

National Aeronautics and  
Space Administration



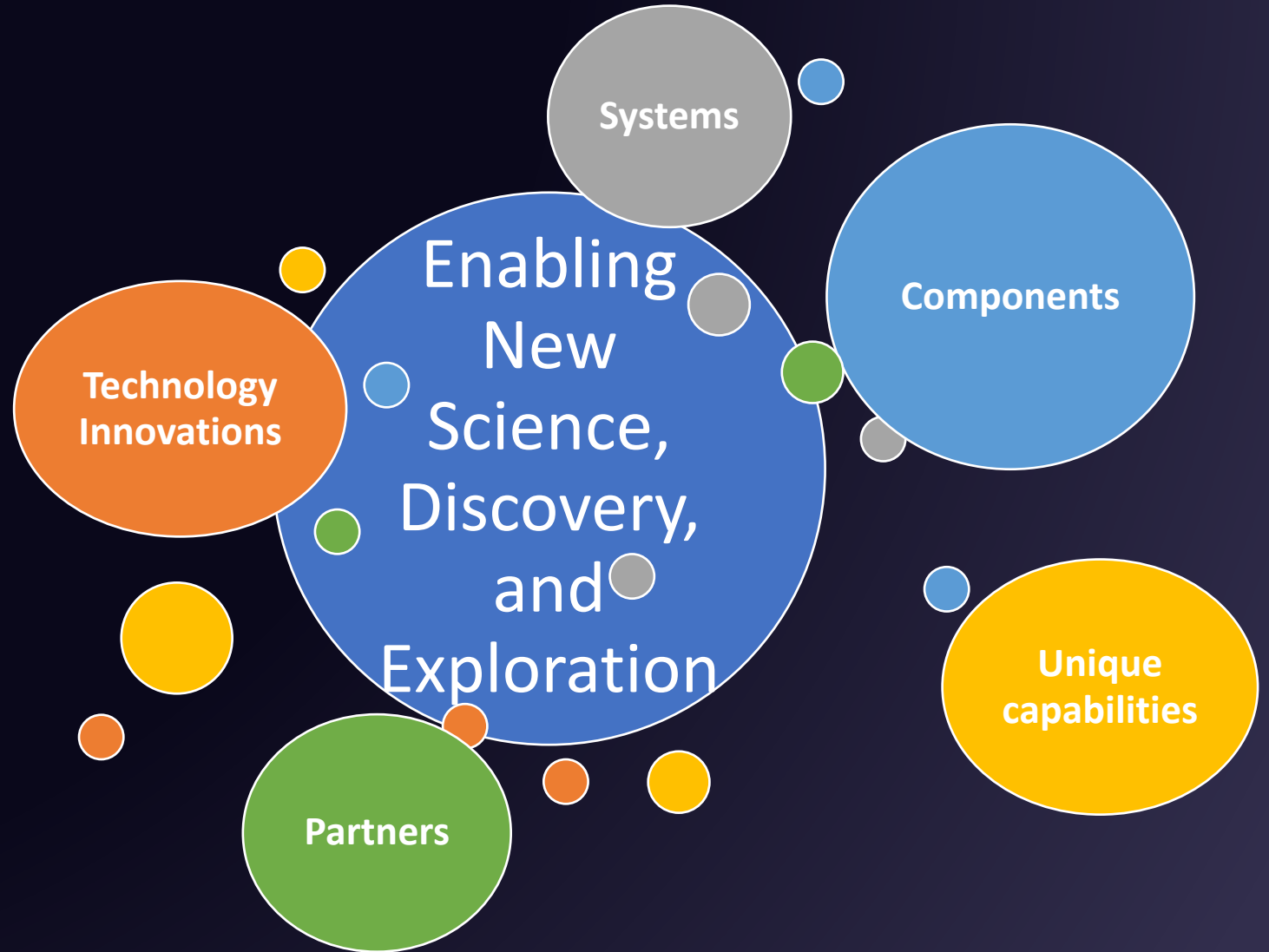
# *An Overview of Small Satellite Activities at NASA's Glenn Research Center*

