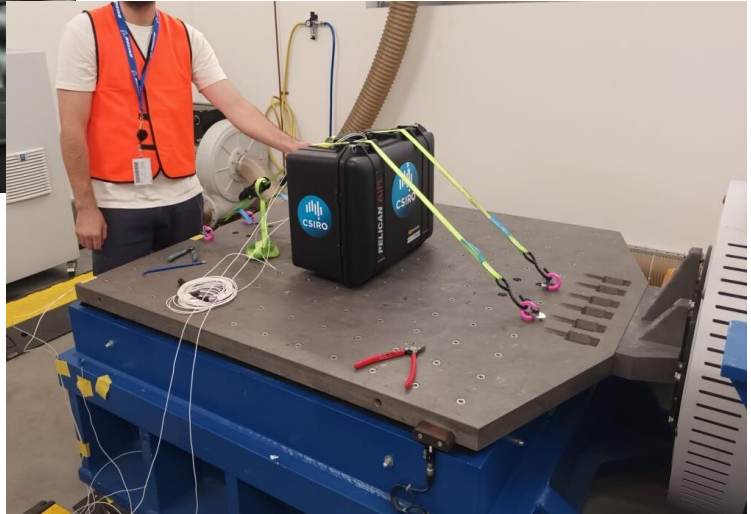
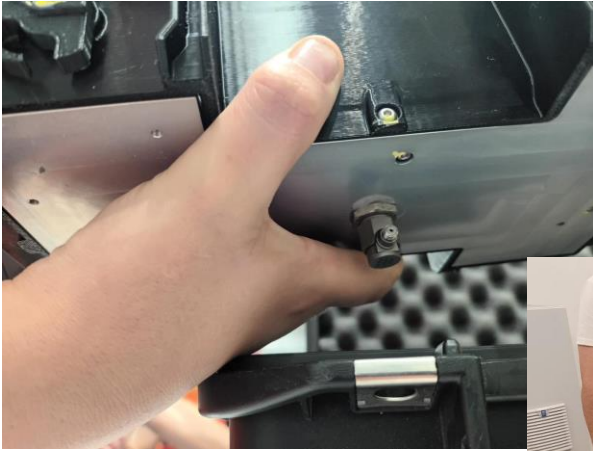
Thermal Testing Phase 2



