

STS-115/12A

FD 04 Execute Package



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035	14 - 15	FD04 Mission Summary (pdf)
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032	21 - 23	1.301 AVU SPEE CAM P3 Truss Install (pdf)
037	24 - 25	Loose SARJ Thermal Blanket Troubleshooting (pdf)
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Approved by FAO: M. Scheib

Last Updated: Sep 12 2006 4:13AM GMT

JEDI (Joint Execute package Development and Integration), v2.04.0003

REPLANNED

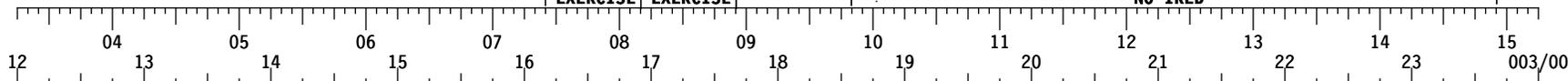
NO ERGOMETER
EXERCISE EXERCISE

09/11/06 21 NO PRCS
NO TRED

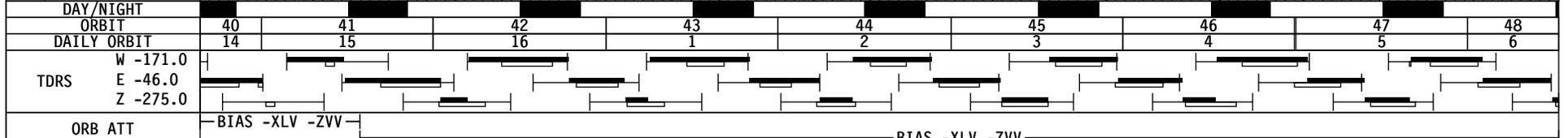
FD04

GMT 09/12/06 (255)

MET Day 002



STS-115	CDR JETT	SLEEP	POST SLEEP	PMCA/G	HYG BRK/HATCH CLS	CAMPOUT EVA PREP	EMURGE	EMU PRE BREATHE	C_LK DPRS	P	P/TV07 EVA OPS	MEAL	P/TV07 EVA OPS	P/TV 07 EVA OPS	EXERCISE						
	PLT FERGUSON	SLEEP	POST SLEEP			EMURGE	SSAS NOM MATE-3 BLT	2-CAP	1-CAP	INAHDI B	CNCT #2	AVU*	AVU*	FES*	CWC XFER (2)	MEAL	EXERCISE	FES*	FES*		
	MS1/EV1 TANNER	SLEEP (IN ISS A/L)	POST SLEEP	HYG BRK/PREBREATHE	A/L REPI	10.2 DPRS	CAMPOUT EVA PREP	EMURGE	EMU PRE BREATHE	C_LK DPRS	P	EVA 1 DEGRESS/SETUP	CNCT P1/P3 LWR UTILITY	RLS AFT BGA RSTRNT	ROTAT	UNSTW AFT SABB	SARJ PREP	CNCT P1/P3 UPR UTILITY	SARJ PREP	ECVLAN 1UP	
	MS3/EV2 PIPER	SLEEP (IN ISS A/L)	POST SLEEP	HYG BRK/PREBREATHE			CAMPOUT EVA PREP	EMURGE	EMU PRE BREATHE	C_LK DPRS	P	EVA 1 DEGRESS/SETUP	RLS AFT SABB RSTRNT	RLS FWD SABB RSTRNT	RLS FWD BGA RSTRNT	UNSTW FWD SABB		SARJ PREP		ECVLAN 1UP	
	MS2/EV3 BURBANK	SLEEP	POST SLEEP		BPSMU AUDIO CONFIG		SRMS EVA1 SUPPORT		PTV07 S/U	C_LK DPRS	P	EVA1 IVA SPPT									
	MS4/EV4 MACLEAN	SLEEP	POST SLEEP		SSRMS P3/4 PRE INSTL	POST SLEEP	MNVR TO RTL	LIMP		UGRPL	SSRMS UNGRPL TO UNLT	PST DK EVA XFR/RCNFG	REBA C/O	EVA 2 PREP TOOL CNFG	E_LK PREP	MEAL	EXERCISE		PS RLEE P		
ISS	ISS CDR	SLEEP - ISS	POSTSLEEP-ISS	PREP WORK	UOP P/D	ПЛЛЕ DWLDPLCMN	ПЛЛЕ	HD DNDLD	TVIS	HD DNDLD	COX MNT	MIDDAY-MEAL	HD DNDLD	XFER TK STOW	DPART PREP	TVIS					
	FE-1	SLEEP - ISS	POST SLEEP-ISS	PREP WORK	SSRMS P3/4 PRE INSTL	HAM*	MNVR TO RTL	1-CAP	2-CAP	SSAS NOM MATE-3 BLT	UGRPL	SSRMS UNGRPL TO UNLT	AVU*	AVU*	MIDDAY-MEAL	IWIS*	DPART PREP	TVIS	IMS EDIT	CEVIS	
	FE-2	SLEEP - ISS	POST SLEEP-ISS	HYG BRK/HATCH CLS	PW	CAMPOUT EVA PREP	EMURGE	EMU PRE BREATHE	C_LK DPRS	P	TVIS	MIDDAY-MEAL	EVA 2 PREP TOOL CNFG	E_LK PREP	CEVIS						



ORB ATT -BIAS -XLV -ZVV

NOTES

*COMM Config *TGT ACQ *ALC-CDR-REPLC *REPRESS INIT *DUMP INIT *DEACT @CMS-TVIS WEEKLY-MNT

*PWRUP *CLNUP *DUMP TERM

*CAM DISASSEMBLY *PWRDN #HTR Deact

1 MSG INDEX

2

3 MSG NO. TITLE

4 27 Contingency P3/4 Installation with Failed SSAS RTL (13-1186)

5 28 Contingency RTL Loss After WR Limping (13-1187)

6 32 1.301 AVU SPEE CAM P3 Truss Install (13-1194)

7 34 FD04 Flight Plan Revision

8 35 FD04 Mission Summary (13-1195)

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10 37 Loose SARJ Thermal Blanket Troubleshooting (13-1196)

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14 41 FD4 Lighting Predicts for AVU Operations (13-1199)

15 42 Robotics Procedure Update Rationale for FD04

16 43 BPSMU Audio Config

17 44 FD04 MMT Summary

18

19 1. We have made some minor changes to the P3/P4 Activation procedure. The updated
20 pages are in message 39 and have revision marks so you can see the deltas. The four
21 updated pages can be swapped one-for-one with the existing pages in the Assembly
22 Ops book. On the ISS side, the entire procedure has been updated electronically in IPV.

23

24 2. Robotics procedure 1.301 AVU SPEE CAM P3 TRUSS INSTALL (MSG 032 /13-
25 1194) has been updated and two Robotics procedures (MSG 027 /13-1186 & 028 /13-
26 1187) have been created for use in the event of specific SSAS RTL failures during P3/P4
27 Install. Details regarding these updates and additions can be found in MSG 042.

28

29 3. The MCC has remotely uplinked the latest HazMat data file to STS3 (KFX) and STS4
30 (DOUG). HAZMAT is currently only up to date on those laptops. The other laptops will
31 be updated from the ground once they are back on the network.

32

33 4. A BPSMU Audio Workaround procedure has been uplinked as Msg 043. MS2 has been
34 scheduled at MET 2/14:35 and is expected to take 45 min.

35

36 5. P1P3 SSAS: The P1P3 SSAS Prep for Mate was completed nominally by OSO
37 overnight. Additionally, the P1P3 MBAs (1, 2, 3 and 4) have been pre-heated in the
38 event they are needed for EVA Manual Override.

39

40 6. REPLACE PAGES 3-32 THROUGH 3-41.