**Concha M. Reid**  
Space Science Project Office  
NASA Glenn Research Center

**Small Satellite Conference, August 5-10, 2023**



# NASA GRC Small Satellite Efforts



Utilizing our unique capabilities and leveraging our partnerships to develop innovative technologies that enable new and exciting discoveries on Earth and In-Space



# SmallSat Ka Operations User Terminal (SKOUT)

POC: Felix Miranda, [felix.a.miranda@nasa.gov](mailto:felix.a.miranda@nasa.gov)

## Based on 5G Technology

*Leverages Ka commercial timescale improvements at mass production costs*

## High-Rate Small Form Factor Payload

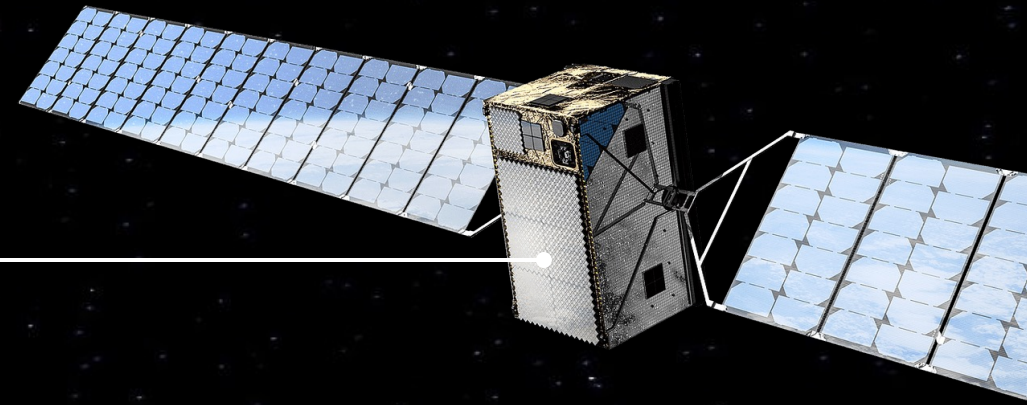
*Fully integrated communications payload in a 1U form factor capable of providing up to Gbps data rates*

## Complete, Modular Communications Solution

*Tiled architecture extensible from CubeSats to traditional large satellite user missions*

## Eliminates Mechanical Gimbal/Body Pointing

*Electronically steered antenna provides near instantaneous pointing and tracking without disturbing spacecraft operations*



***Bringing 5G  
technology to Space***

A Partnership Between



CESIUM



**NON-PROPRIETARY DATA**

**IDENTIFICATION AND SIGNIFICANCE OF INNOVATION**

This SBIR will mature 100G optical transceiver and amplification technology to TRL 6 for GEO, lunar and Lagrange points using mature low-cost and low size, weight, and power (SWaP) technologies. We address specific performance with an emphasis on long mission duration and high reliability vs short LEO missions.

Our link budget indicates that 100G-200G is feasible, small telescopes can support lunar, L1 and L2 ranges.

- Enable GEO-GEO intersatellite link and GEO Lunar, and potentially L1 and L2 ranges direct to earth links.
- Ultra-low SWaP-C and can be expanded to Tbps with multiple modules.
- Enabler to expand NASA human spaceflight operations, telerobotic, HD video and SMD science for GEO, and Lunar missions.
- Enables NASA Heliophysics vision for affordable, sun-sensing SmallSat constellations at L1, L2 for space weather, astronaut safety missions
- Technical approach leverages terrestrial fibercom photonic integrated circuit (PIC) at low cost for space
- Ready for program insertion and available to the space FSO industry in 2025

**TECHNICAL OBJECTIVES AND PROPOSED DELIVERABLES**

This SBIR proposes to develop a very low SWaP, (0.5 U) 100 G transceiver (TRX) card that supports GEO lunar and Lagrange laser communications to Earth and DSM optical inter-satellite links. During Phase I we conducted design and risk reduction activities that established feasibility to develop a TRL 6 prototype in Phase II.

