

# On ESA Flight Software Testing and Its Independent Verification

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NASA IV&V Workshop

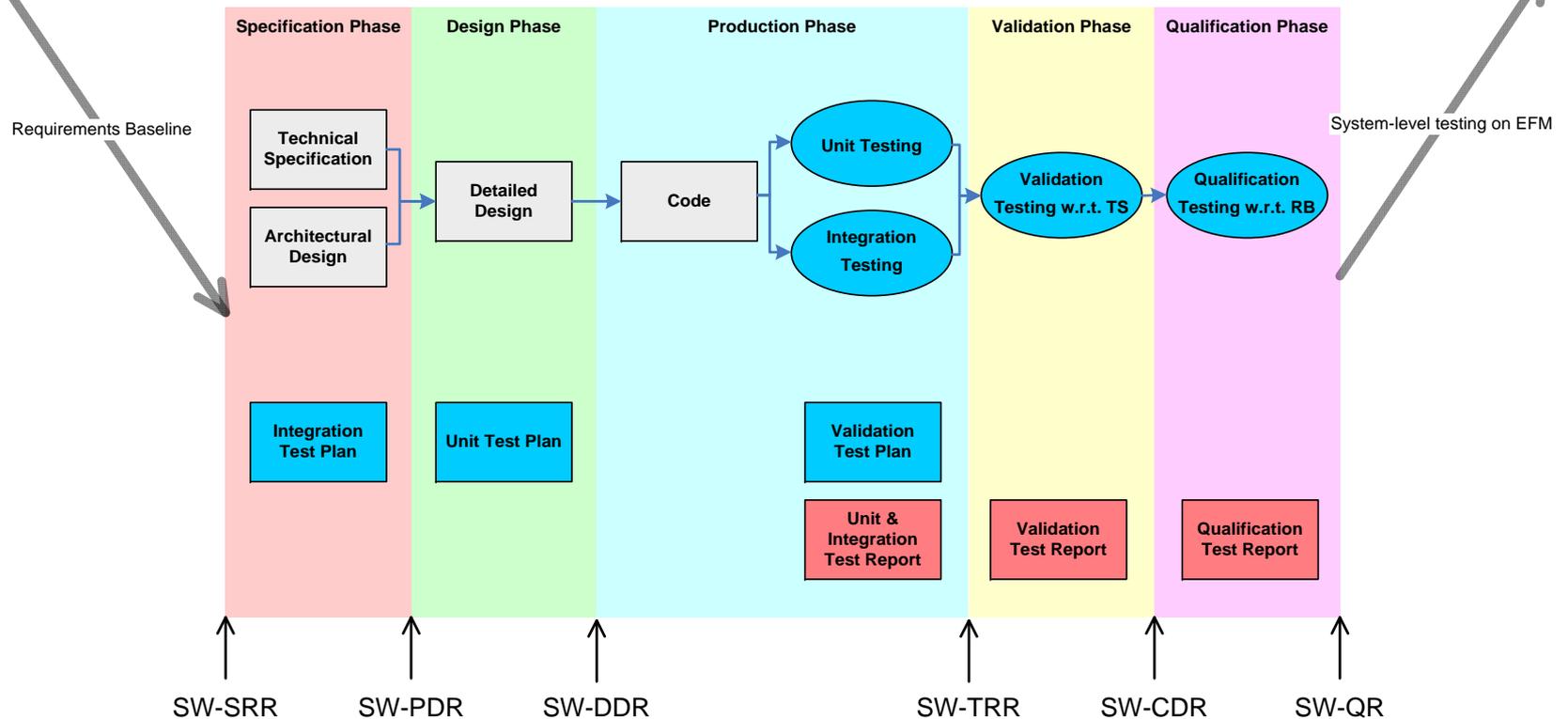
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- ❑ **ESA flight software development lifecycle and the role of testing**
  - ❑ Unit, integration and validation tests
  - ❑ Versioning approach to FSW development
  
- ❑ **Testing in the scope of ISVV**
  - ❑ Tests verification
  - ❑ Independent validation
  
- ❑ **ESA Avionics Test Bench**
  
- ❑ **Advanced techniques to reduce testing effort**
  - ❑ Automatic test generation
  