41

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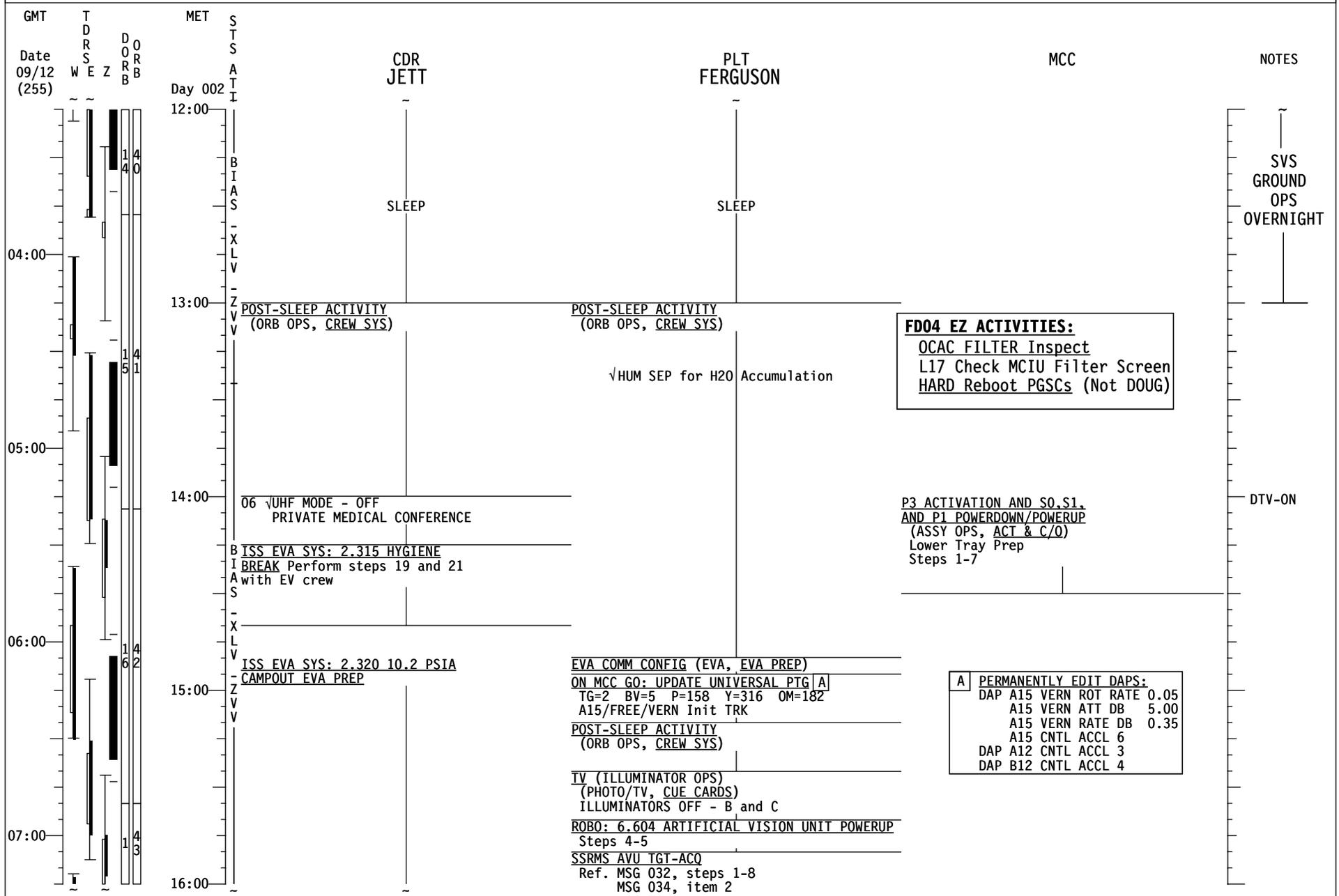
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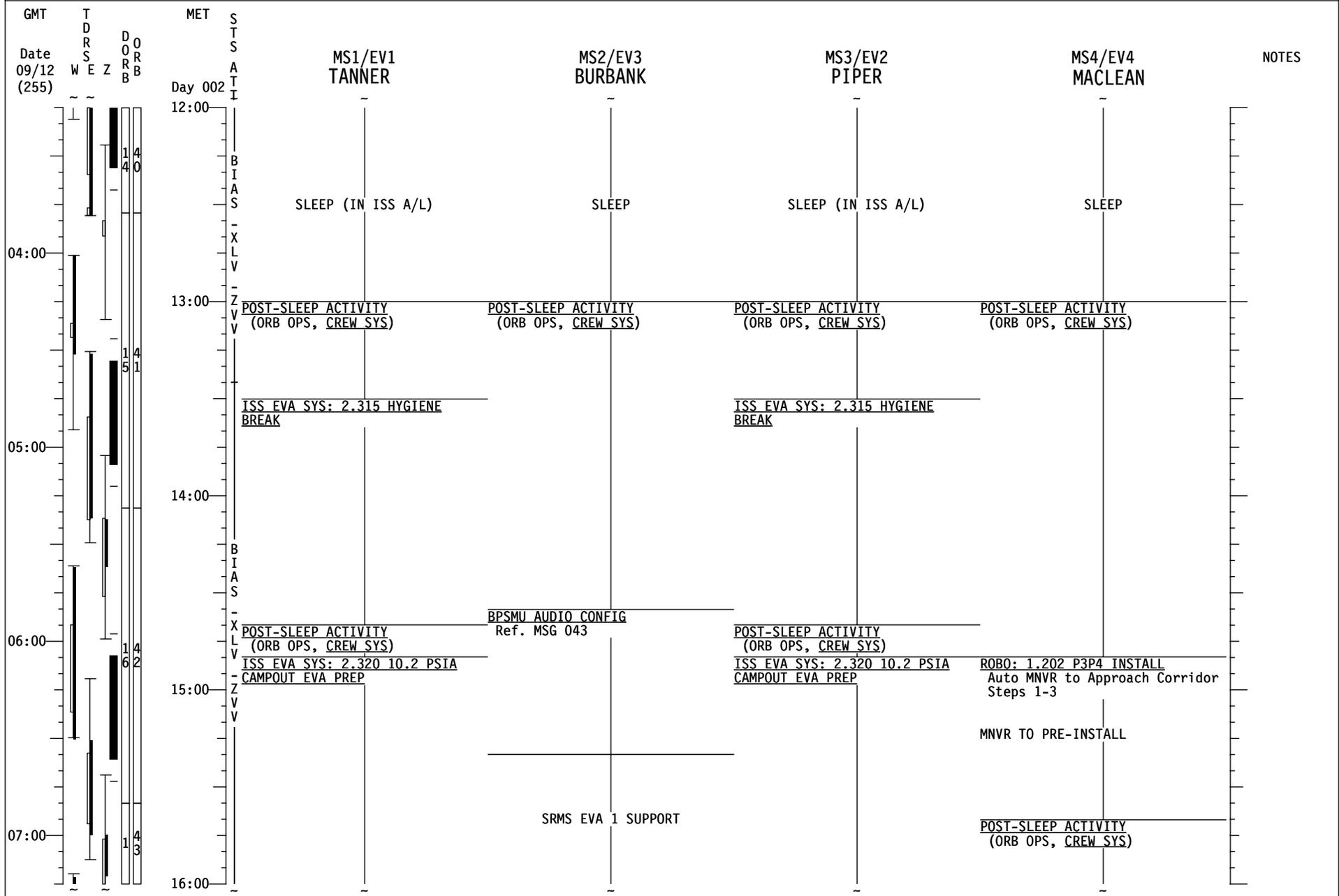
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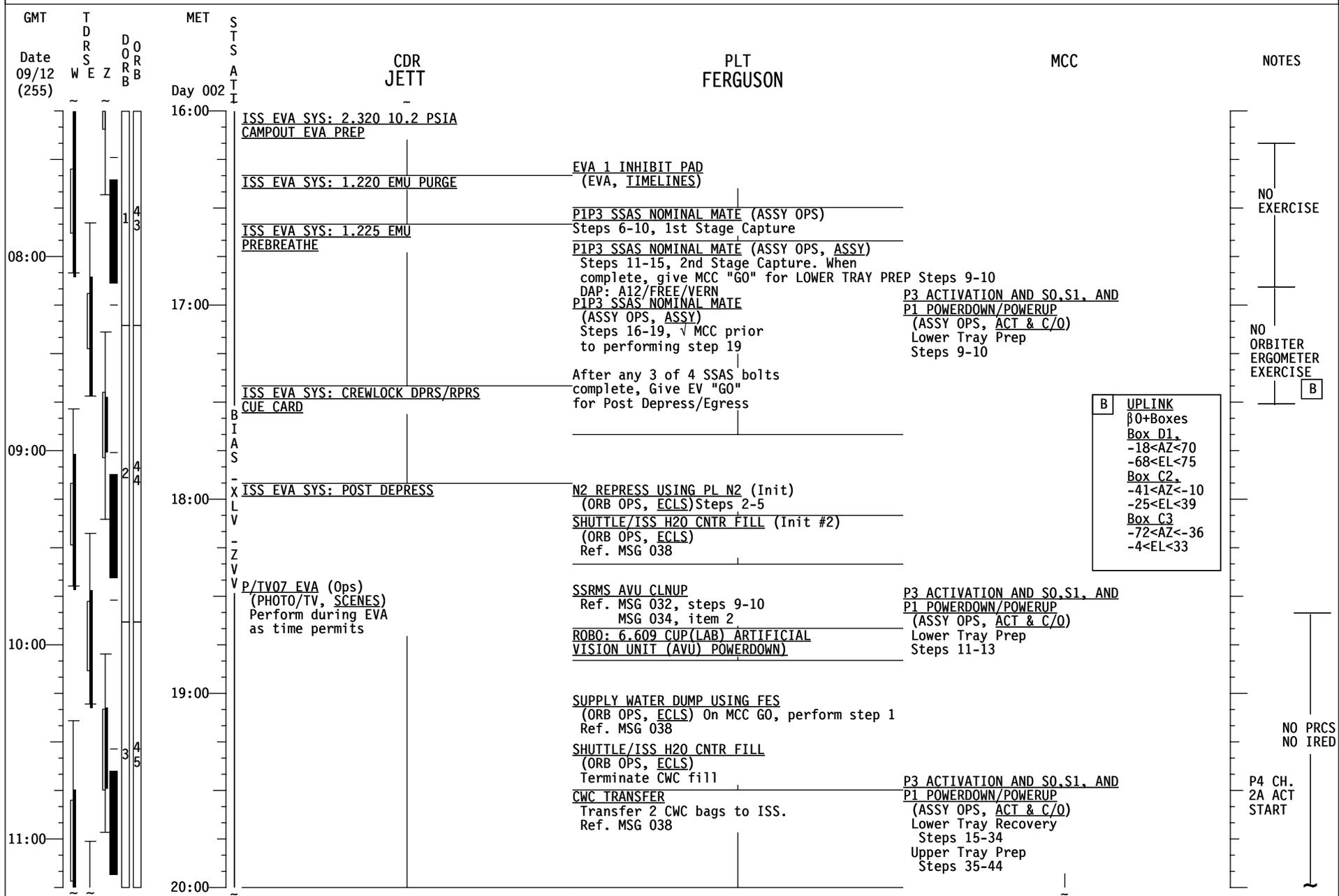
STS-115 (FD04)



STS-115 (FD04)



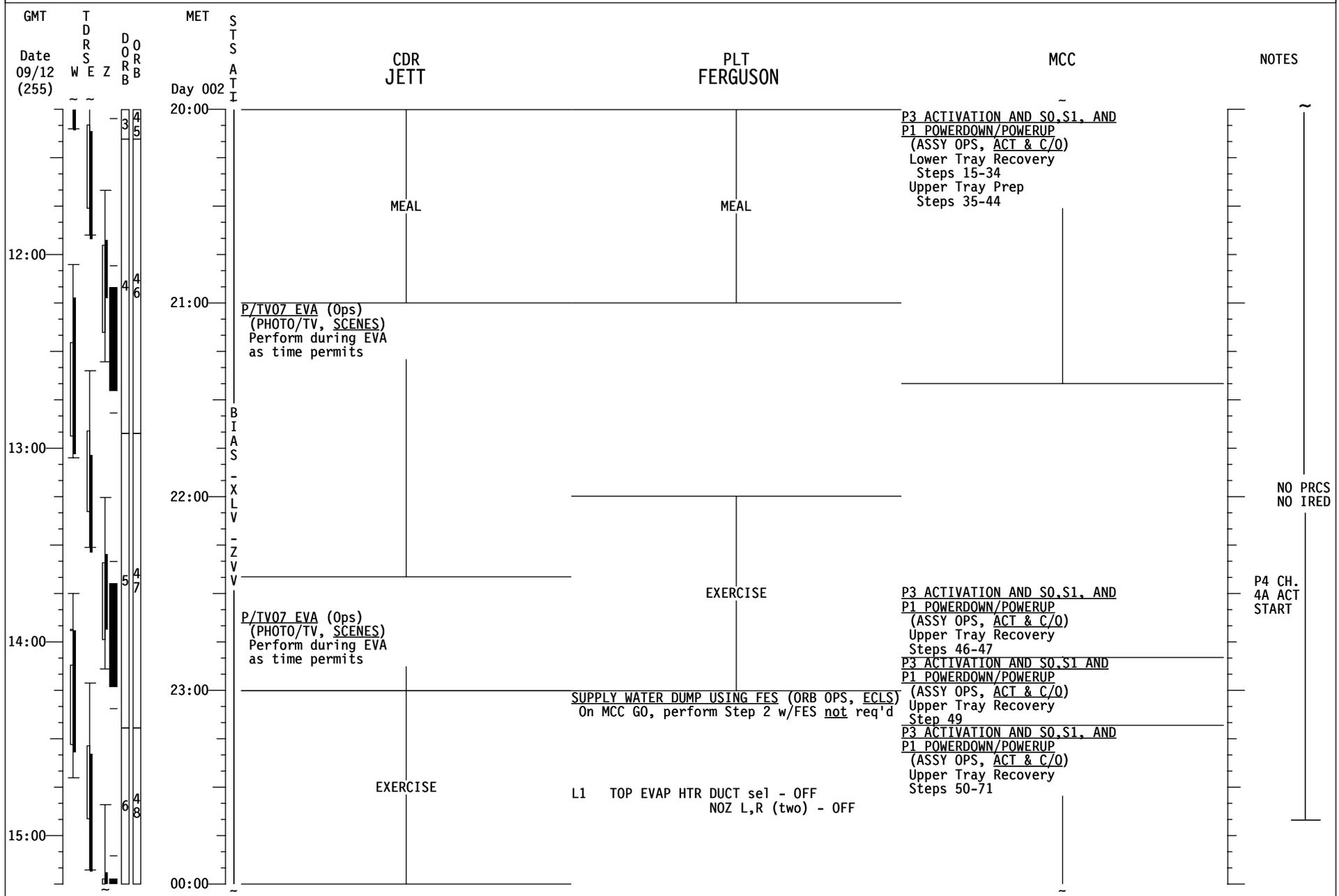
STS-115 (FD04)



