Asteroid Redirect Mission (ARM)
Update to the NASA Advisory Council, Human Exploration and Operations Committee

Michele Gates
Program Director, Asteroid Redirect Mission

July 25, 2016
Asteroid Redirect Mission: Three Main Segments

IDENTIFY
Ground and space based assets detect and characterize potential target asteroids

REDIRECT
Solar electric propulsion (SEP) based system redirects asteroid to cis-lunar space.

EXPLORE
Crew launches aboard SLS rocket, travels to redirected asteroid in Orion spacecraft to rendezvous, study and returns samples to Earth
ARM Contributions to Deep Space Human Exploration

Demonstrate advanced autonomous proximity operations in deep space and with a natural body

Using high-power solar electric propulsion to transport multi-ton masses in space

Demonstrate integrated crewed/robotic vehicle operations in deep space

Use astronaut EVA in deep space for sample selection, handling, and containment on just the second extra-terrestrial body in history
ARM: An Early Mission in the Proving Ground of Cislunar Space

**IN-SPACE POWER & PROPULSION:**
- High efficiency 40kW SEP extensible to Mars cargo missions
- Power enhancements feed forward to deep-space habitats and transit vehicles

**EXTRAVEHICULAR ACTIVITIES:**
- Two in-space EVAs of four hours each
- Primary Life Support System design accommodates Mars missions
- Sample selection, collection, containment, and return

**TRANSPORTATION & OPERATIONS:**
- Capture and control of non-cooperative objects
- Common rendezvous sensors and docking systems for deep space
- Cislunar operations are proving ground for deep space operations, trajectory, and navigation

**High Efficiency Large Solar Arrays**

**Solar Electric Propulsion**

**Deep-Space Rendezvous Sensors & Docking Capabilities**

**Exploration EVA Capabilities**
Asteroid Redirect Mission Progress

- Robotic mission acquisition strategy decisions Aug 2015
- Formulation Assessment and Support Team (FAST) established Aug 2015
- Public comments due on FAST draft report Dec 2015
  - https://www.nasa.gov/feature/arm-fast
- Robotic mission requirements technical interchange meeting Dec 2015
- Update with Small Bodies Assessment Group Jan 2016
- Robotic spacecraft early design study contracts selected Jan 2016
- Formulation guidance updated for ARRM KDP-B Feb 2016
  - Robotic mission launch date, readiness date for crewed mission
- Crewed segment operational requirements meetings at JSC Feb 2016
- ARM leadership team strategy meeting on partnerships and engagement Feb 2016
- FAST final report released Feb 2016
- STMD electric propulsion development contract selection Apr 2016
- Initial ARRM/ARCM safety review at JSC June 22-23
- ARRM spacecraft contractor study final presentations June 24
- ARRM Strategic international partner intent July 31
- ARRM KDP-B July 15
ARM Upcoming Events

• Omnibus BAA Synopsis Release (Appendices A & B)  July 26
• BAA: Investigation Team membership call  August 22
• BAA: Hosted Payloads  August 22
• ARM Industry Update  Sept. 1
• Target launch date moved one year later for the KDP-B estimates and assessment.

• The availability of the asteroid and SEP-based spacecraft for the crewed mission moved into 2026.
Asteroid Redirect Mission Alignment Strategy

### Asteroid Identification Segment
- **2014-2015**: NEO PS-2
- **2016-2017**: NEO WISE

### Asteroid Redirect Robotic Mission (ARRM)
- **2018-2019**: ATLAS
- **2020**: Final target selection
- **2021**: Mission launch & SEP demo (TBD)
- **2022**: Asteroid rendezvous & capture
- **2023**: Planetary Defense Demo
- **2024**: Asteroid redirected to lunar vicinity

### Asteroid Redirect Crewed Mission (ARCM)
- **2025-2026**: NEO

- **2022**: WISE
- **2023**: ATLAS
- **2024**: EM:
- **2025-2026**: EM: Crew to Asteroid

- **EFT-1**: First flight of Orion
- **EM-1**: Un-crewed Orion test beyond the Moon
- **EM-2**: Crew on Orion beyond the Moon
- **EM**: Crew to Asteroid
Human Space Exploration Phases From ISS to the Surface of Mars

Phase 0: Exploration Systems Testing on ISS

Phase 1: Cislunar Flight Testing of Exploration Systems

Phase 2: Cislunar Validation of Exploration Capability

Phase 3: Crewed Missions Beyond Earth-Moon System

Phase 4a: Development and robotic preparatory missions

Phase 4b: Mars Human Landing Missions

Today

Ends with testing, research and demos complete*

Asteroid Redirect-Crewed Mission Marks Move from Phase 1 to Phase 2

Ends with one year crewed Mars-class shakedown cruise

Planning for the details and specific objectives will be needed in ~2020

* There are several other considerations for ISS end-of-life

Mid-2020s 2030
Strategic Principles for Sustainable Exploration

• FISCAL REALISM: Implementable in the near-term with the buying power of current budgets and in the longer term with budgets commensurate with economic growth;

• SCIENTIFIC EXPLORATION: Exploration enables science and science enables exploration; leveraging scientific expertise for human exploration of the solar system.

