#### October 31, 2022 NASA Advisory Council (NAC)



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- Artemis Planning Manifest
- Artemis I
- Artemis II
- Artemis III
- Artemis IV

# **Artemis Planning Manifest**





Icons represent the calendar year in which the launch occurs. | Based on FY23 Presidents budget request. | Does not include impact from FY22 appropriations. | Selected Mars forward elements in SMD and STMD included for context.



# A R/TE M I S

# Artemis I Mission Status

# **MAJOR MILESTONES FOR ARTEMIS I**



**EXPLORATION** FLIGHT TEST-1



MOBILE LAUNCHER CONTRUCTION



CREW MODULE/SERVICE MODULE MATE



MISSION CONTROL CENTER TEAM TRAINING START



START OF BOOSTER STACKING ON MOBILE LAUNCHER



UPGRADES AT

PAD 39B

HEAT SHIELD INSTALL

ON CREW MODULE

ORION STAGE

ADAPTER DELIVERED

LAUNCH ABORT SYSTEM

HANDOVER TO EGS



SLS WIND TUNNEL TESTING



VAB HIGH BAY 3 CONSTRUCTION



CORE STAGE ASSEMBLY



INTERIM CRYOGENIC **PROPULSION STAGE DELIVERED** 



GREEN RUN HOT FIRE

PRESSURE VESSEL

COMPLETE

**MPPF 1ST HAZARDOUS** 

OPERATIONS

**ORION ENVIRONMENTAL** 

TESTING

GREEN RUN TESTING

BEGINS

CORE STAGE

INTEGRATION

CORE STAGE PRODUCTION BEGINS



EUROPEAN SERVICE MODULE SHIPPED TO KSC



SLS STRUCTURAL TESTING



LAUNCH VEHICLE STAGE ADAPTER DELIVERED TO KSC



LAUNCH VEHICLE STAGE ADAPTER/ INTERIM CRYOGENIC PROPULSION STAGE INTEGRATION



VESSEL TESTING

SLS BOOSTER

**QUAL TESTING** 

INTEGRATED TEST LAB

VERIFICATION TESTING

MOTOR SEGMENTS

DELIVERED TO KSC





C3 READY FOR

HAZARDOUS OPERATIONS

**RS-25 ENGINE** 

TESTING COMPLETE

ASCENT ABORT-2

FLIGHT TEST





SERVICE MODULE PROPULSION TESTING



**MOBILE LAUNCHER/VAB** MULTI-ELEMENT

LCC TEAM TRAINING START

**ROTATION PROCESSING &** 



**ORION STRUCTURAL** 

CORE STAGE INTEGRATION



UNDERWAY **RECOVERY TEST 8** 

#### See Next Chart for KSC Flow



**BOOSTER & AFT** SKIRT MATE

TESTING









**VERIFICATION & VALIDATION** 



SLS AVIONICS TESTING

MOBILE LAUNCHER/PAD MULTI-ELEMENT VERIFICATION & VALIDATION



#### **REV W1** As of 10/7/2022

#### **MAJOR MILESTONES FOR ARTEMIS I KSC FLOW** Status – October 7, 2022



START OF CORE STAGE MATE



UMBILICAL CONNECTS

COMPLETE

UMBILICAL REMATES





START OF LVSA MATE

START OF UMBILICAL RELEASE

AND RETRACT TEST (URRT)



**ORION TO LASF** 

WE ARE GOING







STACK ORION STAGE ADAPTER

STRUCTURAL TEST ARTICLE



STACK MASS SIMULATOR

FOR ORION



CLUTCH CHANGEOUT



MODAL TAP TEST COMPLETE SLS PREVALVE



FIRST INTEGRATED POWER UP



STACK ORION TO SLS









WDR RUN #2



COMM END-TO-END

(LIVE) COMPLETE

for Thermal

Conditions

DEMONSTRATION

TEST

Helium Purge

Pressure Resolved

LOX Load Adjusted

INTEGRATED MODAL

TEST (IMT) COMPLETED FOR ORION



DESTACK MASS SIMULATOR



ORION STAGE ADAPTER TO DESTACK ORION STAGE ADAPTER ESTABLISH CORE STAGE VAB HIGH BAY 4 STRUCTURAL TEST ARTICLE ACCESS COMPLETED