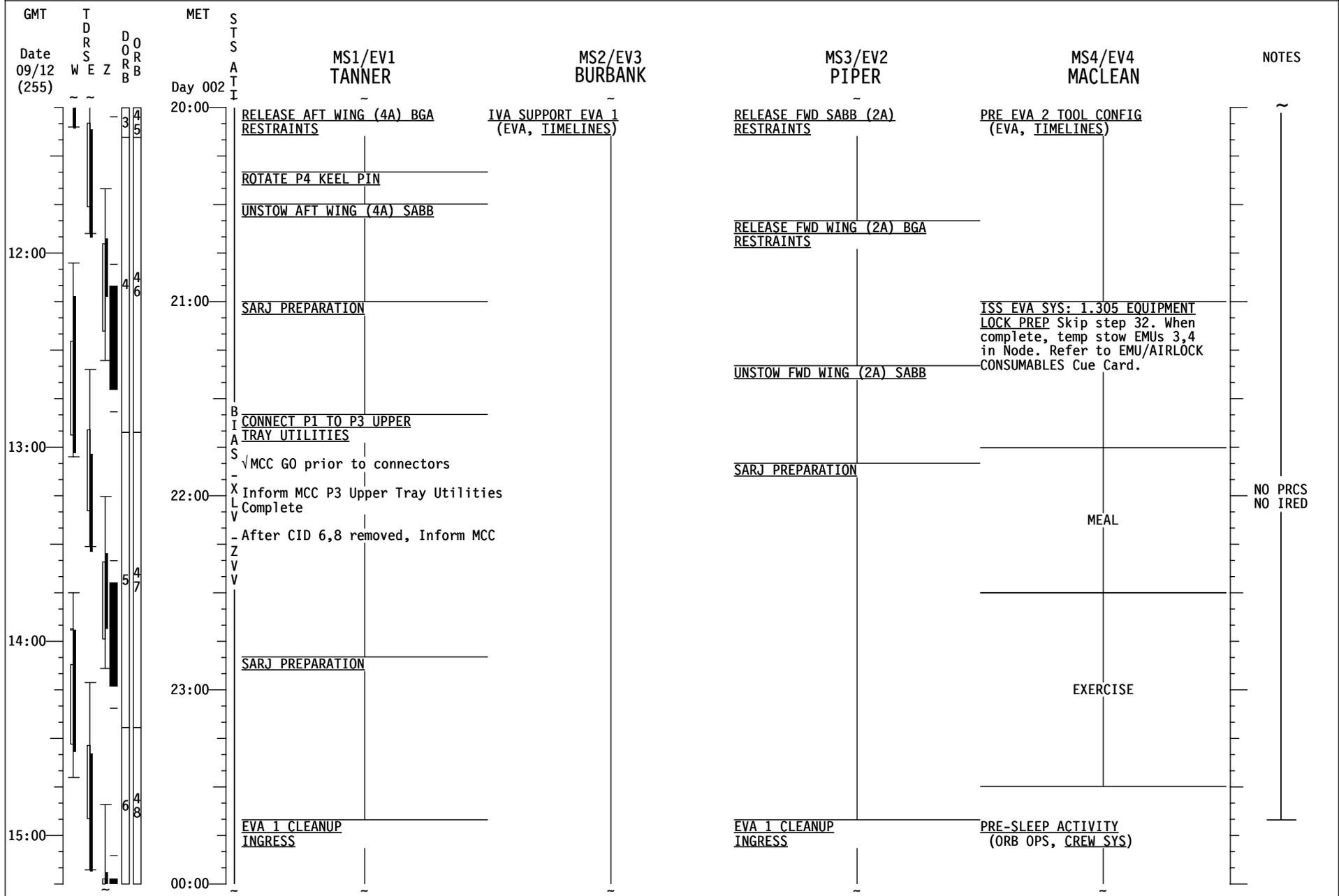
STS-115 (FD04)

GMT	TDRS Z	DORB	MET	STS	MS1/EV1 TANNER	MS2/EV3 BURBANK	MS3/EV2 PIPER	MS4/EV4 MACLEAN	NOTES
Date 09/12 (255)	W	E	Day 002	STAT					
16:00					ISS EVA SYS: 2.320 10.2 PSIA CAMPOUT EVA PREP		ISS EVA SYS: 2.320 10.2 PSIA CAMPOUT EVA PREP	POST-SLEEP ACTIVITY (ORB OPS, CREW_SYS)	
					ISS EVA SYS: 1.220 EMU PURGE	SRMS EVA 1 SUPPORT	ISS EVA SYS: 1.220 EMU PURGE	ROBO: 1.202 P3P4 INSTALL (MNVR to RTL) Steps 5-8	NO EXERCISE
08:00					ISS EVA SYS: 1.225 EMU PREBREATHE		ISS EVA SYS: 1.225 EMU PREBREATHE	ROBO: 1.202 P3P4 INSTALL (Limp) Step 9	
17:00						P/TVO7 EVA (PHOTO/TV, SCENES) Setup			NO ORBITER ERGOMETER EXERCISE
					ISS EVA SYS: CREWLOCK DPRS/RPRS CUE CARD	ISS EVA SYS: CREWLOCK DPRS/RPRS CUE CARD	ISS EVA SYS: CREWLOCK DPRS/RPRS CUE CARD		
09:00								ROBO: 1.203 P3P4 UNGRAPPLE	
18:00					ISS EVA SYS: POST DEPRESS EGRESS EVA 1 SETUP	ISS EVA SYS: POST DEPRESS IVA SUPPORT EVA 1 (EVA, TIMELINES)	ISS EVA SYS: POST DEPRESS EGRESS EVA 1 SETUP	ROBO: 2.101 UNGRAPPLE TO SABB UNLATCH Steps 1-4	
10:00					CONNECT P1 TO P3 LOWER TRAY UTILITIES √ MCC GO prior to connectors Inform MCC P3 Lower Tray Utilities Complete After CID 7-CLOSED, Inform MCC		RELEASE AFT SABB (4A) RESTRAINTS	POST-DOCKING EVA TRANSFER/RECNEG (EVA, AIRLOCK CNFG) Start @ step 10; Configures Dan and Steve's EMUs (refer to EMU/ CONSUMABLES Cue Card)	NO PRCS NO IRED
11:00					RELEASE AFT WING (4A) BGA RESTRAINTS		RELEASE FWD SABB (2A) RESTRAINTS	ISS EVA SYS: 1.403 REBA POWERED HARDWARE CHECKOUT Perform w/gloves only, No EMU TVs	
20:00									

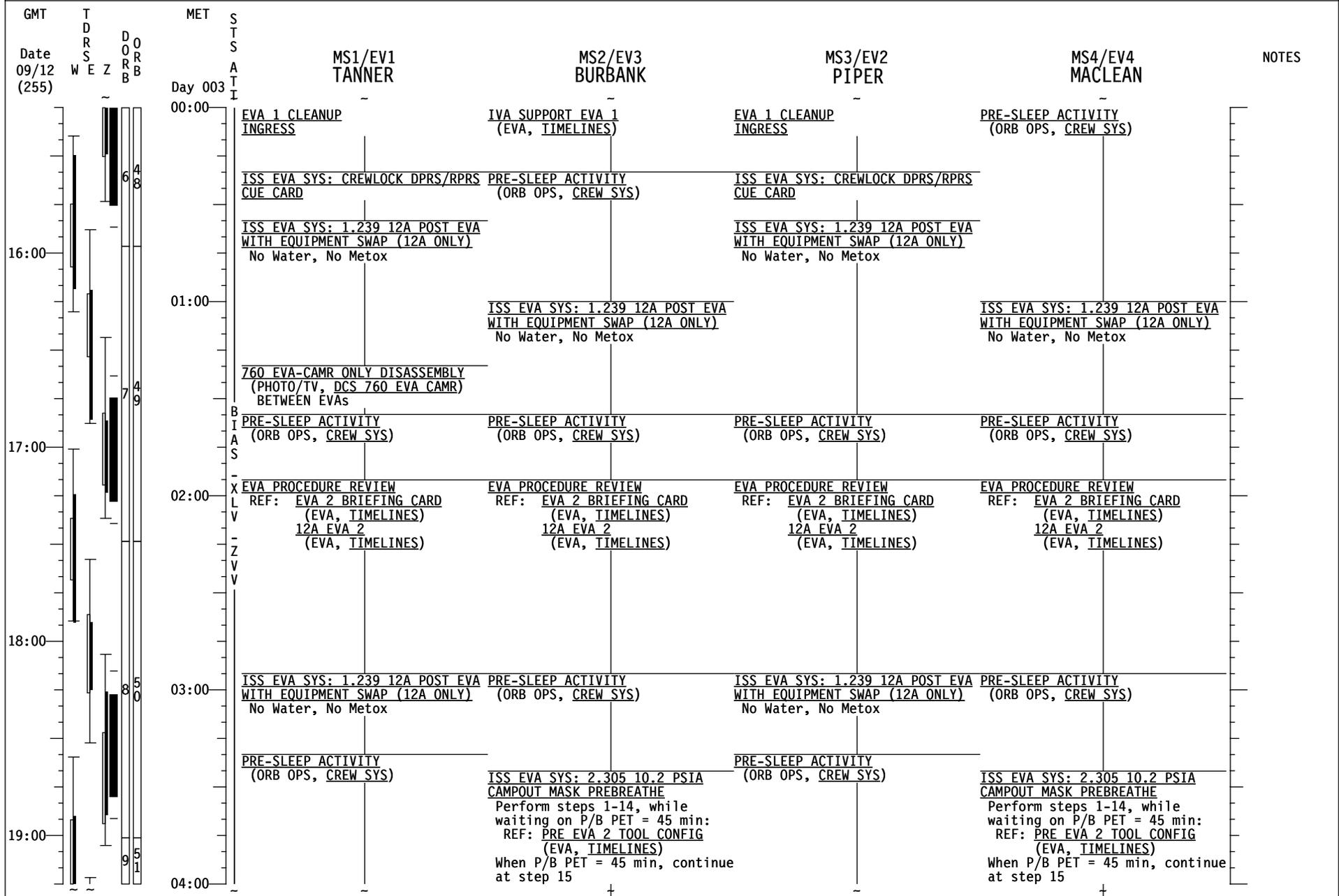
STS-115 (FD04)



