

The RAVAN CubeSat Demonstration of Earth Energy Budget Technologies: Challenges and Successes

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- L-1 Standards and Technology: James Briscoe, Steven Lorentz (also NISTAR), Allan Smith, Yinan Yu
- NASA/GSFC: Warren Wiscombe, Dong Wu
- Blue Canyon Technologies: John Carvo, Tom Golden, and others

As of Oct 2017



Outline

- Brief overview
 - Top-level summary
 - Key papers
 - Notable things about RAVAN
- Science motivation
- Technologies demonstrated
- (Selected) trials and tribulations
- RAVAN on-orbit results
- Lessons learned



RAVAN is an Earth energy budget constellation pathfinder

- RAVAN: <u>Radiometer Assessment using Vertically Aligned Nanotubes</u>
- CubeSat project funded through NASA ESTO's InVEST program (InVEST-2012)
- Principally a technology demonstration
- CubeSat = High-risk
- Led by Johns Hopkins University Applied Physics Laboratory (APL), Laurel, Maryland, USA
- Partners:
 - L-1 Standards and Technology (L-1): Steven Lorentz (NISTAR PI)
 - NASA/GSFC: Warren Wiscombe, Dong Wu
 - Blue Canyon Technologies (BCT)
- Pathfinder for an Earth energy (radiation) budget constellation
- Combines
 - Vertically aligned carbon nanotube radiometer absorber and black body emitter (APL)
 - Gallium fixed-point black body calibration source (L-1)
 - Compact, low-cost radiometer payload (L-1/APL)
 - 3U CubeSat bus, I&T, operations (BCT)
- Operated Nov 2016–Aug 2018







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Vertically aligned carbon nanotubes (VACNTs) are super black and compact—perfect for smallsat applications



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RAVAN proved VACNTs for Earth energy budget measurement from space



RAVAN 3U CubeSat Nov 2016–Aug 2018 demo (and still flying!)

APL,



Principal RAVAN references



RAVAN notables

- Funded as a part of the inaugural opportunity from NASA's Earth Science Technology Office (ESTO)'s In-Space Validation of Earth Science Technologies (InVEST) program, in 2012
- First ESTO InVEST CubeSat to fly
- First JHU/APL "science" CubeSat
- First Blue Canyon Technologies bus to fly
- RAVAN technology demonstration was a success



A small energy imbalance drives climate change



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Outgoing energy (radiation) highly variable, geographically and temporally



Current space-based assets cannot quantify Earth's outgoing radiation well enough to resolve the Earth energy imbalance from space (~1% accuracy...0.1% needed).

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We have been measuring EEB from space for a long time

| TARLE 1 | Selected Satellites a | d Missions Making Sig | nificant Contributions to | Earth Radiation Budget Science |
|-----------|-----------------------|---------------------------|---------------------------|--------------------------------|
| L'ADLU L. | ACIEVICU MALCINES A | IC IVESSIONS IVERNIES OFF | | |

| Satellite Missions | Launch Date(s) | Altitude Range, km | Inclination Angle, deg | Orbit Time | Lifetime(s) | Contributions |
|--------------------------------|-----------------------------|--------------------------|------------------------------|---------------|-------------|---|
| | | F | irst-Generation | Missions | | |
| Explorer 7 | Oct. 13, 1959 | 550-1,100 | 51 | drifter | 7 months | first dedicated satellite providing usable ERB data |
| TIROS 2 | Nov. 23, 1960 | 717-837 | 48 | drifter | 1-5 months | first scanning radiometer with five SW/LW channels |
| TIROS 7 | June 19, 1963 | 713–743 | 58 | drifter | 12 months | provided 1 year of radiation balance observations |
| | | Sec | cond-Generation | Missions | | |
| Research/ESSA | 1960s | ≈1,500 | 102 | 0900/1500 | 3–15 months | global data sets from WFOV nonscanning radiometers |
| Nimbus 3 | April 14, 1969 | 1,100 | 99 | noon | 1 year | detailed global radiation balance for 1 year |
| NOAA | 1970s | ≈1,500 | 102 | 0900 | years | combined data sets provided |
| TIROS-N/NOAA | 1978-1981 | ≈840 | 99 | 1500/0730 | years | 10 years of observations |
| | | T | hird-Generation | Mission | | |
| Nimbus 7 ERB | Oct. 1978 to the present | 950 | 99 | noon | 6+ years | total and spectral solar monitoring; bidirectional reflectance and directional albedo models |
| | | (| Geostationary M | issions | | |
| GOES-E/W (75°-135°W) | 1970s/1980s | 36,000 | 0 | 24 hours | years | diurnal variations of SW/LW exitances and cloud distri- |
| METEOSAT 1/2 (0° longitude) | 1977/1982 | 36,000 | 0 | 24 hours | years | butions; satellite Mission simulations |



Fig. 2. First dedicated ERB satellite, Vanguard 2, mounted on the rocket prior to launch in 1959.