The Phase 2 objectives of this SBIR include the following:

- Engineer a < 1 U CubeSat sized 100G transceiver ready for mission insertion.
- Demonstrates the unit meets unique to GEO, lunar and beyond practical free space links and support a link that tolerate: amplitude fluctuations, low dispersion, doppler shifting, and background noise.
- Demonstrate the unit meets radiation, vibration and thermal vacuum requirement for general aerospace industry use.
- Validate that short pulse 100G signals can be amplified to highpowers with minimal distortion and can support coherent communications

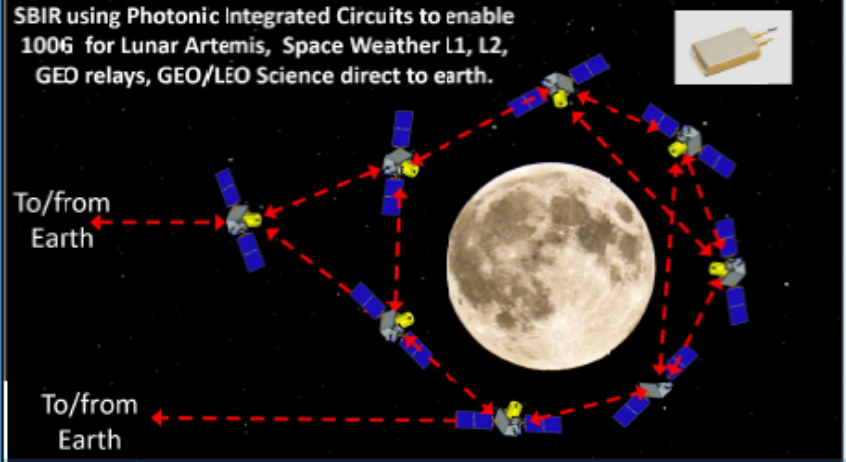
**TRL**

Estimated



**IMAGE TITLE: Small Sat 100G for DSM**

SBIR using Photonic Integrated Circuits to enable 100G for Lunar Artemis, Space Weather L1, L2, GEO relays, GEO/LEO Science direct to earth.



**NASA APPLICATIONS**

- Support Lunar Artemis human exploration missions
- NASA state-of-the-art mesh networked lasercom capability to support ScaN deployment of optical communications
- Deep Space & Heliophysics – Space weather, Sun studies out to L1, L2 at 100 Gbps
- Near Earth science missions – Increased data rate for SmallSat sensors
- Near Earth GEO and cis-lunar, lunar orbital, Lagrange Point L1, L2
- Future NASA SmallSat and CubeSat constellations with science missions that need optical coms to support high data rates including multispectral imaging sensors

**NON-NASA APPLICATIONS**

- DoD and U.S. Government for intelligence imagery. GEO, MEO, LEO
- CLPS commercial lunar payload services companies
- High data-rate, low-cost commercial optical communications from LEO/GEO satellites

**FIRM CONTACTS**

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# WBG LET Charged Particle Telescope

POCs: John Wrbanek, john.d.wrbanek@nasa.gov; Susan Wrbanek, susan.y.wrbanek@nasa.gov