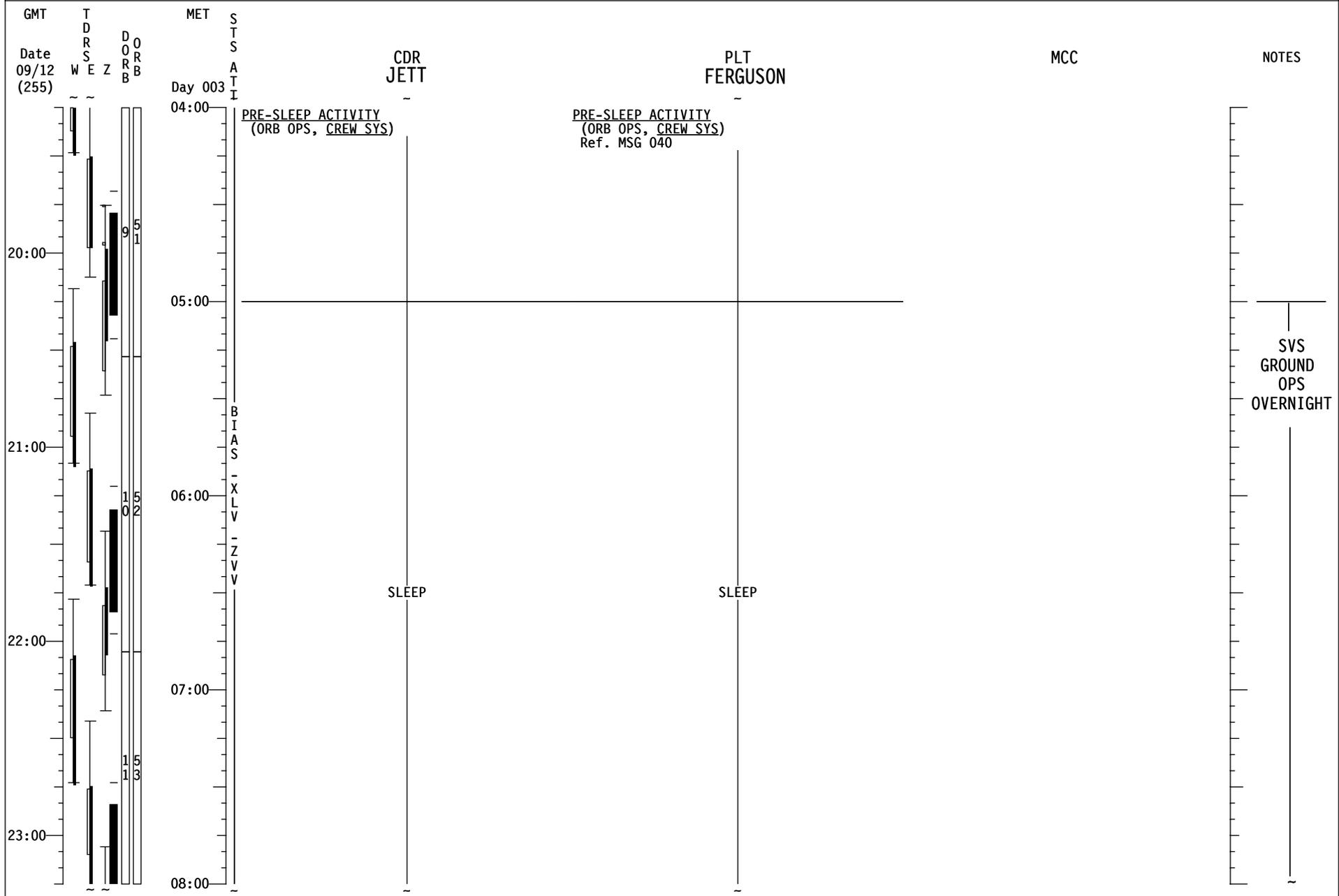
STS-115 (FD04)



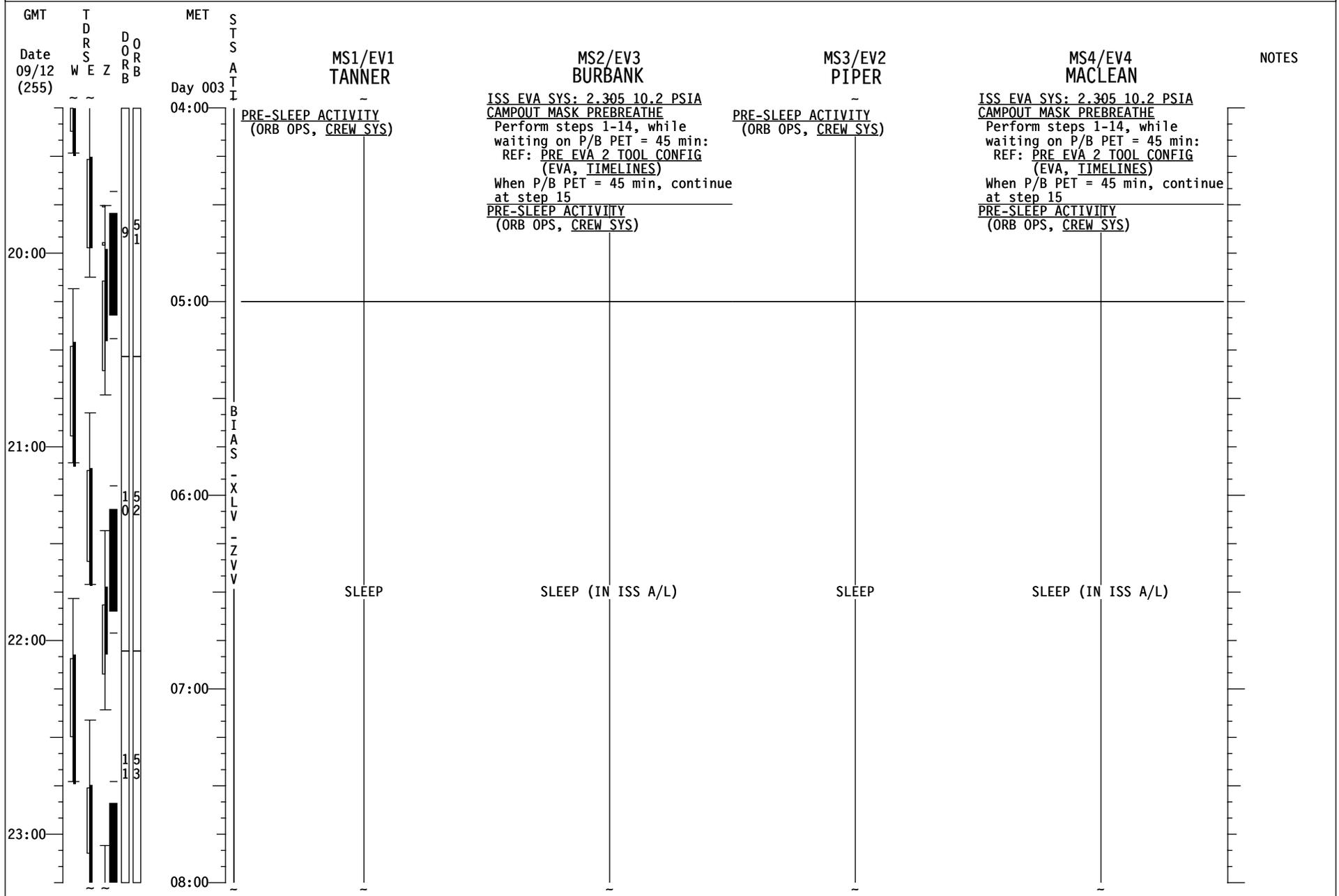
STS-115 (FD04)



STS-115 (FD04)



STS-115 (FD04)



MSG 035 (13-1195) - FD04 MISSION SUMMARY

Page 1 of 2

1
2 Good morning Atlantis!
3
4 Thanks for one more day of superb work on orbit yesterday! Today is construction day and
5 we will be happily following along! Good luck with powering up Station!
6
7
8
9
10
11 YOUR CURRENT ORBIT IS: 189 X 181 NM
12
13 NOTAMS:
14
15 EDWARDS (EDW) – 15/33 ELS ONLY – 18L CLOSED
16 WHITE SANDS (NOR) – RED
17 AMBERLY (AMB) - CLOSED
18 ANDERSEN (GUA) - 06L/24R CLOSED, ALTERNATE 06R/24L
19 ANDERSON BACKUP (UNZ CH 100) OUT OF SERVICE
20 ISTRES (FMI) - RUNWAY 33 DISTANCE MARKERS @ 300, 600, 900M
21 LAJES TACAN (LAJ) CH 45 OUT OF SERVICE
22 ORMOND BEACH (OMN) CH 73 AZIMUTH OUT OF SERVICE
23 RIO GALLEGOS (AWG) - NOT APPROVED
24 SALISBURY (SBY CH 49) OUT OF SERVICE (PRIME TACAN FOR WALLOPS)
25 WAKE ISLAND (WAK) - CLOSED
26 WAKE ISLAND (AWK) CH 82 OUT OF SERVICE
27 WILMINGTON (ILM) - CLOSED
28
29 NEXT 2 PLS OPPORTUNITIES:
30
31 EDW22 ORB 48 – 3/00:03 (FEW120, 030@04P08)
32 EDW22 ORB 64 – 4/00:26 (FEW120, VRB05)
33
34 OMS TANK FAIL CAPABILITY:
35
36 L OMS FAILS: YES (SHALLOW TARGETS)
37 R OMS FAILS: NO
38
39 LEAKING OMS PRPLT BURN:
40
41 L OMS LEAK: BURN OUT-OF-PLANE AN+66 <= TIG <= AN+33
42 OTHERWISE BURN RETROGRADE
43 R OMS LEAK: ALWAYS BURN RETRO
44
45 OMS QUANTITIES(%)
46
47 L OMS OX = 33.5 R OMS OX = 33.0
48 FU = 34.5 FU = 33.7
49
50 SUBTRACT I'CNCT COUNTER FOR CURRENT OMS QUANTITIES
51

MSG 035 (13-1195) - FD04 MISSION SUMMARY

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DELTA V AVAILABLE:

OMS	345 FPS
<u>ARCS (TOTAL ABOVE QTY1)</u>	<u>35 FPS</u>
TOTAL IN THE AFT	380 FPS
ARCS (TOTAL ABOVE QTY2)	68 FPS
FRCS (ABOVE QTY 1)	60 FPS
AFT QTY 1	91 %
AFT QTY 2	54 %

<u>SYSTEM</u>	<u>FAILURE</u>	<u>IMPACT</u>	<u>WORK AROUND</u>
DPS	CRT4 MDU Blank at Powerup - This was an occurrence of MEDS Hardware User Note 2 ("MDU DOES NOT POWER ON"). This is a random timing condition associated with the power on reset release logic.	None. MDU CRT 4 recovered after multiple power cycles at the bezel per User Note response.	<p>If this occurs again, you are go to power cycle the MDU at the bezel 5 times (pause 5 seconds between cycles). Do not power cycle at the IDP without MCC go.</p> <p>To minimize the potential of hitting this timing window, continue to use this preferred sequence for powering IDPs/MDUs:</p> <p>IDP ON 1st, then MDU ON at bezel</p> <p>MDU OFF @ bezel 1st, then IDP OFF</p>
ECLS	MNA Supply H2O Dump Line heater is deactivating on the overtemperature thermostat.	None	Use alternate heater string (MNB H2O LINE HTR B) for remainder of mission.