STACK OSA

BAY 4



INTEGRATED VEHICLE INTERFACE **VERIFICATION TEST (IVT) PT 1** 



COUNTDOWN SEQUENCE TEST (CST) PT 1)



CORE STAGE POWER UP









VEHICLE CLOSEOUT AND PREP FOR ROLL



FOR WET DRESS **REHEARSAL** (WDR)













LAUNCH ATTEMPT 1 (Scrub: Eng Bleed Pressure)

LAUNCH ATTEMPT 2 (Scrub: Hydrogen Leak)



ROLL TO VAB (Hurricane lan; Limited Life Items)

WDR RUN #3

CORE STAGE PROGRAM SPECIFIC ENGINEERING **TESTING (PSET) COMPLETE** LOX Load Technique



LH2 Leak (Ground Side) Discovered





PAD OPS AND LAUNCH COUNTDOWN

**ARTEMIS** LAUNCH (Planned Nov 14)









A STATE OF THE OWNER

THURSDAY .....

WDR RUN #4

































03/28/2022

# ARTEMIS

- Launch attempt 1 on August 27, 2022 was scrubbed due to an inability to confirm proper engine conditioning. Also had small but manageable hydrogen leaks
- Launch attempt 2 on September 3, 2022 was scrubbed due to a large hydrogen leak
- Stayed at pad and replaced seals within umbilical
- Successful tanking test was completed on September 21, 2022
- Set up for launch attempt on September 27, 2022 but had to roll back to the VAB due to Hurricane lan
- Teams will use roll back as an opportunity to replace Flight Termination System batteries, and assess Limited Operational Life Items that may need replacement or servicing
- Agency will prepare for November launch attempt on Nov. 14.





# A RATE MUS

# **Artemis II Mission Status**

## Rev H, Updated as of 8/22/2022

# **MAJOR MILESTONES FOR ARTEMIS II**





PARACHUTES QUALIFIED \*CREW EGRESS FOR FLIGHT TRAINING AT

ORION PRESSURE VESSEL ELEMENTS MACHINED NBL



CONTROLLER

EVAL

\*VACUUM

PRESSURE CREW

TEST

CORE STAGE 2

READY FOR

SHIPMENT TO

KSC

ROLL TO PAD

FOR TANKING

TEST

\*HAND \*DOCKING HATCH

EVAL

PRESSURE VESSEL

COMPLETE

ENVIRONMENTAL

CONTROL SYSTEM

INFRASTRUCTURE

INSTALLED

BOOSTERS

ARRIVE AT KSC

ARTEMIS II

TANKING

TEST



**ORION WATER** 

IMPACT TESTING



\*CREW

EMERGENCY

EGRESS TESTS



\*CREW AT SEA

TEST

A-10 - 100

TESTING AT KSC



ORION

ENVIRONMENTAL

TESTS



HEAT SHIELD

BLOCK INSTALL

COMPLETE

ENGINE

COMPLETION

**ARTEMIS II** 

LAUNCH



SLS BOOSTER

MOTOR

SEGMENTS CAST



RS-25 ENGINES PROCESSED

SLS CORE STAGE PROOFING AND WELDING







\*HUMAN-IN-THE-LOOP TESTS

TRAINING

\*MOBILE

LAUNCHER 1

COMPLETE

SLS CORE

STAGE, ICPS, &

ADAPTERS

INTEGRATION

AT KSC





PROTOTYPE



BOOSTER

STACKING

COMPLETE

CREWED MODS



SLS INTERIM

CRYOGENIC

PROPULSION

STAGE (ICPS)