• TECHNOLOGY PULL AND PUSH: Application of high Technology Readiness Level (TRL) technologies for near term missions, while focusing sustained investments on technologies and capabilities to address the challenges of future missions;

• GRADUAL BUILD UP OF CAPABILITY: Near-term mission opportunities with a defined cadence of compelling and integrated human and robotic missions, providing for an incremental buildup of capabilities for more complex missions over time;

• ECONOMIC OPPORTUNITY: Opportunities for U.S. commercial business to further enhance their experience and business base;

• ARCHITECTURE OPENNESS AND RESILIENCE: Resilient architecture featuring multi-use, evolvable space infrastructure, minimizing unique developments, with each mission leaving something behind to support subsequent missions;

• GLOBAL COLLABORATION AND LEADERSHIP: Substantial new international and commercial partnerships, leveraging current International Space Station partnerships and building new cooperative ventures for exploration; and

• CONTINUITY OF HUMAN SPACEFLIGHT: Uninterrupted expansion of human presence into the solar system by establishing a regular cadence of crewed missions to cislunar space during ISS lifetime.
ARRM Electric Propulsion Status

Demonstrated magnetic-shielding for a long-life, high-power, high-specific impulse Hall thruster

Electric Propulsion Subsystem Procurement

• Contract awarded April 2016
• 2,000 hours of planned wear testing on the technology development unit underway.
Capture Module Prototyping/Testing at Goddard Space Flight Center

7-DOF Robot "Capture" Arm
Restore common development robot arm with Robot Electronics Unit controller
Capture Module Prototyping/Testing at Goddard Space Flight Center
• Full scale welded metal prototype delivered
• Flat-floor testing of landing, extraction, and ascent underway
Technology Update

- Updated design of gripper, drill, drivetrain, and anchor
- Prototypes completed and tested with surrogate asteroid material

- Uses ~2000 independent hooks to opportunistically grip the surface
- Fast release capability
- Integrated rotary-percussive anchoring drill augments Microspine grip capability
- Design update from risk reduction
The acquisition strategy for the ARRM spacecraft leverages existing commercially available U.S. industry capabilities for a high power solar-electric-propulsion (SEP) based spacecraft for the agency's Asteroid Redirect Robotic Mission.

- Align with U.S. commercial spacecraft industry plans for future use of SEP
- Reduce costs and cost risk to ARRM

Strategy includes procurement of the ARRM spacecraft bus through a two-phase competitive process:

- Phase 1 four spacecraft design studies in progress
- Phase 2 competition for development and implementation of the flight spacecraft bus by one of the study participants

JPL selected four companies to conduct Phase 1: Lockheed Martin Space Systems, Littleton, Colorado; Boeing Phantom Works, Huntington Beach, California; Orbital ATK, Dulles, Virginia; and Space Systems/Loral, Palo Alto, California.
Procurement Process for ARRM Spacecraft

NASA Acquisition Strategy Meeting (August 2015) designated JPL as the lead for the Asteroid Redirect Robotic Mission (ARRM) and concurred with JPL’s plan to utilize its proposed two-step acquisition of a commercially-derived spacecraft.

Step 1 study contracts:

- Goals:
  1. To determine the current capability of commercial spacecraft providers as they apply to the ARRM mission.
  2. To influence the ARRM mission design based upon these capabilities such that the spacecraft can be procured at minimum risk and cost.
  3. To investigate how ARRM can align with and enhance future industry needs/opportunities.

- Vendors needed to have relevant capabilities:
  - Experience with large Geosynchronous or Deep Space vehicles
  - Experience with moderate-to-large Solar Electric Propulsion systems
Reduced Spacecraft Bus Implementation Risk

• The study phase of this two-step procurement has allowed for a far greater level of interaction with the potential spacecraft vendors than is possible via the normal acquisition process.