## Phase I

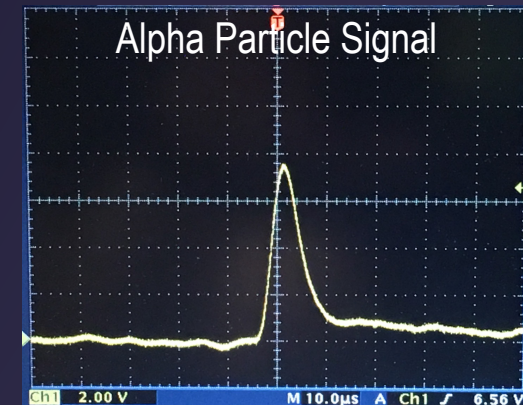
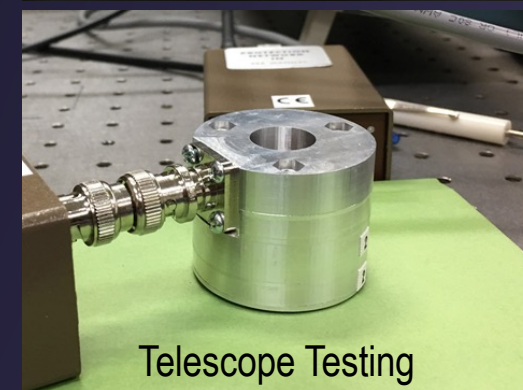
- Designed, fabricated, packaged and tested four LET detectors singly and configured as stacked pairs
- Improved performance and stable operation of the telescope design
- Phase I Telescope Specs:
  - Telescope Size: 4.375 cm dia. x 3.50 cm tall
  - Aperture Size: 200 mm<sup>2</sup>
  - Geometric Factor: 0.5 sr·cm<sup>2</sup>
  - Field of View: 62°
  - Detector: HPSI 4H-SiC, 1000Å Pt/Ti (anode), 1000Å Ni/Ti (cathode)
  - Die Size: 1.778 cm x 1.778 cm square
  - Capacitance: 56.7 ± 1.5 pF

## Phase II currently underway

- Accommodate smaller connectors
- Measure a spectra

## Future plans beyond Phase II

- Integration of charge amplifiers into the package
- Accelerator beam line tests

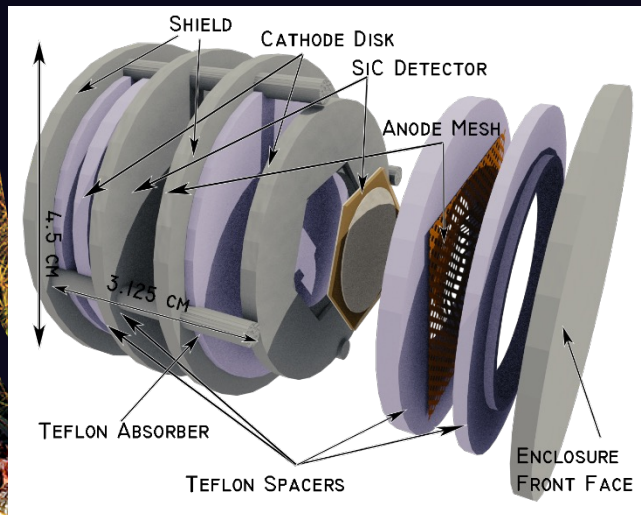
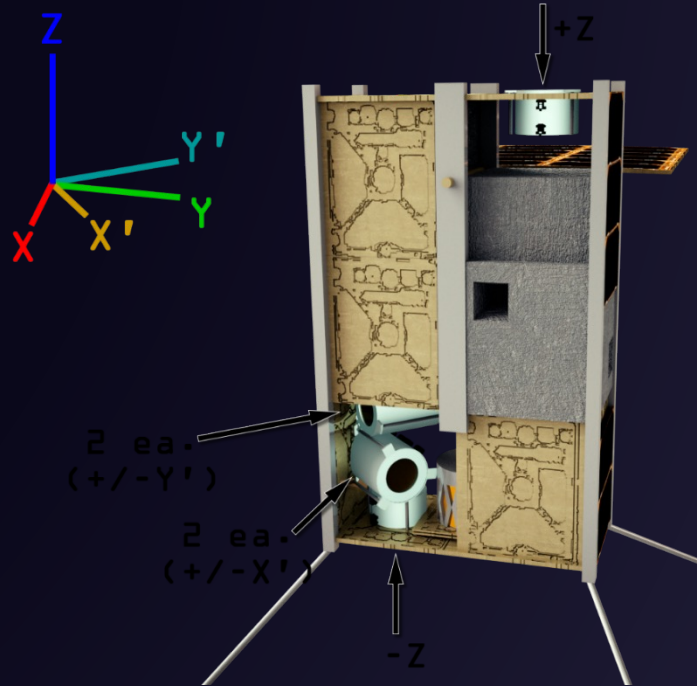
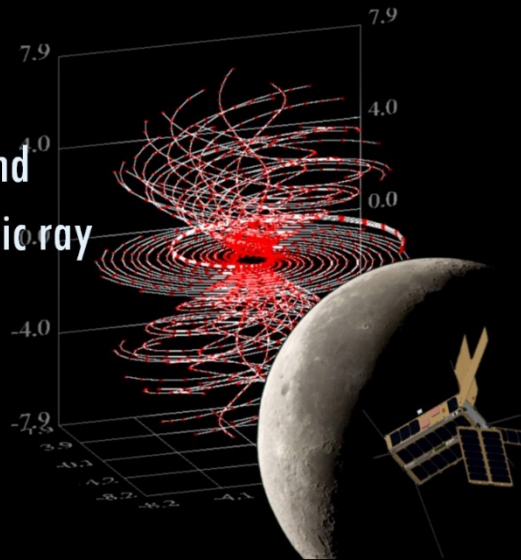