- ❑ **Conclusions**

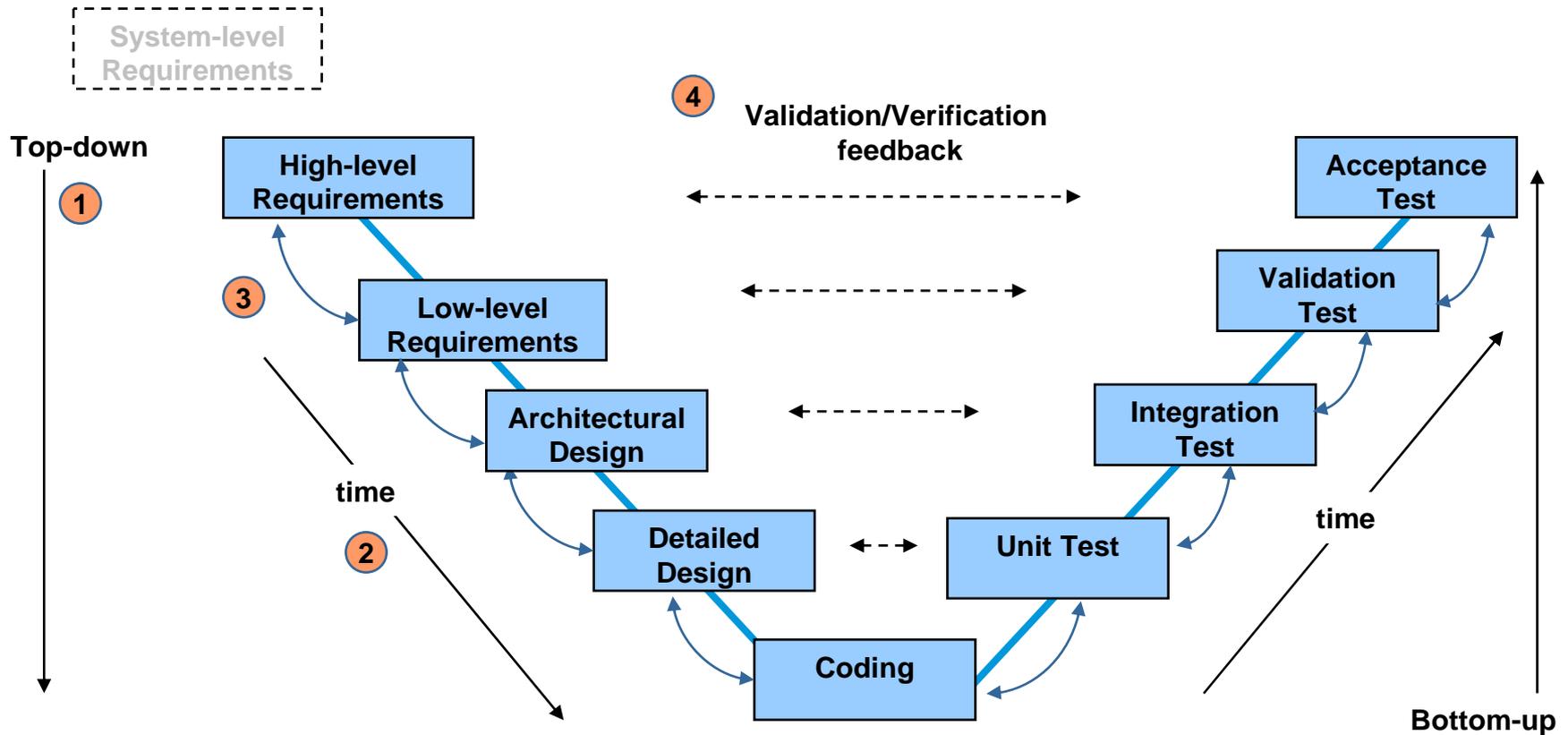
# PHASES OF FSW TESTING

System Engineering

Acceptance



# PHASES OF FSW TESTING: THE V MODEL



## Testing of individual software units

### ❑ Objectives: White box testing during code production phase

- 1) Exercise code using boundaries at  $n-1$ ,  $n$ ,  $n+1$  including looping instructions, while, for and tests that use comparisons
- 2) All the messages and error cases defined in the design document
- 3) Access to all global variables as specified in the design document
- 4) Out of range values for input data, including values that can cause erroneous results in mathematical functions
- 5) Software at the limits of its requirements (stress testing)

### ❑ Metrics:

Code coverage versus criticality	A	B	C	D
Source code statement coverage	100%	100%	0-100%	0-100%
Source code decision coverage	100%	100%	0-100%	0-100%
Source code modified condition and decision coverage	100%	0-100%	0-100%	0-100%
Object code coverage	100%	N/A	N/A	N/A

**Testing in which software components, hardware components, or both are combined and tested to evaluate the interaction between them**

❑ **Objectives:**

- 1) Problems in integrated FSW found early in the lifecycle
- 2) Functional interfaces between all integrated components are tested

❑ **Metrics:**

- 100% interface coverage for Cat A, B

## Testing w.r.t. Technical Specification and Requirements Baseline requirements

- ❑ **Objective:** Verify that all requirements are met
  
- ❑ **Metrics:**
  - 100% RB and TS requirements coverage
  - Traceability matrices between requirements and tests and vice versa

## ❑ Stress testing

- Evaluates a system or software component at or beyond its required capabilities (e.g. TM/TC frequency)
- Used at all levels (unit, integration, validation)

