

\*jwconklin@ufl.edu

The Miniature Optical Communications Transceiver (MOCT)

John W. Conklin\* UF team: Paul Serra, Nathan Barnwell, Tyler Ritz, Myles Clark, Danielle Coogan NASA Ames leads: Anh N. Nguyen, Belgacem Jaroux

> [This presentation contains no EAR- or ITAR-controlled information]

## MOCT Development



CHOMPTT (2013-2021) Currently on orbit AFRL, NASA, FSGC



MOPFA seed laser SDPM prototype

2/13

CLICK tech demo, (2017-23) NASA SSTP

development of MOCT

SmartFusion2 FPGA development board

(2014-18), Low TRL

NASA Early Career Faculty grant



### Some Results from CHOMPTT

- Despite S/C pointing (and other issues) preventing a successful time-transfer, CHOMPTT accomplished several things:
  - First ever measured performance of a CSAC in space (on any platform)
  - Optical tracking of CHOMPTT by an SLR facility via Sun reflections and CHOMPTT's laser beacons
  - Successful laser ranging to CHOMPTT
  - Demonstrated functionality and performance of all elements of the OPTI payload:
    - Photoreceivers
    - CSACs (two)
    - Timing electronics
    - Laser beacons
    - Thermal & Power
      performance



### The Miniature Optical Communications Transceiver (MOCT)

- Concept: Laser communications transceiver for small satellites and/or for long-haul comms based on pulse position modulation (PPM) consuming <15 W</li>
- Innovations:
  - Replace GHz slot clock with low power precision oscillator: 100 mW CSAC
  - Laser pulse timing generated within an FPGA with <10 ps precision</li>
  - Use feedback control to continuously correct timing errors due to PVT
  - Power efficient gain-switched seed laser diode driver
  - Master oscillator power fiber amp to achieve optical link
- Performance goals:
  - ~100 Mbps & <100 ps timing</li>
- Project does not focus on optical Master of font-end, acquisition, structural/thermal

John W. Conklin, STP EXPO, 24 May 2021



**Original MOCT Concept** 

### Software-Defined Pulse Modulator

- Generates electrical pulses with ~ps precision referenced to an external oscillator (CSAC)
- Pulse timing is based on a selectable signal propagation time through a delay chain within the FPGA
- Delay time is continually calibrated against the CSAC
  - U.S. patent (16/086,092) on this technology
  - See [Serra, Conklin, 2019]
- Demonstrated using COTS 80 FPGA (Smartfusion2) with flight ~equivalent (RTG4)
- Measured performance in a lab environment
  - 2 ps timing precision
  - 10 ps timing accuracy
  - 14 ns range

John W. Conklin, STP EXPO, 24 May 2021





#### Timing compensation circuit

### 100 ps Gain-switching Seed Laser Driver

- Fast Laser Advanced Switching High-Frequency Emitter (FLASHE)
- Capable for generating 500 mA pulses at 50 MHz
- Uses double FET configuration
  - One for laser ON, one for laser OFF separated by 100 ps
  - Transistors are 'slow', operated in linear mode to conserve power and eliminate the need for an RF amp





#### Triggered on CSAC reference oscillator



Triggered on measured rising edge

### Pulse Detection and Timing

- Time-to-Digital Converter 2 (TDC-GPX2) used for time stamping
  - Measures rising and falling edges based on selected threshold
  - Consumes less power than a fast ADC, but ADC has better link margin with matched filtering
  - Data rate of 70 MS/s with 20 ps timing precision (<40 ps measured)
- Avalanche photodiode (APD) receiver
  - Selectable reverse bias for gain control
  - Active thermal control for stable breakdown voltage
  - Minimum received power: 100 nW
  - Measured timing jitter: 50 ps



MOCT APD photo receiver (left) and mechanical enclosure (center)

### **MOCT Breadboard**



### Measured Performance in the Lab

- EDFA output attenuated by an amount equivalent to LEO-to-Earth link
  - Tx peak power: 1.4 W
  - Rx peak power: 180 nW
- Pulse modulation scheme
  - 32 PMM
  - Pulse width: 2.5 ns
  - Guard time: 50 ns
  - Slot width 0.4 ns
- Uncorrected BER: 10<sup>-2.6</sup>
- Data rate: 78 Mbps
- See
  - [Barnwell et al. Aerospace, 2018]



# **ITTUF** CLICK: CubeSat Laser Infrared CrosslinK



- NASA-MIT-UF CLICK B/C mission will demonstrate CubeSat space-to-space laser comms
  - CLICK A is a space-to-ground laser comm risk reduction mission for B/C, launching later this year
- Elements of MOCT will fly on CLICK B/C
  - FPGA modulator with CSAC frequency reference
    - Operated in a different way than the SDPM
  - APD photoreceiver
  - Time-to-digital converter
    - ADC-based receiver also flown for comparison
- B/C launch: 2022-23





- Pair of **1.5 U, ~1700 g, <25 W** payloads
- 200 mW avg. Tx, 70 μrad divergence
- 2.5 cm receive aperture
- 500 mW beacon at 976 nm for PAT
- Full-duplex **PPM** crosslinks 1537/1565 nm
  - 50 Mbps, 4-PPM, <450 km</li>
  - 25 Mbps, 16-PPM, <920 km
- 200 ps timing accuracy & time transfer
- Primary success criteria:
  - >20 Mbps full-duplex @ 580 km
  - <0.5 m ranging w/o GPS @ 580 km</p>

Credit: P. Serra



### Conclusions

- CHOMPTT/OPTI
  - Ground → space time-transfer to <200 ps</li>
  - <5 W
  - ~0.3U without redundancy
  - Partial flight demo; ongoing

- MOCT
  - ~10s Mbps laser comm
  - <200 ps time-transfer</p>
  - Ideal for long haul or small sats
  - More easily scalable to longer links but not higher data rates
  - ~15 W, ~1.5U
  - Lab demo only

- CLICK B/C
  - ~10s Mbps laser comm
  - <200 ps time-transfer</p>
  - More easily scalable to higher data rates
  - ~25 W, ~1.5U
  - Near future flight demo