COMPLETION

ORION MASS

SIMULATOR

MATE

ORION MISSION

CONTROL

SIMULATIONS

\*MOBILE ENVIRONMENTAL LAUNCHER 1 60% CONTROL SYSTEM DESIGN REVIEW CHILLERS INSTALLED





PRESSURE VESSEL \*DISPLAY AND ARRIVES AT KSC CONTROL EVAL





\*CREW MODULE

UPRIGHT

SYSTEM TEST



MOTOR QUALIFIED

ATTITUDE CONTROL MOTOR OUALIFIED

CORE STAGE 2

FORWARD

JOIN

\*MOBILE

LAUNCHER 1

MULTI-

ELEMENT V&V

COMPLETE

ROLL TO PAD

FOR LAUNCH









\*FES MOCKUP EVALUATION

EGRESS SYSTEM 60% DESIGN REVIEW

\*PAD EMERGENCY















TRAINING ARTICLE

TRANSPORTED TO

LETF







SLS LAUNCH VEHICLE STAGE ADAPTER COMPLETION

**ORION CREW** MODULE PART 1 COMPLETE

CORE STAGE 2 4/5ths JOIN





EGS ORION OFFLINE PROCESSING START





**ZT** 

 $\triangleleft$ 

- No core stage hot-fire at SSC or post hot-fire refurbishment required at KSC
- No modal test required
- Bleed line re-sizing and Pre-Valve clutch R&R are not anticipated



AIRBUS

CREW MODULE

COMPLETE

ROLL TO VAB

FOLLOWING

TANKING

TEST

EUROPEAN SERVICE MODULE ASSEMBLY AT



HEAT SHIELD INSTALL ON CREW MODULE

CREW AND

SERVICE

MODULE

MATE

ORION

TO VAB



TO KSC

\*PAD

UPGRADES

COMPLETE

ORION

INTEGRATION

TO SLS





MODULE MATE

VAB ECS

UPGRADES

COMPLETE

CONDUCT FINAL

INTEGRATED

TESTING



# **ARTEMIS II**

First Crewed Test Flight to the Moon Since Apollo

1 LAUNCH Astronauts lift off from pad 39B at Kennedy Space Center.

9

**2** JETTISON ROCKET BOOSTERS, FAIRINGS, AND LAUNCH ABORT SYSTEM

CORE STAGE MAIN **ENGINE CUT OFF** With separation.

**OPERIGEE RAISE** MANEUVER

**Prox Ops** Demonstration

**5** APOGEE RAISE BURN Begin 24 hour checkout of spacecraft.

6 PROX OPS DEMONSTRATION **Orion proximity** 

operations demonstration and manual handling qualities assessment for up to 2 hours.

- **INTERIM CRYOGENIC PROPULSION STAGE** (ICPS) DISPOSAL BURN
- TO HIGH EARTH ORBIT 🕕 HIGH EARTH ORBIT

Life support, exercise, and habitation equipment evaluations.

> **11** TRANS-LUNAR INJECTION (TLI) BY ORION'S MAIN ENGINE

CHECKOUT

Lunar free return trajectory initiated with European service module.

**0** OUTBOUND TRANSIT TO MOON

**ICPS Earth** disposal

4 days outbound transit along free return trajectory.

**1** LUNAR FLYBY 4,000 nmi (mean) lunar farside altitude.

12 TRANS-EARTH RETURN **Return Trajectory Correction** (RTC) burns as necessary to aim for Earth's atmosphere; travel time approximately 4 days.

- **13** CREW MODULE SEPARATION FROM SERVICE MODULE
- 11 ENTRY INTERFACE (EI) Enter Earth's atmosphere.
- **15** SPLASHDOWN Ship recovers astronauts and capsule.

PROXIMITY **OPERATIONS** DEMONSTRATION SEQUENCE



## **Artemis II Mission Status**



Four Artemis II engines in their shipping containers and the RS-25 "pathfinder" engine



# NASA ATEMIS

#### Orion

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- ~70% of all Crew Module hardware has been installed
- Service Module initial power up successfully completed and functional testing is underway

#### SLS

- Artemis II Core Stage progress at Michoud Assembly Facility (MAF) is on plan to complete by March 2023
- All four Core Stage Engines were delivered to MAF and engine installation is planned for December
- All solid rocket motor segments for Artemis II have been cast and are in storage in Utah
- Artemis II Interim Cryogenic Propulsion Stage has been manufactured and is at ULA's facilities at Cape Canaveral in Florida