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13-1186 (MSG 027) CONTINGENCY P3/4 INSTALLATION WITH FAILED SSAS RTL

Page 1 of 3

NOTE

This procedure should only be performed if 2 RTLs are present and all available cues (camera views, SVS, SSRMS digitals) indicate P3/4 is properly aligned such that the third RTL should be present.

1. RTL VISUAL VERIFICATION

If either RTL-1 or RTL-3 is missing go to Step 2. Camera visual verification of Coarse Alignment Pin crossing entry plane of Coarse Alignment Cup is not possible.

If RTL-2 is missing perform Step 1.1 only.

1.1 RTL-2 VISUAL VERIFICATION

Route the SSRMS Base Elbow camera to one of the RWS monitors and adjust zoom as required.

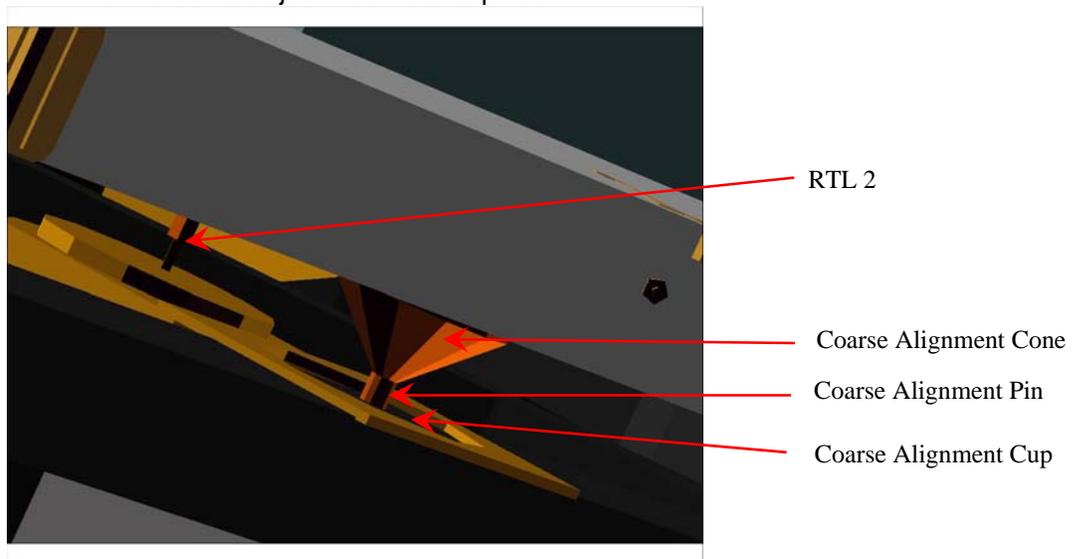


Figure 1.- 7cm Separation Reference View
(22: Base Elbow: -51, -10).

If Coarse Alignment Pin 2 has crossed the entry plane of the Coarse Alignment Cup:

Go to {1.202 P3P4 INSTALL}, step 8 (SODF: RBT FS: P3P4 INSTALL).

Else:

√MCC-H

13-1186 (MSG 027) CONTINGENCY P3/4 INSTALLATION WITH FAILED SSAS RTL

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2. SETUP

Route the SSRMS Base Elbow camera to one of the RWS monitors and adjust zoom as required.

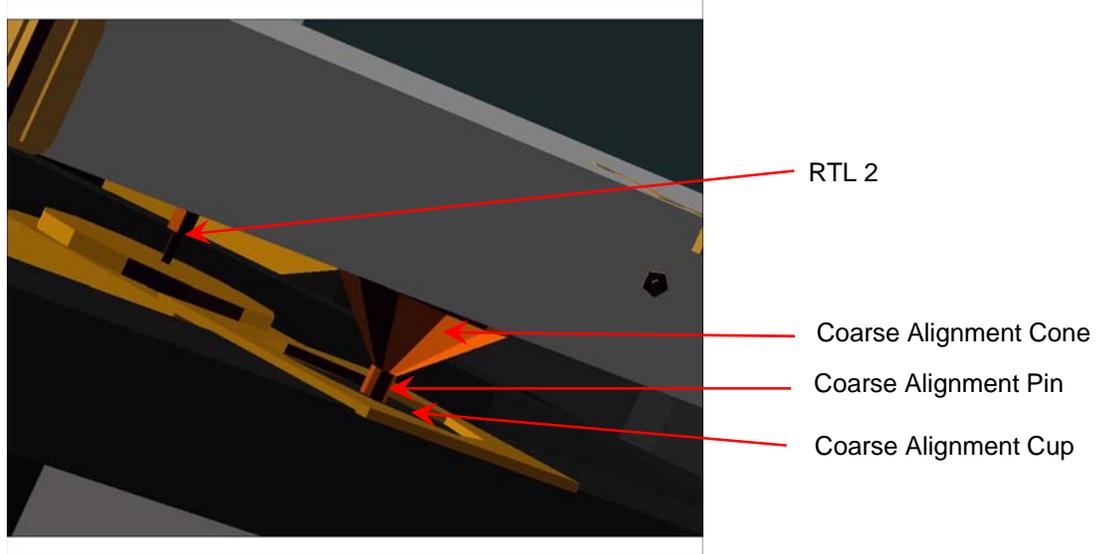


Figure 2.- RTL-2 7cm Separation Reference View
(22: Base Elbow: -51, -10).

Verify 'Loaded Parameters' (two) – P3P4 Mate, SY Held

√Display – W7>P3P4 Mate

√Command – SSRMS>Internal

√Mode – Manual (blue)

√SY - Locked

√Vernier

PCS MSS: SSRMS: Rate:

√'Scale' 'Vernier' Current: 0.5

3. VISUAL PAYLOAD ALIGNMENT

CAUTION
<ol style="list-style-type: none">1. Operator should monitor Cmd vs. Actual FOR rate on RWS monitor, & stop motion if actual rate goes to zero.2. Operator should ensure positive P1 to P3/4 interface clearance using available camera views.3. The SVS wobble must be no greater than 0.5 degrees during the maneuver in step 3. A valid SVS solution must be available throughout the maneuver.4. Step 3 should only be performed when S-band AOS so that MCC-H can monitor SSRMS joint motor currents.

On **MCC-H** Go:

Perform THC +X input until Coarse Alignment Pin 2 is no longer visible.
(Reference figure 3)

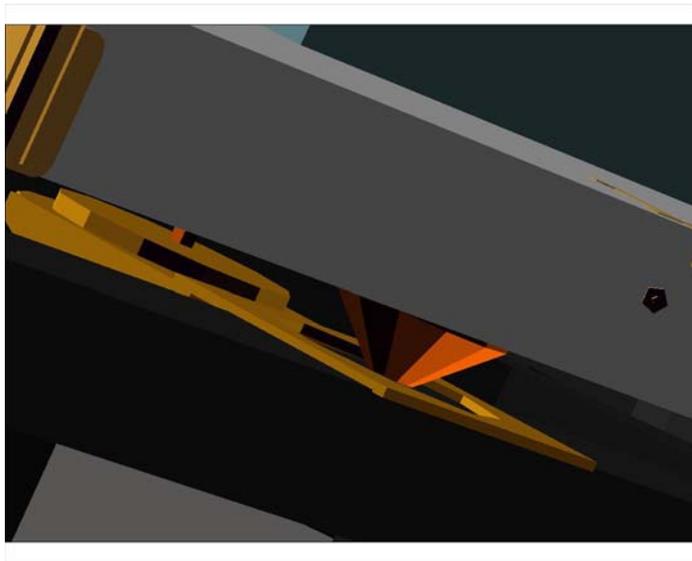


Figure 3.- RTL-2 4cm Separation
(22: Base Elbow: -51, -10).

4. ENTER STANDBY MODE (NO WR LIMPING)

PCS

MSS: SSRMS:

On **MCC-H** Go:

Enter Mode – Standby (Verify blue)

Notify SSAS Operator, “Maneuver to RTL complete – Go for Modified First Stage Capture.”

Go to {1.202 P3P4 INSTALL}, step 9 (SODF: RBT FS: P3P4 INSTALL).

13-1187 RTL LOSS DUE TO SSRMS WR LIMPING

Page 1 of 2

NOTE

1. This procedure should be performed in the event that one or more RTLs are lost after the SSRMS WR joint was limped.
2. Crew & Ground should have recorded the SVS and/or SSRMS digital position after RTL acquisition prior to WR limping.

CAUTION

1. Operator should monitor Cmd vs. Actual FOR rate on RWS monitor, & stop motion if actual rate goes to zero.
2. Operator should ensure positive P1 to P3/4 interface clearance using available camera views.

1. RTL REACQUISITION ATTEMPT

1.1 SETUP

PCS

MSS: SSRMS

Verify 'Loaded Parameters' (two) – P3P4 Mate, SY Held

√Display – W7>P3P4 Mate

√Command – SSRMS>Internal

√Vernier

MSS: SSRMS: Limp:

cmd None Limp (Verify Standby – blue)

MSS: SSRMS:

Enter Mode – Manual (Verify blue)

MSS: SSRMS: SSRMS Manual: Joint Lock:

cmd Shoulder Yaw (Verify SY – Locked)

1.2 RTL REACQUISITION

RHC, THC

Perform required HC inputs to realign per recorded position data.

Verify RTLs (three) – green

THC

Perform THC +X input to 2 cm closer than recorded position data.