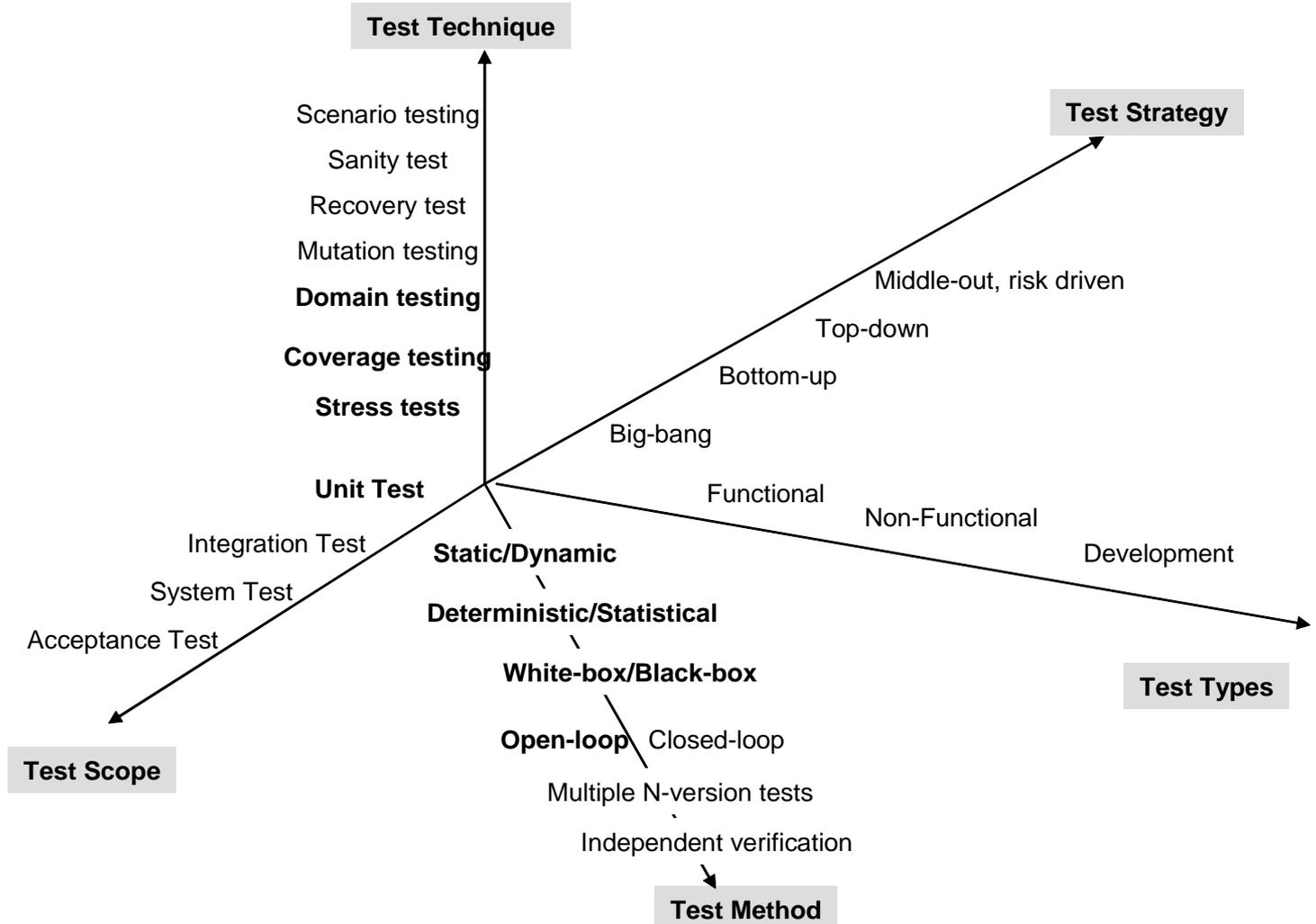
## ❑ Robustness testing

- Testing beyond specification (dependability, resource sharing, division by zero, pointers, run-time errors, etc.)
- Used at all levels (unit, integration, validation)

## ❑ Alternative verification means

- When full validation on the target computer is not feasible or where performance goals are difficult to achieve
- Only if it can be justified that validation by test cannot be performed, it is performed by either analysis, inspection or review of design
- Use static analysis for errors that are difficult to detect at run-time
- Use models & proofs
  - An example: Schedulability analysis
    - Mathematical model
    - Worst-case execution time (WCET)
    - Application of heavy stress scenario