#### EGS

 New 1.4Mgal LH2 Sphere at Pad 39B is essentially complete (remaining work after Artemis I Launch)

LH2 Tank at KSC



# A RATE MUS

# Artemis III Mission Status

# **Artemis III Baseline**

#### **FIRSTS:**

- Human landing in south pole region
- Orion to Human Landing System
   direct mission
- Near Rectilinear Halo Orbit
- 4 crew to lunar orbit (2 to surface)

#### **NEW ELEMENTS:**

- Human Landing System (Starship)
- xEVA Suits and tools (Axiom)
- Orion full up Rendezvous, Prox Ops, and Docking systems

#### **COMMON ELEMENTS:**

- Crewed Orion on Block 1 SLS with Interim Cryogenic Propulsion Stage
- Enhanced ESA Service Module
- Mobile Launcher 1



# **ARTEMIS III**

#### Landing on the Moon

- 1 LAUNCH SLS and Orion lift off from Kennedy Space Center.
- 2 JETTISON ROCKET BOOSTERS, FAIRINGS, AND LAUNCH **ABORT SYSTEM**
- CORE STAGE MAIN ENGINE CUT OFF With separation.
- Inter Earth Orbit Perform the perigee raise maneuver. Systems check and solar panel adjustments.
- 5 TRANS LUNAR INJECTION BURN Astronauts committed to lunar trajectory, followed by ICPS separation and disposal.
- ORION OUTBOUND TRANSIT TO MOON
  - **Requires several outbound** trajectory burns.

- **ORION OUTBOUND POWERED FLYBY** 60 nmi from the Moon.
- **NRHO INSERTION BURN** 8 Orion performs burn to establish rendezvous point and executes rendezvous and docking.
- LUNAR LANDING PREPARATION Crew activates lander and prepares for departure.
- **10** LANDER UNDOCKING AND SEPARATION
- **11** LANDER ENTERS LOW LUNAR ORBIT Descends to lunar touchdown.
- 12 LUNAR SURFACE EXPLORATION Astronauts conduct week long surface mission and extra-vehicular activities.
- 13 ORION REMAINS IN **NRHO ORBIT** During lunar surface mission.

14 LANDER ASCENDS TO LOW LUNAR ORBIT

6

LANDER PERFORMS 15 RENDEZVOUS AND DOCKING

DESCEND

16

- **CREW RETURNS IN ORION** 16 Orion undocks, performs orbit departure burn.
- **ORION PERFORMS RETURN** 17 **POWERED FLYBY** 60 nmi from the Moon.
- FINAL RETURN TRAJECTORY 18 **CORRECTION (RTC) BURN** Precision targeting for Earth entry.
- 19 CREW MODULE SEPARATION FROM SERVICE MODULE
- 20 ENTRY INTERFACE (EI) Enter Earth's atmosphere.
- 21 SPLASHDOWN Ship recovers astronauts and capsule



10

ASCEND

8

NEAR-

RECTILINEAR

HALO ORBIT

(NRHO)

9

### **Artemis III Candidate Landing Regions**



KEY LANDING REGION CHARACTERISTICS

Close proximity to the geographic South Pole

Gentle slope for landing and moonwalks

Constant view to Earth for communications

Continuous sunlight throughout the surface expedition of about 6.5 days

Landing Accuracy

Surface data resolution

Combined mission vehicle capabilities: Space Launch System, Orion spacecraft, Starship Human Landing System

A landing *region* is approximately 15 km<sup>2</sup>. Each landing region includes multiple potential landing sites.



# AS OF 10/17/2022 MAJOR CESD MILESTONES FOR ARTEMIS III



# Initial Human Landing System

NASA will use the HLS Starship for use on Artemis III, the mission that will put the next two Americans on the surface of the Moon.