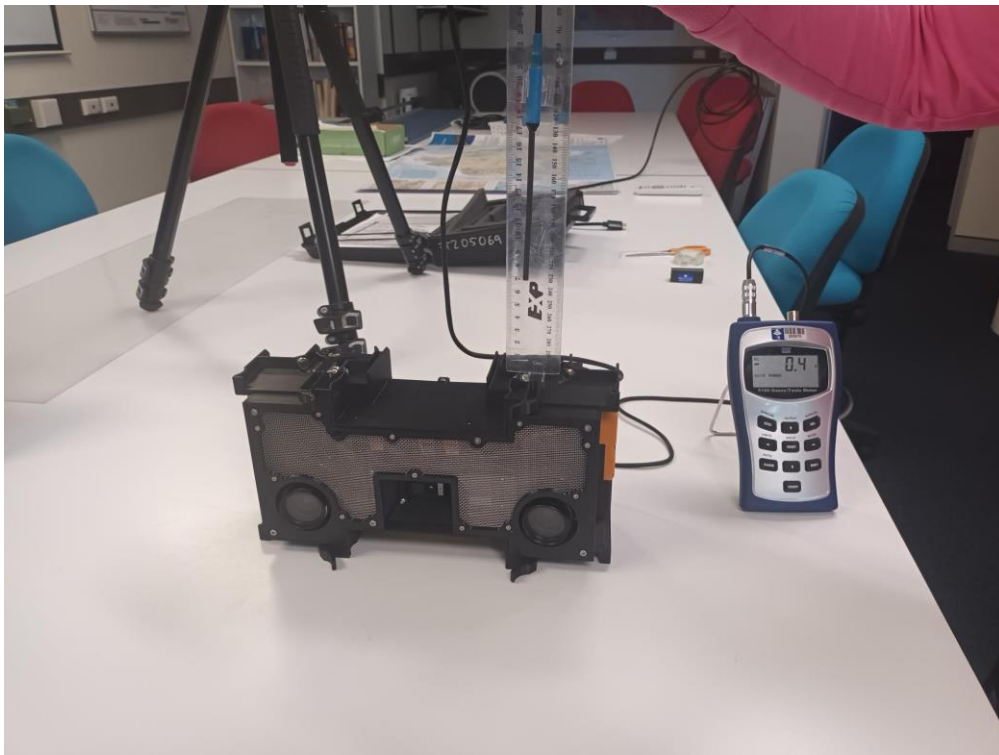
EMI Testing Phase 2



Vibration Testing Phase 2

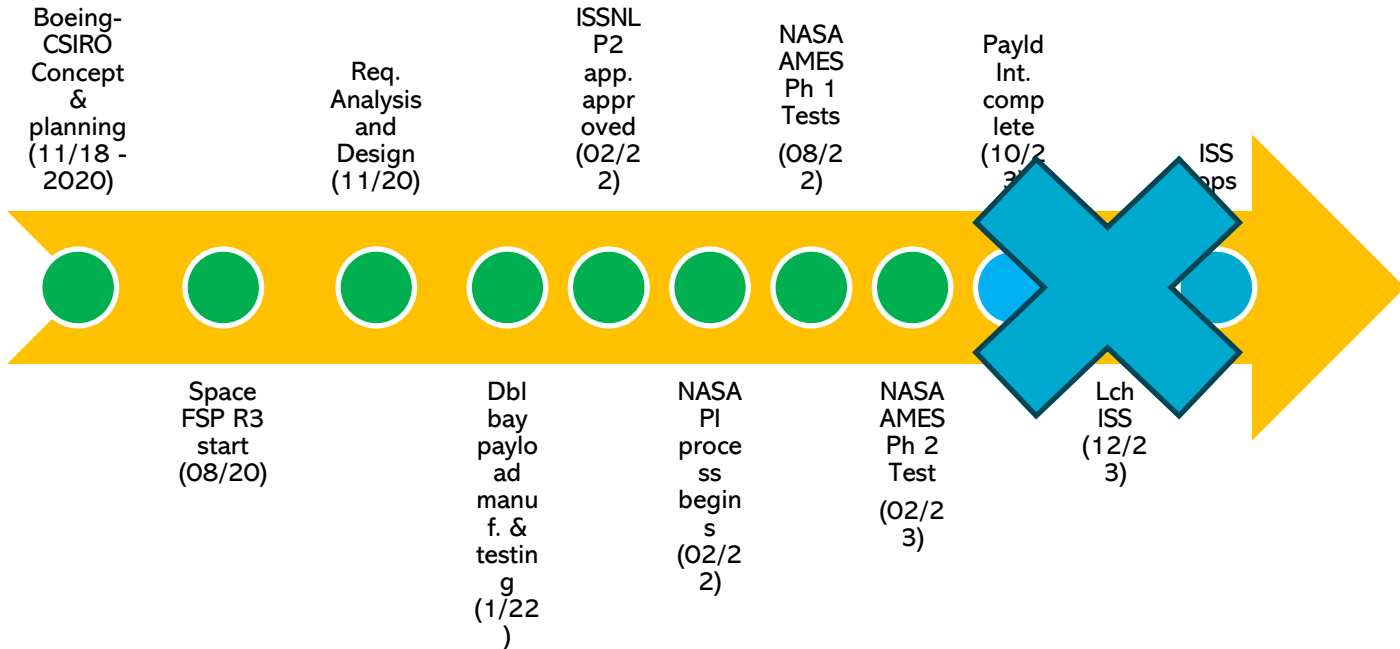


Static magnetics



Challenges

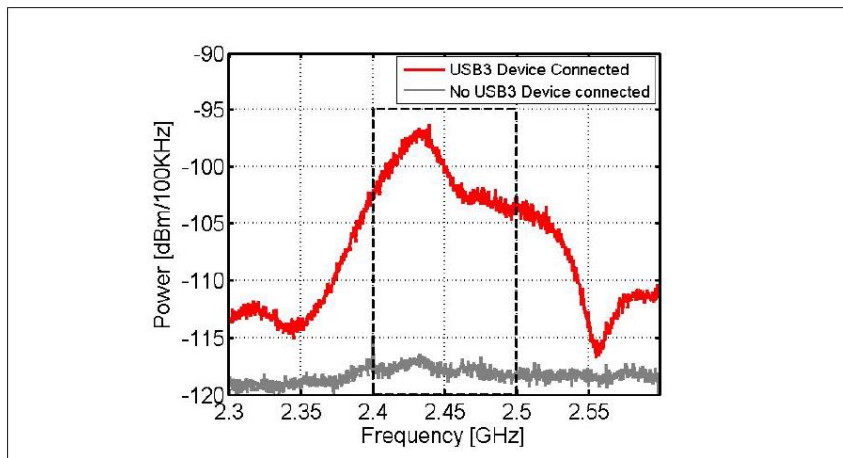
Timeline



Challenges

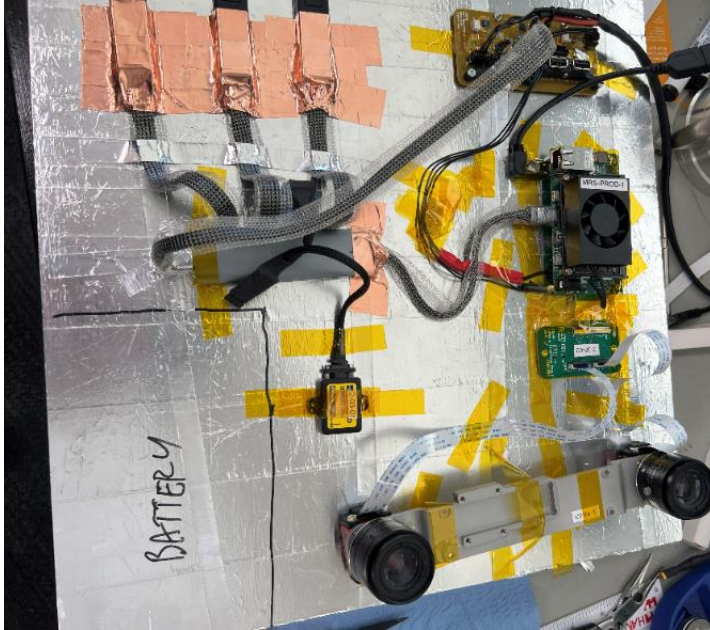
- Manufacturer specs aren't always accurate! Average vs Peak power.
- USB3 is noisy!

Figure 3-3. Noise from External USB 3.0* Hard Disk Drive



USB 3.0* Radio Frequency Interference Impact on 2.4 GHz Wireless Devices
Intel White Paper Document: 327216-001

Challenges

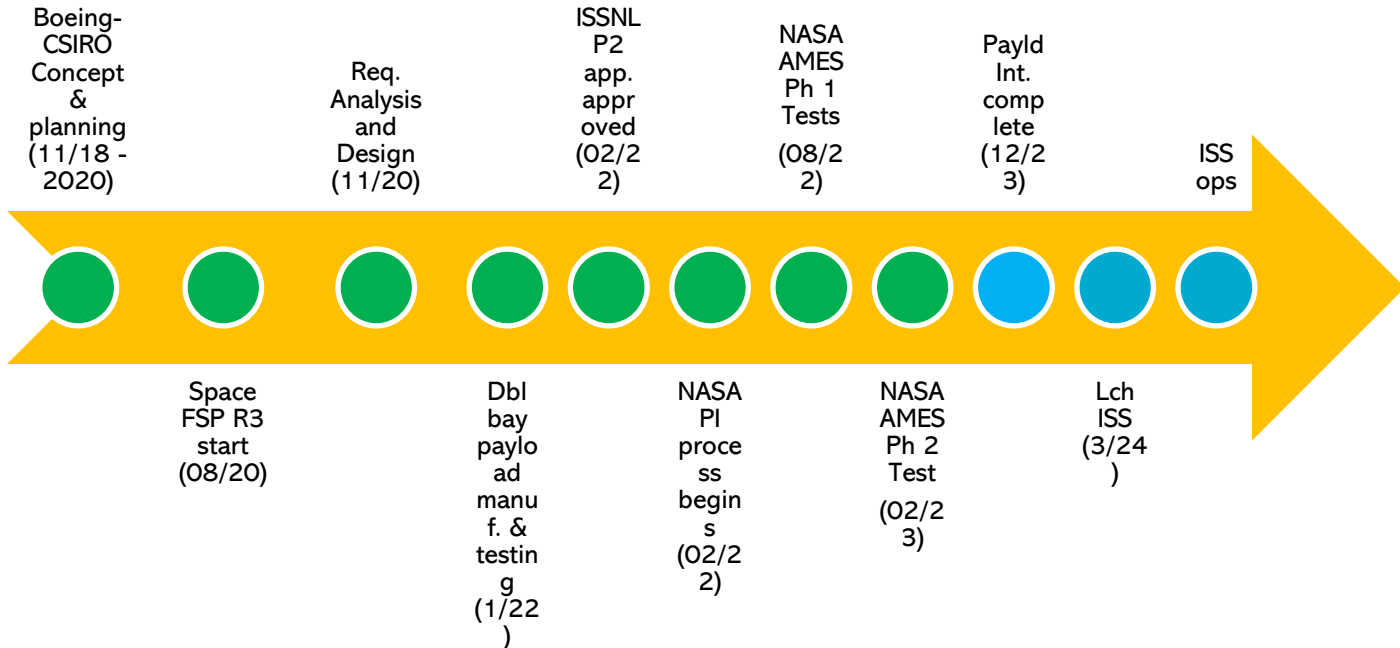


Next steps

Next steps

- Phase 3 – final safety review and acceptance
- AMES testing & HFIT
- CONOPS finalisation

Timeline



Acknowledgements

- Special thanks to the entire NASA team including Jose Benevides, Jose Cortez, Cristian Garcia, Ruben Garcia, Jonathan Barlow, Simeon Kanis, Aric Katterhagen, Gerald Readore, Henry Orosco, Antonius Widjokongko, Tameka Stewart, Katrina Whitlock, Melissa Boyer, Larry Kim, Alla Shuhatovich
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Thank you



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