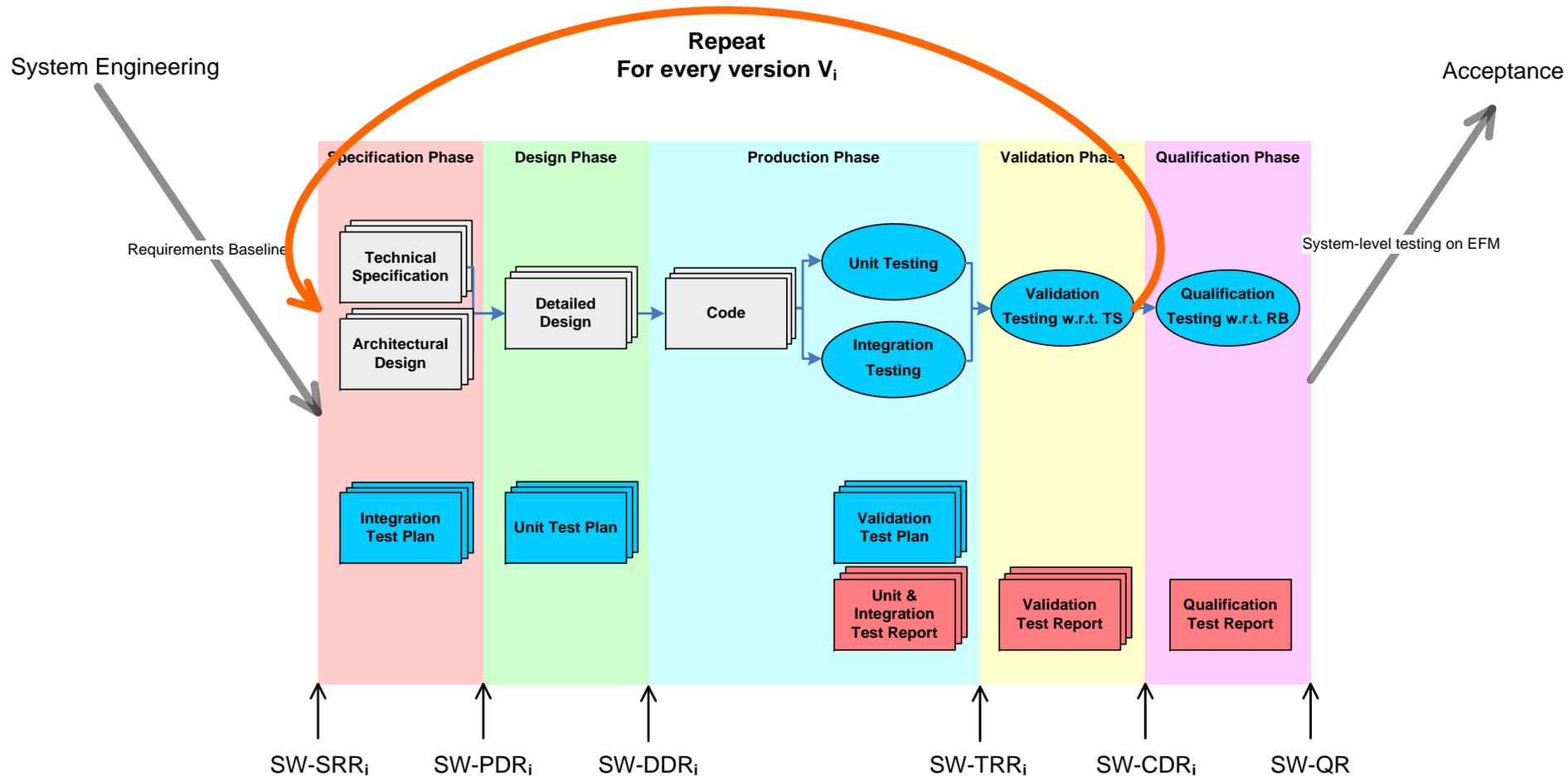
# DIMENSIONS OF TESTING



## Getting things more complicated: Versioning of FSW

- ❑ **Introduced to enable early releases of well-defined versions to AIT**
- ❑ **Incrementally built versions reflecting system development**
- ❑ **A typical FSW versioning:**
  - V1: System startup and initialisation, Basic Software (operating system, BSP, hardware drivers), elementary TM/TC handling, simple interaction with equipments (switch on/off), no AOCS
  - V2: Full data handling, AOCS Initial Acquisition Mode and Safe Mode allowing to perform close-loop tests, full interaction with equipment
  - V3: Full AOCS (all modes), FDIR, monitoring, reconfiguration sequences
- ❑ **Gains in the schedule of system development** due to parallelisation of AIT and further software development
- ❑ **But the software development effort increases!**
  - More versions, more test effort

# PHASES OF FSW TESTING REVISITED



## ESA is constantly re-assessing its processes, methods and standards...

- ❑ **Different approaches to testing for non-nominal software lifecycle**
  - E.g. to which extent is unit testing needed for reverse engineering?
  
- ❑ **Examples of non-agreed recent tendencies to relax the testing burden**
  - 1) **Achieving code coverage by a combination of validation and integration/unit testing**
    - Increases the risk that bugs are found too late
  - 2) **Interface coverage as part of the validation test campaign**
    - Two executions of the test suite would be necessary (code instrumentation)
    - Too late, defeats one of the two objectives of integration tests
    - Not testing interfaces & out-of value parameters which are avoided by nominal operational scenarios (reflected in validation tests)
    - What in case that software is patched or nominal behaviour is changed?
  - 3) **Very limited or no integration tests**
    - Unit tests cover all interfaces anyway?

**Unit tests also examine all interfaces & out-of-boundary values, so why repeating this effort again?**

❑ **Consider the following examples:**

Assumptions:

- Unit X is unit tested, 100% code coverage
- Unit Y is unit tested, 100% code coverage
- Unit Y is integrated with unit X

1) Missing functionality example

- Y requires one more interface, not provided by X!

2) Mismatch in communication protocol example

- X expects `init()` called first, and other methods called only after this
- However Y does not call `init()` before calling other methods!

❑ **Bottom line: Integration tests also address interface completeness, semantics and the protocols of interactions between units**

# PROJECT-SPECIFIC TESTING PLATFORMS



Testing facility	CPU	Central flight computer	Equipment & Environment	Loop	Applicable Tests
<b>Unit Test Bench</b>	Simulated	Simulated	Simulated	N/A	Unit testing
<b>Software Validation Facility</b>	Simulated	Simulated	Simulated	Open Closed	SW-SW integration testing Validation testing AOCS
<b>Software Test Bench</b>	Target Processor	Breadboard	None Simulated	Open	SW-HW integration testing Validation testing AOCS
<b>Functional Validation Test Bench</b>	Simulated	None	Simulated Models	Closed	AOCS
<b>Electrical Functional Model</b>	Real	Real	Simulated Models	Closed	Qualification testing Acceptance testing

- ❑ **Electrical Functional Model (EFM) is further used for System-level testing**
  
- ❑ **Operations validation is performed at system level after the software acceptance**
  - An option is an early System Validation Test with an earlier version of software

## There are several activities dedicated to testing defined in the ESA ISVV Guide

### ❑ Independent Test Verification

- Integration Test Specification and Test Data Verification
  - “Integration testing” here includes validation testing
- Unit Test Procedures and Test Data Verification

### ❑ Independent Validation

## ❑ Integration Test Specification and Test Data Verification

- Verify consistency with Technical Spec and Architectural Design
- Verify Test Procedures correctness and completeness (are tests testing what they are expected to test?)
- Verify models
- Test reports verification

## ❑ Unit Test Procedures and Test Data Verification

- Verify consistency with Detailed Design
- Verify Test Procedures correctness and completeness (are test testing what they are expected to test?)
- Test reports verification
- Verify models (automatic code generation)

## □ **Tasks:**

- Identification of test cases
  - Complementary tests
  - Non-nominal test scenarios (e.g. FDIR, worst case)
- Independent validation test plan, test procedures, test scripts
- Independent validation test execution
- Independent validation test reports
- Analyse failed test cases