The SpaceX Option A contract includes two lunar surface missions:

- SpaceX Uncrewed Lunar Demo-A
- SpaceX Crewed Lunar Demo-A





- Initial HLS for Artemis III (SpaceX)
  - Completed SpaceX Option A contract milestones in CY22 including:
    - Landing Software and Sensor Demo
    - Software Architecture Review
    - Sea-level Raptor Engine Cold Start Demo
  - Advancing lander system design maturity; examples include:
    - Raptor development/build: demonstrated production of 7 engines in one week
    - Cryo Fluid Management: Plan in-space Propellant Transfer and Long Duration Flight Tests
  - Prep for first Starship orbital flight
- Sustaining HLS for Artemis IV+
  - Conducting Sustaining HLS studies and risk reduction tasks ("Appendix N" contracts)
  - Sustaining HLS Acquisitions
    - Impending SpaceX Option B contract mod
    - Issued HLS Sustaining Lunar Development "Appendix P" solicitation



# Path to First Starship Orbital Flight Test



#### Booster 7



### Starship 24



7-Engine Static Fire

6-Engine Static Fire

#### Stacked vehicle (S24 atop B7) at Boca Chica



### **Booster 7**

- Completed 1-engine, 3-engine, and 7-engine static fire tests
- Completed 33-engine spin prime tests after incorporating repairs and corrective actions from July 11 high-energy event

## **Starship 24**

• Completed 2-engine and 6-engine static fire tests

### **Stacked Vehicle**

 Conducted LOX & LCH4 single species prop system test and de-stacked

## **Remaining Path to Flight (dates TBD)**

- Booster 7 / Starship 24 re-stacking
- Additional single-species prop system testing
- Additional de-stacked static fire tests (including Booster 7 33-engine test)
- Wet dress rehearsals
- FAA license processing in work

## **Artemis III EVA System Overview**

#### **Advanced suits**

- Portable Life Support Subsystem; Oxygen, cooling, power, CO<sub>2</sub> removal
- Pressure Garment Subsystem

#### **Tools and equipment**

- Geology sampling tools
- Maintenance tools
- Handrails and umbilicals





Portable Life Support Subsystem







- Axiom Space and Collins Aerospace selected to compete for task orders to demonstrate an EVA System and certify the system to support NASA programs
  - Task order for Artemis development and demonstration on Artemis III awarded to Axiom Space
  - Task order for ISS development and demonstration award planned for the near future
- Developed government reference design (DVT) EVA system (xEMU) to validate requirements and reduce development risks
  - All xEMU technical details and test results provided to suit vendors for potential incorporation into provider EVA systems designs
- New, government reference design thermal control system, the ISS SERFE Payload, was demonstrated for ~2 years on ISS including 25 simulated EVAs
- Initiated EVA testing in the Iceland Highlands, the Suit System Lab, ARGOS, the Desert RATS, and the NBL



# **EHP EVA Development Achievements**



**ISS SERFE Payload** 

~2 years of on-orbit science on xEMU PLSS Thermal Loop performance
Returned in August, disassembled, inspected and evaluated
Post-flight findings available for Vendors

Completed design, build, and integration of complete xEMU system

DVT xEMU Functional Testing

**B7** Space Suit System Lab

July 2022

 Performed over 18 months of Development and Verification Tests at the component through full assembly levels Joint Test Team NBL Runs with new facility capabilities, EVA Tools, xEMU DVT

CT III



# A RATE MUS

# Artemis IV Mission Status

# **Artemis IV Baseline**

#### **FIRSTS:**

- SLS Block 1B Flight with Exploration Upper Stage (EUS)
- Launch from ML-2
- Orion/SLS Co-Manifested Payload (I-HAB); Orion to deliver to Gateway
- Gateway Initial Operational Capability (IOC) with PPE+HALO
- Orion and HLS Docking at Gateway

#### **NEW ELEMENTS:**

- SLS Block 1B with EUS, Universal Stage Adapter, Payload Adapter Fitting
- ML-2 with supporting EGS physical and comms upgrades
- Sustaining Human Landing System
- Gateway IOC (PPE + HALO), I-HAB

#### **COMMON ELEMENTS:**

- Crewed Orion with docking capability
- Common Block1 / Block 1B SLS Elements
- xEVA Suits and tools

# **ARTEMIS IV**

International Habitation Module delivery to Gateway followed by Crewed Lunar Landing

#### LAUNCH

SLS with I-HAB co-manifested payload and Orion with 4 crew members lift-off from Kennedy Space Center.