# SPAGHETI: Deep-Space CubeSat Concept

POCs: John Wrbanek, john.d.wrbanek@nasa.gov; Susan Wrbanek, susan.y.wrbanek@nasa.gov

Solar  
Proton  
Anisotropy and  
Galactic cosmic ray  
High  
Energy  
Transport  
Instrument



SiC LET Detector Stacks

Exploration of the transient variations in ion flux anisotropy in deep space and near the lunar surface

## Features:

- 6 packages of SiC Linear Energy Transfer (LET) detector stacks, arranged to provide simultaneous multidirectional measurements
- Detector insensitivity to temperature changes will allow compact, low-power operation



# NanoSonic Inc., NASA Ames Research Center, and NASA Glenn Research Center Technology Demonstration for Small Spacecraft

GRC POC: Susan Wrbanek, [susan.y.wrbanek@nasa.gov](mailto:susan.y.wrbanek@nasa.gov)

## New lightweight composite structural and shielding material

- Developed by NanoSonic Inc. under a NASA Small Business Innovative Research (SBIR) contract

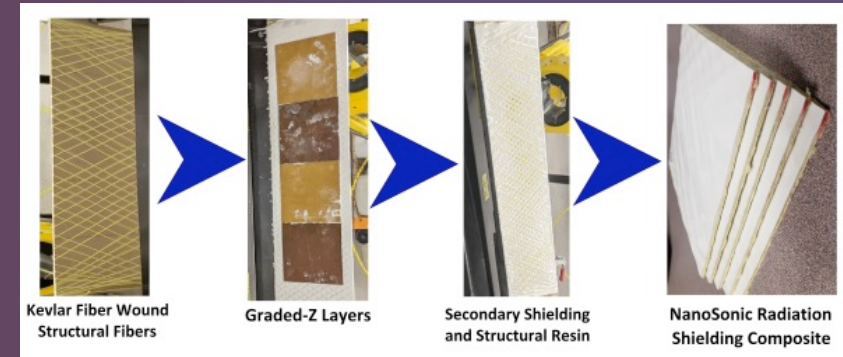
## May offer increased mission lifetime and improved functionality for small satellites