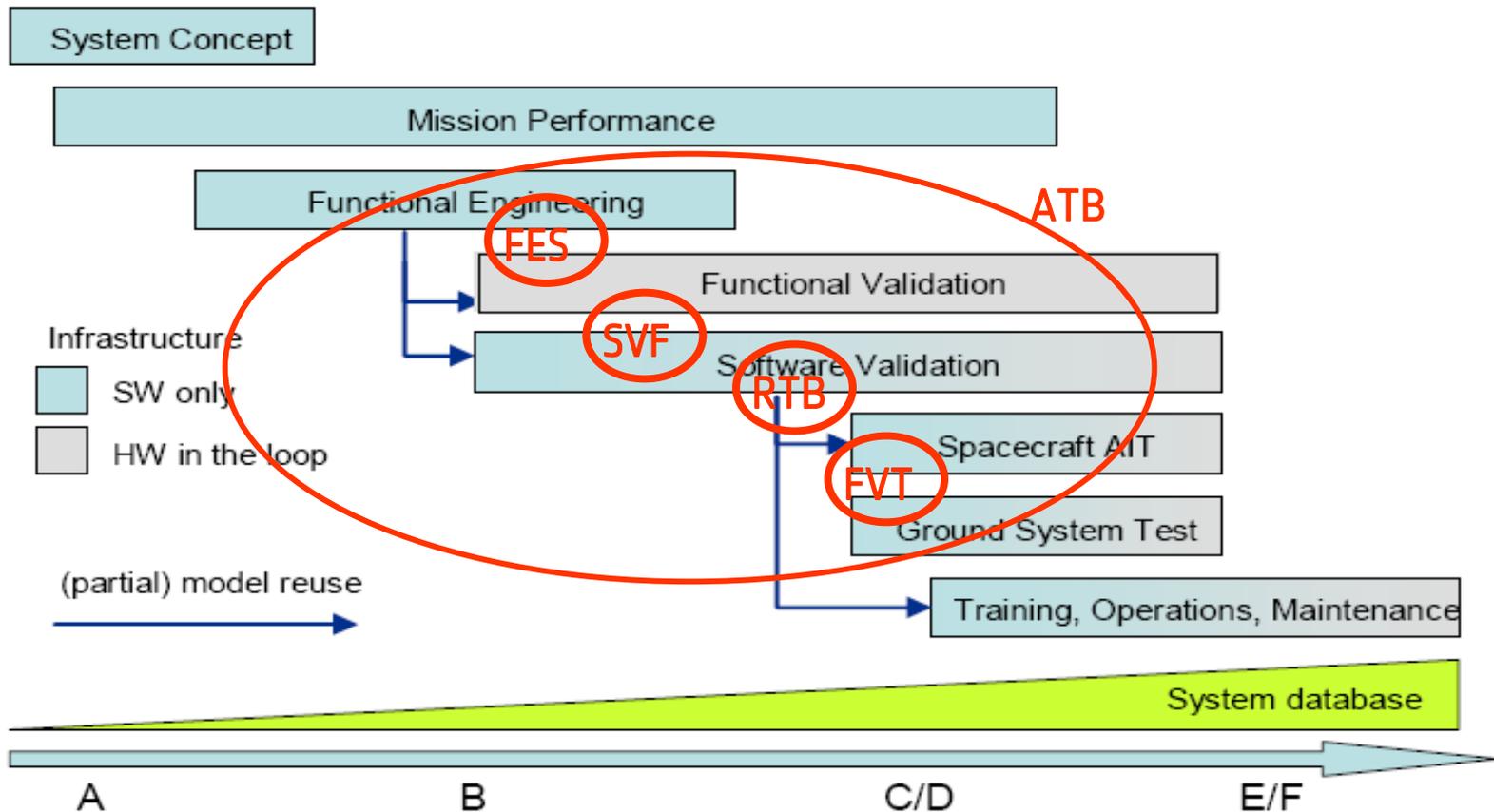
## □ **Platform used for independent validation: Software Validation Facility**

- Nominal project-specific SVF
- A generic SVF

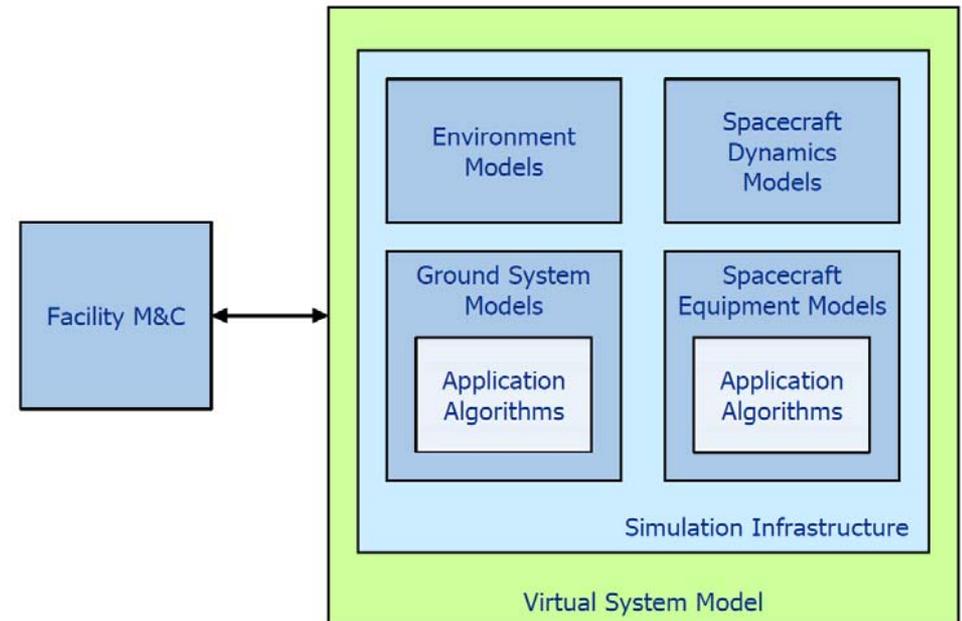
# ESA AVIONICS TEST BENCH (ATB)



The ATB is an open facility that allows for technology and standards assessment, validation and demonstration in simulated environment



- ❑ **Verification of critical elements of baseline system design**
- ❑ **Building blocks:**
  - Models of sensors/actuators, spacecraft kinematics, dynamics and environment
  - Simulation infrastructure (e.g. Matlab/Simulink, Eurosim)
  - Monitoring and control
- ❑ **Use cases:**
  - Mission analysis verification
  - (Sub)system verification
  - Equipment modelling
  - Functional modelling
  - Demonstration of autonomous mission management