House et al. [1986]



RAVAN is a LEO non-scanning (WFOV) radiometer

Table 3. Non-Scanning Broadband Earth Radiation Budget space instruments. IKOR: Short-wave outgoing radiation monitor; ERM: Earth Radiation Monitor; RAVAN: Radiometer Assessment using Vertically Aligned Nanotubes.

| Period | Instrument | Reference | ces |
|--------------|--------------------------------|-----------|------------------|
| 1975-1978 | ERB on Nimbus 6 | [25] | |
| 1978-1987 | ERB on Nimbus 7 | [25] | |
| 1984-1999 | ERBE on ERBS | [26] | |
| 1985-1990 | ERBE on NOAA 9 | [26] | |
| 1986-1994 | ERBE on NOAA 10 | [26] | |
| 1994 | IKOR (SW only) on Meteor-37 | [27] | LEO non-scanning |
| 1998 | IKOR (SW only) on Resurs-1 | [27] | |
| 2008-2011 | ERM NS on FY3A | [28] | (WFOV) |
| 2009-2014 | IKOR-M (SW only) on Meteor-M 1 | [29] | |
| 2011 | ERM NS on FY3B | [28] | |
| 2013-present | ERM NS on FY3C | [28] | |
| 2014-present | IKOR-M (SW only) on Meteor-M 2 | [30] | |
| 2016 | RAVAN | [31] | |

Table 4. Scanning Broadband Earth Radiation Budget space instruments on Low Earth Orbit satellites. CERES: Clouds and the Earth's Radiant Energy System; ScaRaB: Scanning Radiometer for Radiation Balance; TRMM: Tropical Rainfall Measuring Mission; NPP: National Polar-orbiting Operational Environmental Satellite System Preparatory Project.

| Period | Instrument | Referen | nces |
|------------------|-----------------------------|---------|--------------|
| July-August 1975 | ERB on Nimbus 6 | [25] | |
| 1978-1980 | ERB on Nimbus 7 | [25] | |
| 1984-1989 | ERBE on ERBS | [26] | |
| 1985–1987 | ERBE on NOAA 9 | [26] | |
| 1986-1989 | ERBE on NOAA 10 | [26] | |
| 1994-1995 | ScaRaB-1 on Meteor-3 7 | [32] | |
| 1997-1998 | CERES on TRMM | [33] | |
| 1998-1999 | ScaRaB-2 on Resurs-1 | [34] | LEO scanning |
| 2000-present | CERES FM1 on Terra | [33] | |
| 2000-present | CERES FM2 on Terra | [33] | |
| 2003-present | CERES FM3 on Aqua | [33] | |
| 2003-present | CERES FM4 on Aqua | [33] | |
| 2008-2010 | ERM on FY3A | [28] | |
| 2011–present | ScaRaB-3 on Megha-Tropiques | [35] | |
| 2011-present | CERES FM5 on Suomi NPP | [33] | |
| 2011 | ERM on FY3B | [28] | |
| 2013-present | ERM on FY3C | [28] | |
| | | | |

Table 5. Broadband Earth Radiation Budget space instruments on geostationary satellites. GERB:Geostationary Earth Radiation Budget; MSG: Meteosat Second Generation.

| Period | Instrument | References | |
|--------------|---------------|------------|-----|
| 2003-present | GERB2 on MSG1 | [39,41] | |
| 2007-2012 | GERB1 on MSG2 | [39,41] | 000 |
| 2012-present | GERB3 on MSG3 | [39,41] | GEU |
| 2015 | GERB4 on MSG4 | [39,41] | |

Dewitte et al. [2017]



Energy budget measurement requires an extensive oceanobserving system like Argo



Motivation for RAVAN: "Argo in space"

- The small imbalance (~1 W/m²) between incoming solar irradiance and Earth outgoing energy (solar reflected + Earth's thermal emission) drives climate change
- Current space-based assets cannot quantify Earth's outgoing radiation well enough to resolve the Earth energy imbalance from space (~1% accuracy...0.1% needed)
- RAVAN is an Earth radiation (energy) budget constellation pathfinder