- Protection of electronics from damage by space radiation

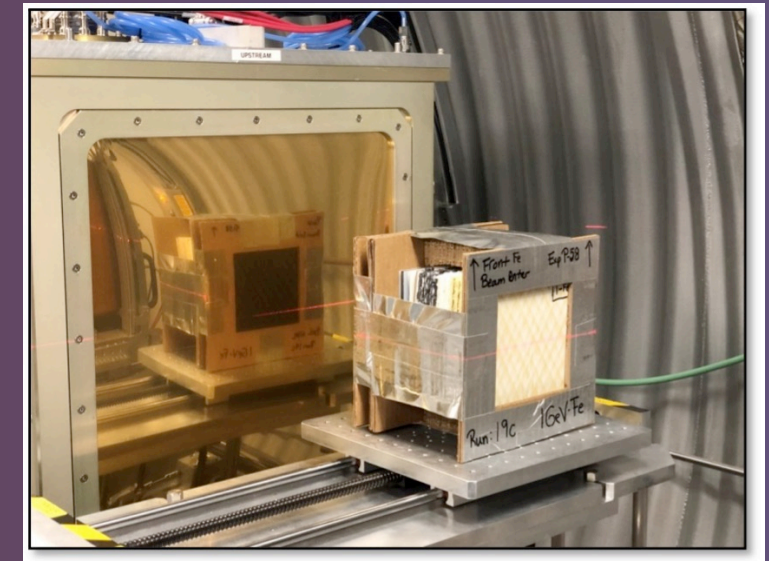
## Planned for a CubeSat demonstration flight in Autumn 2024

- NanoSonic material development PI: Dr. Jennifer Lalli
- ARC CubeSat mission personnel: Ali Guarneros Luna, Avery Brock (Wyle Services LLC), Dr. Rudolphe De Rosee
- GRC COR and Science Demonstration PI: Susan Wrbanek

**Applications: Commercial subsystems, Gateway, long missions to the Moon, Mars or Jupiter, and human missions**



Material Development



Ground testing



# Satellite Hosting Atmospheric and Littoral Ocean Water Sensors

POC: John Lekki, john.d.lekki@nasa.gov

Software Defined Hyperspectral Imaging System for large or small Earth Science and Planetary Science missions

## Features:

- Small size
- Low power requirement
- Rapid deployment

## Characterization of:

- Coastline water changes due to global warming
- Water/ice interfaces
- Atmospheric content and distribution (Earth and outer planets)
- Surface composition (Earth and outer planets)
- Ice content of planetary surfaces (Earth, outer planets, moons)



Platforms: Aircraft, UAVs, drones (Earth and outer planetary), high altitude balloons, CubeSats, small and large satellites

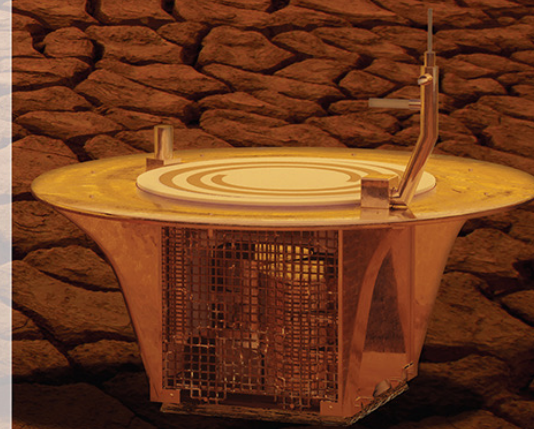


# Long-lived In-situ Solar System Explorer (LLISSE)

National Aeronautics and  
Space Administration



- Small, long-duration lander for Venus
  - Designed to be an independent platform with all the needed subsystems (power, communication, sensors, ...)
- < 10 kg, < 20 cm/side at the base
- Operates for 60 days or more on Venus
- Enable compelling science by returning first ever temporal in-situ data
  - Meteorology (temp, pressure, radiance, wind speed and direction)
  - Atmospheric species abundances and variability
  - Future enhancements: seismometer, inclinometer, reaction chemistry analysis, accelerometer, camera system
- Will transmit measurements to a supporting orbiter to relay data to Earth



Principal Investigator: Tibor Kremic, [tibor.kremic@nasa.gov](mailto:tibor.kremic@nasa.gov), Program Manager: Nathan Funk, [nathan.w.funk@nasa.gov](mailto:nathan.w.funk@nasa.gov)