# FUNCTIONAL VALIDATION BENCH



## ❑ Test bench for hardware and software prototyping

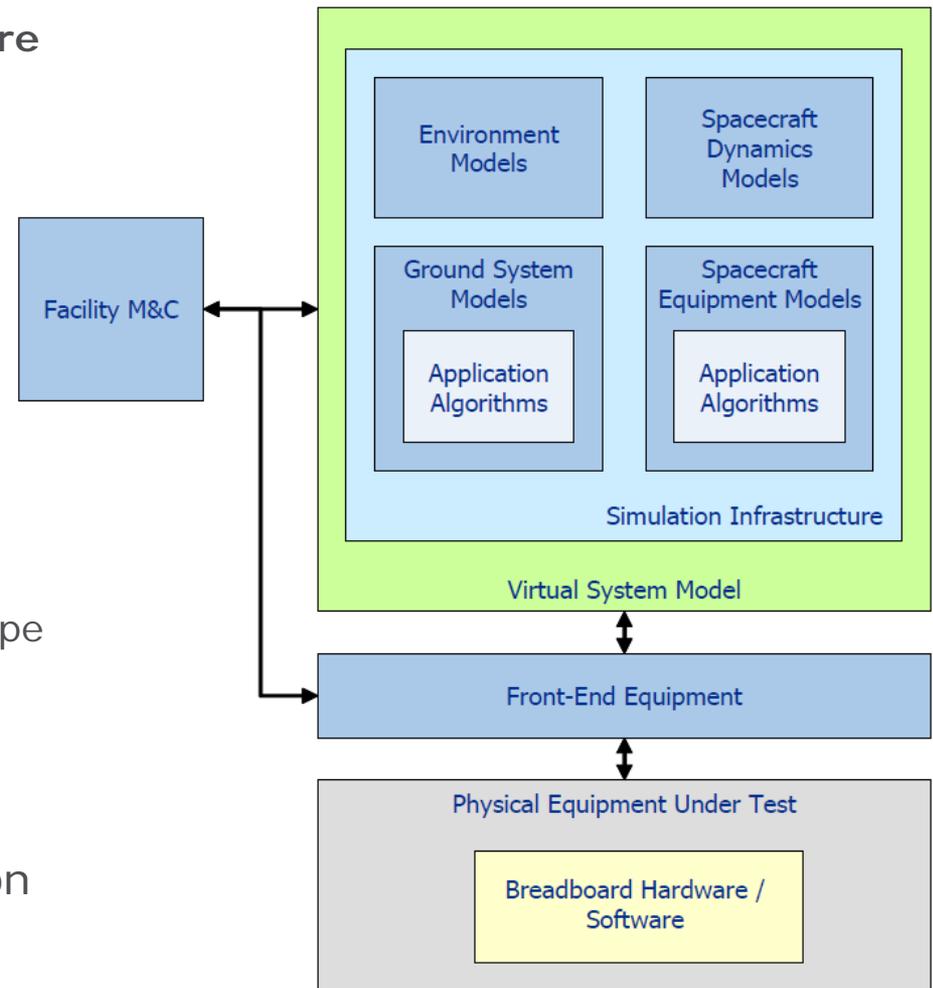
- Performance analysis and validation of critical elements / subsystems (e.g. AOCS)

## ❑ Building blocks:

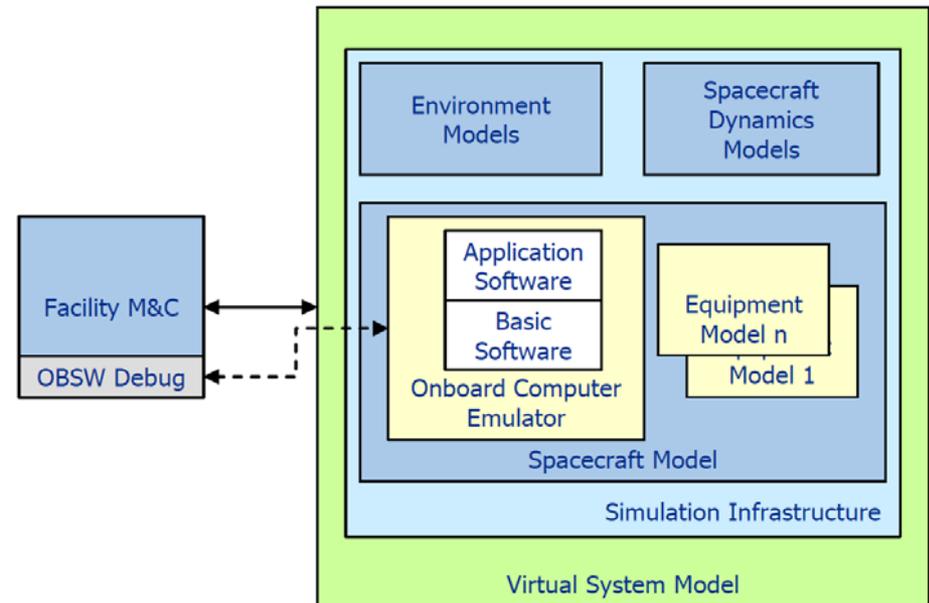
- Simulation infrastructure
- Monitoring and Control
- Functional models
- Models for protocols and electrical interfaces
- Support equipment for prototype / breadboard under test
- Prototype / breadboard under test