• The vendors have responded to the limited set of requirements they were given (a six-page spacecraft specification of functional requirements instead of the typical 100-page set).
  – This approach has not over-constrained their designs or unintentionally disallowed their heritage
  – Vendors are able to apply heritage (sometimes from multiple product lines) to address issues

• In turn, vendors have asked for more information in the areas where they need it.
  – In response, the ARRM project team has responded with additional documentation and changed some requirements

• The vendors have a much better understanding of what JPL is asking for and the result should be a much better proposal in Step 2
  – The typical scope revision that occurs shortly after issuance of a contract should be eliminated
  – Cost risk should be greatly reduced through this process
Progress on Spacecraft Studies

- Step 1 studies are underway
- Mid-term briefings completed
- There are significantly different designs from the different vendors
  - ARRM requirements and mission design allow adequate design flexibility (Goal 2)
- Vendors possess and are clearly applying significant commercial heritage (Goal 1)
  - Structures, Attitude Control, Computing, Power Distribution, Deployment Mechanisms, Propulsion
- Vendors are prepared to bring contributions (Goal 1,3)
  - New capabilities in electronics, fuel tanks, data interfaces, and solar arrays that were developed with internal corporate funds
- ARRM-derived designs are compatible with future commercial and Government applications (Goal 3)
  - Electric Propulsion Space Tug, Satellite Servicing/Refueling, Next-Gen Communication Satellites, and Deep Space Habitat and Cargo transportation
The FAST final report and complete set of public inputs was released on February 18, 2016 and can be found at: https://www.nasa.gov/feature/arm-fast

Report provides:

- Overview of the FAST effort.
- Responses to a set of high-priority questions that were derived from the ARRM engineering team’s risk analysis and needed to help design and develop the ARRM mission, spacecraft, and capture system.
- An initial list of potential investigations which could be performed by ARM (ARRM and/or ARCM).
- List of additional findings by the FAST in combination with public input.
- A summary of relevant public inputs and comments received by the deadline of December 4, 2015.
- Appendices with more detailed responses to each of the ARRM Project questions.

The Formulation Assessment and Support Team (FAST) identified ARM as a unique opportunity to gain a wide range of valuable knowledge beyond other asteroid missions or from what is available in current meteorite collections.
Provided information on the nature of the asteroid target and boulders.

Provided ideas and recommendations to enhance the scientific return, including a wide range of observations and instruments.
Planetary Defense Demonstration

• Enhanced Gravity Tractor (EGT)
  – Uses the mass of the collected boulder to augment the mass of the spacecraft and increase the gravitational attraction.
  – ARRM PD demo is moving toward a stationary in-line mode (vs. halo orbit) to decrease GN&C complexity and provide better range of mass applicability (boulder and asteroid) with the planned Ion Propulsion System (IPS) thruster throttle range.
  – Low IPS throttling requires additional thruster validation and verification and also adversely impacts the cathode lifetime.
  – In-line GN&C is simpler to implement, but will require large IPS gimbal range-of-motion.

• Small, but measurable deflection will be imparted on hazardous size asteroid
  – 2008 EV₅ deflection can be verified using ground-based radar
  – Other targets may require the Asteroid Redirect Vehicle (ARV) to loiter near the asteroid for deflection verification via differential ranging to the ARV

• Actual EGT planetary defense mission could collect much more mass, increasing the effectiveness of this technique.
Community Interest in Reference Target 2008 EV₅

- Significant interest from the science and small bodies communities with well-documented investigation opportunities.
  - Primitive, C-type (carbonaceous) believed to be rich in volatiles (water and organics).
  - Target of ESA’s Marco Polo-R candidate sample return mission (not selected by ESA).

- Largest asteroid with lowest mission ΔV (6.3 km/s) of the 1756 objects (as of 6/17/16) in the Near-Earth Object Human Space Flight Accessible Targets Study (NHATS) list.
  - Potential human mission target in late 2030s.
  - Possible target for resource collection.

- Large, hazardous-size asteroid provides a representative target for a planetary defense demonstration.
Enabling In-Space Resource Utilization

- Characterization and proximity operations with a human-class spacecraft provides relevant experience for future mining operations.
- Collection of surface regolith samples provides geological context of the boulder and allows correlation with Earth-based observations.
- Examples of opportunities for contributed payloads at the target asteroid:
  - CubeSat prospector technology demonstration
  - Material extraction or processing demonstrations
- The evolving asteroid mining community expects that material and operations provided by ARM will accelerate commercial asteroid mining efforts.

Image Credits: Deep Space Industries

Image Credits: Planetary Resources

Image Credits: Deep Space Industries
Scientific Characteristics of ARM Targets

• Preferred target asteroids are primitive C-type (carbonaceous) asteroids. The ARM reference target 2008 EV$_5$ is a C-type asteroid.

  – Primitive, C-type (carbonaceous) asteroids are believed to be rich in volatiles (water and organics). Carbonaceous meteorites found on Earth have up to 20% water by weight and are thought to be from C-type asteroids.

  – The study of these asteroids and their associated meteorites has the potential to further understanding of early Solar System evolution (volatile content, thermal history, etc.).

  – Carbonaceous asteroids may have played a key role in the formation of Earth’s atmosphere and oceans via delivery of volatiles. They may also have delivered key ingredients for the formation of life (astrobiology).

• The ARM reference target 2008 EV$_5$ appears to be representative of a distinct population of carbonaceous asteroids unlike other sample return targets

  – Opportunity to obtain a large multi-ton sample from a completely different parent carbonaceous asteroid (new world).