- 2 JETTISON ROCKET BOOSTERS, FAIRINGS, AND LAUNCH ABORT SYSTEM
- 3 CORE STAGE MAIN ENGINE CUT OFF With separation.
- ENTER EARTH ORBIT Exploration Upper Stage (EUS) performs circularization of Low Earth Orbit. Systems check and solar panel adjustments.
- 5 TRANS LUNAR INJECTION BURN EUS commits astronauts in Orion and I-HAB to lunar trajectory.
- 6 ORION TUGS I-HAB TO MOON Orion separation from Universal Stage Adapter (USA), ejection of USA, Orion docking with I-HAB for extraction from EUS/ Payload Adapter Fitting (PAF) followed by Orion tug of I-HAB to Gateway Orbit and EUS disposal.
- ORION OUTBOUND TRANSIT TO MOON
   Requires several outbound trajectory burns.
- ORION OUTBOUND POWERED FLYBY
   60 nmi from the Moon.

#### GATEWAY ORBIT INSERTION BURN Orion performs burn to establish Orion performs burn to establish

23

rendezvous point and executes rendezvous and docking.

INTERNATIONAL HABITATION MODULE ARRIVAL AT GATEWAY

I-HAB docking with Orion to Power and Propulsion Element (PPE) and Habitation and Logistic Outpost (HALO) module.

- I-HAB ACTIVATION AND CREW INGRESS Astronauts ingress, activate and utilize I-HAB as part of larger Gateway complex.
- LUNAR LANDING PREPARATION Crew activates Lander and prepares for departure.
- **13** LANDER UNDOCKING AND SEPARATION
- LANDER ENTERS LOW LUNAR ORBIT Two astronauts descent to lunar touchdown.
- LUNAR SURFACE EXPLORATION Astronauts conduct week long surface mission including moon walks, rover ops, and surface science.
- 6 ORION REMAINS IN LUNAR GATEWAY ORBIT

Other two astronauts tend to Gateway during lunar surface mission.

- LANDER ASCENDS TO LOW LUNAR ORBIT
- 18 LANDER PERFORMS RENDEZVOUS AND DOCKING
- CREW RETURNS IN GATEWAY / ORION Crew transfers science samples to Orion for return, undocks, performs departure burn.
- ORION PERFORMS RETURN POWERED FLYBY Lunar gravity assist, fly 60 nmi from the Moon.
- FINAL RETURN TRAJECTORY CORRECTION BURN Precision targeting for Earth entry.
- 22 CREW MODULE SEPARATION FROM SERVICE MODULE
- ENTRY INTERFACE Enter Earth's atmosphere.

24

SPLASHDOWN Astronaut crew, science sample and capsule recovery by ship.





# Status: SLS Exploration Upper Stage and Associated Capabilities/ Block 1B Variant





- The Exploration Upper Stage (EUS) enhances SLS lift-capability, enabling Orion orbit flexibility, can deliver co-manifested payloads to the Gateway, supports lunar exploration, and future Mars mission.
- The EUS has unique human rated capability for agile repositioning, dwell, and docking maneuvers in deep space with precisions guidance, cryogenic management, and real time communications and control.
- EUS enables the SLS Block 1B variant to accomplish a 42% increase in TLI mission capture.
- Tremendous progress has been made with advance manufacturing, tooling, and preparations for structural testing.
- The Block IB CDR board is schedule for November 3, 2022 and represents a major milestone in program advancement.
- The Michoud Assembly Facility outside of New Orleans is being prepared for full production of the design and the overall status is on schedule to support the Artemis IV mission.

# Status: EGS Mobile Launcher 2 (ML-2)



- NASA's Exploration Ground Systems program is working with contractor Bechtel National Inc. to build a second mobile launcher (ML2) for NASA's Space Launch System (SLS) Block 1B and Block 2 configurations
- The ML2 project is a key component for future exploration, providing the assembly and launch platform for the Artemis IV mission and beyond that enable the power Block IB SLS operations.
- Significant progress has been made with the design with 90% design completion over the next few months
- However, the project has experienced development and production issues:
  - Completion of steel fabrication drawings has also been delayed by revisions to reduce overall weight
  - Steel supply chain issues have caused delays of both steel fabrication and delivery on the completed design
  - Significant improvements have been made in the design to accommodate weight, changes in management approach to improve performance, and additional work is being performance to address supply base issues.