## ❑ Use cases:

- Complete system simulation with prototyped / breadboarded elements



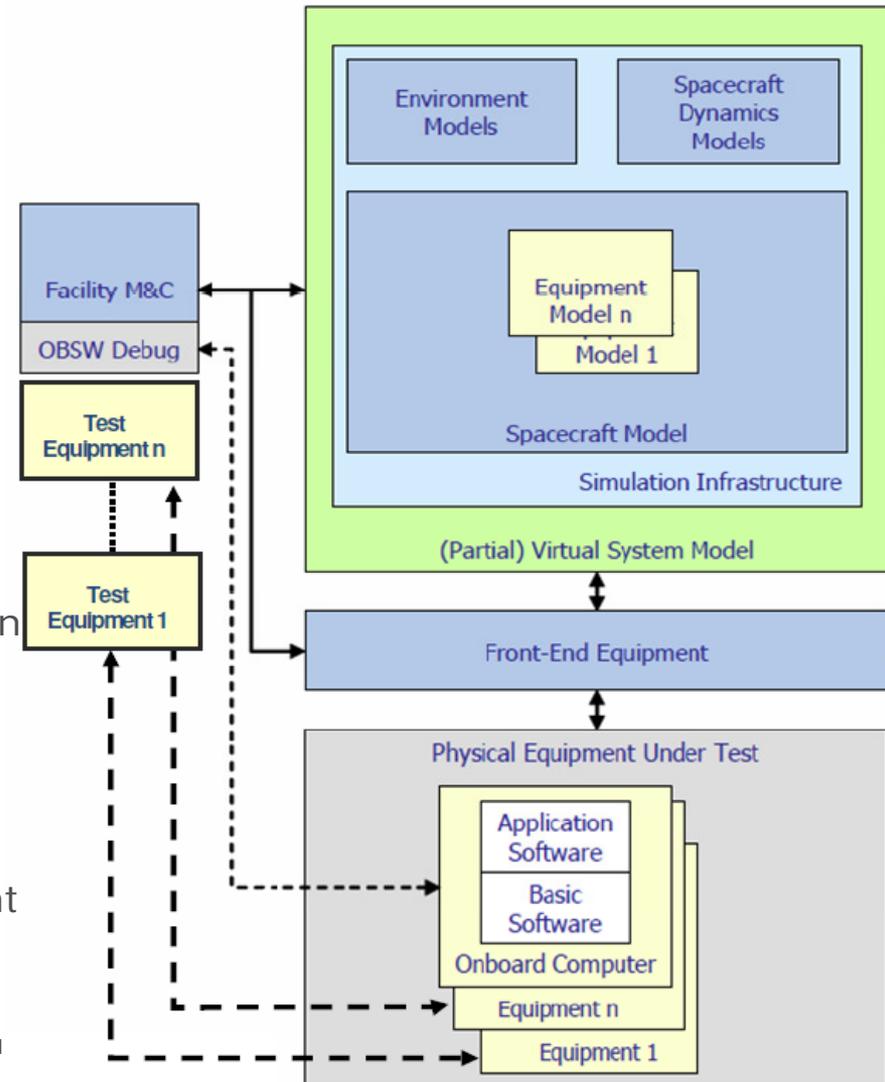
- ❑ **Validation of flight software in context**
  
- ❑ **Building blocks:**
  - Functional models
  - Simulated TM/TC interface
  - Fully functional simulation of HW, including performance and dynamic behaviour
  
- ❑ **Use cases:**
  - Verification of transitions between FSW modes
  - FDIR design verification
  - FSW validation including regression testing
  - ISVV