TOE = Total Outgoing Energy



TSI = Total Solar Irradiance



RAVAN is part of a progression of effort toward a new Earth energy budget measurement



• 2011/APL: Earth Radiation Imbalance System (ERIS)

- Proposed to NASA Earth Venture program (not selected)
- Fly ~69 radiometer payloads on Iridium NEXT constellation
- Science: "The accurate ERIS measurements of TOR and ERI at high spatial and temporal resolution are highly relevant to NASA Earth science activities and science in general. There is no way that NASA, by itself, could reproduce these measurements for anything close to the proposed cost." (from Earth Venture debrief)
- Weaknesses:
 - Calibration not demonstrated (including Ga blackbody source)
 - Payload cost (thought to be too low, but there's no precedent)

2012/APL: Radiometer Assessment using Vertically Aligned Nanotubes (RAVAN)

- Proposed to NASA ESTO/InVEST (selected)
- Fly single radiometer payload on a CubeSat: Technology demonstration
- Pathfinder for an Earth radiation imbalance mission
- Directly addresses ERIS weaknesses



(Vertically aligned) carbon nanotubes are at the heart of RAVAN

RAVAN carbon nanotubes developed at APL with previous internal/external investments





VACNTs further developed and tested under RAVAN



Gallium phase-change black bodies also demonstrated

• Demonstrate the use of a gallium closed-cell source for stability monitoring



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Compact (1U) payload hosts radiometers, Ga black bodies



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Earth, solar, and cold space views





Near-death experiences (selected) ["Just wait. It will get worse."—Phil Huang

• Proposed APL CubeSat bus not really an option



Bus was a moving target



Quoting Y1 Annual Review (mid-2014):

- State-of-the-art advances since MBD (ORS Tech) design/development
- Radios need updating (frequency allocation)
- Attitude control (pointing limitations)



Preliminary BCT concept



Payload fits, with room to spare



APL

Near-death experiences (selected) ["Just wait. It will get worse."—Phil Huang

- Proposed APL CubeSat bus not really an option
- Choice: Gamble on early, uncertain launch (with "success-oriented" schedule) vs. wait two years for CSLI
- Payload delivery delay and late-in-the-game bus problems use every bit of launch integrator schedule margin



RAVAN at Blue Canyon

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RAVAN flight spacecraft





RAVAN in perspective



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RAVAN, ready for launch vehicle integration



(b) Credit: Tyvak/Cal Poly



Near-death experiences (selected) ["Just wait. It will get worse."—Phil Huang

- Proposed APL CubeSat bus not really an option
- Choice: Gamble on early, uncertain launch (with "success-oriented" schedule) vs. wait two years for CSLI
- Payload delivery delay and late-in-the-game bus problems use every bit of launch integrator schedule margin
- Obtained FCC license one day before dis-integration from launch vehicle

Atlas V rocket, before original launch day (9/15/16)



Source: Spaceflight Now

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Launch scrub #2 (wildfires): Sun 9/18/16



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Launched Nov 11, 2016



RAVAN 3U CubeSat Blue Canyon Technologies bus

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Launch 11/11/16

Near-death experiences (selected) ["Just wait. It will get worse."—Phil Huang

- Proposed APL CubeSat bus not really an option
- Choice: Gamble on early, uncertain launch (with "success-oriented" schedule) vs. wait two years for CSLI
- Payload delivery delay and late-in-the-game bus problems use every bit of launch integrator schedule margin
- Obtained FCC license one day before dis-integration from launch vehicle
- Communication problems
 - None for first 11 days
 - Winter weather at ground station (Boulder, CO)
 - Ground interference



"First Light": The VACNT and cavity radiometers track very well



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Near-death experiences (selected) ["Just wait. It will get worse."—Phil Huang

- Proposed APL CubeSat bus not really an option
- Choice: Gamble on early, uncertain launch (with "success-oriented" schedule) vs. wait two years for CSLI
- Payload delivery delay and late-in-the-game bus problems use every bit of launch integrator schedule margin
- Obtained FCC license one day before dis-integration from launch vehicle
- Communication problems
 - None for first 11 days
 - Winter weather at ground station (Boulder, CO)
 - Ground interference
- Bus SD card failure



Payload data not continuous but provided what we needed for tech demo



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Gallium melt provides repeatable reference



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Gallium melt temperature consistent

Gallium melt observed by radiometer



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BB: Instrument long-term stability, but short-term fluctuations





Solar: Instrument long-term stability, but short-term fluctuations



Solar (eclipse) observations



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Earth-viewing dataset is episodic (not by design!)



