## **Title and Research Team**

Project Title: Multi-scale Modeling of Plume-Spacecraft Interactions for Novel Propellants

Solicitation: SpaceTech-REDDI-2018 Early Stage Innovations Topic 1 - Modeling for Small Satellite Electric Propulsion

Two Opposite Polarity

Electrospray Thrusters

Low Energy

Ions

SmallSat

Sputter/

Secondary

Electrons

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## **Approach**

Propose new electrospray thruster plume model based on kinetic, particle simulations of cation/anion interactions at multiscales.

- At the 100 nm 1 micron scale, coarsened MD-like simulations and PIC will model jet growth for ILs such as [EMI][BF4] and [EMI][IM].
- 2 more length scales, micron mm and mm m using AMR/Octree grid-free in KMC and DSMC/PIC methods. to include Coulomb interactions and sputtered species.
- Fundamental experiments to measure secondary emissions and sputtered products from surfaces bombard by IL electrospray plumes.

## **Research Objectives**

Ground-based Vacuum

Sputtered Wall

Propellant

Fragments

Secondary

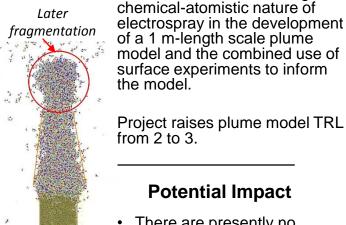
Electrons

Material

-Chamber Testing

Plume Ions impact Facility Walls

- Develop a model that will be used to interpret recent ground-based experiments and have the capability to extrapolate and predict plume-spacecraft interactions in the space environment.
- Study interactions of electrospray anions and cations with facility walls in terms of degree of secondary electron emission and sputtered wall material and propellant fragments. Innovation is in maintaining the



Early Taylor

Cone

formation

Project raises plume model TRL from 2 to 3.

## **Potential Impact**

- · There are presently no electrospray plume models.
  - Model will provide a tool that can be used at NASA research centers to interpret ground based experiments.
- Proposed research will allow us to predict adverse spacecraft charging and sputtering due to espray thrusters in LEO and GEO orbits.
- While focusing on electrospray thruster plumes, the model development will be sufficiently general such that spacecraft effects from both ion and Hall thruster plumes and "cross-device" plume interactions may be assessed.