# REAL-TIME TEST BENCH

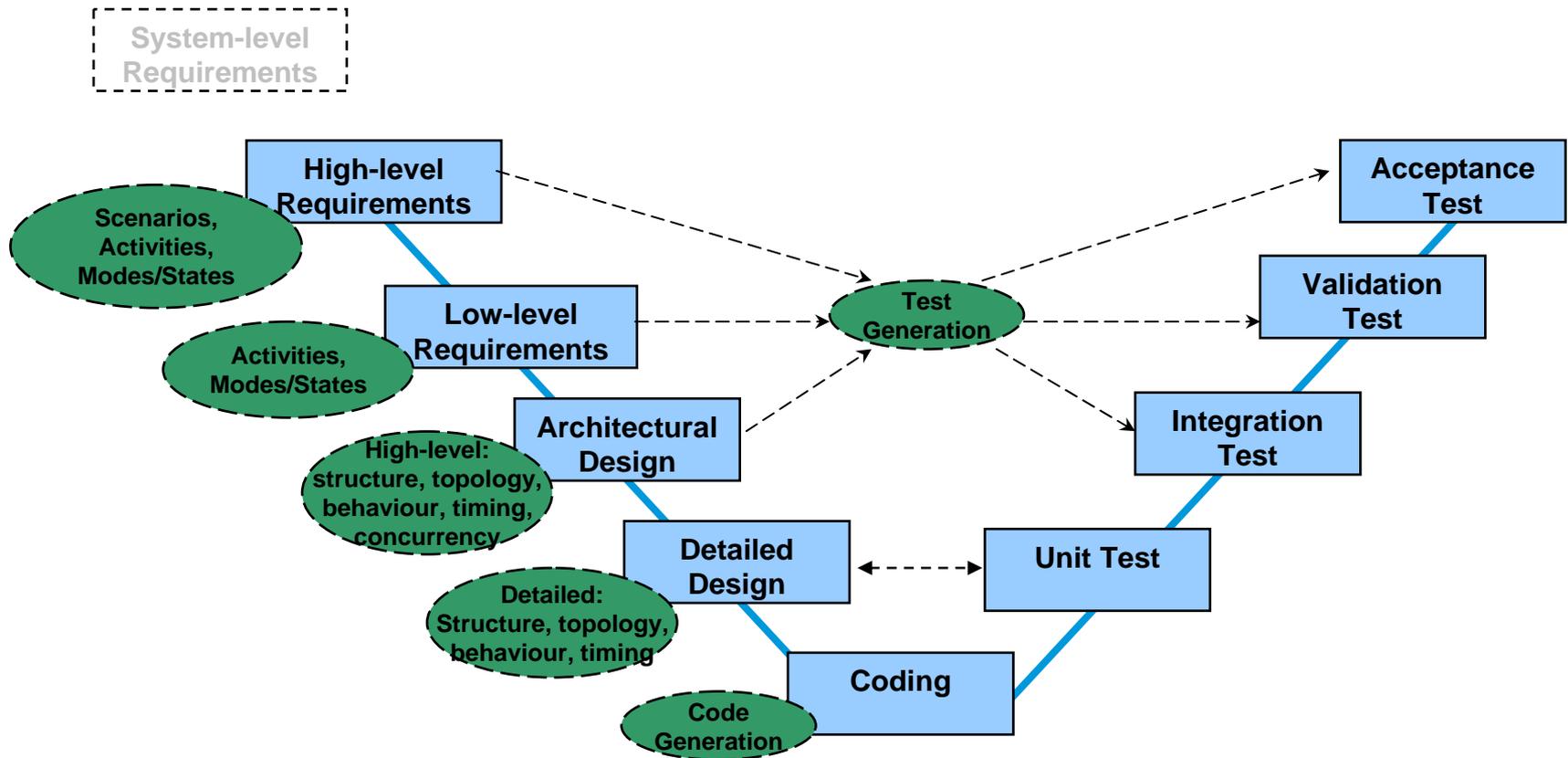


- ❑ **HW in the loop testing**
- ❑ **Building blocks:**
  - Functional models
  - On boards computer
  - Simulated TM/TC link
  - Representative TM/TC reference facility (SCOS2000-based)
  - Simulated sensors/actuators connected by MIL-STD-1553 bus
- ❑ **Use cases:**
  - HW–SW integration and verification
  - HW function verification
  - System validation with HW in the loop
- ❑ **An extension: End-to-End ATB**
  - Database to store/retrieve different configurations



- ❑ **Careful risk assessment**
  
- ❑ **Correctness-by-design & formal methods**
  - Especially for extra-functional properties
  - In some cases this is already in use (e.g. schedulability analysis – mathematical model rather than testing)
  
- ❑ **Testing logistics**
  - Configurable test benches
  - Avionics Test Bench
  
- ❑ **Saving time/effort by limiting the involvement of humans**
  - Automatic test case identification
  - Automatic test generation
  - Automatic test execution
  - Automatic test results analysis
  - Automatic test report generation

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- ❑ Consistency between models
- ❑ Multi-formalism, model translations (horizontal)
- ❑ Multi-formalism, model refinement (vertical)
- ❑ Code w.r.t. the higher-level models

## Key idea: Automate test design by creating a system model and generate integration test cases out of behaviour diagrams

- ❑ **System model:** Formal modelling of requirements facilitates V&V of their completeness and correctness early on in the software development lifecycle
- ❑ **Test case identification and development:** Automatic test case generation from the test models improves efficiency and effectiveness of test case identification and development, possibly with improved test coverage
- ❑ **Testing automation:** Automatic test execution makes the testing phase more efficient, with support for test re-execution and regression testing
- ❑ **Quality assurance of the test model:** Static analysis, inspection, simulation
- ❑ **Improved traceability** from system requirements through the system model and test model to test cases and their results
- ❑ **Applicability to nominal as well as Independent V&V**
- ❑ **Weaknesses:** Not capturing of dynamic behaviour (e.g. interrupts)

- 1) **System modelling:** Model the System under Test (SUT) and model the environment around the SUT
  - Sequence-based specification
  
- 2) **Abstract test generation:** Use system models to generate abstract tests
  - I.e. tests which are independent of both the test programming language used and the test environment
  - Statistical operational testing
    - Markov chain usage model
  
- 3) **Concrete test generation:** Make tests executable
  
- 4) **Test execution:** Execute the tests on the SUT respecting the pass/fail criteria
  
- 5) **Test results analysis**

## Key idea: Generate some units tests directly from the source code (instead from the specification)

### ❑ Not the same as static analysis!

- Two ways of detecting a deviation
  - Rule-based: violation of a rule
  - Symptom based: observation of a symptom (e.g. unhandled exception)
- Static analysis cannot detect unanticipated faults

### ❑ Expected benefits:

- Source code provides a good start to generate the test environment
- Stimuli are generated from the source code (path-driven stimuli identification)
- Comparison of the stimuli set to the specification
  - Specification maybe incomplete if the code represents what is really needed (or vice versa)
- Easier finding of test cases to achieve full code coverage (e.g. MC/DC)

- ❑ **Apply to real FSW of existing ESA satellite (cat B criticality)**
  - Model-based testing: Focus on integration/validation testing
  - Source code-based testing: Focus on unit testing
  
- ❑ **Compare with manual testing**
  - Cost/effort
  - Regression testing
  - Measure efficiency to find bugs
    - Fault injection
    - Comparison with FSW version with known bugs
    - Finding bugs which no other method found
  - Coverage metrics
  - Sensitivity to particular fault types?
  - Nominal or complementary method?
  
- ❑ **Assess applicability to our domain**
  - Full spacecraft FSW
  - Only some subsystems (especially model-based testing)
  - Nominal and independent V&V
  
- ❑ **Assess scalability**

- ❑ **Testing of FSW takes a lot off time/effort/cost**
  - Testing at different levels serves its purpose and it is very sensitive to reduce the effort by re-defining the objectives and/or combing more levels
  - Software lifecycle change may cause testing effort to increase (e.g. versioning)
  
- ❑ **Advanced methods are being developed to**
  - Substitute testing by other means (e.g. proofs)
  - Reduce testing effort
  - Increase testing efficiency
  - Increase confidence in testing
  
- ❑ **ESA is exploring several ways of reducing this effort**
  - Constantly assessing existing processes
  - Finding innovative methods
    - SW development (models)
    - V&V (formal proofs)
    - Testing
  - Applicable both to nominal and independent V&V

**THANK YOU**

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