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Nadir observations of outgoing flux well correlated: Old tech vs. new tech





VACNT and cavity radiometers well correlated, but with absolute differences of 3 and 6%





| | | | CERES SW | | VVI. | |
|--------------------------------|-------------------------------|--|---|---|--|-----|
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| RAVAN Lo | ngwave (Total-SV | | ┿ <mark>间加空格10-year mean(</mark> | ([0-9])×([0-9]) CERES EBAF Flux, | 1×12 Dewitte et al. [20 | 17] |
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"Single-pixel camera" contains spatial information



Single day (June 27, 2017) of RAVAN LW flux (Total – SW)



...however, wide FOV (130°)



Single day (June 27, 2017) of RAVAN LW flux (Total – SW)



Spatial reconstruction from a single day of data



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More satellites provide greater spatial (and temporal) resolution and less error







APL



Improvements for a future mission

- Climate-level thermal knowledge and control are challenging in a CubeSat
- Significant improvements in future RAVAN-type sensors on a larger small satellite (more volume and power)
 - Better active and passive temperature control of the electronics and the structures around the radiometers
 - <u>Better radiometer baffles</u>, with more space afforded by an even modestly larger small satellite bus and/or fewer radiometers in close proximity, these baffles could be made deeper, which would help to minimize the glint associated with passing out of eclipse. Radiometers built with a narrower field of view, such as CERES and scanner radiometers in general, would allow for smaller baffles and better thermal control.
 - <u>Better gallium black bodies</u>, with more space along the optical axis allowing the doors to accommodate a cavity emitter and a longer melting transition.

More frequent calibration

- UHF interference often limited how often we could make observations, but capturing and understanding shorter-term changes would enable us to account for more payload variability.

Extensive pre-launch calibration

- Such calibration impossible for RAVAN given compressed launch schedule



RAVAN technology incorporated into subsequent projects



VACNT low-TRL development (2003–2012) APL IRAD; NASA









VACNT radiometer/ BCT 6U bus LASP CTIM (2022) NASA ESTO InVEST VACNT BB emitter; Ga BB LaRC/APL Trutinor next-gen "CERES" LaRC IRAD (2018–2019)

VACNT radiometer array LASP Black Array of Broadband Absolute Radiometers for Imaging Earth Radiation (BABAR) NASA ESTO IIP-2019

VACNT radiometer LASP Libera (2027) NASA EVC-1

The CSIM CubeSat

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NASA (APL) EZIE mission leverages RAVAN's bus heritage



- Electrojet Zeeman Imaging Explorer (EZIE)
- Measurement of Electrojet Temporal Evolution:
 - 3 6U CubeSat flying in a pearls-on-a-string formation with varying separation managed by differential drag.
- Measurement of Electrojet spatial structure:
- A compact payload consisting of four identical O₂ 118-GHz spectropolarimeters to remotely measure and image electrojet induced magnetic fields
- Deployment orbit:
 - Circular, 525- to 625-km altitude
 - Near Sun-Sync, 09:00–11:00 or 22:30–00:30 LTAN
 - Launch Date: Late 2024 or early 2025

Summary

- RAVAN (InVEST-2012) launched Nov 11, 2016 for 20-month mission
 - Four radiometers worked well
 - One (of two) gallium black bodies failed; the second performed throughout mission
- Primary conclusions
 - NASA ESTO technology demonstration success
 - Earth radiation budget science measurements harder problem
- The good
 - Carbon nanotubes ("VACNTs") work in space, specifically as radiometer absorbers
 - Gallium phase-change black bodies for calibration monitoring
 - Long-term stability demonstrated -
 - Qualitative (at least) agreement with analysis and CERES
 - Reconstruction of spatial information wFOV "single pixels"
- The "less good"
 - Short-term fluctuations problematic (for 0.1% climate-level observations), most likely due to inadequate thermal knowledge and control
- RAVAN serves as a benchmark for future EEB science missions that use RAVAN technologies and/or smaller spacecraft

Swartz, W. H., et al. (2019), RAVAN: CubeSat demonstration for multi-point Earth radiation budget measurements, Remote Sens., 11, 796.



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