# Future Concept: Compact Full-Field Ion Detector System (CFIDS)

POCs: John Wrbanek, john.d.wrbanek@nasa.gov; Susan Wrbanek, susan.y.wrbanek@nasa.gov



## Space radiation detector with spherical geometry

- Technology covered by U.S. Patents 7,872,750 and 8,159,669

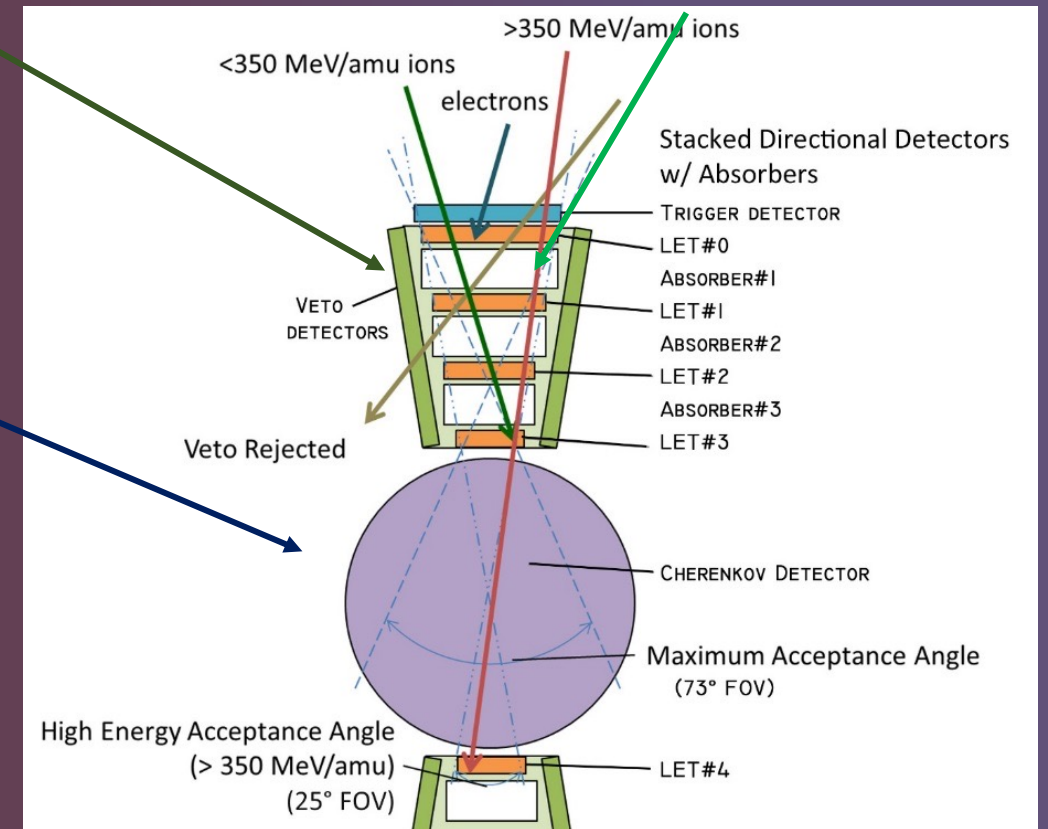
ZnO UV Sensor for Solid-State Cherenkov Detector  
(U.S. Patent 10,054,691)



Large Area SiC Diode for LET Detectors



GaP Diode for Solid-State Coincidence/Anticoincidence Detectors  
(U.S. Pat. 10,429,521)



## Wide Band Gap (WBG) detectors enable:

- Mapping of heavy ions  $> 100 \text{ MeV/amu}$
- High radiation flux rates for 10+ year missions
- Low noise, multi-directional measurements at single locations



## NASA GRC Small Satellite Efforts

- **Developing components, technologies, and systems**
- **Innovations to enable new science and exploration in harsh and unique environments**
- **Planetary Science and Earth Science Applications**



# Questions?

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