13-1187 RTL LOSS DUE TO SSRMS WR LIMPING

Page 2 of 2

1.3 SSRMS WR LIMPING

PCS

MSS: SSRMS:

Enter Mode – Standby (Verify blue)

MSS: SSRMS: Limp:

sel WR

cmd Execute (Verify WR – Limped)

NOTE

Positive Ready to Latch (RTL) Indication is when you have at least one of two switches per RTL location showing green on the P1P3 SSAS Graphical Display, which indicates a positive RTL.

2. CONTINUE WITH P3/4 INSTALL

If 3 of 3 RTL indications are green

Notify SSAS Operator, "Maneuver to RTL complete – Go for Modified First Stage Capture."

If 2 of 3 RTL indications are green

Notify SSAS Operator, "Maneuver to RTL complete – Go for Modified First Stage Capture."

Monitor missing RTL indication during first stage capture.

After first stage capture is complete, √ **MCC-H** if missing RTL was not green at any point during first stage capture.

If less than 2 of 3 RTL indications are green

| √ **MCC-H**

Go to {1.202 P3P4 INSTALL}, step 9 (SODF: RBT FS: P3P4 INSTALL).

- RWS 1. [CONFIGURING VIDEO](#)
Route video (if required)

Table 1. Camera Routing

MONITOR SELECT	CAMERA SELECT
MON 1	94
MON 3	94

Verify P3/P4 UNGRAPPLE, step 4 (PDRS, NOM P3/P4 HANDOFF) complete.

- MON 1 2. [LOAD SCRIPT FILE](#)
cmd Browse [Alt-B]

sel /csvs/scripts/001_P3_Inst_SPEE.act

cmd Load [Alt-L] [script proceeds to '14: READY (P)']

Table 2. Required Operator Action

MON 1 MON 2	3	<p>cmd Start [Alt-S] [script proceeds to '24: AIM CAM AT P1 (P)']</p> <p><u>NOTE:</u> Preset windows will not be aligned over targets due to FOV offset noted on FD3.</p>
MON 1	3.1	<p>'Task 1:' sel window status indicator '1'</p> <p>cmd Exclude</p> <p>√ 'Task 1:', window status indicator '1' – GREY</p>
MON 1 MON 2	4	<p>cmd Start [Alt-S] [script proceeds to '70: VERIFY TRACKING P1 (P)']</p> <p>√ Windows tracking targets correctly (except window #1)</p> <p>*****</p> <p>* If windows not tracking correctly, or if script proceeds to</p> <p>* '263: ACQUIRE P1 MANUALLY (P)'</p> <p>* Go to {1.302 AVU SPEE CAM P1 MANUAL TARGET</p> <p>* ACQUISITION} (SODF: RBT FS: 12A: P3P4 INSTALL).</p> <p>*****</p>
MON 1 MON 1 MON 2	5	<p>cmd Start [Alt-S] [script proceeds to '113: STORE SPEE CAM SCT? (P)']</p> <p>√ 'Task 1:' – Green</p> <p>√ PSEC – Green</p>
MON 1 MON 2	6	<p><u>NOTE:</u> Target #2 is most critical in array. Ensure good lighting on targets 2 and 6 prior to commanding Start.</p> <p>cmd Start [Alt-S] [script proceeds to '142: MNVR P3 TO PRE-INST (P)']</p> <p>√ P3 truss target backgrounds completely visible on MON 2</p> <p>Do not adjust SRMS SPEE Camera position for remainder of SVS operations.</p> <p><u>NOTE:</u> saved database: /csvs/sys/P1_Aim_SPEE_Complete.dat</p>
MON 1 MON 2	7	<p>cmd Start [Alt-S] [script proceeds to '182: VERIFY TRACKING P3 (P)']</p> <p>√ Windows tracking targets correctly</p> <p>*****</p> <p>* If windows not tracking correctly, or if script proceeds to</p> <p>* '274: ACQUIRE P3 MANUALLY (P)'</p> <p>* Go to {1.303 AVU SPEE CAM P3 MANUAL TARGET</p> <p>* ACQUISITION} (SODF: RBT FS: 12A: P3P4 INSTALL).</p> <p>*****</p>
MON 1	8	<p>cmd Start [Alt-S] [script proceeds to '201: INSTALL P3 – SPEE CAM (P)']</p> <p>√ PSEC 2 < 0.004 throughout install</p> <p>*****</p> <p>* If SVS in HOLD during step 8 due to loss of P3 tracking</p> <p>* Repeat step 8.</p> <p>*****</p>

Verify {1.202 P3P4 INSTALL}, step 9 (SODF: RBT FS: 12A: P3P4
INSTALL) complete before performing steps 9 and 10

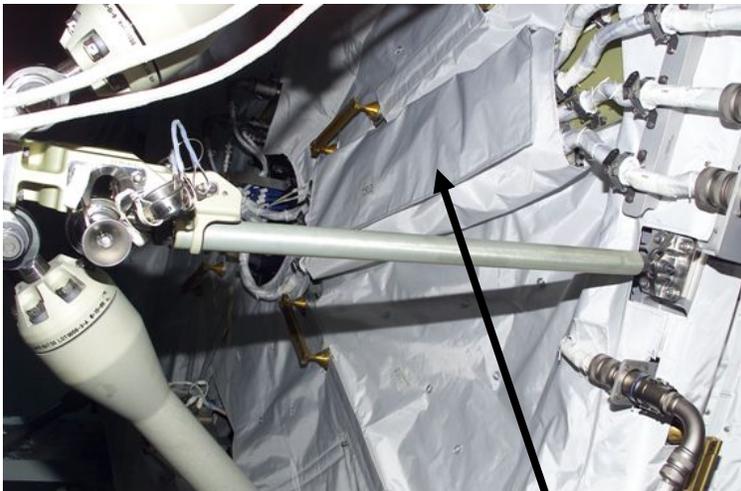
Table 3. Required Operator Action

MON 1	9	cmd Start [Alt-S] [script proceeds to '218: SVS OPS COMPLETED? (P)'] √SVS ops complete
	10	cmd Start [Alt-S] [script proceeds to '227: -----END-----']

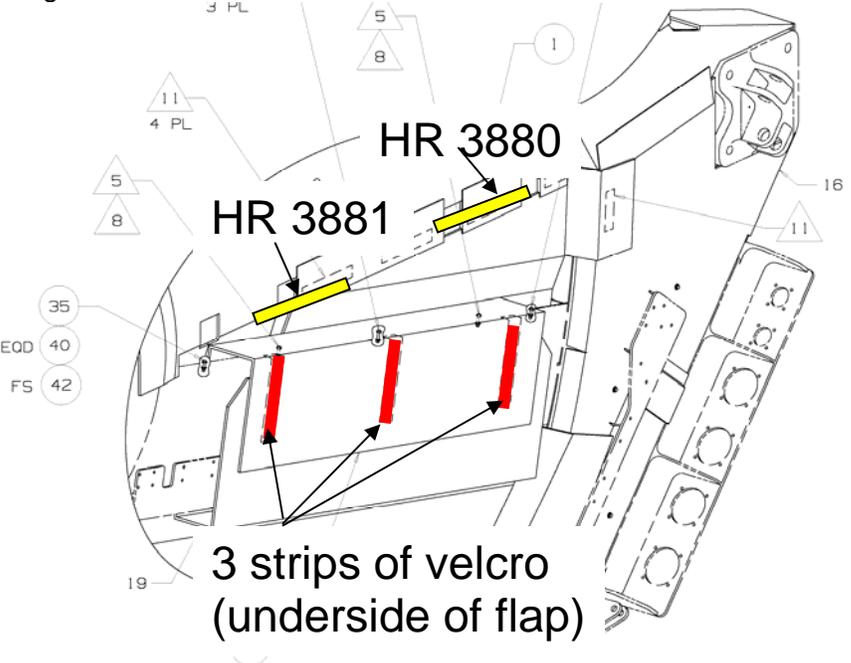
- During review of Flight Day 2 video, a thermal blanket on the P3 outboard bulkhead was observed to be loose. Review of payload bay walkdown photography revealed that this happened on the ground due to 1-G effects and the blanket was reattached.
- We will plan for you to reattach the thermal blanket during EVA 2 during SARJ Prep Tasks.
- Your best access to the protruding thermal blanket will be from face 1; this will avoid interference from struts and power cabling. Once inside the worksite, reattach the opened thermal blanket to the attached thermal blanket. There are three 1.5" wide Velcro strips on the open flap that must be mated to corresponding Velcro flaps on the attached thermal blanket. See the third photo below



MLI FLAP LOOSE



MLI FLAP SECURE



MSG 038 - FD04 WATER ACTIVITY SUMMARY

1 Today there will be one CWC fill and a Supply Water Dump using the FES. Once the FES
2 dump is complete, the FES will be deactivated to save cryo. For today's CWC fill, use any
3 of the following CWCs: 1029, 1053, 1066, 1072, 1073, 1077 (on Middeck), or 1030, 1035,
4 1043 (pregathered from ISS).

5
6
7 **FD4 Shuttle/ISS H2O Container Fill Details - Shuttle Crew**

8
9 The Shuttle/ISS H2O Container Fill initiation scheduled for PLT at MET 002/18:05 should
10 contain the following details:

11
12 **SHUTTLE/ISS H2O CONT FILL INIT #2**

13 (ORB OPS, ECLS)

14 Ag Biocide is req'd. (Use syringe from Ag biocide kit SN 1001)

15 Sample is req'd.

16 Fill Duration: ~52 minutes

17 Report Serial Number and Barcode to MCC.

18
19 After the fill is complete, insert green label in CWC window; apply decals to CWC ends; and
20 mark CWC serial number on the end decals. Transfer the two filled CWCs to NOD1P3. If
21 there is insufficient room for the bags at this ISS location, stow CWCs at NOD1P1 and
22 report to MCC.

23
24 **Supply Water Dump using FES Details – Shuttle Crew**

25
26
27 Today you will perform a supply water dump using the FES. Dump duration will be
28 approximately 4 hours. Due to constraints with ISS momentum management, check with
29 MCC prior to the Init and the Term (as requested in the timeline). The following details will
30 be required for the dump:

31
32 Initiate SUPPLY WATER DUMP USING FES (ORB OPS, ECLS) p 5-9 Step 1 DUMP
33 INITIATION, using FES Pri B.

34
35 Terminate dump on MCC call using Step 2 DUMP TERMINATION, FES is not req'd.

cmd UB_SEPS_P3_23 – Ena **Execute** (Verify – Ena)

Wait 1 minute, 40 seconds.

CDH: P3 2 MDM

P3 2 MDM

Verify UB_SEPS_P3_23 – BC

23.7 Verifying P3 2B/3A RPCM Integration Counters

P3: EPS

P3: EPS

sel RPCM P32B A

RPCM P32B A

Verify Integ Cnt – incrementing

sel RPCM P33A [X] where [X] = C E F

RPCM P33A [X]

Verify Integ Cnt – incrementing

Repeat

23.8 Enabling PTCS Application in P3 2 MDM

NOTE

The SEPS_P3_23 application should be enabled in the P3 2 MDM before the PTCS application is enabled since PTCS is dependent on SEPS for closed loop mode control (default).