  – Furthers understanding of the variety of conditions and materials that existed among carbonaceous parent asteroids.
Asteroid Redirect Mission

SBAG supports and appreciates the continued engagement of the small bodies community by the Asteroid Redirect Mission (ARM), through mechanisms such as the recent Formulation Assessment and Support Team (FAST). **SBAG supports the plan as presented by the ARM team to create opportunities for hosted payloads on the ARM spacecraft and to have a competitively selected Investigation Team, both of which would maximize the science return of the mission.**
Two Upcoming Opportunities

• An announcement of payload opportunities for **PARTNER-PROVIDED SYSTEMS** to be hosted on the robotic ARM spacecraft.
  – Following a successful KDP-B, issue a broad solicitation as early as July 2016 for partner-provided payloads.
  – Intent is to provide the resources for payload accommodation with partners providing systems or instruments for investigations or experiments toward partner goals and awarded through no-cost contracts.

• An opportunity for **PARTICIPATION IN THE ARM INVESTIGATION TEAM (IT)**, which will be a multidisciplinary group of U.S. industry, academia, and government, and international members.
  – With an initial term of 3-5 years, the IT will assist with the definition and support of investigations for the robotic and crew segments of the ARM in the science, planetary defense, in-situ resource utilization, and capability and technology demonstration domains, as well as provide Program-level and Project-level support.
Call for Partner Provided Payloads

- Following a successful KDP-B, NASA plans to issue a call for partner provided payloads to fly on the ARM Robotic Segment
  - NASA will provide accommodations including integration with ARM robotic spacecraft
    - Expected total allocations of 50 Kg mass, 200 W operational/50 W keep-alive power plus 32 GB data storage and downlink
  - Partner provides payload consistent with ARM goals, objectives, robotic mission implementation schedules, interfaces and operations
    - Payloads assumed non-mission critical and may include remote sensing or deployable systems
    - Priority to payloads which may help characterize asteroid or otherwise reduce mission operations risks
  - Successful proposers will be offered representation on ARM IT

- Open to US Industry, academia, Government agencies, NASA Centers and International participation
Preliminary Prioritization for Hosted Payloads from FAST Findings:

- The following is a preliminary ARRM prioritization for hosted payloads based on inputs from the FAST.
  1. Thermal imager for basic boulder/regolith thermal inertia properties
  2. Multispectral imaging (broad-band filters) for morphology/composition
  3. Visible-near infrared spectrometer (~0.4-4 micron) to assess hydration
  4. Regolith geotechnical property estimation via surface interaction
  5. Acoustic/ultrasonic sensors to provide boulder characterization during and/or after collection during the ARRM
  6. Free-flyer camera and/or orbit determination asset
  7. Surface interaction deployable payloads
  8. In Situ Resource Utilization deployable assets/experiments
  9. Small carry-on impactor
  10. Global radar tomography and near-surface structure characterization with Ground Penetrating Radar

- Payloads that can perform multiple investigations and reduce risk for the ARRM are preferable.
ARM Investigation Team (IT)

• Two-phase solicitation for IT membership planned in coordination with call for hosted payloads on the ARRM.
  – Phase I solicitation planned for this July/August with announcements in March/April 2017 in coordination with selection of contributed payloads.
  – Phase II solicitation after selection of additional payloads and 6-month integration study.

• IT will participate in the definition, design, development, and operations phases of the ARRM with the goal of maximizing the probability of mission success and the knowledge return from the mission.

• IT will lead or assist in the implementation of mission investigations focused on the following four main areas as they support the robotic and crewed segment objectives:
  – Science
  – Planetary defense
  – Asteroidal resources and in-situ resource utilization (ISRU)
  – Capability and technology demonstrations

• NOIs and proposals will be submitted electronically via NSPIRES.
The IT will support ARM Program-level and Project-level functions, provide technical expertise, and support NASA Headquarters interactions with the technical communities.

Some examples of duties of the selected members:

- Provide technical expertise to ARRM and ARCM project teams to assist in the mission operations and vehicle development (e.g., NEA/surface/boulder properties estimates, test suites, support engineering, etc.) and extensibility/technology opportunities.
- Perform analyses to support spacecraft and mission design.
- Support concept of operations development for the mission and hosted payload suite, including characterization and boulder collection phases.
- Provide input into asteroid target selection.
- Provide inputs to planning and implementation of planetary defense demonstration.
- Provide inputs into post-capture characterization and EVA sampling and investigation sites (e.g., planning and procedures).
- Provide input for curation, contamination control, and the types of samples to be obtained.
- Support education and public outreach efforts.

Support for post-docs, research, travel planned.
Mission formulation spans NASA centers across the United States. Additional international collaborations are being developed.