# **GATEWAY** Artemis IV – Initial Capability

The two foundational elements of Gateway will launch together, prior to Artemis IV establishing a platform where astronauts can live and work in lunar orbit

#### **Power and Propulsion Element (PPE)**

- High-power solar electric propulsion used to transfer PPE/HALO to lunar orbit
- Provides power for the Gateway
- Maintains the Gateway's orbit

#### Habitation and Logistics Outpost (HALO)

- Houses up to 4 crew for up to 30 days (with Orion)
- Provides high-rate lunar communication relay to support lunar surface activities and command and control systems for Gateway
- Docking port for visiting spacecraft and future modules

# PPE and HALO are launched together as a Co-Manifested Vehicle (CMV) on a SpaceX Falcon 9 Heavy







- Conducted the Program PDR-informed Sync Review in May. Targeting PDR-informed Sync Review technical closeout for December.
- Finalizing updates to the Maxar PPE contract to incorporate major changes for CMV, risk mitigation, and 15-yr life. This is the last major contract change, all other IOC elements under contract to flight.
- HALO and PPE manufacturing underway.
  - Completed HALO Habitable Element Longitudinal Welds. Circumferential welds in progress.
  - PPE central cylinder manufacturing and assembly complete.

## **Gateway Hardware Progress**





PPE Central Cylinder Assy Complete



Completed 12kw thruster end-to-end string Phase 1 testing

Completed 6kw thruster end-to-end string Phase 3 testing





#### HALO Habitable Element (HE) Longitudinal Welds Completed



# GATEWAY



# Artemis IV – I-HAB and Logistics

- International Habitation Module (I-HAB)
  - Enhances Gateway capabilities for scientific research, life support systems and crew living quarters and imagery components
  - Enables longer crewed mission durations
  - Includes additional docking ports
  - Will be launched with Orion on SLS as co-manifested payload on Artemis IV

#### Logistics

- SpaceX was selected under the Gateway Logistics Services contract to deliver cargo, experiments and other supplies to the Gateway in lunar orbit
- 5.4 6.0 MT delivered cargo capability
- Automated docking/undocking
- Independent power and thermal





## Glossary



Acronym	Definition	Acronym	Definition
VAB	Vehicle Assembly Building	PLSS	Portable Life-Support System
MAF	Michoud Assembly Facility	ML2	Mobile Launch 2
LH2	Liquid Hydrogen	EUS	Exploration Upper Stage
KSC	Kennedy Space Center	I-HAB	International Habitat
xEVA	Exploration EVA	HALO	Habitation and Logistics Outpost
ESA	European Space Agency	NRHO	Near Rectilinear Halo Orbit
SLS	Space Launch System	TLI	Translunar Injection
HLS	Human Landing System	CDR	Critical Design Review
LOX	Liquid Oxygen	PPE	Power and Propulsion Element
LCH4	Liquid Methane	CMV	Co-Manifested Vehicle
EVA	Extravehicular Activity	PDR	Preliminary Design Review
xEMU	Exploration Extravehicular Mobility Unit	ECP	Engineering Change Package
NBL	Neutral Buoyancy Lab	HALO HE	HALO Habitable Element
SERFE	Spacesuit Evaporation Rejection Flight Experiment	GERS	Gateway External Robotic System
DVT	Development and Verification Tests	ISS	International Space Station

#### **Gateway Integrated Spacecraft** ARTEMI SPACEX Gateway External **IX**A Robotic System (GERS) Logistics Module 2 May **HTV-XG** logistics MAXAR resupply capability Power and Propulsion Element (PPE) Airlock (provider TBD) ESPRIT-Refueler esa International Habitat (I-HAB) Habitation and Logistics Outpost Cesa 🚜 🛪 A (HALO) NORTHROP GRUMMAN European Orion Spacecraft Service Module Human Landing System (HLS) esa (govt. reference concept shown)