CDH: P3 2 MDM

sel Applications

P3 2 Applications

cmd PTCS – Ena **Execute** (Verify – Ena)

23.9 Inhibiting Utility Rail Survival Heaters

NOTE

Scan 44 identified multi-element heaters that are susceptible to heater burnthrough. These include the Utility Rail Heaters on P3. PPLs that would prevent this configuration have not been delivered. Therefore, user action is required to mitigate this risk of heater burnthrough.

P3: EPS: P3 Rail Heaters

P3 Rail Heaters

'P3 1A Survival (Htr 1)'

cmd Inhibit – Arm (Verify Avail Status – Inh Arm)

cmd Inhibit – Inhibit (Verify Avail Status – Inh)

√ RPC Position – Op

'P3 4B Survival (Htr 1)'

cmd Inhibit – Arm (Verify Avail Status – Inh Arm)

cmd Inhibit – Inhibit (Verify Avail Status – Inh)

√ RPC Position – Op

24. S0 2 MDM CHECKOUT

24.1 Enabling S0 2 MDM RT

Task: S0 Act-Deact

S0 Act Deact

'Primary EXT MDM'

sel LB SYS P2 RT Status

LB SYS P2 RT Status

cmd 27 S0 2 MDM RT Status – Enable **Execute** (Verify – Ena)

24.2 Verifying S0 2 MDM Status in Wait State

Task: S0 Act-Deact

S0 Act Deact

'S0 2 MDM'

Verify Frame Count – incrementing

Verify Processing State – Wait

sel S0 2 MDM

S0 2 MDM

√ Sync Status – In Sync

```
*****
* If BST not blank
* | P3: CDH: P3 1 MDM: MDM Utilities
* | P3 1 MDM Utilities
* | 'Clear Latched Data in BST A'
* |
* | cmd Clear Execute
*****
```

55.3 Transitioning P3 1 MDM in Normal State

P3 1 MDM

sel Processing State

P3 1 MDM Processing State Transitions
 'Current State'

cmd Transition to Normal State Execute

Wait 60 seconds.

√Current State – Normal

55.4 Verifying P3 1 MDM in Normal State

CDH: P3 1 MDM

P3 1 MDM

Verify Frame Count – incrementing
 Verify Processing State – Normal

55.5 Checking P3 1 MDM RT FDIR Enabled

CDH: Primary EXT MDM

Primary EXT MDM

sel LB SYS P1
 sel RT Status

LB SYS P1 RT Status

√16 P3 1 MDM RT FDIR Status – Enable

55.6 Enabling P3 1 MDM SEPS Control

CDH: P3 1 MDM

P3 1 MDM

'Software Control'

sel Applications

P3 1 Applications

cmd UB_SEPS_P3_14 – Ena Execute (Verify – Ena)

P3 ACTIVATION AND S0, S1 AND P1 POWERDOWN/POWERUP

CDH: P3 1 MDM

Wait 1 minute, 40 seconds.

Verify UB_SEPS_P3_14 – BC

55.7 Verifying P3 1A/4B RPCM Integration Counters

NOTE

Integration counters may take up to 2 minutes to increment after UB_SEPS_P3_14 – Ena.

P3: EPS

sel RPCM P31A A

Verify Integ Cnt – incrementing

sel RPCM P34B [X] where [X] =

Verify Integ Cnt – incrementing

Repeat

55.8 Inhibiting P3 Utility Rail Survival Heaters RPCs

NOTE

Scan 44 identified multi-element heaters that are susceptible to heater burn through. These include the Utility Rail Heaters on P3. PPLs that would prevent this configuration have not been delivered. Because these individual heaters can not be inhibited prior to enabling the P3-1 PTCS Logic Thread, these heaters RPCs are close inhibited.

P3: EPS: P3 Rail Heaters: RPCM P34B C RPC 02

cmd Close Command – Inhibit (Verify – Inh)

√ [RPC Position – Op](#)

P3: EPS: P3 Rail Heaters: RPCM P34B C RPC 01

cmd Close Command – Inhibit (Verify – Inh)

√ [RPC Position – Op](#)

1 Delta to Orbit Ops Checklist Pre-Sleep Activity. Add new Step 13 on Page 3-6

2
3 13. Shuttle Video Ch 92 Crew Sleep Configuration

4
5

6 NOTE: 7 This procedure will configure ISS Loopback capability from CVIU 6 to VTR 2 8 using the shuttle's Monitor 2 and VPU capability.
--

9
10 A3 TV MON 1,2
11 FILTER CLEANING CCTV MON
12 (IFM, SCHEDULED MAINTENANCE)
13 Page 4-9

14
15 A7 √TV DNLK-ENA
16 PWR CNTL - PNL
17 CONTR UNIT - MNA
18 CNTL - CMD (wait 10 sec for system initialization)

19 A3 TV MON 2
20 PWR - On

21 A7 VID OUT MON 2 pb - push
22 VID IN pb - PL2

23 R12 (VPU) √ PWR - ON
24 Green Jumper - ISS

25 A3 TV MON 2 - Verify Good ISS Image
26 Brightness Knob - Turn Counter CW

13-1199 (MSG 041) – FD4 Lighting Predicts for AVU Operations
Page 1 of 1

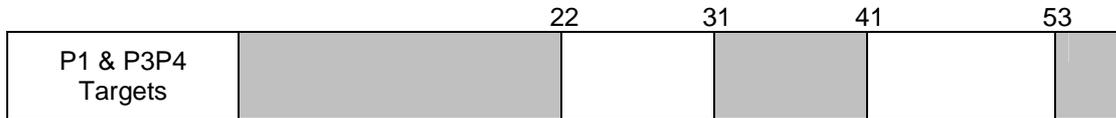
The table below indicates the predicted lighting for SVS operations based on the consecutive day passes during the P3P4 Truss Install. In the lighting descriptions and the graphical DAY PASS TIMELINE below, 0 references dawn.

Dawn MET times for day passes in the neighborhood of the P3P4 Truss Install:

- 002/13:50
- 002/15:21
- 002/16:53



002/13:50



002/15:21



002/16:53



NON-UNIFORM LIGHTING

UNIFORM LIGHTING

NIGHT PASS LIGHTING INFORMATION:

OBSS RSC and SRMS Elbow Camera

Adequate lighting for SVS without shadowing using MSS lower and upper elbow camera lights and (optional) P1 Outboard Nadir camera light. P1 Outboard Nadir camera light can cause SVS target washout, shadows, and camera glare if pointed directly at SVS targets or SVS camera. PLB floods may cause shadowing on SVS targets.

FLIGHT NIGHT 3 TESTING RESULTS:

OBSS RSC performance and SVS database parameters verified in daylight conditions with target P1_E4 manually excluded as per the updated install procedure. The sync telemetry status line causes changes in target centroid data as it scrolls through the tracking windows and it is recommended to perform SVS ops without the line present in the video.

MSG 042 - ROBOTICS PROCEDURE UPDATES RATIONALE FOR FD04

1 We have updated procedure 1.301 AVU SPEE CAM P3 TRUSS INSTALL for the
2 following reasons:
3

1. Video from the RSC camera will have to be routed through the SCU time-based correction capability in order to ensure that it contains good sync for the SVS. The procedure has been updated to call out camera ID 94 instead of camera ID 91 for the initial video routing.

2. Added a note for the crew to expect the P1 target windows to not be lined up with the P1 targets in step 3 due to the RSC camera field of view offset observed on FD3

3. Added step 3.1 to exclude P1 target 1 from the solution because we discovered that it is partially blocked by a cable while examining the RSC video from FD3.

4 The updates are in message msg 13-1194 (MSG 032). This new message was
5 linked in OSTPV for the PLT activity to acquire AVU targets.
6

7 Additionally, two contingency procedures have been put onboard:
8

9 13-1186 (MSG 027) Contingency P3P4 Installation with Failed SSAS RTL should be
10 used if 1 of 3 RTLs is missing after P3P4 has been maneuvered to the expected
11 RTL position, and attempts to acquire the third RTL are not successful. This
12 procedure will use visual verification of RTL 2 positioning in addition to your other
13 cues, and maneuver P3P4 to the Topological Capture position while looking for stall
14 indications (using video and telemetry). If P3P4 reaches the Topological capture
15 position with good alignment and no stall indications, first stage capture will be
16 initiated after a go from MCC.
17

18 13-1187 (MSG 028) Contingency RTL Loss After WR Limping should be used if 3
19 RTLs are acquired, then one or more RTLs are lost when the SSRMS Wrist Roll joint
20 is limped. This procedure will have you re-acquire the lost RTL and drive 2 cm
21 further in X than your original position, then limp the Wrist Roll joint again.
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MSG 043 - BPSMU AUDIO CONFIG

1 This configuration will provide BPSMU audio capabilities in the ISS lab in place of
2 the planned BPSMU Audio Only string referenced in the P/TV05 ISS INTERNAL
3 OPS scene in the Photo/TV Checklist. This configuration will not provide the same
4 reach that the original configuration provided into ISS. It will provide at least 27ft of
5 cable from the Airlock CCU. It will not utilize the Drag-through QD box. It will utilize
6 one of two on-board BPSMU-CCU adapters that is currently deployed on the flight
7 deck.

8 We recommend using the BPSMU that was borrowed from ISS.

9 The following hardware will be connected in series to the airlock CCU.

<u>Hardware</u>	<u>Stowage Location</u>
10 BPSMU (p/n: SED16102419-301)	Deployed on Flight Deck (PLT)
11 BPSMU-CCU Adapter (p/n: 528-21225-1)	Deployed on Flight Deck
12	connected to BPSMU (PLT)
13	
14 14ft Comm Cable Assy (p/n: 10108-10079-01/02)	MF28K (in cushion w/VLHS)
15 4ft Comm Cable Assy (p/n: MC409-0005-0240)	MF28K (in cushion w/VLHS)
16	

17 For additional length, the pre-routed cable assemblies below may be utilized. It is
18 expected that only one of the two 14 ft Comm Cables listed would be needed to
19 reach the RWS. If only using one of the 2 cables below, the pre-routed cable to seat
20 5 should be easier to remove than the pre-routed cable to seat 6. To use these
21 cables, the Comm Cable should be separated from the O2 lines **without**
22 **disconnecting the O2 line QDs from MO32M**. This can be performed by
23 disconnecting the Comm Cable from the MHA and pulling the Comm Cable through
24 the O2 Comm Assy sleeving (refer to attached picture). For entry, the Comm
25 Cables can be re-routed outside of the sleeving using the black velcro cable ties.

<u>Hardware</u>	<u>Stowage Location</u>
26 14ft CEE O2/Comm Assy (p/n: 528-20976-1)	Pre-routed from the MHA
27	connected to MO39M to seat 5
28	
29	
30 14ft CEE O2/Comm Assy	Pre-routed from the MHA
31	connected to MO39M to seat 6
32	
33	

34 Operations Notes:

35 Overnight the above assembled cabling should be returned to the orbiter side of the
36 interface and then redeployed for station use in the mornings.

37
38 Should a Joint Expedited Undock and Separation or a Joint Emergency Egress
39 become required the above assembled BPSMU cabling should be pulled back over
40 to the shuttle side prior to hatch closure.

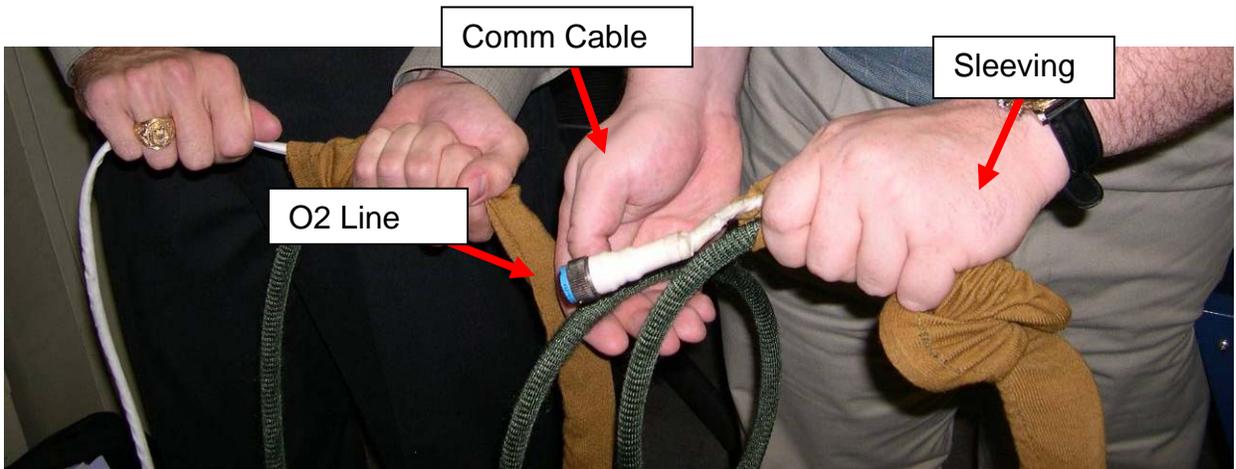
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MSG 043 - BPSMU AUDIO CONFIG

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MSG 044 (13-1201) - FD04 MMT SUMMARY

Page 1 of 4

1 There were only a few items of significant discussion at the FD 3 MMT which included a
2 continued review of the ascent imagery and a very preliminary look at the FD 2 TPS
3 inspection and FD3 RPM imagery. The MMT left the decision as to whether to perform a
4 focused inspection with MOD, the Orbiter Project, and the Engineering debris/imagery
5 teams.
6

7 Several orbiter systems items were discussed and brief summaries are provided in
8 no particular order:

9 **1) Cryo margins-** No change-mission can be extended by two days if required.

10 **2) Fuel Cell 1-** The pumps continue to operate within limits on the remaining two
11 phases (B&C). FDF pen and ink changes to the AC VOLTS, AC OVERLOAD, and
12 THREE PHASE MOTOR STOP in the OPCL and EPCL were sent to you during FD
13 03 execute package.

14 **3) Payload Bay Door Microswitches** - No change in the status provided yesterday. The
15 Starboard PLBD Closed B microswitch has remained in the failed ON status over the last 24
16 hours and will continue to be monitored. (Update: The microswitch toggled a few times and
17 is now indicating correctly.)

18 **4) Speedbrake FCS Channel 3-** The MCC continues to monitor ASA 3 during the
19 crew sleep periods. To date, the ASA 3 performance on all actuators has been
20 nominal in these sleep periods with no re-occurrence of the intermittent driver
21 outputs seen during ascent. The preliminary plan is to still bypass Speedbrake
22 Channel 3 during FCS checkout.

23 **5) H2O Dump Line Heater A Thermostat** -At the end of the water dump, the MNA
24 H2O LINE HTR A began cycling at the overtemp thermostat upper limit. There were
25 4 cycles to 106 deg Fahrenheit which is in the 90 –110 Fahrenheit overtemp control
26 band. The thermostat normally operates in the 70 -90 deg Fahrenheit range. Since
27 swapping to the B string (MNB H2O LINE HTR), the temperatures have returned to
28 the nominal range. The plan is to continue using the B heaters. The A heaters are
29 available using the overtemp logic should that be required.

30 **6) Battery Powered Speaker Mike (BPSMU) Cable** - This missing cable was
31 returned inadvertently on STS-121. Of course the biggest impact is loss of A/G 2
32 and Orbiter caution and warning monitoring on the ISS. The MCC teams are
33 reviewing alternate configurations and IFM options to provide a workable cable.

34 **7) Starboard Manipulator Retention Latch (MRL) B microswitch-** This
35 microswitch failed to indicate CLOSED at OBSS stow on flight day 2. The software
36 did not terminate latch motor drive due to missing indication but the A microswitch
37 indicated correctly. This A microswitch drives the cockpit talkback. A similar
38 phenomenon was seen on STS-114 and there is no impact to planned future
39 operations.

40 **8) Wing Leading Edge Sensors** - There has been no significant change since
41 yesterday's MMT. Video for the earliest event of interest at ~14 seconds MET (200
42 mph velocity) on the port wing Panels 6-7 has been reviewed and there is no
43 evidence of impact at that time (See Figure 1).
44
45

MSG 044 (13-1201) - FD04 MMT SUMMARY

Page 2 of 4

1 **Figure 1: High Definition Video of Port Wing indicates no impact**



2
3

4 **Ascent Data Review** - The MMT reviewed the preliminary pad debris environment
5 walkdowns and ascent vehicle propulsion and trajectory performance. Pad B is in
6 very good shape after a slight sand blasting by the SRBs. No vehicle flight hardware
7 has been found and the Pad environment was considered to be very nominal based
8 on previous flight experience. The orbiter liftoff lateral acceleration data and visual
9 inspections indicated no stud hang-ups. Additionally, the preliminary ascent data
10 shows that powered flight and post MECO were very nominal with no issues. All
11 four LH2 and LO2 ECO sensors worked nominally during ascent. The ET umbilical
12 and handheld photography also showed that the tank performed very well and a
13 summary of those pictures will be provided later in the mission for your review.

14

15 **Focused Inspection CHIT Meeting**

16

17 **TPS Imagery Analysis** - The imagery and TPS teams have reviewed all of the FD 2
18 TPS inspection data and the FD 3 RPM photography. Based on this review all of the
19 RCC including both port and starboard wings and nosecap and all tile have been
20 cleared for entry and **no OBSS focused inspections on FD 5 are required.**

21

22 In terms of RCC, the only items found in the surveys were blackspots from the SRB
23 sep motor deposition. In term of tile, there are 2 tile damages on aft ET door
24 structure side with the largest being 1.5 inches and the other is being 1.0 inches.
25 These are only significant because they are in the cone of influence of two of three
26 protruding gap fillers that are greater than the 0.25" criteria. The largest is on the left
27 hand ET door, has a height of 1.2 inches and is shown Figure 2. The right hand ET
28 Door contains a small piece of gap filler that protrudes about 0.8 inches and the port
29 elevon has a protruding gap filler of about the same height.

30

MSG 044 (13-1201) - FD04 MMT SUMMARY

Page 3 of 4

1 Based on all of the inspection data to date there are only two items of interest
2 remaining. These items are the protruding gap filler on the left hand ET door (See
3 Figure 2) and the thermal blanket protrusion of the right hand OMS pod (See Figure
4 3). The overall plan is to collect additional non-OBSS based imagery data using
5 ISS, RMS, or SSRMS assets. More details on this plan will be provided as it
6 develops over the next 24 hours.

7

8 Additionally, thermal and structural analysis largely based on the conservative Mach
9 25 boundary layer trip work from STS-121 is underway in an attempt to clear the ET
10 door gap filler. A debris liberation and transport analysis for this gap filler will also be
11 performed. More details from this analysis will be provided as they become
12 available.

13

14

15

16

17

Figure 2: Left Hand ET Door Protruding Gap Filler

**Protruding approx. 1.7" between tiles V070-395025-020/395055-072
X0 1347 Y0 -72 tile thickness 1.6"**

- Actions in Work**
- Review installation data**
- Review STS-121 analytical work**
- Define impact to heating/Thermal/Subsystems**
- Assess need for EVA/Focused Inspection**



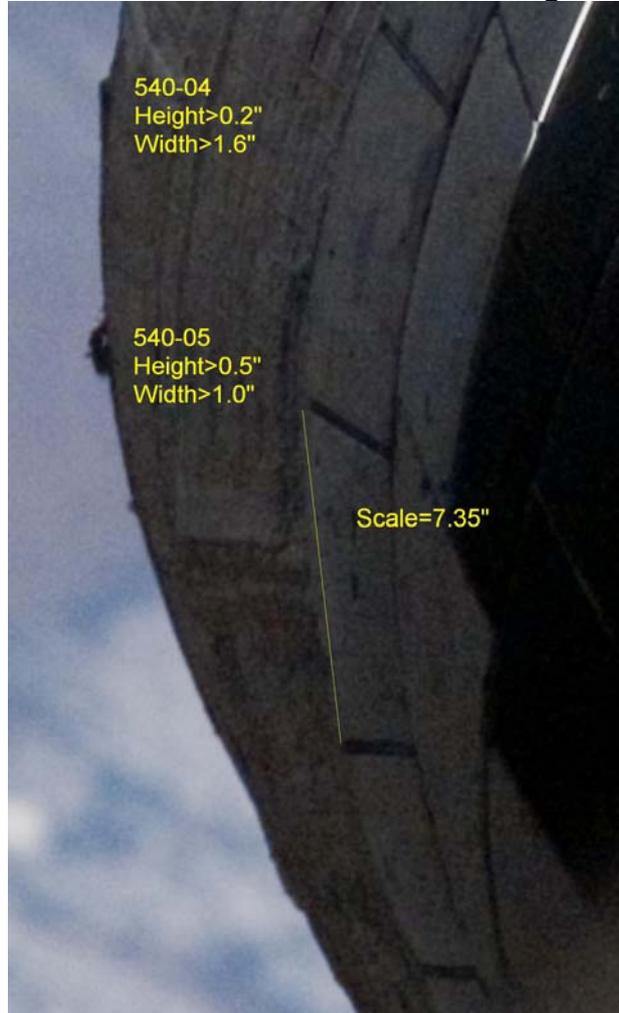
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MSG 044 (13-1201) - FD04 MMT SUMMARY

Page 4 of 4

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Figure 3: Starboard OMS Pod Protruding Blanket



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